



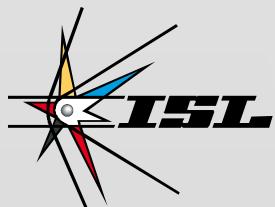
Trajectory Deflection of Fin- and Spin-Stabilized Projectiles Using Paired Lateral Impulses

Pierre Wey

Pierre.Wey@isl.eu

Daniel Corriveau

Daniel.Corriveau@drdc-rddc.gc.ca

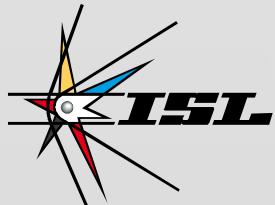


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Presentation Overview



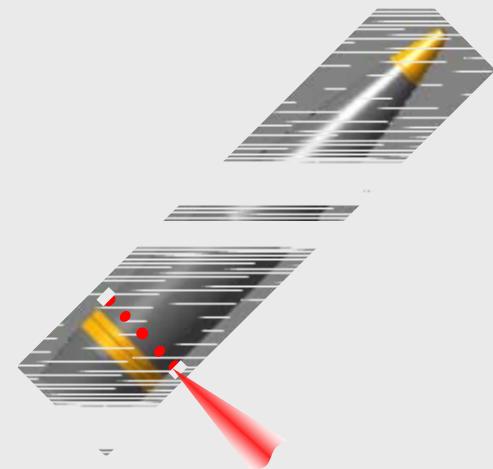
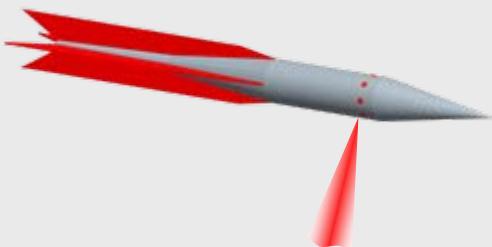
- Objective
- Effect of a Lateral Impulse
- Basic Mathematical Model and Analytical Solution
- Pairing the Impulses
- Examples
- Conclusions



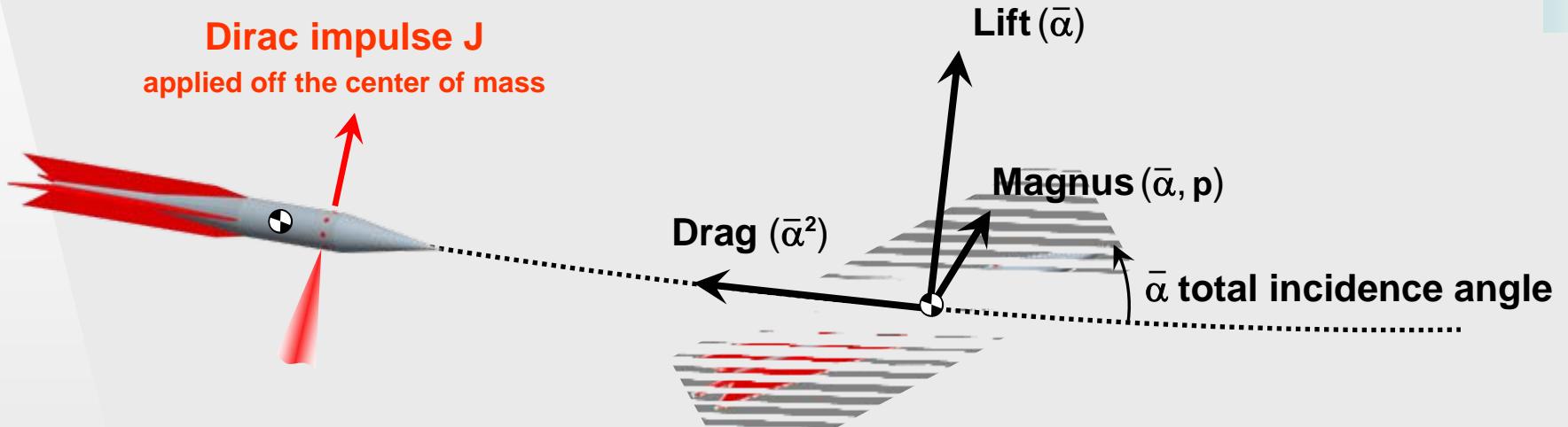
Objective



- To developed a well defined procedure for using paired impulses on fin- and spin-stabilized projectiles in order to achieve enhanced drift corrections



Effect of a Lateral Impulse (1/2)



Trajectory deflection: total lateral impulse = $J + J_L + J_Y$

$$\begin{aligned} \int \text{Magnus}(\bar{\alpha}) dt \\ \int \text{Lift}(\bar{\alpha}) dt \end{aligned}$$

