

Heterogeneous System-on-a-Chip Design for Self-Powered Wireless Sensor Networks in Non-Benign Environments

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| 14. ABSTRACT A new dimension of system architecture design is emerging where hundreds to thousands of ultralight (<10g) sensor nodes will collectively perform a spectrum of wireless sensor network missions in a distributed fashion. To support this architecture, high volume production of sensor nodes at low cost is required. This proposed basic research project is aimed at the development of a technique to design and fabricate self-powered wireless sensor nodes monolithically as a system-on-a-chip (SoC) with a commercially available complementary metal-on-silicon (CMOS) very large scale integration (VLSI) process. Two essential technologies specifically targeted at non-benign environments have been investigated and reported on: integrated solar cells in CMOS and radiation hardening by design of asynchronous logic. A first-ever design for integrated solar cells in commercial CMOS is presented. Two prototype designs have been designed, fabricated, and tested. The average efficiency of the first prototype is 2.4%, compared to an estimated, but unverified 1% from previous work. The actual efficiency of the junction is 8.3%, without considering the metallization overhead. An improved design demonstrates 3.44% efficiency, a 40% improvement. The junction efficiency alone is 11.3%. However, power from these first two prototypes cannot be harnessed properly in the current implementation. A final design, overcoming this limitation, has been submitted for fabrication and will be reported in a later publication. This novel development has potential widespread application to a rapidly growing number of solar self-powered SoC designs of any type. The application of radiation hardening by design (RHBD) to asynchronous logic is suggested as a unique approach for bare die SoC implementations in hostile environments. A case study is presented using a common microcontroller design. Starting with a common synchronous microcontroller design implemented with commercial logic gates, the application of RHBD results in an expected 200% core area increase and requires 160% more energy. The most significant result is that the application of asynchronous design reduced the energy penalty to 85% (from 160%) for a 6% area increase with no performance impact. Additionally, electromagnetic interference is greatly reduced. This approach provides environmental tolerance to radiation and temperature extremes. A suggested next step would be the monolithic integration of the developed solar cells and microcontroller with a single-chip radio design and simple sensor. The focus of this work would be to minimize or eliminate the traditional external components and establish self-powered wireless interconnectivity. To date this complete monolithic approach has not been demonstrated in the literature and would make a great impact on a number of technology applications. | | | | | |
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Summary

A new dimension of system architecture design is emerging where hundreds to thousands of ultra-light (<10g) sensor nodes will collectively perform a spectrum of wireless sensor network missions in a distributed fashion. To support this architecture, high volume production of sensor nodes at low cost is required. This proposed basic research project is aimed at the development of a technique to design and fabricate self-powered wireless sensor nodes monolithically as a system-on-a-chip (SoC) with a commercially available complementary metal-on-silicon (CMOS) very large scale integration (VLSI) process. Two essential technologies specifically targeted at non-benign environments have been investigated and reported on: integrated solar cells in CMOS and radiation hardening by design of asynchronous logic.

A first-ever design for integrated solar cells in commercial CMOS is presented. Two prototype designs have been designed, fabricated, and tested. The average efficiency of the first prototype is 2.4%, compared to an estimated, but unverified 1% from previous work. The actual efficiency of the junction is 8.3%, without considering the metallization overhead. An improved design demonstrates 3.44% efficiency, a 40% improvement. The junction efficiency alone is 11.3%. However, power from these first two prototypes cannot be harnessed properly in the current implementation. A final design, overcoming this limitation, has been submitted for fabrication and will be reported in a later publication. This novel development has potential widespread application to a rapidly growing number of solar self-powered SoC designs of any type.

The application of radiation hardening by design (RHBD) to asynchronous logic is suggested as a unique approach for bare die SoC implementations in hostile environments. A case study is presented using a common microcontroller design. Starting with a common synchronous microcontroller design implemented with commercial logic gates, the application of RHBD results in an expected 200% core area increase and requires 160% more energy. The most significant result is that the application of asynchronous design reduced the energy penalty to 85% (from 160%) for a 6% area increase with no performance impact. Additionally, electromagnetic interference is greatly reduced. This approach provides environmental tolerance to radiation and temperature extremes.

A suggested next step would be the monolithic integration of the developed solar cells and microcontroller with a single-chip radio design and simple sensor. The focus of this work would be to minimize or eliminate the traditional external components and establish self-powered wireless interconnectivity. To date this complete monolithic approach has not been demonstrated in the literature and would make a great impact on a number of technology applications.

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Related Publications

The incremental results of this work are reported in the following publications. The first two publications are not sponsored, as they are before the contract start date.

1. D. J. Barnhart, T. Vladimirova, and M. N. Sweeting, "Satellite-on-a-Chip: A Feasibility Study," in *Proc. Fifth Round Table on Micro/Nano Technologies for Space Workshop*, Nordwijk, The Netherlands, 2005, pp. 728-735. <https://escies.org/GetFile?rsrcid=1698>
2. D. J. Barnhart, T. Vladimirova, and M. N. Sweeting, "Satellite-on-a-Chip Development for Future Distributed Space Missions," in *Proc. CANEUS Micro-Nano Technologies for Aerospace Applications Conf.*, Toulouse, France, 2006, Paper CANEUS 2006-11045. <http://store.asme.org/>
3. D. J. Barnhart, T. Vladimirova, and M. N. Sweeting, "System-on-a-Chip Design of Self-Powered Wireless Sensor Nodes for Hostile Environments," in *Proc. IEEE Aerospace Conf.*, Bozeman, MT, 2007, Paper 7.05.01. <http://ieeexplore.ieee.org/>
4. D. J. Barnhart, T. Vladimirova, M. N. Sweeting, R. L. Balthazor, L. C. Enloe, L. H. Krause, T. J. Lawrence, M. G. Mcharg, J. C. Lyke, J. J. White, A. M. Baker, "Enabling Space Sensor Networks with PCBSat," in *Proc. USU/AIAA Small Satellite Conf.*, Logan, UT, 2007, Paper SSC07-IV-4. <http://www.smallsat.org>
5. D. J. Barnhart, T. Vladimirova, and M. N. Sweeting, "Very Small Satellite Design for Distributed Space Missions," *AIAA Journal of Spacecraft and Rockets*, vol. 44, no. 6, Nov-Dec 2007, pp. 1294-1306. <http://www.aiaa.org/>
6. D. J. Barnhart, T. Vladimirova, M. N. Sweeting, "Design of Self-Powered Wireless System-on-a-Chip Sensor Nodes for Hostile Environments," in *Proc. IEEE International Symp. on Circuits and Systems*, Seattle, WA, 2008, Paper 1869. <http://ieeexplore.ieee.org/>

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1 Introduction

This report satisfies the final requirement of EOARD Grant FA8655-06-1-3053. The work presented in the report was carried out from the 15th September 2006 to 29th February 2008 (17.5 months). A 5.5-month extension (15th September to 29th February) was granted at no cost due to chip manufacturing delays.

1.1 Scope and Objectives

A new dimension of system architecture design is emerging where hundreds to thousands of ultra-light (<10g) sensor nodes will collectively perform a spectrum of remote sensing missions in a distributed fashion. To support this architecture, high volume production of sensor nodes at low cost is required.

This basic research project is aimed at the development of a technique to design and fabricate self-powered wireless sensor nodes monolithically with commercially available complementary metal-on-silicon (CMOS) technology. The goal is to realize a novel system-on-a-chip (SoC) component integration on a single silicon die. Until now, integration of optical, radio frequency (RF), solar power, and data handling technologies have necessitated the use of other system-level integration approaches such as system-in-package (SiP), multi-chip module (MCM), wafer-scale integration (WSI). These approaches have been used in the DARPA-sponsored “Smart Dust” effort. A feasibility study, using the space application as an example, had already been completed showing great promise for the project [6]. The feasibility study highlighted that optical sensors, solar power, wireless communication, and data processing can conceivably be integrated on one CMOS die. This preliminary work is directly fed into the work described here.

Defense interests parallel academia, where this technology could be potentially used to support a variety of military missions. New terrestrial, atmospheric, and space-based missions have been envisioned for distributed remote sensing networks. Potential missions include: signals intelligence, environment monitoring, close inspection, and numerous other envisioned and yet-to-be envisioned applications.

Finally, the novelty of this work clearly stands out in the literature—no one has ever before integrated a sensor technology with wireless communication, data processing, and solar/self-powered technology integrated on one CMOS die. This technology will meet the demand created by the recent explosion of distributed mission proposals over the past decade—new aerospace

applications alone have increased by 850% over the last decade. However, no system architecture exists yet to support them.

Objectives:

1. Develop key technologies suitable for mass-production of integrated heterogeneous self-powered SoC sensor nodes to enable distributed missions for non-benign environments.
2. Experimentally verify selected key emerging technologies on commercial bulk-CMOS to demonstrate the approach via manufacture and testing of VLSI circuits.

1.2 Project Milestones

The project milestones are listed below:

| | |
|-------------------|---|
| 15 September 2006 | Contract Awarded |
| 3-10 March 2007 | IEEE Aerospace Conference (1 st published & presented paper) |
| 5 March 2007 | Test VLSI Chip #1 submission for manufacture |
| 21 March 2007 | Submission to EOARD of the Window on Science Trip Report |
| 21 March 2007 | Submission to EOARD of the Interim report |
| 13-17 August 2007 | USU/AIAA Small Satellite Conference (2 nd published & presented paper) |
| 11 September 2007 | Test VLSI Chip #2 and #3 submission for manufacture |
| 1 November 2007 | AIAA Journal of Spacecraft Rockets (3 rd published paper) |
| 19 November 2007 | Test VLSI Chip #4 submission for manufacture |
| 8 February 2008 | IEEE Int. Symposium on Circuits and Systems (4 th paper accepted for publication, presentation is in May 2008) |
| 3 March 2008 | Contract effort completion and submission of the final report to EOARD |
| 3 March 2008 | Test VLSI Chip #5 submission for manufacture |

1.3 Schedule of Test VLSI Chip Fabrication

Originally, two test VLSI chips were proposed for fabrication. Ultimately, five chips were fabricated as shown in Table 1. Results from these test chips are reported on in Chapters 3 and 4.

Table 1. Test Chip Fabrication

| Test Chip | Date | Size (mm ²) | Purpose |
|-----------|-----------|-------------------------|--|
| 1 | 5 Mar 07 | 2 | Solar cells (1 st design) |
| 2 | 11 Sep 07 | 3.5 | Synchronous/commercial cells + test structures |
| 3 | 11 Sep 07 | 3.5 | Synchronous/hardened cells |
| 4 | 19 Nov 07 | 7 | Asynchronous/hardened cells + solar cells (2 nd design) |
| 5 | 3 Mar 08 | 2 | Solar cells (final design) |

1.4 Outline of Report Contents

Chapter Two, *Mission Needs Statement*, first presents the compelling need for this research followed by a brief literature review. The chapter concludes with a technology development roadmap for this and future work.

Chapter Three, *SiGe BiCMOS Solar Cells*, presents the first of two technology focus areas for this research. The novel invention of monolithically integrated solar cells on CMOS is presented, complete with the basic theory, design, and test results.

Chapter Four, *Synergy of Radiation Hardening by Design of Asynchronous Logic*, presents the second focus of this research. A case study of harmoniously combining two existing technologies is presented, which results in CMOS designs that are tolerant to radiation and temperature extremes, in addition to voltage and process parameter fluctuations. The case study gives the background of these two technologies, describes the core design for comparison, then presents the simulation and hardware test results.

Chapter Five, *Conclusions*, summarizes the results of the research and proposes directions for future work.

2 Mission Needs Statement

2.1 The Satellite on-a-Chip Concept

The satellite-on-a-chip idea has sparked a lot of interest in the space community, since the first known mention of the concept in the 1993-1994 timeframe [1]-[5]. An initial satellite-on-a-chip feasibility study was completed, based on a monolithic system-on-a-chip (SoC) design, but the lack of viable applications discouraged further development initially [6]. In response, the future need for low-cost mass-producible very small satellites for distributed space missions was examined [7].

The smallest silicon-based mass-producible technique for satellite fabrication was proposed by Janson and Helvajian from the Aerospace Corporation in starting in 1993 [1]-[4]. Well beyond the scope of SoC, the vision was to build satellites out of stacks of silicon wafers processed by complementary metal-on-silicon (CMOS), microelectromechanical system (MEMS), and photovoltaic foundries. Their team has since pioneered a range of small satellite manufacturing technologies [8]. The high cost of commercializing these processes has prevented widespread implementation.

The Surrey Space Centre set a long-term goal in 1999 of developing and flying the world's first satellite-on-a-chip, based on a true stand-alone SoC approach. Since that time, they have facilitated numerous research efforts towards that goal. The monolithic SoC approach has been challenged by various packaging alternatives, including traditional printed circuit board (PCB), multichip module (MCM), system-in-package (SiP), and now system-on-package (SOP); however, SoC's attraction is its low cost and mass-producibility.

Related prototyping design activities have been undertaken, targeting a system mass less than one kilogram, leading to a 70 g satellite-on-a-PCB prototype. This very small satellite design, named PCBSat, has given insight into various aspects of satellite system development on a very small scale. Although developed as a prototype, it gives rise to a promising cost-effective mass-producible solution for certain large-scale distributed space missions [9].

Satellite-on-a-chip has gained new appeal in the context of space sensor networks [10]. Nearly all wireless sensor network applications to date have been for relatively benign terrestrial environments, with a few exceptions where thermal extremes are concerned.

2.2 Wireless Sensor Networks

The wireless sensor network concept emerged in the early 1990's, with academic roots that can be traced through an original group of researchers at the University of California, Los Angeles [11]. Various terms have been used to describe this concept over the past decade, but “wireless sensor networks” has endured. In addition to developing the theory and supporting software, three hardware solutions for sensor nodes, sometimes called motes, were initially pursued: Smart Dust, commercial off-the-shelf (COTS) Dust, and Wireless Integrated Sensor Networks (WINS).

Although the actual idea of Smart Dust was born at a 1992 U.S. military workshop, Pister [12] is credited with coining the phrase and the first major development, shortly after leaving UCLA for Berkeley. The first Smart Dust implementation was a battery-powered MCM featuring a MEMS corner cube reflector for optical communications [13]. Pister's team went on to demonstrate a solar-powered variant soon after, as shown in Figure 1 [14].

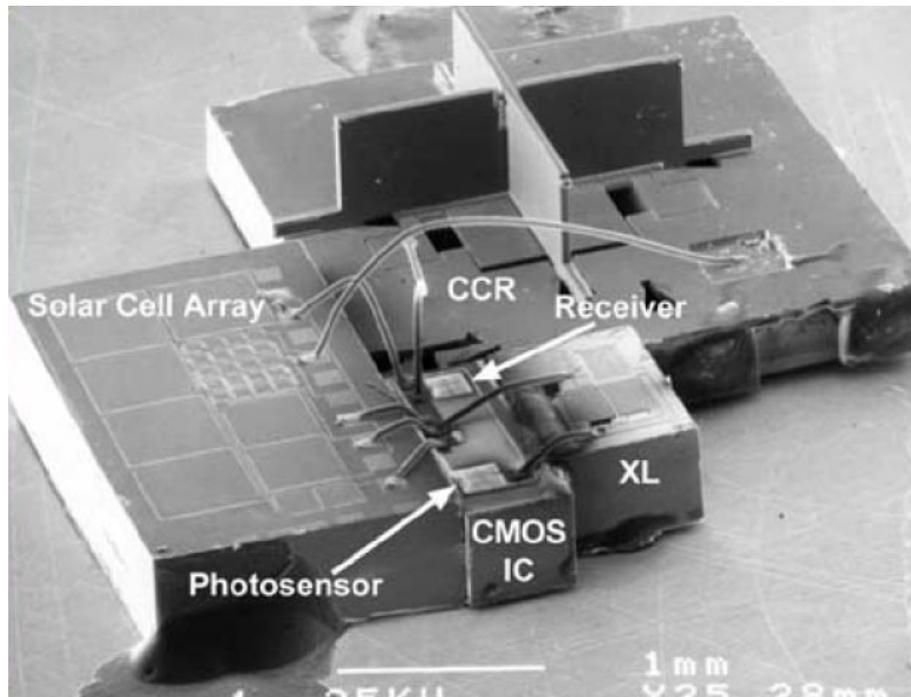


Figure 1. Smart Dust

The new Berkeley team developed COTS Dust in parallel to Smart Dust. As shown in Figure 2, this concept was based on a PCB substrate, with three versions utilizing radio frequency (RF) communications while one used optical [15]. Spin-off companies emerged, such as Crossbow, which now sell the popular MICA family of motes. To simplify their implementation, the TinyOS operating system is now widely used in most motes.

While Smart Dust was in development, four of the original UCLA academics, led by Kaiser [16], pursued an RF-based SoC called WINS. Upon closer inspection, their approach was actually

based on MCM integration of a sensor, microprocessor, and transceiver; which is similar to optical Smart Dust, but uses an RF link. Kaiser went on to lead further integrated RF work in CMOS; however, no recent work on WINS has been published in the literature.

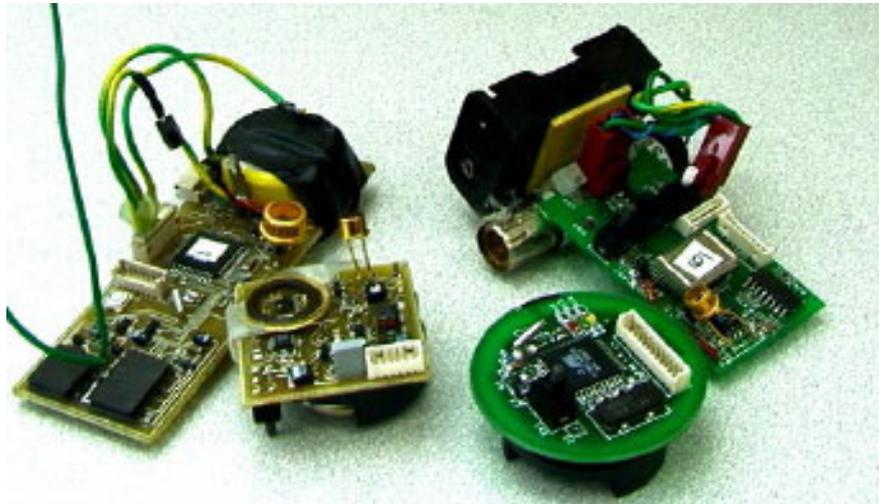


Figure 2. COTS Dust

One of the most promising SoC projects is WiseNET, which has successfully integrated a radio, microprocessor, data storage, power control, and analog interface, as shown in Figure 3 [17]. Although closer to a true SoC solution, the WiseNET sensor node still requires numerous external components, including a power source, passive devices, an antenna, and sensor.

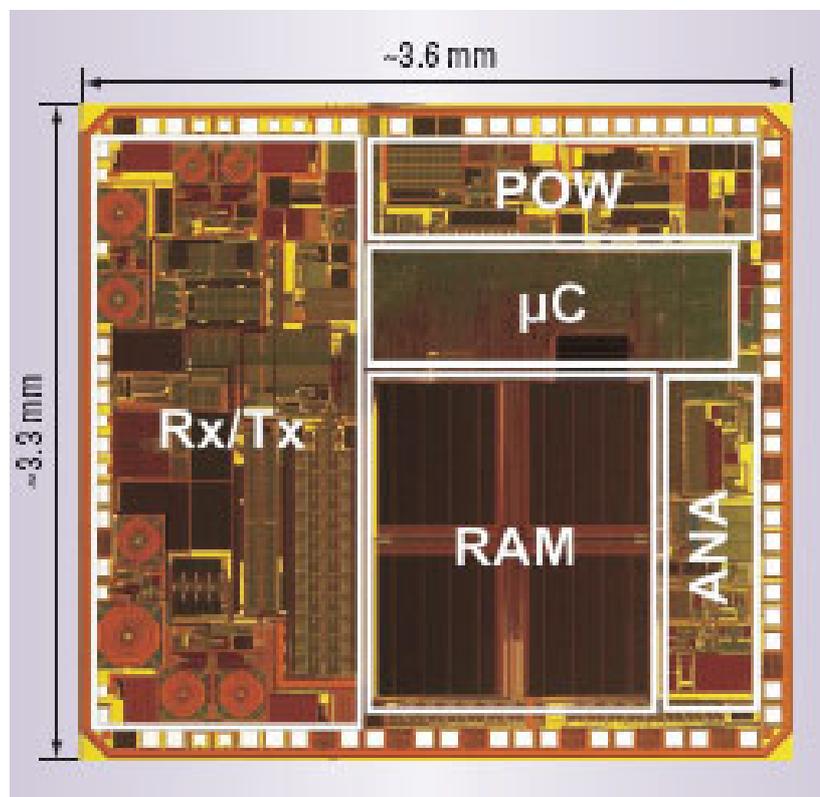


Figure 3. WiseNET SoC Sensor Node

In response to WiseNET, The Smart Dust team recently published a comprehensive investigation of an RF-based SoC approach [18]. It includes a discussion on the remaining work to realize a complete stand-alone SoC implementation. They concluded that although recent SoC solutions have demonstrated increased monolithic integration, many large off-chip components are still required, such as a sensor, battery, passives, crystal clock source, and RF antenna. Completed during the same period, our initial satellite-on-a-chip feasibility assessment, with similar objectives, arrived at the same conclusions [6].

2.3 Survivable SoC Node Requirements

In this section, we discuss functional requirements for a self-powered SoC sensor node design aimed at hostile environments. A range of potential solutions for a generalized set of functional requirements is presented, focusing on the following aspects:

- Missions and sensors
- System configuration
- Power generation, storage, distribution, and control
- Data handling, processing, and storage
- Wireless communications with other nodes
- Environmental operability and survivability

2.3.1 Missions and Sensors

The chosen SoC approach greatly limits payload options. Considering a case study mission presented in [19], no on-chip sensors are possible to detect plasma, due to the physical geometries required. However, the following sensors are routinely manufactured in CMOS [20]:

Table 2 Typical CMOS Sensors

| | | | |
|-------------|---------------|---------------|------------|
| • Visible | • Infrared | • Ultraviolet | • Magnetic |
| • Radiation | • Temperature | | |

CMOS imagers are growing in popularity and may eventually replace charge-coupled devices (CCD) for most imaging applications [21]. Unlike CCDs, CMOS imagers use mainstream semiconductor fabrication techniques, require less power, and can be integrated monolithically with image co-processors. Complete camera-on-a-chip devices are now emerging [21]. Typically, a separate lens is required to focus the image on the sensor, but microlenses can now be integrated monolithically [22].

Recently, a wide range of sensors has emerged, based on “CMOS-MEMS” technology. CMOS-MEMS requires custom pre-, front-end, and/or back-end processing of the CMOS wafer. Of these three methods, back-end bulk micromachining of CMOS has been the most successful. Due to its growing popularity, a few commercial foundries now offer limited CMOS-MEMS processing, such as X-FAB. Table 3 lists some sensors that have been demonstrated [23].

Table 3. Typical CMOS MEMS Sensors

| | | | |
|-------------|----------------|--------------|-----------|
| • Pressure | • Chemical | • Thermal | • Tactile |
| • Proximity | • Flow | • Force | • Neural |
| • Vacuum | • Acceleration | • Gyroscopic | • Audio |

2.3.2 System Configuration

With a goal of a SoC implementation, the configuration is essentially fixed to the planar nature of a silicon chip. CMOS technology is the most widely used microelectronics fabrication technology, due to its low cost at high volume. A maximum-sized prototype integrated circuit (IC) design, using a multi-project vendor such as MOSIS or EURO PRACTICE, start at \$2,400 per die depending on the technology, while a production run would cost about \$300 each. Currently, feature sizes of 45 nm are possible, which will only shrink in time [24]. CMOS technology options have broadened over the past decade with the introduction of processes optimized for radio frequency (RF), optical sensors, integrated bipolar transistors (SiGe BiCMOS), and non-volatile “flash” memory.

The primary advantage of a monolithic approach is the manufacturing simplicity. However, it does not allow the attachment of discrete components or the merging or various elements into a hybrid assembly, which imposes considerable limitations. Most notably, the design cannot exceed the reticle size, which is a physical area limit imposed by the photolithography process used in the particular semiconductor process line. This caps the maximum circuit area to approximately 400 mm² (20×20 mm) for modern CMOS processes [24]. Assuming a silicon density of 2330 kg/m³ and wafer thickness of 0.75 mm, the die mass is approximately one gram.

In 1967, a technique called wafer-scale integration (WSI) was proposed to overcome the reticle limit [25]. WSI allows multiple reticle-sized designs to be co-located on the same wafer, and then connected together using various interconnection techniques. This would allow a final product that in theory could be as large as the entire wafer, which could be as large as 300 mm in diameter [24]. Unfortunately, inherent defects in the semiconductor manufacturing process have prevented WSI from becoming widely adopted [26].

Multichip-Module (MCM) technology eventually replaced WSI for designs requiring more area [26]. MCMs integrate unpackaged “known-good-die” on a range of substrates, such as printed

circuit boards (PCBs), thin films, and ceramics using fine line interconnects. MCM technology, including three-dimensional variants, has already been used in satellite applications [27]. MCMs or other system-in-package (SiP) techniques are typically used in applications where integrated density or performance is essential [28]. For less demanding applications, evolutionary advancements in IC packaging make traditional PCBs a cost-effective choice.

Despite a growing number of packaging alternatives, SoC technology is rapidly advancing. Popular MCM-based miniaturization efforts, such as “Smart Dust,” are now looking to SoC for further miniaturization of their terrestrial wireless sensors [18].

2.3.3 Power Generation, Storage, Distribution, and Control

Power distribution, regulation, and control aspects of an EPS can be met with basic wiring, switching, and regulation circuitry that are routinely implemented in CMOS [29]. Recent “micro power” research has presented several new integrated options for SoC applications, presented in Table 4 [30].

Table 4. Micro Power Sources

| | | | |
|--------------------|-------------------|----------------|-------------|
| • Solar cells | • Fuel Cell | • Vibration | • Induction |
| • Chemical battery | • Nuclear battery | • Microturbine | |

Power generation via integrated solar cells on CMOS is the most straightforward solution, but has not yet been demonstrated successfully. Typically, solar cells are fabricated with optimized silicon (Si) or gallium arsenide (GaAs) processes, optimized for efficiency and distinctly different from commercial CMOS. Integrating solar power with digital circuitry has not been of interest until recently. The first Smart Dust prototype was implemented as a MCM and attached to an external battery [13], then later used MCM integration to incorporate solar cells [14], and finally demonstrated a monolithic solution using a custom silicon-on-insulator (SOI) process [31]. Although SOI is growing in popularity, it is not yet widely available.

Truly monolithic self-powered devices in CMOS have been proposed. Three such examples are a sensor network processor [32], an implantable device to cure human blindness [33], and other basic research [34]. Unfortunately, none of these efforts reported any success in hardware, including efficiency results. In private correspondence with Blaauw [32], it was revealed that their efficiency was less than 1%. Castañer explains that the CMOS process imposes some restrictions that drastically reduce the efficiency of solar cells. His approach is similar to other efforts, using advanced packaging techniques to create self-powered SiP designs [35]. Obviously, with a maximum efficiency of 1%, commercial CMOS presents a challenge. A novel solar cell design is presented later in this report.

A monolithically integrated chemical fuel cell has been demonstrated with an operating time of 170 hours and mean open-circuit voltage of 0.533V [36]. Unfortunately, it relies on an oxygen-rich atmosphere, which is not suitable for space, but will work terrestrially. In addition, no under load performance data is presented. Other micro chemical power supplies, such as thin-film batteries [37], nuclear batteries, and microturbines have been investigated, but none can be monolithically integrated.