Additional velocity decrease: axial impulse = J_{D2}

$$\int \text{Drag}(\bar{\alpha}^2) dt$$

More challenging to implement properly

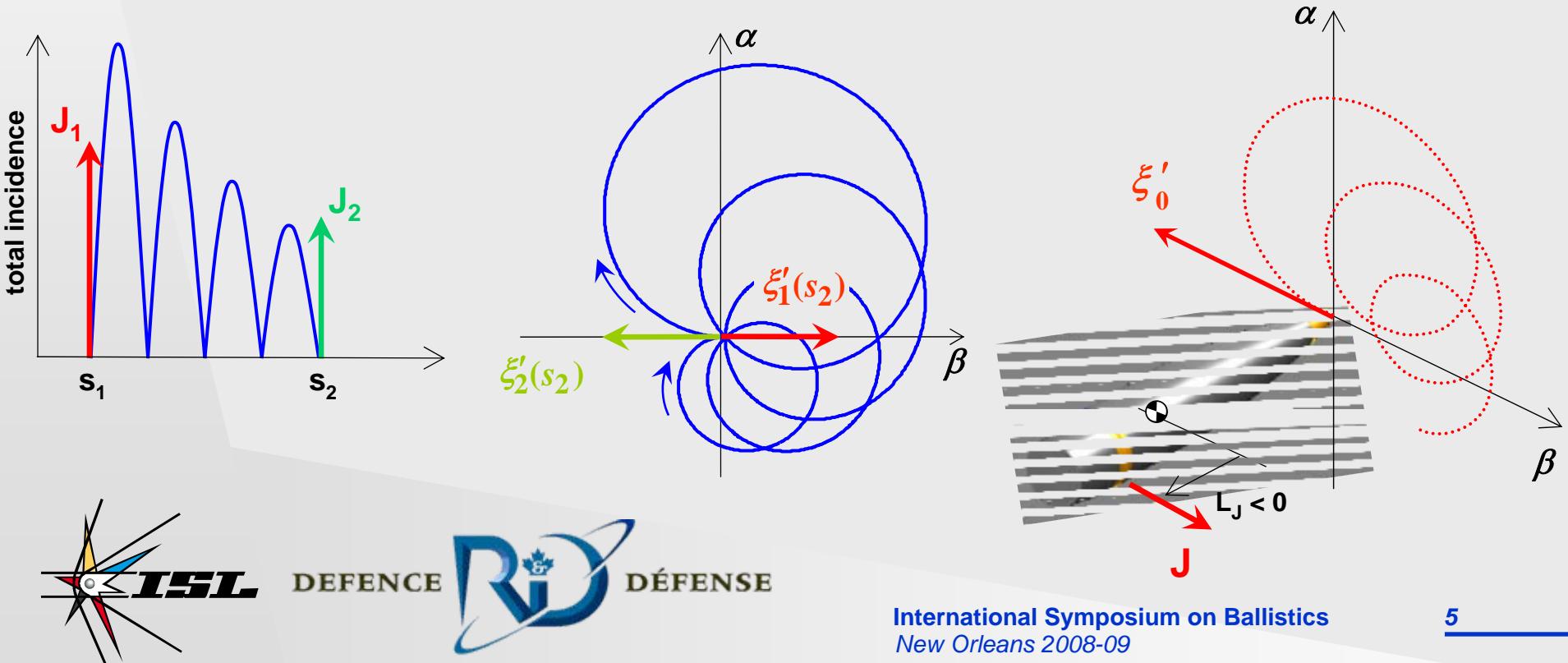


Effect of a Lateral Impulse (2/2)

Pairing lateral impulses off the center of mass

- Basic idea:
- J_1 triggers the angular motion at s_1
 - J_2 stops the angular motion at s_2

- Advantages:
- the lateral correction is enhanced by the lift impulse resulting from the angular motion
 - the range lost is minimized by stopping the angular motion



Basic Mathematical Model (1/4)



Linearized equation of the complex incidence motion:

$$\xi'' + (H - iP) \xi' - (M + iP T) \xi = \xi'_0 \delta(s)$$

Dirac impulse

Magnus moment
+ vel. change

$$T = \frac{\rho A d}{2m} \left(C_{L\alpha} + \frac{m d^2}{I_x} C_{np\alpha} \right)$$

Overturning moment
(restoring or destabilizing)

$$M = \frac{\rho A d^3}{2I_y} C_{m\alpha}$$

Gyroscopic effect $P = \frac{I_x}{I_y} \frac{pd}{V}$

Damping factor
+ vel. change

$$H = \frac{\rho A d}{2m} \left(C_{L\alpha} - C_D - \frac{m d^2}{I_y} C_{mq} \right)$$



Basic Mathematical Model (2/4)

Motion = sum of two rotating arms:

$$\xi = K_{F_0} e^{i\phi_{F_0}} e^{(\lambda_F + i\phi'_F)s} + K_{S_0} e^{i\phi_{S_0}} e^{(\lambda_S + i\phi'_S)s}$$