Mechanical energy is typically converted by electromechanical generators, but piezoelectric power generation is also possible. Work is underway in piezoelectric micro power sources, but not yet for SoC [38]. Another promising source of integrated electrical power is through inductive energy transfer. This has been shown in a monolithic SoC for medical implants [39].

2.3.4 Data Handling, Processing, and Storage

The DH subsystem provides a range of on-board computing services. It receives, validates, decodes, and distributes commands from the ground, payload, or a subsystem to other spacecraft subsystems. It also gathers, processes, and formats spacecraft housekeeping and mission data for downlink or use on board. DH subsystems are usually the most difficult to define early in the design due to the initially vague requirements of the payload and subsystems.

At a minimum, the DH subsystem is composed of a central processing unit (CPU) and supporting memory elements. The difficult part of the design is the hardware interface to the other systems, typically using a digital data bus and analog-to-digital converters (ADC). For monolithic sensor nodes, a minimal reduced instruction set (RISC) CPU design is all that can be supported by the available power. An on-chip ring oscillator with selectable frequency output and power up reset can be used to run the CPU. Some introductory thought has already been given to miniaturizing flight computer components to a single chip, reflecting a growing trend in SoC development [40].

One issue that plagues data handling systems operating in space is the extreme radiation and thermal environment, especially considering that the proposed system architecture is a bare die in space with no shielding. Additionally, low power operation is essential, considering the small surface area for integrated solar cells as discussed. A unique solution for this issue is the second focus of this research effort and is presented in the next section. It combines two atypical design techniques to enable low-power operation in hostile environments.

2.3.5 Wireless Communications with Other Nodes

The original Smart Dust design presented in [13] used optical communications to take advantage of its power efficiency. Optical links are also free of regulatory issues and can use simple on/off keying (OOK) modulation schemes. This approach is only effective in line-of-sight situations

where the alignment is controlled. For sensor networks within a larger spacecraft, line of sight would be difficult. For free-flying nodes, the alignment problem becomes the predominant issue.

Low-power on-chip transceivers have become the preferred choice for sensor nodes. SoC transceivers, which were a novelty only a few years ago are now commercially available, some even with an integrated microcontroller [41]. The commercial availability of RF CMOS and SiGe BiCMOS processes has offered increased capabilities, including a wider selection of operating frequencies. SoC transceivers still require external passive elements, crystal oscillators, and an antenna. In an effort to eliminate external antennas, on-chip antennas have been investigated. The maximum range achieved is approximately five meters, as demonstrated by Lin [42] and O [43]. Due to a 20×20 mm reticle size, most experiments use frequencies over 3.75 MHz, which gives a quarter-wavelength antenna size of 20 mm or smaller. On-chip antennas for the 900 MHz and 2.4 GHz Instrumentation, Scientific, and Medical (ISM) bands are not feasible as they are 12.5 cm and 3.1 cm respectively. 5.8 GHz ISM fits at 1.3 cm. Unfortunately, higher frequencies require more power given the same desired range than lower frequencies.

Another technology related to wireless sensor networks is RFID. The basic concept was explained in 1948 and arguably was envisaged before this time [44]. This technology was not used much until the 1970s, when it saw some widespread use in automated vehicle identification for various purposes, such as road tolls. Technology has allowed miniaturization to the point where RFID “tags” can be made monolithically, including an antenna, with a range of a few meters [45].

2.3.6 Environmental Operability and Survivability

Emerging wireless sensor network applications for hostile environments has prompted an investigation into survivable sensor node design techniques, which currently do not exist. The following five environmental hazard categories are introduced and discussed further.

- (1) Mechanical (shock, vibration, acceleration)
- (2) Atmospheric (corrosion, debris, vacuum)
- (3) Thermal (extremes, limited heat transfer)
- (4) Energetic (radiation, including charged particles)
- (5) Dynamic (free-fall orbit, high velocity mobility, attitude disturbance torques)

Mechanical (shock, vibration, acceleration)—Fragile MEMS structures are not suitable for applications where excessive shock, vibration, and/or acceleration may exist. These hazards are seen in the space launch segment and industrial process plants. The mechanical rigidity of a monolithic SoC is far superior in this case.

Atmospheric (corrosion, debris, vacuum)—Corrosion is an issue for low-Earth orbit (LEO), industrial/chemical, and biomedical applications. Any exposed aluminum on a SoC must be covered, either by gold plating or by passivation. Space debris is normally considered a hazard for satellites, but for a mission where thousands of objects are put in space, they become a big concern to other systems. The only realistic way to solve this problem is to confine these missions to LEO, where the orbital lifetime is very short, essentially making these missions disposable. Finally, the vacuum of space introduces several issues, such as cold welding and outgassing, but for SoC, the only concern is limited heat transfer.

Thermal (extremes, limited heat transfer)—Thermal extremes and cycling are exacerbated in a vacuum, as thermal radiation is the only method available for heat transfer. Silicon wafer thermal properties are well understood and certain packaging materials can be used to manage the temperature extremes for a SoC. For example, space-qualified paraffin can be used to absorb heat during the sunlit portion of an orbit, and then keep the system warm during eclipse, effectively narrowing the temperature range the SoC will experience.

Energetic (radiation, including charged particles)—Extreme radiation conditions are usually experienced in nuclear power plants, certain industrial process plants, and in space. Ionizing radiation causes gradual system degradation as the dose accumulates. In addition, high-energy particles, such as electrons, protons, and heavy ions, can cause single event phenomena. This environment is discussed in detail in Chapter 4.

Dynamic (free-fall orbit, high velocity mobility, attitude disturbance torques)—Terrestrial sensor networks are composed of relatively fixed nodes. In contrast, orbital velocity in low Earth orbit (LEO) is approximately 7.5 km/s. Natural, but undesirable perturbations change the orbit over time, altering the arrangement of nodes, which is called a constellation. This factor must be fully understood, so key parameters like communication range can be selected properly. The freefall environment also presents unique challenges. The dominant effect is that objects in orbit “float” and change their orientation or “attitude” based on perturbations from solar pressure, gravity gradients, magnetic fields, and aerodynamic drag. This may not be an issue if the sensor technology does not have pointing requirements. However, if attitude control is required, SoC solutions are very challenging at this scale.

2.4 Analysis

The ultimate SoC vision for any application is a stand-alone product that can be used directly off the CMOS process line without any additional components, packaging, or interfaces. A survivable SoC has additional features and functional requirements as just outlined. Based on our experience with very small satellite design we have identified the following research areas that are worth

pursuing further in order to realize a true SoC implementation of a survivable wireless sensor node:

- (1) Sensors
- (2) Power generation and storage
- (3) Asynchronous system architecture
- (4) Transceivers and antennas
- (5) Attitude control
- (6) Location and time determination
- (7) Propulsion
- (8) Environmental extremes tolerance

Our aim is to not only help achieve the vision of a stand-alone SoC, but to design a system that can withstand hostile environments, particularly those encountered in space missions. A notional system configuration is illustrated in Figure 4.

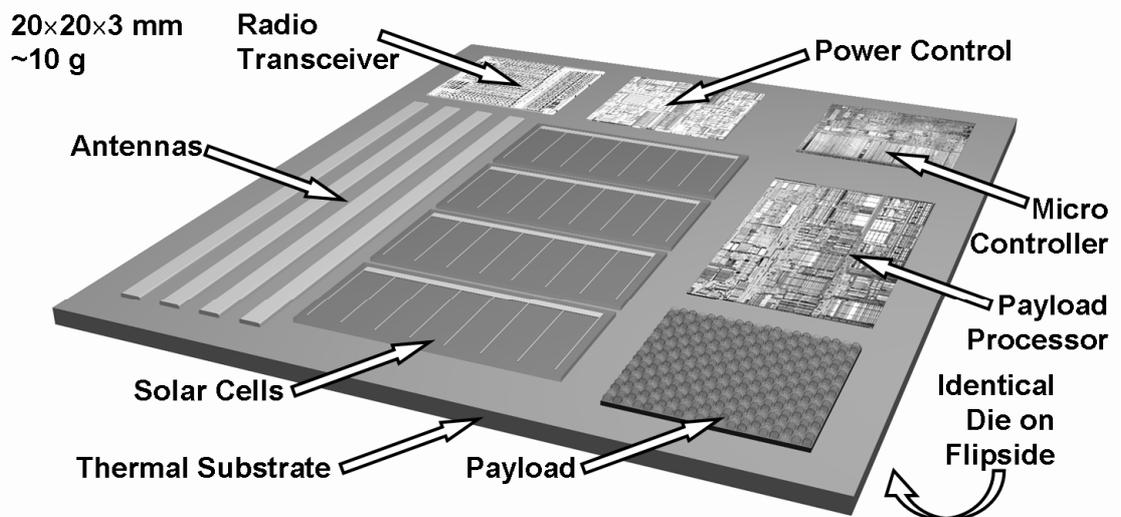


Figure 4. Notional System Configuration

In the above context the focus of this research is on developing and demonstrating two key capabilities:

Power Generation via Solar Cells—Numerous integrated power sources have been studied, but all have remained elusive for commercial CMOS. Integrated solar cells seem to be the most relevant approach. Of the few published attempts, only one has achieved an estimated and unverified efficiency of one percent. If successful, this development could be applied to rapidly growing number of standalone SoC applications. Progress in this area is reported on in Chapter 3.

Radiation Hardening by Design of Asynchronous Logic—Asynchronous design for hostile environments through the application of radiation hardening by design has only been briefly considered previously. Most efforts focus on implementing redundant logic to overcome upsets from radiation sources. The challenge is that the asynchronous power efficiency gain is partially offset by power and area hungry design hardening techniques. A case study on this issue is presented in Chapter 4.

3 SiGe BiCMOS Solar Cells

As discussed in the previous chapter, solar cells are typically fabricated with dedicated silicon or gallium arsenide processes optimized for efficiency, then strung together externally with the appropriate series and parallel connections to achieve the desired voltage and current output. Regarding monolithically integrated cells, CMOS does not provide insulating features, as SOI, which facilitates series connections, but is not widely available. Consequently, monolithic CMOS solar cell research is limited to a few attempts with unreported results [32]-[34], apart of an estimated efficiency of 1% [32]. A novel approach to monolithic solar cell design in CMOS is presented here, which aims to overcome the limitations of previous implementations. This technology development can be applied to a rapidly growing number of SoC applications.

3.1 Basic Solar Cell Theory of Operation

Solar or *photovoltaic* cells are devices that convert light energy or *photons* into electric current. Although modern day solar cells are derived from semiconductor technology made popular by the invention of the transistor in 1947, crude photovoltaic cells have been in use before 1900. The basis of a modern photovoltaic cell is the *p-n junction* of a crystalline semiconductor material, such as Germanium (Ge), Silicon (Si), Gallium Arsenide (GaAs), or numerous other compounds.

In silicon for example, the p and n regions are created by introducing dopant materials, such as boron (B) or phosphorous (P), respectively. Boron has one less valence electron than Silicon, so its introduction in the crystal lattice creates an absence of an electron, called a hole (+). Similarly, Phosphorous has one more valence electron than Silicon, creating an excess electron (-). The p-n junction is created from a single crystal. Under normal conditions, excess holes from the p-type material migrate to the n-type material while excess electrons in the n-type material migrate to the p-type material, where electron-hole recombination takes place until equilibrium is reached [47]. Under illumination, most of the photon energy is absorbed at the surface of the material, creating excess electron hole pairs, reversing this migration condition as illustrated in Figure 5.

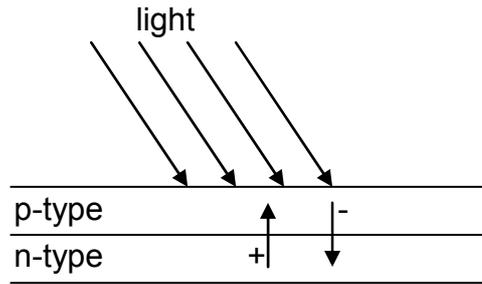


Figure 5. Illuminated p-n Junction Photovoltaic Effect

To harness the photovoltaic energy, an ohmic contact is placed on either side of the p-n junction as shown in Figure 6. The left side of the figure indicates the accepted voltage polarity convention, where the ground (-) probe of the voltmeter is placed on the n-type material and the positive (+) probe is placed on the p-type material. Under illumination, the open circuit voltage is a positive. On the right side of the figure, the short circuit current is illustrated, where the current flow is positive, indicating the flow of holes in the direction shown.

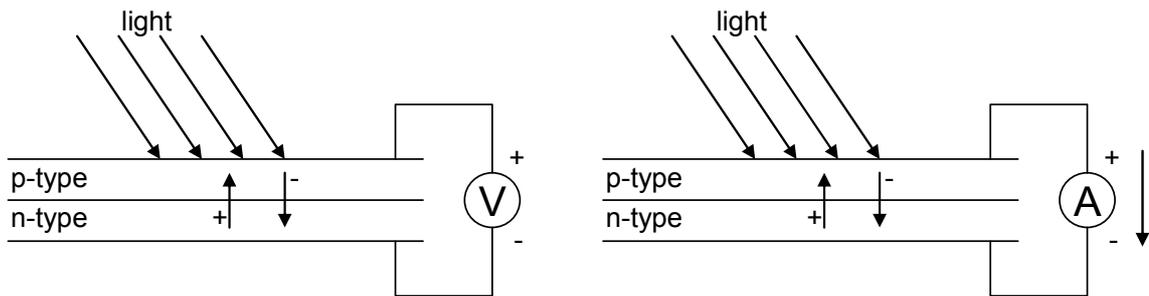


Figure 6. Photovoltaic Voltage and Current Direction Conventions

3.2 Integrated SiGe BiCMOS Solar Cell Design

The 0.35 μm SiGe BiCMOS (S35) process from austriamicrosystems (AMS) is used throughout this work due to its cost effectiveness, lack of light-blocking layers, and support for integrated radio in future work. However, the NPN SiGe bipolar junction transistor (BJT) structure is the primary reason for selecting this technology, as it provides a semi-isolated p-n junction at the surface. Bulk CMOS only supports an n-well based n-p junction, which cannot be isolated for series connections and produces a negative voltage with respect to ground, which is the p-type wafer. This experimental photocell design is investigated and reported on. Not every detail of the AMS process is given due to the academic non-disclosure agreement in force.

The novel photocell design utilizes NPN SiGe large area transistors, which are thin and close to the surface. The standard NPN SiGe BJT structure is modified to maximize the collector-base (CB) interface and minimize the emitter (E) contact area, which is left floating. A conceptual side view drawing (not to scale) is shown in Figure 7.

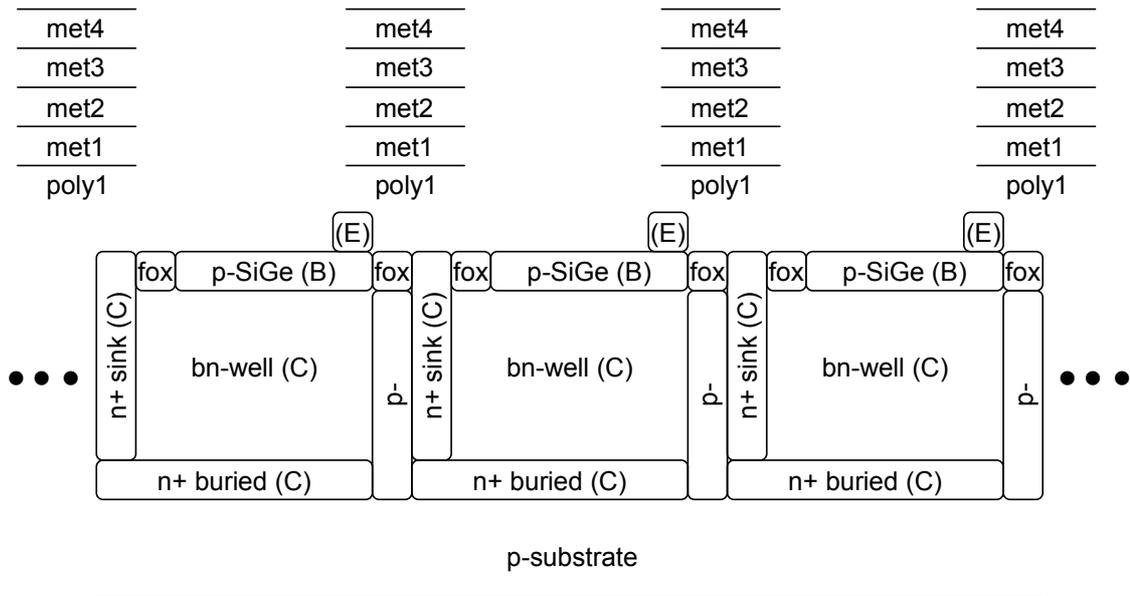


Figure 7. Photocell Design Concept (Side View)

A closer inspection of Figure 7 reveals the essential physical elements of the design. Starting from the bottom, the AMS S35 technology uses a p-type substrate, common to most commercial CMOS processes. To create the collector (C), a n+ sinker and buried layer are required to contact the buried n-well. On top of the collector, the base (B) is formed of a thin p-type SiGe layer, where polysilicon (not shown) is used to make the base contact. Field oxide (fox) insulates the base from the surrounding elements. The emitter (E) is a small amount of n-type material connected by polysilicon (not shown) to create the complete NPN structure. The emitter is left floating and is kept as small as possible to maximize incident light while satisfying the process design rules. Finally, the polysilicon (poly1) through metal layer four (met4) are shown to illustrate that regular placement of these layers is required to satisfy the coverage and slotting rules of the process. Unfortunately, these layers reduce the overall efficiency.

The advantageous placement of field oxide (fox) in the NPN design is what makes series connections possible in SiGe BiCMOS and not bulk CMOS. Making the series and parallel cell connections is straightforward with this single-cell design. As shown in Figure 7, these cells are arranged for a series connection, raising the voltage at each increment. The base (B) of one cell is connected to the neighboring collector (C) through vias to the metal layers above (not shown). Looking at the cell design from the top, Figure 8 illustrates how the field oxide completely isolates the p-type SiGe base (B) from the adjacent material. However, this design is not as efficient as a similar one in SOI, as there is no insulating layer available between the bottom n+ buried layer and the p-substrate as shown in Figure 7. Figure 9 illustrates the physical layout in the Cadence computer aided design (CAD) package, mirroring the view in Figure 8.

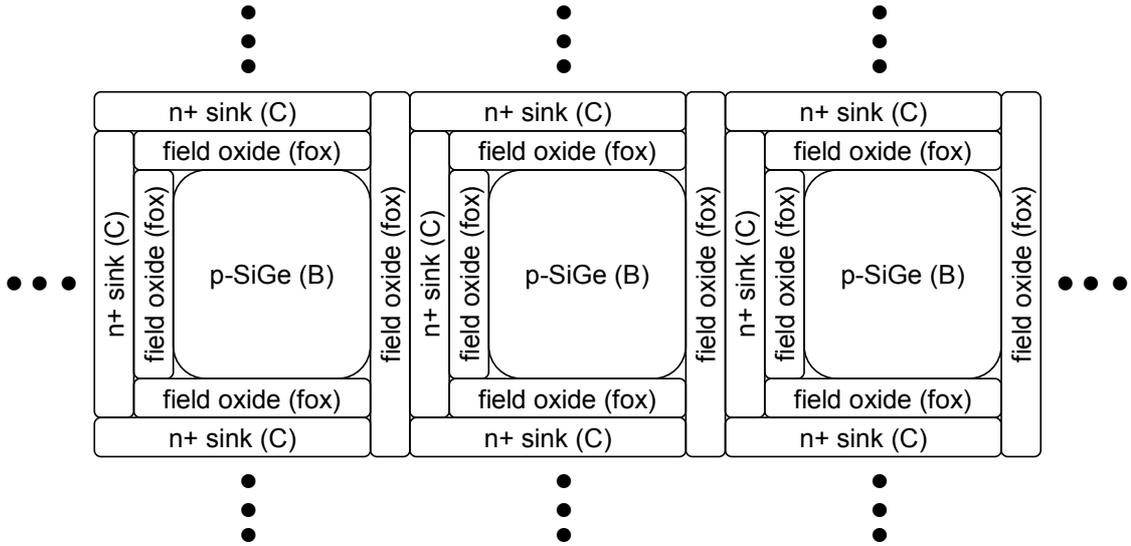


Figure 8. Photocell Design Concept (Top View)

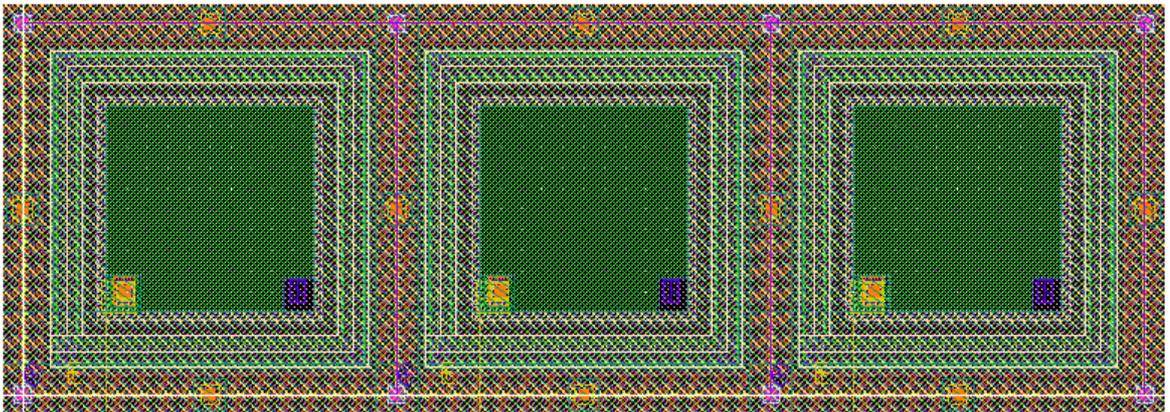


Figure 9. Photocell Design Concept (Cadence Layout View)

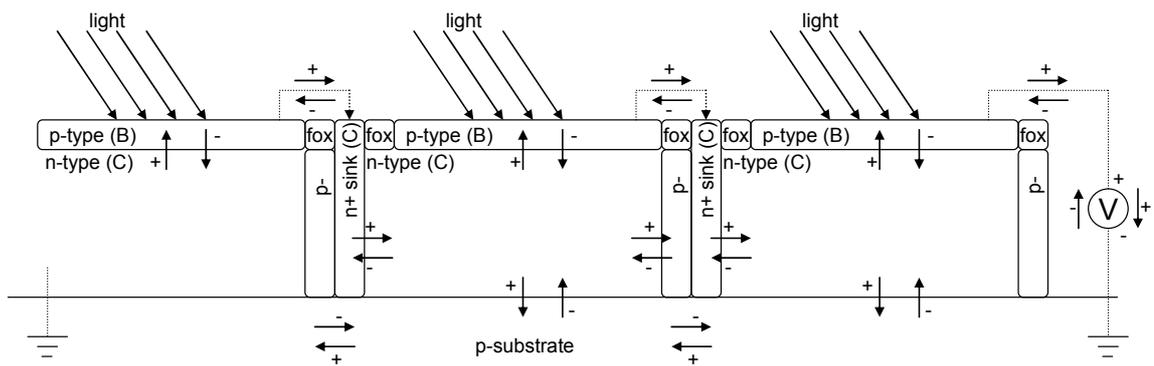


Figure 10. Photocell Design Concept (Schematic View)

Figure 10 is a hybrid view of the layout and schematic. It is essential to understand that while most light is absorbed at the top layer, some penetrates into the material and activates the lower n-p junction at the substrate, as well as the n-p junction of the sidewalls. All electron hole migrations are illustrated, giving the desired positive bias with respect to the substrate.

Figure 11 is the final layout in Cadence of the first test chip from run 1550 and a micrograph of an unpackaged die after fabrication. The die is $1420 \times 1420 \mu\text{m}$ ($1.42 \times 1.42 \text{ mm}$), which is the minimum-sized test chip in the AMS S35 process. Two more experimental designs are reported on in the next section, with a final design still in fabrication at the time of writing.

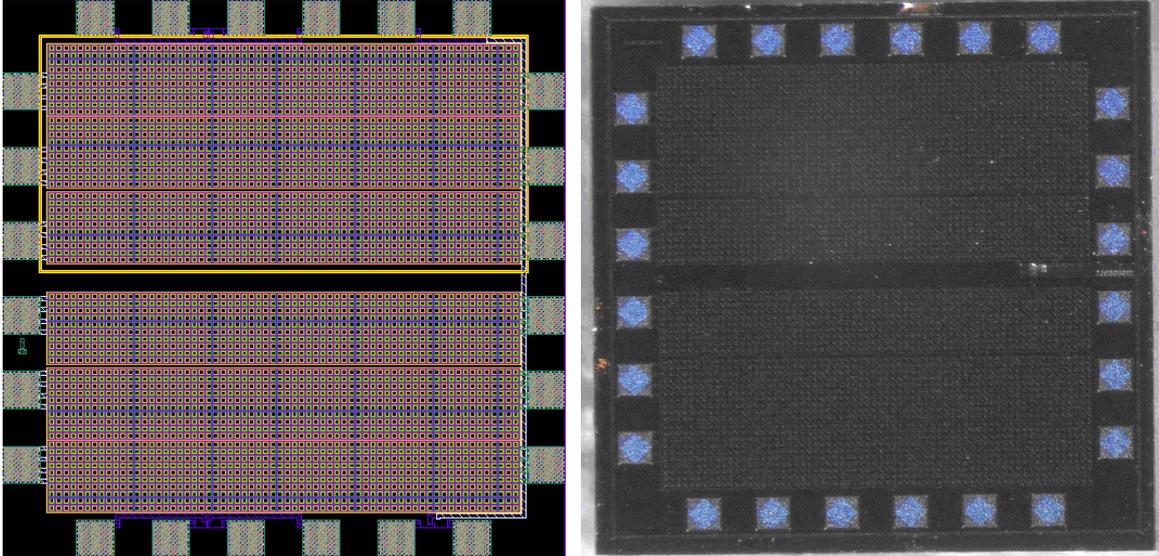


Figure 11. Cadence Layout and Die Micrograph of Solar Cell Test Chip #1

3.3 Integrated SiGe BiCMOS Solar Cell Test Results

Results from three experimental photocell designs are reported on. The first design and test chip is shown in Figure 11. The schematic of the design is similar to that in Figure 10; however, the base (instead of the emitter) is floating on each cell, based on a photocell design given in [21]. Secondly, there are six banks of photocells in parallel, clearly seen in Figure 11, three on the top and three on the bottom, with a large channel in between the sets, and smaller channels within the sets of three. The various channel spacings are to evaluate the isolation qualities. Additionally, the six banks of photocells have all collectors (left) and emitters (right) connected to the adjacent test pads. This allows for selectable external series connections of the cells. The pads at the top and bottom are for transistor test structures and are discussed in section 4.4.

Test chip results reveal that the NPN CB junction is not activated as expected. Upon closer investigation, the reference photocell design [21] is not appropriate for this application, as the BE interface acts as a diode, preventing current from flowing through this interface. However, the test chip allows examination of the underlying n-well to p-substrate junction and its performance, which has some value as efficiency results from this straightforward approach is not reported in the literature. As expected, this junction has a negative bias with respect to the substrate, which prevents direct application of the power from the cells.

Solar cells from AMS S35 run 1550 test chips are subjected to AM0 solar conditions per ASTM E-490 (1366.1 W/m^2). Summary current and power measurements are presented in Figure 12. The average efficiency is an encouraging 2.4%, vs. 1% from previous work [32]. The actual efficiency of the interface is 8.3%, without considering the metallization overhead.