Angular frequencies

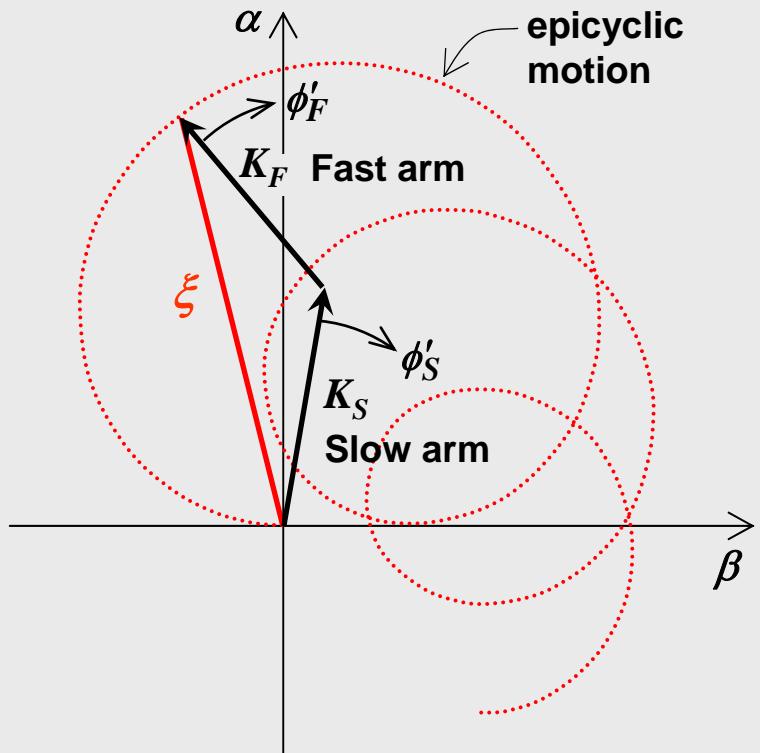
$$\phi'_F = \frac{1}{2} \left(P + \sqrt{P^2 - 4M} \right) \quad \phi'_S = \frac{1}{2} \left(P - \sqrt{P^2 - 4M} \right)$$

Damping factors

$$\lambda_F = -\frac{1}{2} \left(H - \frac{P(2T - H)}{\sqrt{P^2 - 4M}} \right) \quad \lambda_S = -\frac{1}{2} \left(H + \frac{P(2T - H)}{\sqrt{P^2 - 4M}} \right)$$

Initial arms

$$K_{F_0} e^{i\phi_{F_0}} = - \frac{i\xi'_0 + \phi'_S \xi_0}{\phi'_F - \phi'_S} \quad K_{S_0} e^{i\phi_{S_0}} = \frac{i\xi'_0 + \phi'_F \xi_0}{\phi'_F - \phi'_S}$$



Basic Mathematical Model (3/4)



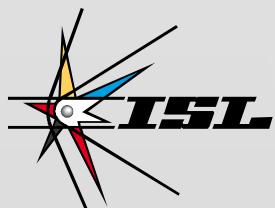
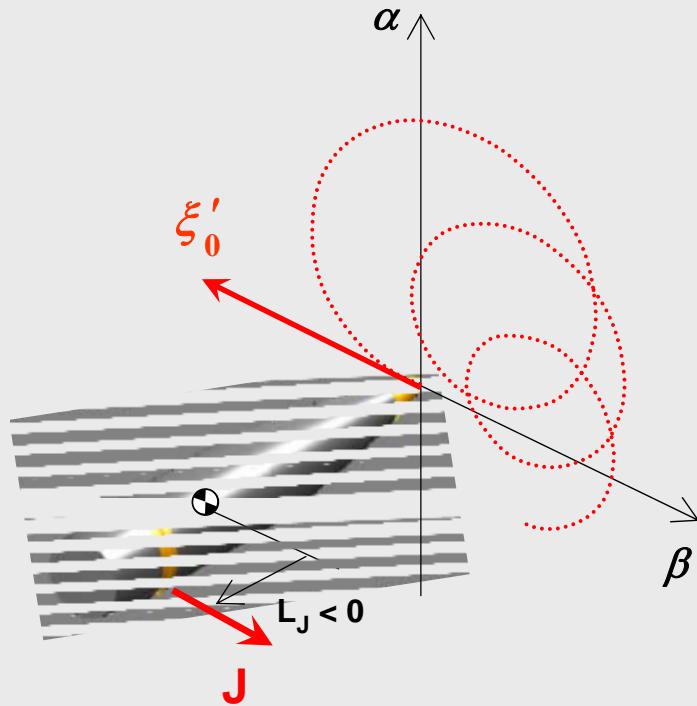
Initial conditions forced by the Dirac impulse:

$$\xi_0 = -\frac{J}{mV}$$

↳ negligible in supersonic mode

$$\xi'_0 = \frac{J L_J d^2}{I_y V}$$

↳ main cause of angular motion
(unless $L_J \rightarrow 0$)



Basic Mathematical Model (4/4)