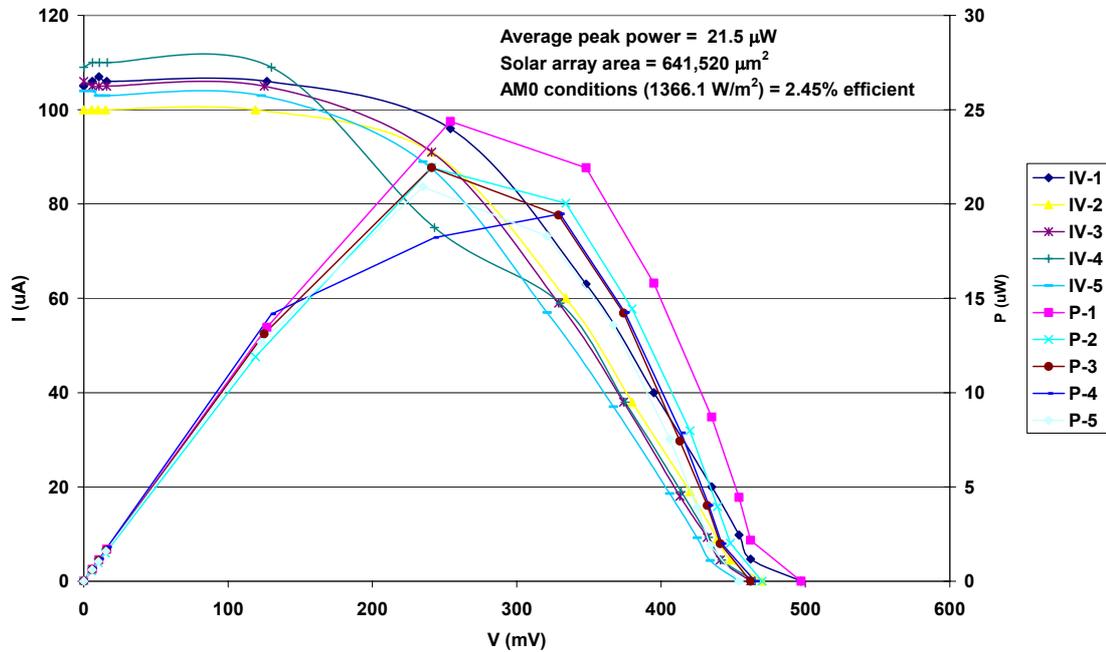


Figure 12. Solar cell current vs. voltage, AM0, test chip 1550

As the cause for the unexpected results was not immediately discovered, further examination of n-well based photocells took place. To potentially improve efficiency, the SiGe layer was removed to allow more light to penetrate down to the lower n-well junction. The improved cells were included with other work on run 1791, discussed in the next section. They demonstrate 3.44% efficiency as shown in Figure 13, which is a 40% improvement over the first attempt. The interface efficiency alone is 11.3% without considering the metallization overhead. Additionally, these experiments confirmed that integrated solar cells can be integrated with CMOS logic with no adverse effects. This is verified in the hardware testing discussed in the next chapter.

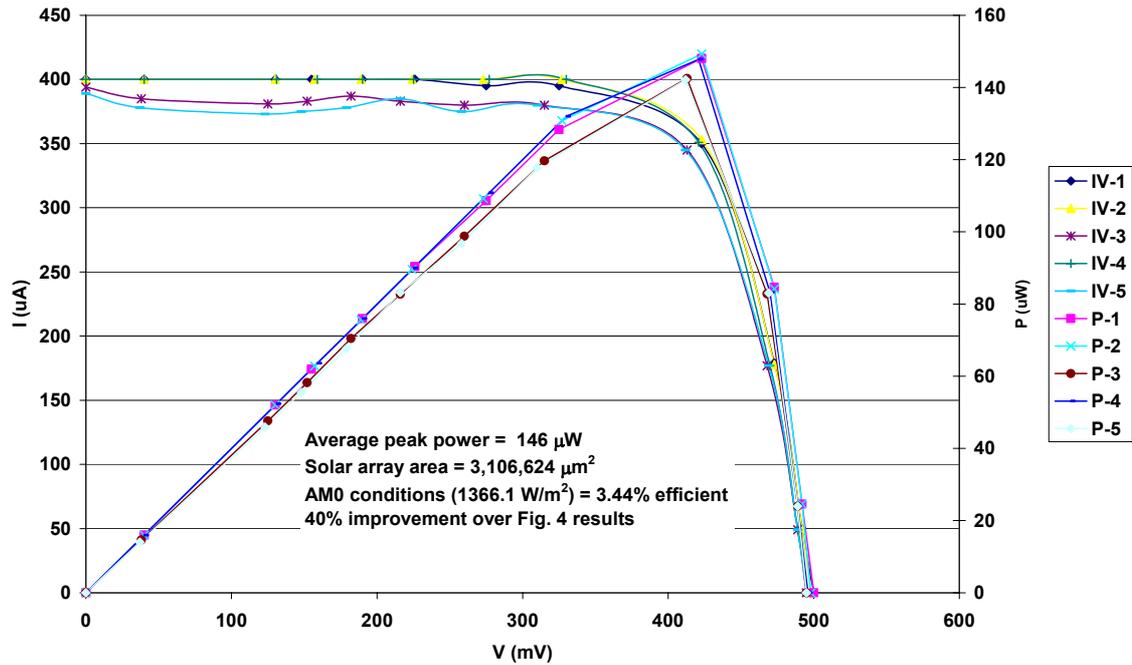


Figure 13. Solar cell current vs. voltage, AM0, test chip 1791

Until the cause of the unexpected results on the first test chip was discovered, an on chip charge pump was considered. Recently, a conventional single-chip (without external passives) charge pump design has been reported for energy harvesting applications using external solar cells [48]. Charge pumps are commonly used on flash memory devices to provide the required higher voltage for write operations. Not only can they boost the voltage considerably above the power supply levels, they can invert the bias. This is particularly interesting in the case of n-well based photocells, as this would allow for both bias inversion and boosting to minimum operating levels. Unfortunately, charge pumps will not work in this situation, as the minimum bias achieved is 0.5V, which is half of the required charge pump start up voltage of 1.0V. Now that the original SiGe BiCMOS design is corrected, a final test chip has been submitted on run 1875, which will provide a positive bias at the process operating voltage of 3.3V. This is the best solution overall, as the associated inefficiency of charge pumps is avoided.

4 Radiation Hardening by Design of Asynchronous Logic

A case study supporting the development of environmentally tolerant logic designs is presented. The synergy of a unique asynchronous/hardened design approach improves the tolerance to radiation, semiconductor processing variations, voltage fluctuations, and temperature extremes. Radiation hardening by design (RHBD) has been recognized for over a decade as an alternative open-source circuit design approach to mitigate a spectrum of high-energy radiation effects, but has significant power and area penalties. Similarly, asynchronous logic design offers potential power savings and performance improvements, with a tradeoff in design complexity and a lesser area penalty. These side effects have prevented wider acceptance of both design approaches.

4.1 Radiation Hardened by Design Background

Extreme radiation conditions are usually experienced in nuclear power plants, some industrial process plants, and in space. Surprisingly, in the early days of IC development, alpha particles from impurities in plastic packaging caused mysterious anomalies in terrestrial systems. Neutrons occasionally cause errors in airplane avionics systems flying at normal cruising altitudes [49]. Space and various nuclear environments are more challenging, where the total ionizing dose (TID) of radiation causes gradual system degradation, resulting in an increase in power consumption. In addition, high-energy particles, such as electrons, protons, and heavy ions/galactic cosmic rays (GCRs), can cause single event effects (SEE), predominantly upset (SEU), latchup (SEL), and recently, transient (SET). Unnatural effects, such as enhanced dose rate, prompt neutron dose, and system electromagnetic pulse (System EMP) are not discussed, as they are only concerns for hardened military systems.

Mitigating these effects has historically been accomplished with a system-level approach, which can become quite expensive. Heavy shielding of various types can be used to reduce TID and System EMP, but is ineffective against SEE. SEE are tolerated and detected, typically through triple (or more) modular redundancy (TMR) or voting schemes. At the IC level, dedicated semiconductor foundries for military purposes only are used to produce hardened components. These hardened foundries are typically several generations behind their commercial counterparts. One open source radiation-hardening solution at the IC level is the application of RHBD, which can be used on any generation process, including the most recent [50]. The guiding principle

behind RHBD is to mitigate as many of the radiation effects as possible by using unconventional layout techniques at the transistor device and circuit level.

Beginning with TID, the degradation mechanisms must first be understood before they can be mitigated. CMOS circuits slowly degrade due to the total accumulated dose of ionizing radiation. This degradation is seen as a negative shift in the transistor threshold voltage and decrease in gain. With enough voltage threshold shift, the circuit will start consuming power even when not switching. The decrease in gain causes the transistors to become harder to switch. After extended exposure to radiation, the circuit will cease to function [51].

The main source of degradation comes from the interaction of ionizing radiation with the gate and field oxides (SiO_2) in the device structure. The gate oxide is a thin high-quality oxide used to insulate the gate contact from the transistor channel. The field oxide is a thick low-quality oxide used to isolate metal traces from one another [49].

Ionizing radiation causes the formation of electron-hole pairs in the gate oxide. Electrons have a much higher mobility than holes in SiO_2 and are attracted to and swept out of the gate in a nMOS transistor. The holes become trapped and migrate toward the transistor channel. This results in the eventual buildup of positive charge above the transistor channel and acts like the charge that is present when voltage is applied at the gate. As more charge is trapped, the voltage threshold of the nMOS transistor becomes more negative, which means it becomes easier to turn on. With enough shift in threshold voltage, the transistor will be turned on without a gate voltage applied. Conversely, a pMOS transistor becomes more difficult to turn on. Figure 14 shows how the gate voltage versus drain current curve changes resulting from exposure to radiation in an nMOS transistor [49].

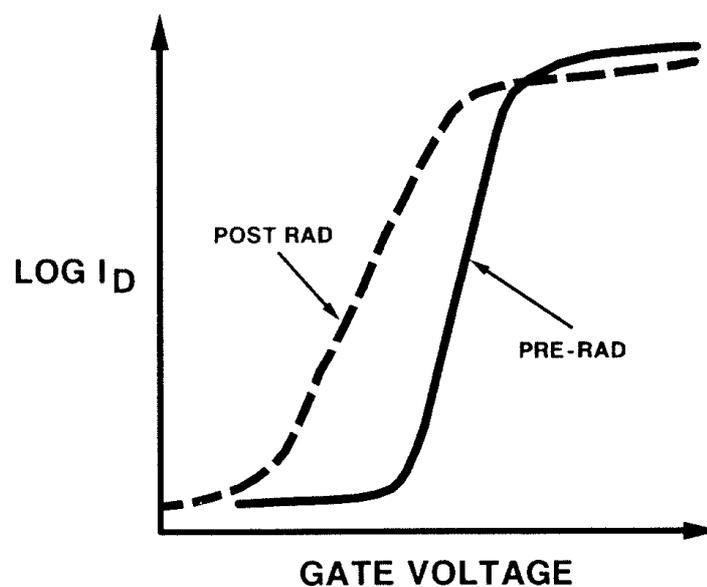


Figure 14. Total Ionizing Dose Effect on nMOS Threshold Shift [49]

The field oxide also traps charge due to ionizing radiation. The trapped positive charge along the edges of the nMOS transistor creates a leakage channel. Leakage paths can also form between transistors through the field oxide. This constant leakage contributes to increased power consumption [49]. Figure 15 illustrates how a circuit exposed to a radiation environment slowly increases power consumption and reduces the operating frequency. Eventually, the circuit will cease functioning when the power required by the degraded electronics exceeds the output capability of the power supply. Premature failure can also occur when the output voltage swing of the transistors becomes insufficient to drive successive stages or when the timing is degraded to the point where the circuit does not operate properly.

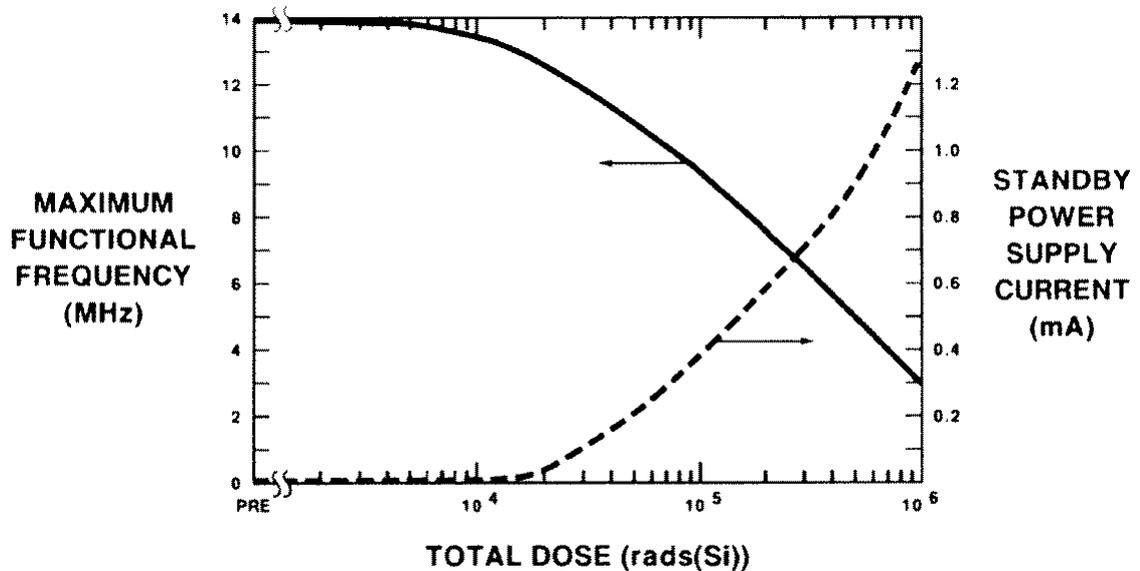


Figure 15. Total Ionizing Dose Response of Maximum Frequency and Supply Current [49]

When a high-energy particle passes through a circuit and causes a disruption in circuit operation, it is classified as a single event effect (SEE). For example, a proton or ion passing through a latch could change the value of a stored bit. This event is called a single event upset (SEU). Protons and high-energy heavy ions typically cause SEUs. Space vehicles passing through the South Atlantic anomaly, where there is a high concentration of protons, can experience SEU activity in that region. These particles create a temporary presence of an abundance of free carriers in the transistor channel region. The free carriers in effect turn the channel on.

If a channel is turned on in a combinational logic circuit, the effect is seen as a glitch in a data or control line, which normally does not affect system operation unless the glitch occurs during a clock transition. However, if a channel is turned on that is part of a memory structure, such as a latch, it can upset the state of the latch. Upset can only occur if enough carriers are present in the transistor channel to turn it on strongly enough to change the state of the latch. SEU can be corrected by refreshing memory locations on a periodic basis.

Another effect seen in CMOS is single event latchup (SEL). SEL describes the phenomenon that occurs when inactive parasitic transistor regions (pnpn structure) are turned on by a high-energy particle. These pnpn regions are formed in CMOS layouts due to the close placement of nMOS and pMOS transistors and have the characteristics of a silicon controlled rectifier (SCR). If a particle with enough energy passes through the controlling pn junction of the SCR, it can switch the SCR on. The only way to turn the SCR off is with a power cycle.

Radiation tolerance to total ionizing dose and single event effects can be achieved through layout. A radiation tolerant inverter is shown in Figure 16. Total ionizing dose effects are minimized by the use of annular geometry nMOS transistors. This geometry minimizes the threshold voltage shift preventing the buildup of trapped charge near the active region and eliminates edge leakage. The transistors are surrounded with highly doped guard rings, which prevent leakage through the field oxide separating the transistors and nearly eliminate SEL. The inherent increased drive strength (width) of the transistors, due to meeting minimum design rules for the annular nMOS then balancing with pMOS, increases the SEU threshold and reduces SET. Additional redundancy techniques can be applied where higher SEE hardness is required.

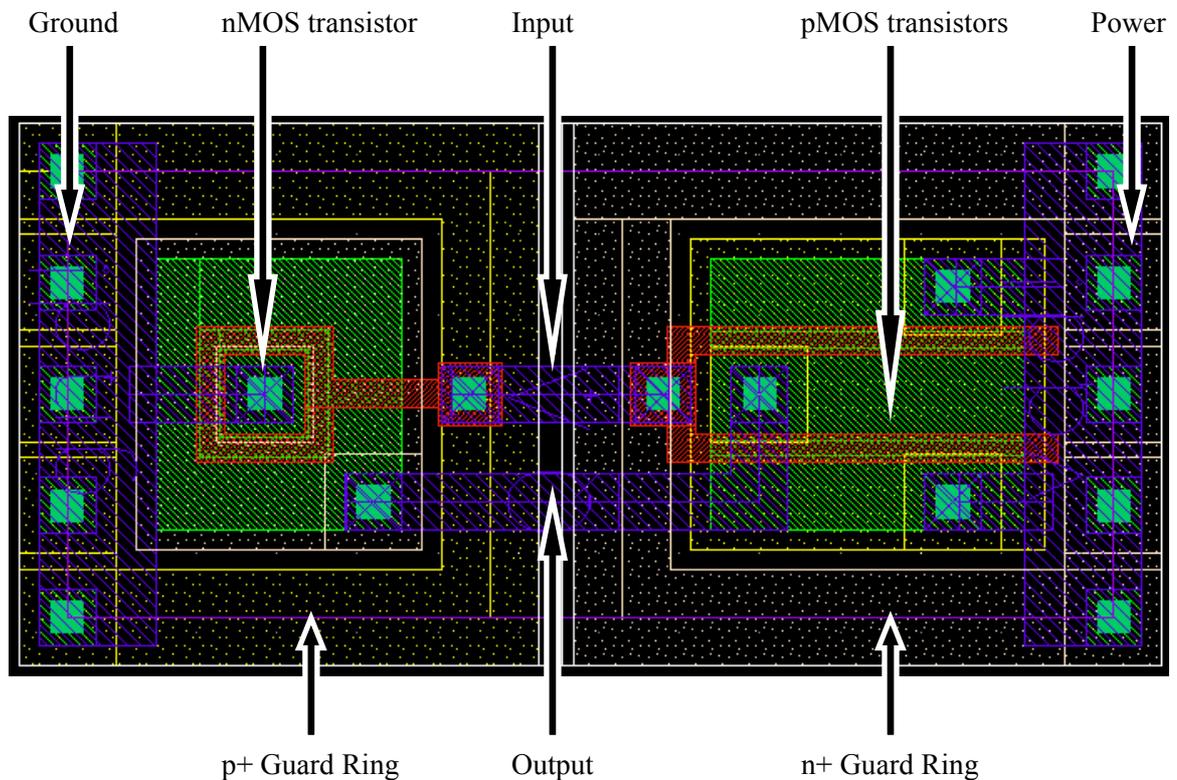


Figure 16. RHBD Layout of an Inverter and Key Features

Numerous RHBD efforts have demonstrated considerable radiation hardness. For example, a recent design and test campaign in 0.25 μm CMOS achieved these results, which far exceed envisioned mission requirements [52]:

- TID > 1 MRad(Si)
- SEL > 110 MeV-cm²/mg @ 125 °C (no latch-up)
- SEU < 1x10⁻¹² errors/bit-day @ 2.25V

Despite the many advantages of this relatively straightforward approach to mitigating radiation effects, there are two primary drawbacks. First, the basic sea of gates or gate array approach does not lend itself to compact designs, so there is a significant area penalty. Also, there is a power penalty, as the transistor length is much longer than the minimum size for the technology.

4.2 Radiation Hardened Library Design

A digital cell library is designed for the AMS S35 process (HITKIT 3.70) in the Cadence DFII framework (2006-2007 5.1.41). A simplified overview of the development process is presented in Table 5. It should be noted that each step involved a significant time investment due to the required learning curve of the complex, yet powerful, commercial tools involved. The most simple cell in the library is the INV0, shown in Figure 16, illustrates clearly the design and features of the nmos4 and pmos4 pcells discussed in steps two and three in Table 5. A complete list of cells required to complete all designs are listed in Table 6 and Table 7.

Table 5. Radiation Hardened Library Design Development Process

| Step | Tool | Action |
|------|----------------------|--|
| 1 | Library Manager | Copy CORELIB, GATES, IOLIB, and PRIMLIB to *_RHBD |
| 2 | Virtuoso (Pcell) | Create/compile nmos4 and pmos4 pcells in PRIMLIB_RHBD |
| 3 | CDF | Edit descriptions of nmos4 and pmos4 in PRIMLIB_RHBD to match |
| 4 | Virtuoso (Schematic) | Verify/update width and length parameters in GATES_RHBD |
| 5 | Virtuoso (Schematic) | Design synthesis to Layout XL |
| 6 | Virtuoso (XL) | Manually place and route pcells, label terminals |
| 7 | Assura | Copy/edit extract.rul file to extract annular nmos properly |
| 8 | Assura (DRC) | Run design rule check, correct errors as needed |
| 9 | Assura (LVS) | Run layout versus schematic, ensure designs match |
| 10 | Assura (RCX) | Run parasitic extraction and verify av_extracted view |
| 11 | DFII (Export Stream) | Create gdsII files from layout view |
| 12 | Library Manager | Create functional (Verilog) |
| 13 | Abstract Generator | Complete abstract generation process for each cell |
| 14 | Virtuoso (Layout) | Manually convert nmos devices in IOLIB to equivalent annular |
| 15 | Voltage Storm* | Characterize and create timing libraries for Verilog and Encounter |

Table 6. Radiation Hardened Library Core Cells

| Cell | Description | Standard Size (μm) | RHBD Size (μm) |
|------------|---------------------------------|---------------------------------|-----------------------------|
| AOI210 | 2-Input AND into 2-Input NOR | 5.6 \times 13 | 16.8 \times 13 |
| AOI220 | 2x2-Input AND into 2-Input NOR | 7 \times 13 | 22.4 \times 13 |
| AOI310 | 3-Input AND into 2-Input NOR | 7 \times 13 | 22.4 \times 13 |
| BUF2 | Buffer | 4.2 \times 13 | 11.2 \times 13 |
| DF1 | D Flip Flop | 21 \times 13 | 67.2 \times 13 |
| DFC1 | D Flip Flop w/active low clear | 23.8 \times 13 | 78.4 \times 13 |
| DFP1 | D Flip Flop w/active low preset | 23.8 \times 13 | 78.4 \times 13 |
| INV0 | Inverter | 2.8 \times 13 | 5.6 \times 13 |
| MUX21 | 2:1 Multiplexer | 8.4 \times 13 | 33.6 \times 13 |
| NAND20 | 2-Input NAND | 4.2 \times 13 | 11.2 \times 13 |
| NAND30 | 3-Input NAND | 5.6 \times 13 | 16.8 \times 13 |
| NAND40 | 4-Input NAND | 7 \times 13 | 22.4 \times 13 |
| NOR20 | 2-Input NOR | 4.2 \times 13 | 11.2 \times 13 |
| NOR30 | 3-Input NOR | 5.6 \times 13 | 16.8 \times 13 |
| NOR40 | 4-Input NOR | 7 \times 13 | 22.4 \times 13 |
| OAI210 | 2-Input OR into 2-Input NAND | 5.6 \times 13 | 16.8 \times 13 |
| XOR20 | 2-input XOR | 9.8 \times 13 | 28 \times 13 |
| TIE0/1 | Tie lo and hi logic | 2.8 \times 13 | 5.6 \times 13 |
| Fill cells | Fill cells for SOC Encounter | Various | Various |

Table 7. Radiation Hardened Library Input/Output Cells

| Cell | Description | Standard Size (μm) | RHBD Size (μm) |
|-------|-------------------------|---------------------------------|-----------------------------|
| BBC1P | 1 mA bi-directional pad | 95 \times 334 | same |
| BU1P | 1 mA output buffer | 95 \times 334 | same |
| ICP | Input buffer | 95 \times 334 | same |

4.2.1 Asynchronous Logic Background

Traditional synchronous circuit designs feature a global clock that drives latches surrounding combinational logic, which as a system, performs a particular function. The clock rate is determined by the critical path through the system. This approach has remained an industry standard largely due to the entrenched design flow, which includes design synthesis from hardware description languages (HDLs). However, synchronous designs have periodic power peaks, which produce EMI. Additionally, the global clock tree consumes a significant fraction of the required power.

Asynchronous SoC architecture, which offers numerous advantages, has only recently been considered by this niche community [53]. Typically, asynchronous implementations can potentially require a fraction of the power of their clocked counterparts and produce very little electromagnetic interference (EMI). Asynchronous designs are event triggered, processing new data using the minimum number of gate transitions possible. Asynchronous SoC design also promises to solve the global clock delay problem, which increases as the size of SoCs grow with increased functionality and performance.

Asynchronous logic concepts have existed since the 1950's, offering potential power savings and performance improvements depending on the application [54]. Analogous to RHBD's shortfalls in power and area penalties, asynchronous logic design is more complex when compared to the synchronous commercial standard and carries a potential area penalty. However, recent advances in automating the asynchronous design process have made the idea more attractive, resulting in new commercial offerings.

Asynchronous designs work on the concept of modular functional blocks with intercommunication using handshaking protocols. The overall function of the circuit resembles that of the synchronous one. Recently, considerable progress has been made to improve the design automation of this particular asynchronous characteristic, complete with a new term, de-synchronization [55].

However, de-synchronization does not yet realize all the potential advantages of asynchronous logic. Although removing the global clock tree and replacing it with a fabric of handshaked interconnections does flatten the power spectrum and reduce EMI generation, it is generally accepted that the opportunity is missed to significantly lower the energy requirements and improve the performance. This can be achieved by recognizing that most synchronous circuits often have redundant operations depending on the system state and that not all operations take the same amount of time. Unfortunately, automating this process has not been achieved due to the variety of power and latency reduction techniques that can be applied, and each one design dependent.

A custom design approach was chosen for this work to demonstrate the best-possible benefits of asynchronous logic, leveraging the assumption that others are continuing to improve asynchronous design automation. The paragraphs to follow describe the general asynchronous design methodologies used in this work. The next section discusses the integration of the RHBD and asynchronous design concepts and presents the comparative results.

The asynchronous building blocks used in this effort fall into four typical categories, briefly reviewed in the following paragraphs [56]. The fundamental mode bounded delay methodology is used for blocks with relatively fixed completion times. The delay insensitive design methodology applies to functional blocks with widely varying completion times. Burst mode design methodology applies to components that serve as controllers or asynchronous finite state machines (AFSMs). Finally, the speed independent model specifies the handshaking protocols between major functional blocks.

The fundamental mode bounded delay methodology was used for functional blocks that had little variation in completion time, such as a latch. This methodology assumes that the delay time through a functional block is known and constant. Worst-case delay, with a margin of safety, is

used similar to a clocked circuit. Difficulty arises in synthesizing this structure since timing information cannot be synthesized from behavioral HDL, but can be back-annotated from layout simulations. Figure 17 illustrates a delay element used to model the latch completion time. An acknowledge (ACK) signal is asserted when the data is latched after the request (REQ) is generated.

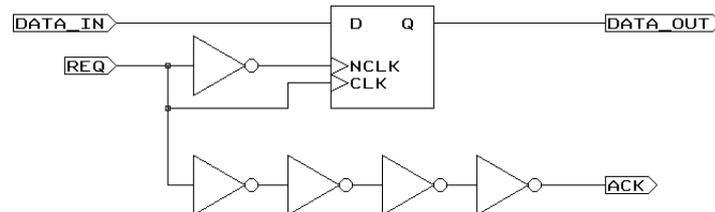


Figure 17. Fundamental Mode Bounded Delay Applied to a Latch

A delay element is not suitable for functional blocks with widely varying completion times, since the average critical path latency can be much lower than the synchronous counterpart. Additional logic can be added to this type of block to detect when its execution is complete. Synthesis tools do not yet have the ability to generate the completion detection circuit for a particular functional block, such as a basic add/subtract unit, shown in Figure 18.