Lateral impulses: Lift + Magnus

$$J_L = \int_0^\infty L(t) dt$$

$$J_L = \frac{1}{2} \rho A V d C_{L\alpha} \int_0^\infty \xi ds$$

$$\int_0^\infty \xi ds = - \frac{K_{F_0} e^{i\phi_{F_0}}}{\lambda_F + i\phi'_F} - \frac{K_{S_0} e^{i\phi_{S_0}}}{\lambda_S + i\phi'_S}$$

$$J_Y = \int_0^\infty Y(t) dt$$

$$J_Y = i \frac{C_{Yp\alpha}}{C_{L\alpha}} \frac{pd}{V} J_L$$

Axial impulse: additional Drag

$$J_{D_2} = \int_0^\infty D_2(t) dt$$

$$C_D = C_{D0} + C_{D2} \sin^2 \bar{\alpha}$$

$$J_{D_2} = \frac{1}{2} \rho A V d C_{D_2} \int_0^\infty |\xi|^2 ds$$

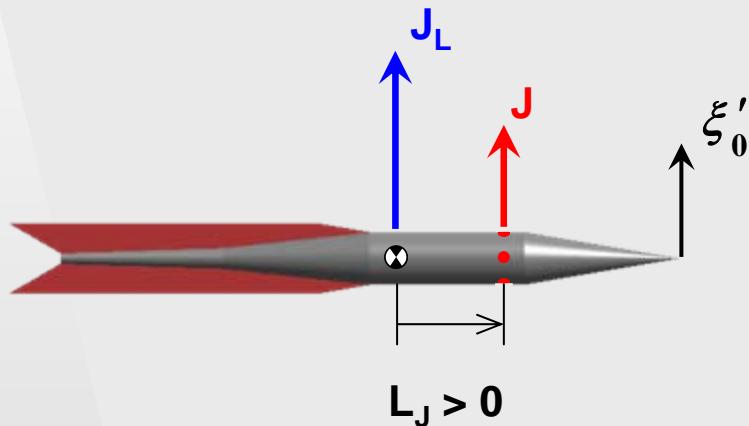
$$\int_0^\infty |\xi|^2 ds = - \frac{K_{F_0}^2}{2\lambda_F} - \frac{K_{S_0}^2}{2\lambda_S} - \frac{2 K_{F_0} K_{S_0} \left[(\lambda_F + \lambda_S) \cos(\phi_{F_0} - \phi_{S_0}) + (\phi'_F - \phi'_S) \sin(\phi_{F_0} - \phi_{S_0}) \right]}{(\lambda_F + \lambda_S)^2 + (\phi'_F - \phi'_S)^2}$$



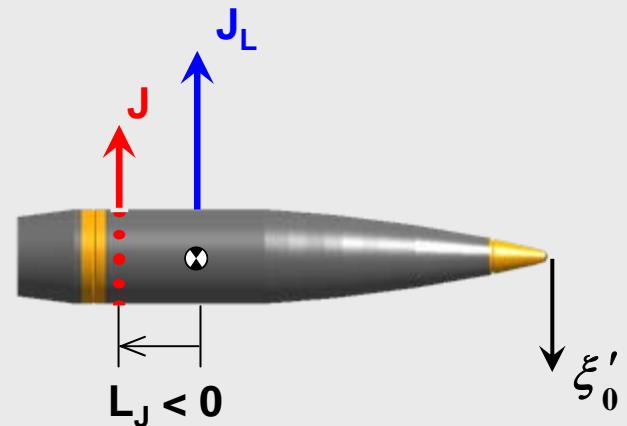
Pairing the Impulses: Location (1/2)

Goal: maximizing $J + J_L$

Fin-stabilized projectile



Spin-stabilized projectile



Rule #1: the lateral impulse must be applied

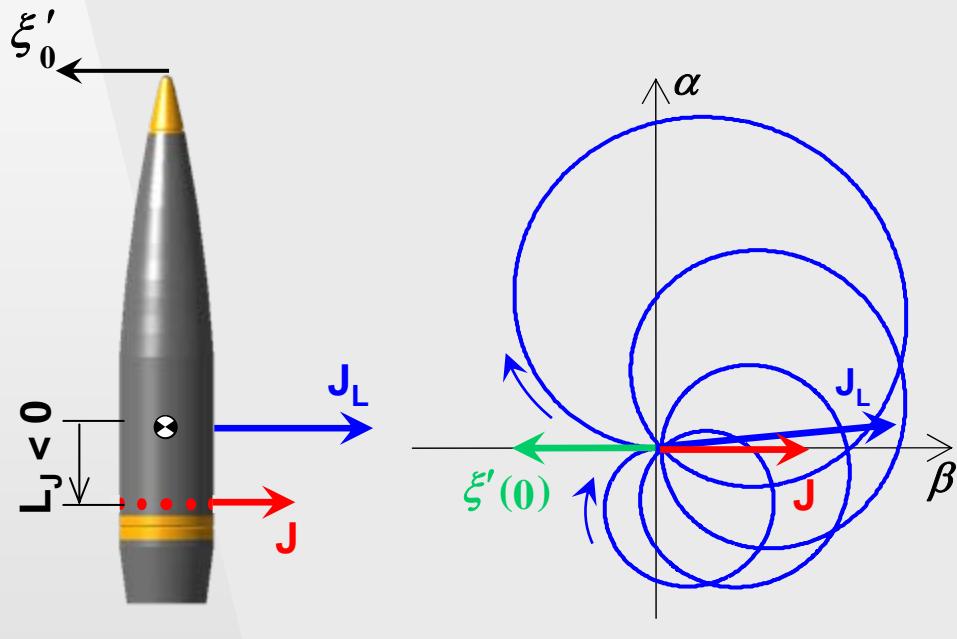
- ahead of the center of mass for fin-stab. shells
- behind the center of mass for spin-stab. shells