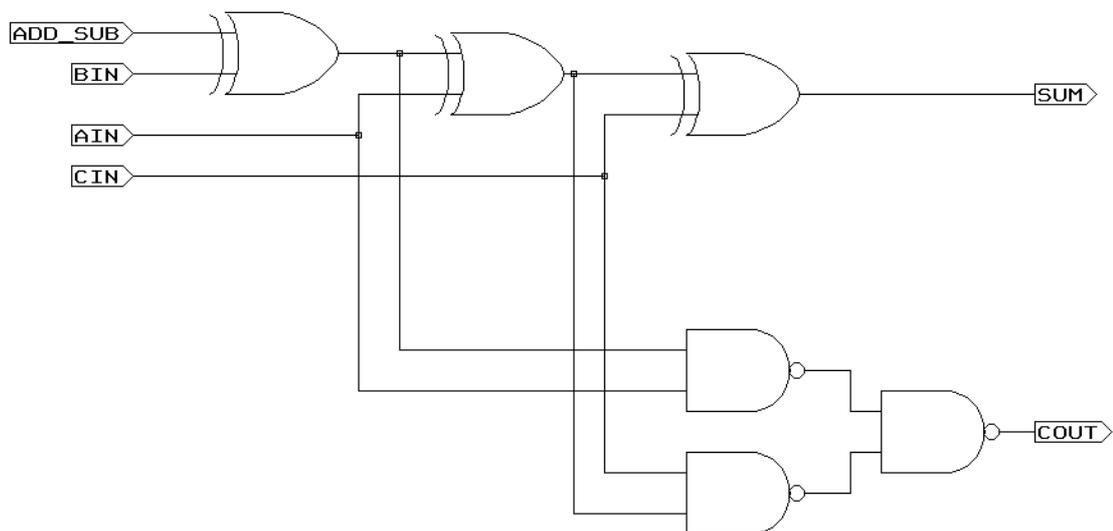


Figure 18. One-Bit Adder without Completion Detection

A dual-rail adder scheme similar to the Manchester adder can be used to implement completion detection [57]. The dual rail adder works on the principle that each stage will have either a carry out (COUT) or no carry out (NOCOUT) condition based on the inputs to the stage. Adding 0 and 0 will never result in a carry out, even if there is a carry in. Likewise, adding 1 and 1 will always result in a carry out, even if there is a carry in of 0. Therefore, the carry condition in these cases can be determined by the data to be summed alone and gives early completion detection. Adding a

0 and 1 or 1 and 0 may or may not have a carry out depending on the carry in condition. In this case, the stage must wait for either a carry in (CIN) or no carry in (NOCIN) value. The end result is the completion detection circuit simply becomes the NOR of the COUT and NOCOUT values. Whenever one of these conditions exist, it indicates that all input values necessary for evaluating the sum are present and DONE is asserted. A design with improved throughput is shown in Figure 19.

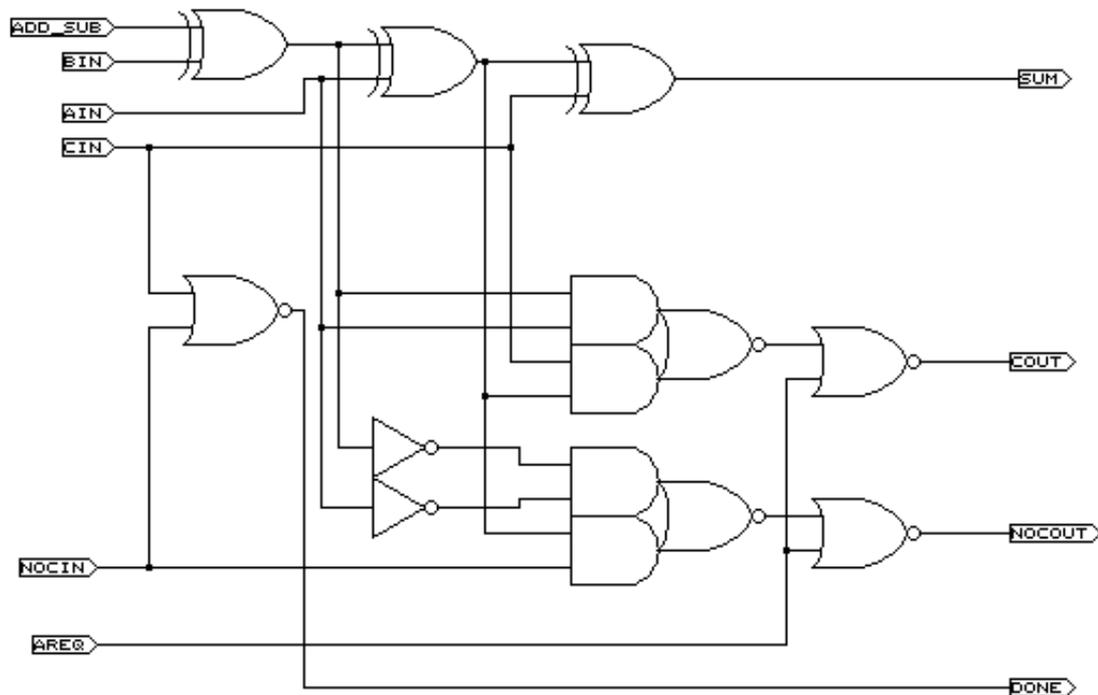


Figure 19. One-bit Adder with Completion Detection

The burst mode design methodology is used to design asynchronous controllers or finite state machines. Synchronous finite state machines are easily synthesized by using latches, flip-flops and clock circuitry. Asynchronous controllers or AFSMs must be synthesized using specialized design tools, such as 3D [58].

A user-specified state table of entry and exit conditions for the state machine is provided to 3D. An example state table is shown in Table 8 for a Johnson counter (00→01→11→10). 3D converts the state table to positive logic equations. These equations are then manually converted into behavioral HDL. A logic synthesizer (with structuring and Boolean optimization disabled) can be used to convert the positive logic behavioral HDL into negative logic structural HDL. After the structural HDL is generated, reset circuitry and corrections for fanout are added manually to the controller circuit. The final two-bit Johnson counter circuit is shown in Figure 20, which includes the reset circuit.

Table 8. 3D State Table of a Two-Bit Johnson Counter

| Present State | Next State | Entry Conditions | Exit Conditions |
|---------------|------------|------------------|-----------------|
| 0 | 1 | COUNT+ | |
| 1 | 2 | COUNT- | BIT0+ |
| 2 | 3 | COUNT+ | |
| 3 | 4 | COUNT- | BIT1+ |
| 4 | 5 | COUNT+ | |
| 5 | 6 | COUNT- | BIT1+ |
| 6 | 7 | COUNT+ | BIT0- |
| 7 | 0 | COUNT- | BIT1- |

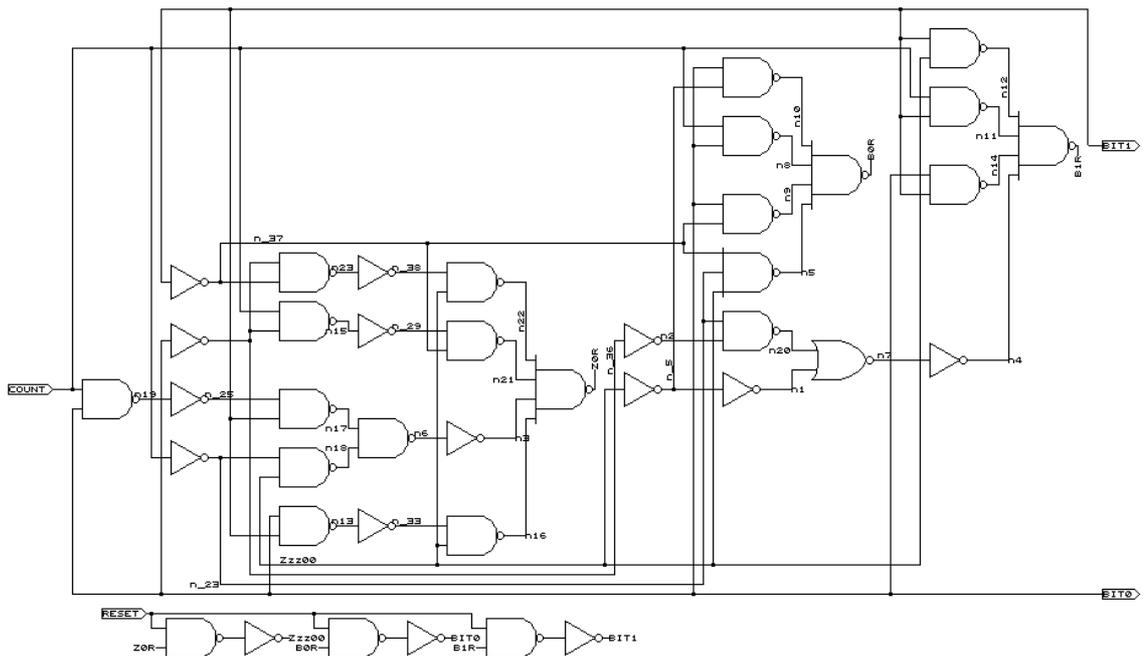


Figure 20. Gate Level Schematic of a Synthesized Two-Bit Johnson Counter

Depending on the complexity of the AFSM, 3D may not be able to synthesize the controller. The controller must then be broken down, using Shannon decomposition, and resynthesized. The two-bit Johnson counter example is used to illustrate how asynchronous synthesis tools work, but it highlights how automated AFSM synthesis does not always produce the most elegant solution [59]. A simpler implementation (but nearly the same transistor count) of the two-bit Johnson counter is accomplished by using two D-registers, as shown in Figure 21. It is also important to note that the advantage of the Johnson counter is the fact that it changes only one bit each clock cycle, avoiding possible data hazards.

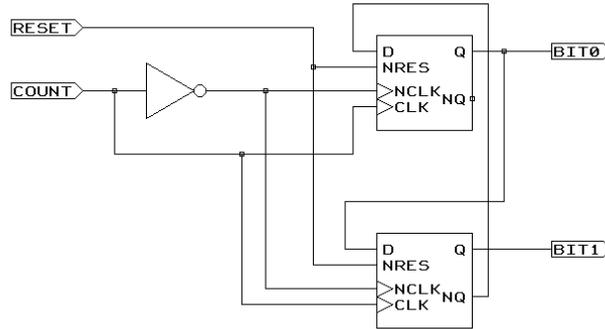


Figure 21. Improved Two-Bit Johnson Counter

Functional blocks in an asynchronous design must have a standard handshaking protocol in order to interface with other blocks. A generic functional block in an asynchronous design is shown in Figure 22. The REQIN signal represents the external request to the block to input new data. The ACKIN signal is asserted when the new input data is fully latched or accepted. The REQOUT signal represents the request of the functional block to send processed data out. The ACKOUT signal is the external acknowledgement from the next block that the processed data was latched or accepted.

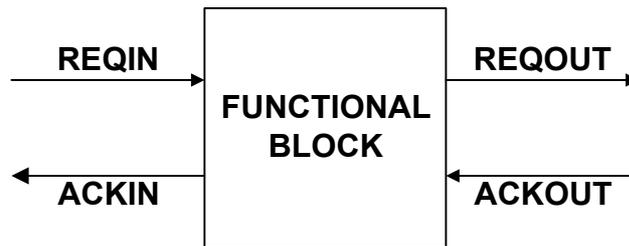


Figure 22. Asynchronous Functional Block

The speed independent methodology describes two standards for handshaking between connecting blocks. It does not assume any pre-defined delays but relies on a set of handshaking signals between the blocks. The two-phase model is illustrated in Figure 23. It is a scheme that senses signal transitions to complete the handshake cycle. The first exchange is signaled by a low to high transition on REQ (1). ACK (2) responds by acknowledging the request. The second cycle uses the complementary set of transitions to complete the cycle.

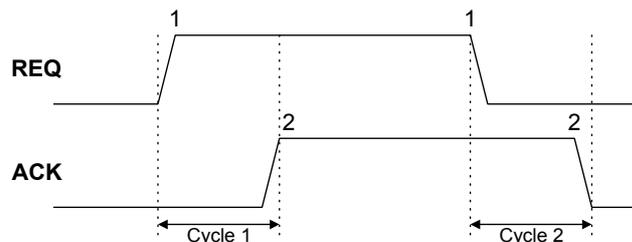


Figure 23. Asynchronous Two-phase Handshaking Model

The four-phase model is illustrated in Figure 24. It has a four-cycle handshake for each data exchange. Although the four-phase model appears to be more difficult to implement, its detection circuit is actually smaller than the two-phase mode. The four-phase model is the primary interface standard used throughout this design.

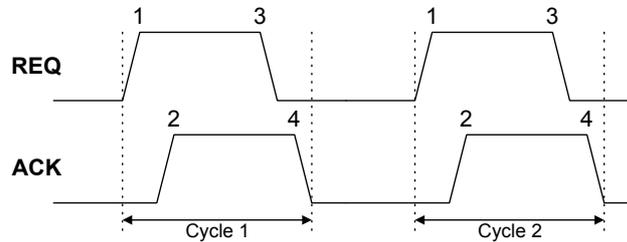


Figure 24. Asynchronous Four-phase Handshaking Model

4.2.2 Case Study of RHBD and Asynchronous Logic Synergy

The basic idea behind this case study was to demonstrate the advantages of using RHBD and asynchronous together. Although area is sacrificed, the hope was that these techniques would offer higher performance, a flatter power spectrum, and similar energy consumption when compared to a synchronous design. Although an obscure application, this approach is not completely novel, as various elements have been presented before in [60] and greatly expanded upon in [61] with test results in [62]. Unfortunately, these efforts failed to influence the community due to the lack of a convincing case study, which is the purpose of this work.