$$J_L = -J L_J \frac{C_{L\alpha}}{C_{m\alpha}} e^{i\Delta\phi}$$

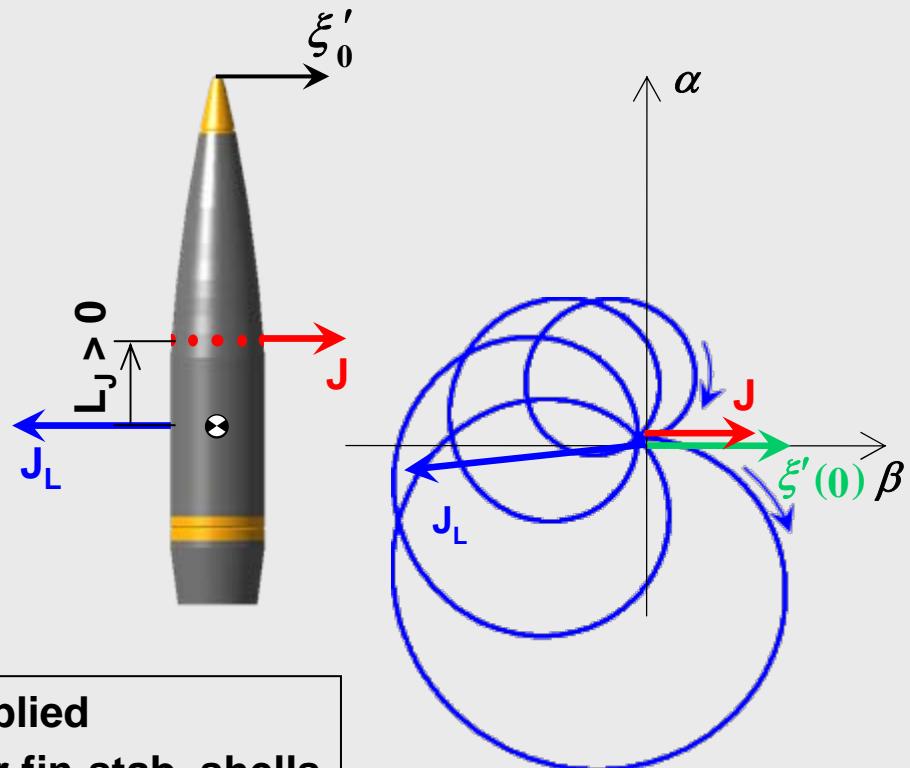
Pairing the Impulses: Location (2/2)

Goal: maximizing $J + J_L$ for spin-stabilized projectile

How to do it:



How not to do it:



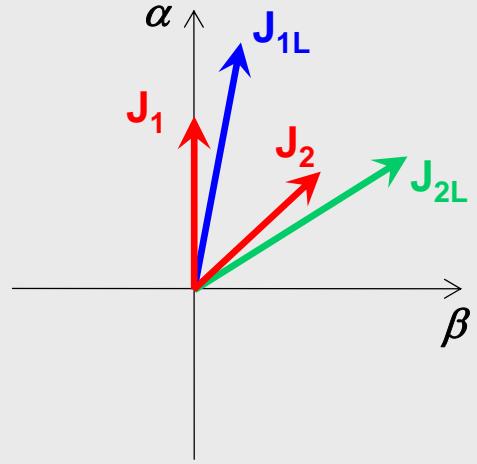
Rule #1: the lateral impulse must be applied

- ahead of the center of mass for fin-stab. shells
- behind the center of mass for spin-stab. shells

Pairing the Impulses: Orientation

$$\text{Total lateral impulse} = (J_1 + J_{1L}) + (J_2 + J_{2L})$$

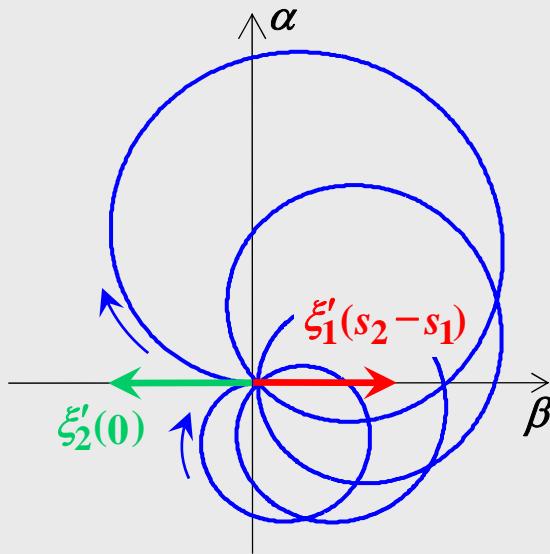
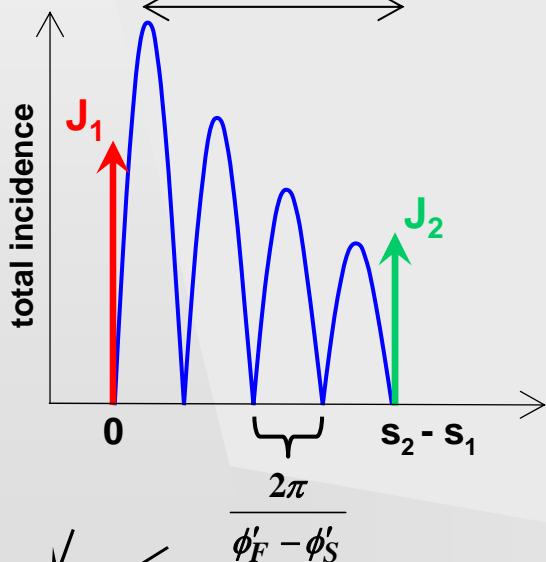
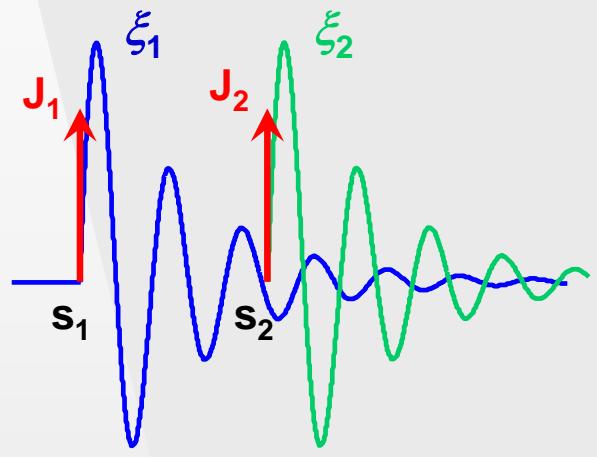
- independent of $(s_2 - s_1)$
- maximum if J_1 and J_2 are aligned



Rule #2: J_1 and J_2 must be triggered at the same roll angle



Pairing the Impulses: Timing



Linearized equation of motion:

$$\xi = \xi_1 + \xi_2$$

$$\forall s \geq s_2, \xi_1(s) + \xi_2(s) = 0$$

Motion strictly opposed if:

$$\xi'_1(s_2 - s_1) = \xi'_2(0) = 0$$

$$\xi'_1(s_2 - s_1) = -\xi'_2(0)$$

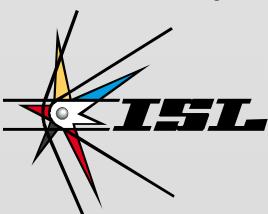
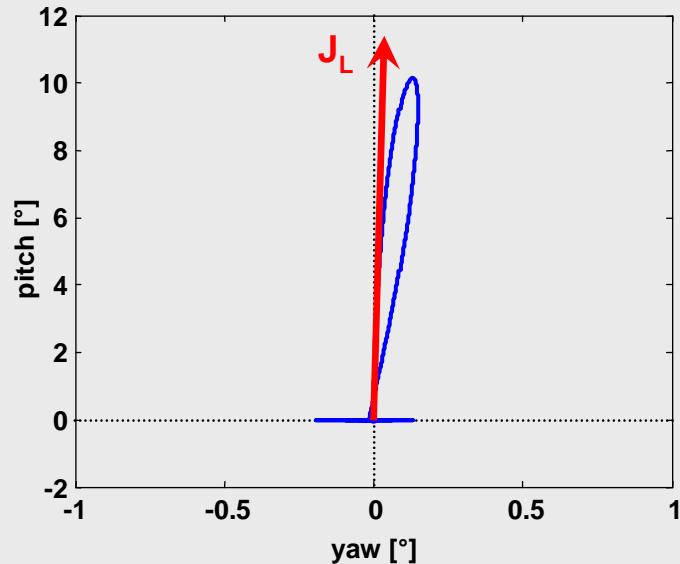
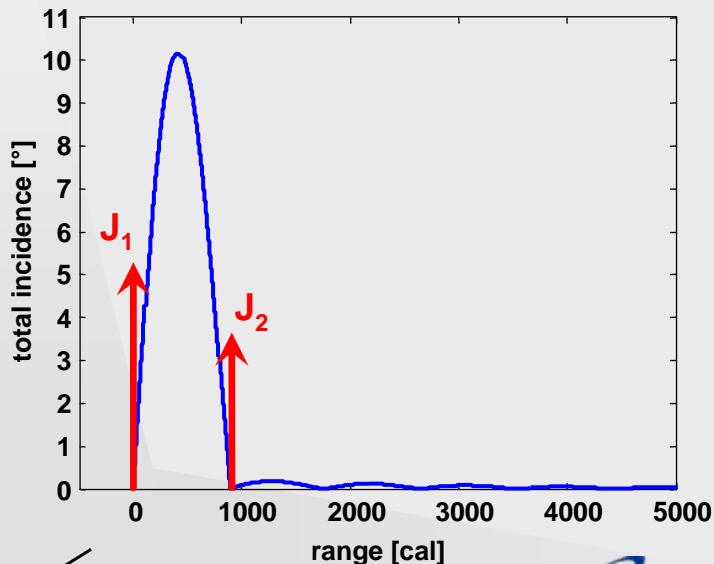
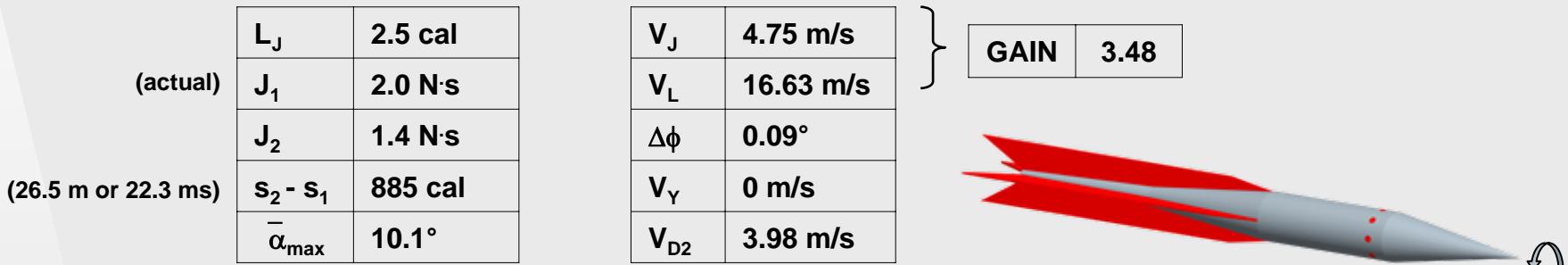
Rule #3:

$$s_2 - s_1 = k \frac{2\pi}{\phi'_F - \phi'_S}$$

$$k = \text{nearest integer to } \frac{\text{sign}(\phi'_S)}{2} \left(\frac{\phi'_F}{\phi'_S} - 1 \right)$$

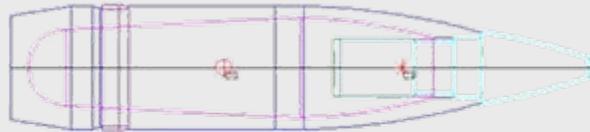
Example: GSP Shell

d	m	I_y	Mach	p	CD_0	$CL\alpha$	$CYp\alpha$	$Cm\alpha$	Cmq
30 mm	0.7 kg	5.04e-3 kgm ²	3.5	22 Hz	0.19	7.6	~ 0	-5.3	~ -300



Example: 105 mm Artillery Shell _(1/2)

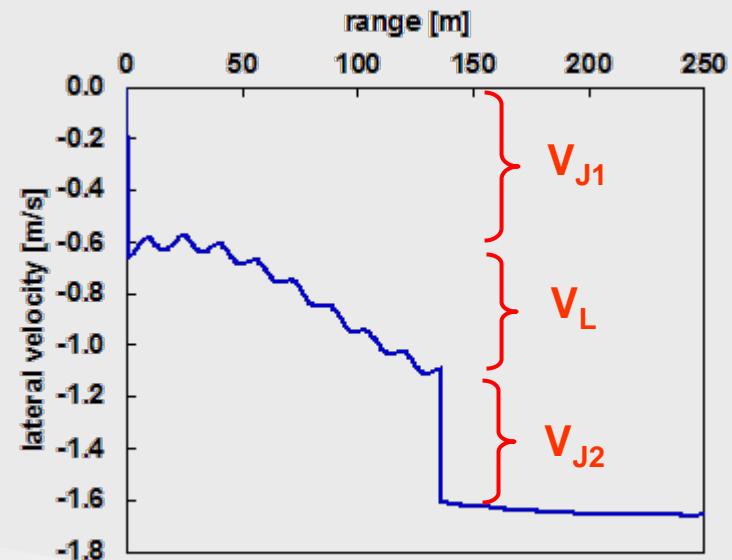
d	m	I_y	Mach	p	CD_0	$CL\alpha$	$CYp\alpha$	$Cm\alpha$	Cmq
105 mm	15.05 kg	2.19e-1 kgm ²	1.5	310 Hz	0.375	2.12	-0.8	3.6	-17



L_J	-0.51 cal
J_1	10 N·s
J_2	7.83 N·s
$s_2 - s_1$	1293 cal
$\bar{\alpha}_{max}$	1.5°

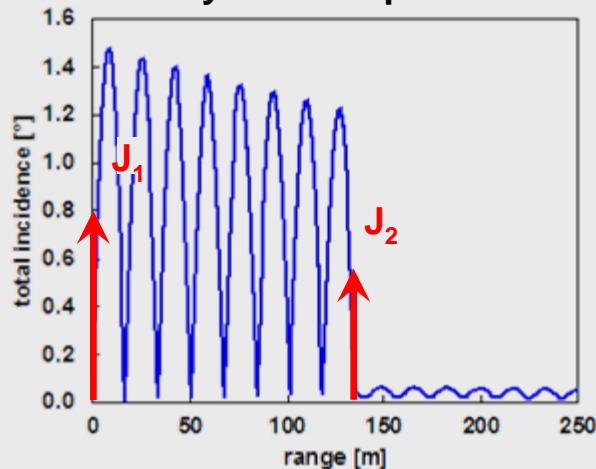
V_J	1.18 m/s
V_L	0.36 m/s
$\Delta\phi$	-3.5°
V_Y	0.05 m/s

(135.6 m)

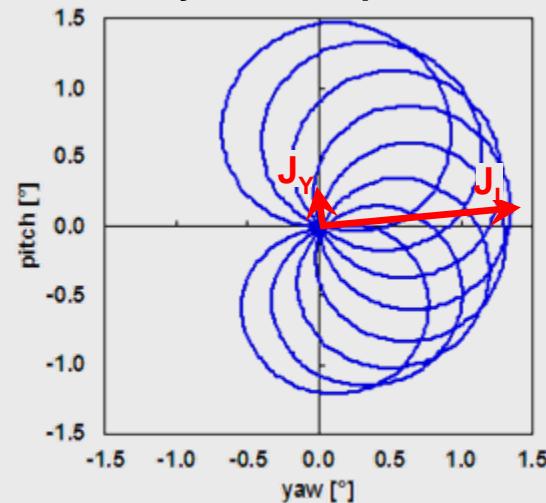


Example: 105 mm Artillery Shell (2/2)

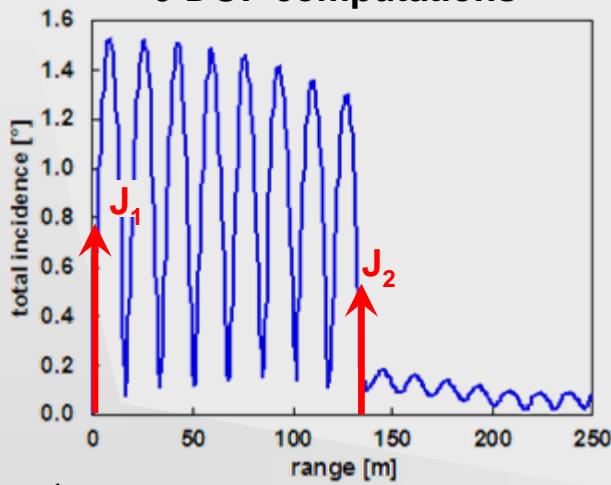
Analytical computations



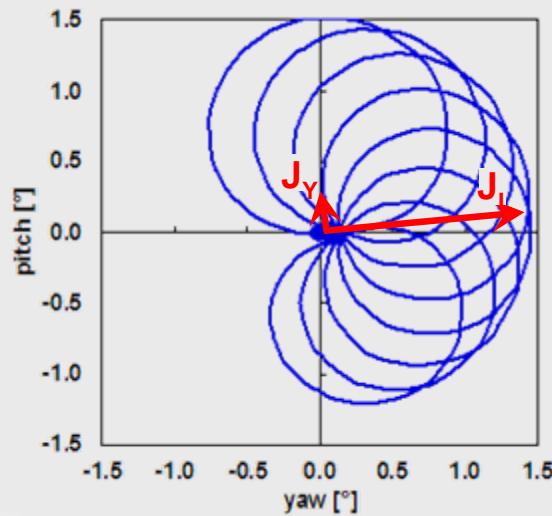
Analytical computations



6-DOF computations



6-DOF computations



Conclusions



- An analytical model was developed to predict the angular motion of a projectile subjected to impulse thrusters
- The analytical model predicts the projectile's angular motion very well
- A procedure to properly paired impulses in order to minimize the drag while maximizing the lateral velocity was developed
- The gain in lateral velocity obtained from the induced angular motion is significant