It should be noted that other approaches have been investigated for space applications of asynchronous logic. For example, fault tolerance and deadlock have been addressed by works such as [63]-[65]. These approaches focus on logic gate and circuit level redundancy techniques to improve SEU hardness. However, they exclude TID and SEL considerations, which are mitigated through RHBD. However, they can be used in addition to RHBD for mission critical applications in very harsh radiation environments.

To make a convincing argument, a common design is implemented in three ways: synchronous with a commercial cell library, synchronous with a RHBD cell library, and asynchronous with the same RHBD library. The textbook “MIPS” multi-cycle microprocessor architecture is used as the baseline design as illustrated in Figure 25 (adapted from Fig. 5.28 [66]). To keep the size small and affordable, a 16-bit fixed-point 4-register variant (versus 32-bit floating point 32-register) is implemented with a simplified instruction set shown in Table 9. The functional block descriptions are given in Table 10.

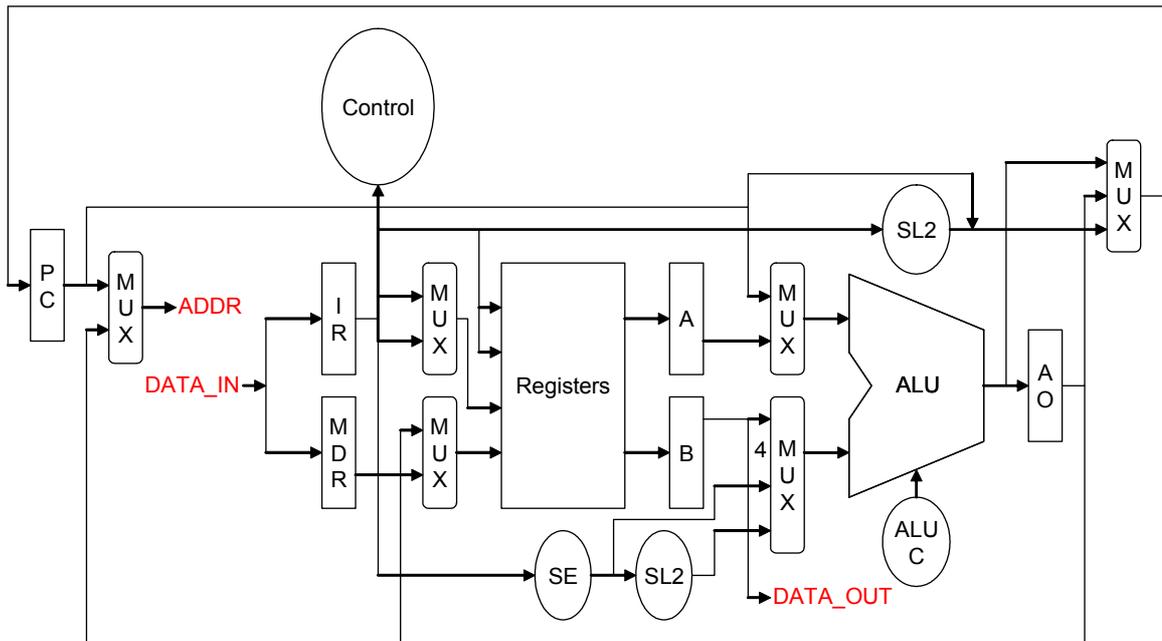


Figure 25. MIPS Conceptual Block Diagram

Table 9. Simplified MIPS Instruction Set

| Instruction | Meaning | 16-bit Instruction | Cycles |
|------------------|-------------------------------|--------------------|--------|
| add | $rd = rt + rs$ | 0000rsrtrd000000 | 4 |
| subtract | $rd = rt - rs$ | 0000rsrtrd000010 | 4 |
| logical AND | $rd = rt$ (bitwise and) rs | 0000rsrtrd000100 | 4 |
| logical OR | $rd = rt$ (bitwise or) rs | 0000rsrtrd000101 | 4 |
| set on less than | set $rd = 1$ if $rt < rs$ | 0000rsrtrd001010 | 4 |
| load word | $rt = mem[rs + addressx]$ | 0001rsrtaddressx | 5 |
| store word | $mem[rs + addressx] = rt$ | 0010rsrtaddressx | 5 |
| branch on equal | if $rs = rt$ go to $addressx$ | 0011rsrtaddressx | 3 |
| jump | jump to $addressx$ | 0100000000000000 | 3 |

The entry of the baseline synchronous/standard cell design into Cadence is outlined in Table 10. The baseline design is then copied and renamed as the synchronous/RHBD variant. The synchronous/RHBD variant is simply modified by using a global search and replace of the cell library name, beginning at step 14 of Table 10. Steps 15-22 were repeated to complete the design. Both synchronous variants were submitted for fabrication on AMS S35 run 1725. The final layout and micrograph of the fabricated chips are shown in Figure 26 and Figure 27.

Table 10. Cadence Design Flow

| Step | Tool | Build Action(s) |
|------|-----------------------|--|
| 1 | Library Manager | New design library |
| 2 | Virtuoso (Schematic) | 16-bit multiplexors (MUX): 2:1, 3:1, 4:1 |
| 3 | Virtuoso (Schematic) | Arithmetic Logic Unit (ALU) basic block: 1-bit add/sub |
| 4 | Virtuoso (Schematic) | 16-bit ALU blocks: add/sub, and, or, slt, zero detect |
| 5 | Virtuoso (Schematic) | Top-level ALU |
| 6 | Virtuoso (Schematic) | ALU control (ALU C) |
| 7 | Virtuoso (Schematic) | 16-bit registers: Program Counter (PC), Memory Data Register (MDR), Instruction Register (IR), A, B, ALUOut (AO) |
| 8 | Virtuoso (Schematic) | Hardwired blocks: Shift Left 2 (SL2), Sign Extend (SE), Four (4), Zero (0) |
| 9 | Virtuoso (Schematic) | Top-level register file (3 registers + hardwired 0) |
| 10 | RTL Compiler | Synthesis of Control block from Verilog description |
| 11 | DFII (Import Verilog) | Import synthesized logic into schematic |
| 12 | Virtuoso (Schematic) | Top-level MIPS |
| 13 | NC-Verilog | Verilog testbench of all instructions with accurate timing |
| 14 | Virtuoso (Schematic) | Top-level chip (adding I/O pads) |
| 15 | NC-Verilog | Reverify testbench, export netlist |
| 16 | RTL Compiler | Pass-through of netlist to satisfy SOC Encounter format |
| 17 | SOC Encounter | Import netlist, place I/O and core, route, clock tree synthesis (CTS), export netlist, export gdsii stream |
| 18 | NC-Verilog | Import layout netlist to schematic, reverify testbench |
| 19 | DFII (Import Stream) | Import gdsii stream to layout |
| 20 | Virtuoso (Layout) | Inspect layout |
| 21 | Assura | Run DRC, LVS, RCX |
| 22 | UltraSim | Run full-chip simulation, compare results with Verilog |
| 23 | DFII (Export Stream) | Export gdsii file for fabrication, submit design |

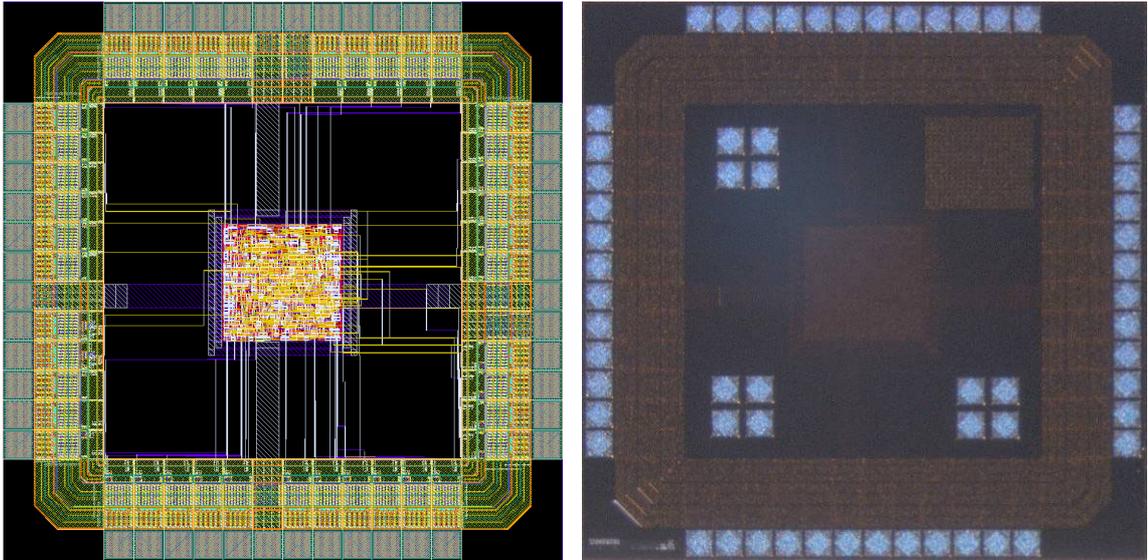


Figure 26. Synchronous/Commercial Layout and Micrograph (400×400µm Core)

Notice the four test structures in the micrograph in Figure 26. Three of the structures are basic RHBD structures intended for use at a micro probe station: nMOS, pMOS, and an inverter. The fourth test structure is in the upper right hand corner, which is a small bank of photocells with the same initial design structure.

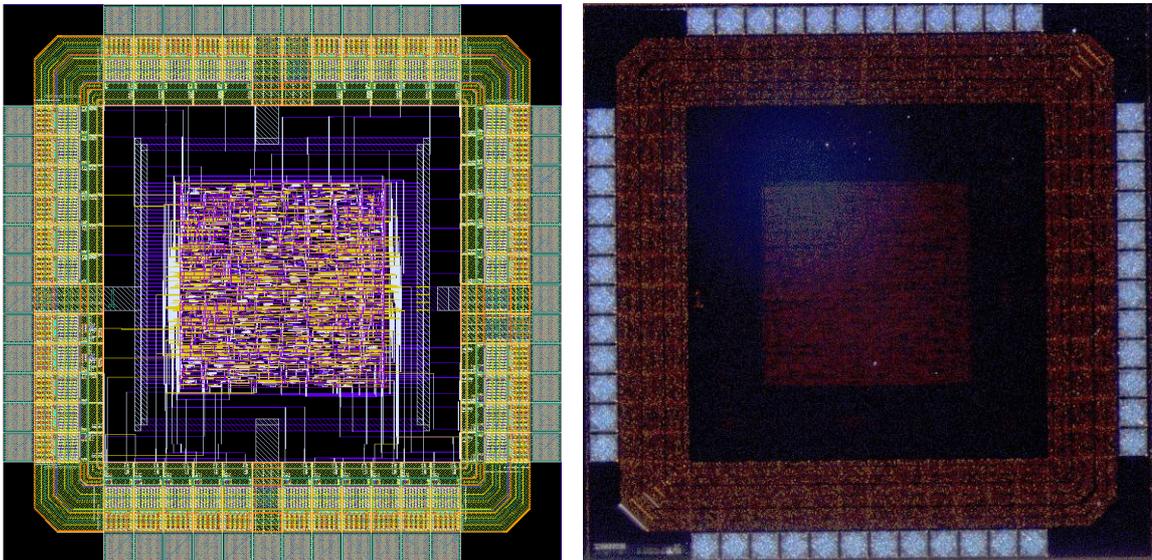


Figure 27. Synchronous/RHBD Layout and Micrograph (700×700µm Core)

Recall that the RHBD library is a layout modification only of the AMS HIT KIT 3.70. The original thought was to use Signal Storm to generate HDL and timing libraries. However, this idea was abandoned due to realizing that this approach would result in reduced drive strength during the various optimization stages. To maintain radiation hardness to SEU and SET particularly, keeping the drive strength and fanout ratios at the same proportion to the standard cell library is required. Therefore, the best approach is to use the standard cell timing libraries.

There are some minor differences between the two designs just presented, regarding the RHBD cell library. Due to time constraints, the RHBD library does not have the full array of buffer and inverter cells that are used during clock tree synthesis (CTS). However, the CTS process compensated for this appropriately, as the sum of the transistor widths is the same. In addition, the I/O cells are the unmodified commercial version, also due to time constraints. This does not affect the simulation or hardware results significantly, as the nMOS transistor widths are equivalent.

The final design in the case study is an asynchronous/RHBD variant. Asynchronous logic offers potential power savings and performance improvements with a tradeoff in design complexity and usually small area penalty. In its purest form, this circuit design approach aims to minimize transistor switching. Due to the variety of circuit types and implementation techniques, the design process can be quite complex.

The unpipelined MIPS architecture may not be the best for demonstrating dramatic power reductions, but it does offer the observer direct insight to the design process. For example, it does not make sense to break down the architecture into smaller blocks where handshaking can be applied. Instead, the MIPS circuit should be thought of as a design block in a larger asynchronous SoC, as in the envisioned sensor node architecture. The external interface of the asynchronous MIPS implementation is shown in Figure 22 with four-phase handshaking as in Figure 24.

Several asynchronous design methodologies are applied to the synchronous MIPS architecture. This approach is not to be confused with de-synchronization as defined in [55], but rather a unique focus on overall power reduction and flattening the power spectrum. The global clock is removed, but instead of replacing the flip-flops with master-slave latches and delay elements as in de-synchronization, a phased sequence of latching with tailored delay elements is carefully applied across the latches and multiplexers in the data path, as shown in Figure 28. Care is taken to ensure a hazard-free sequence and no double-switching of elements. The synchronous FSM control block is improved to minimize latching of the MDR and ALUOut registers. Additionally, a form of clock gating is applied within all registers, which allows the use of basic D-latches without enables. This also requires latches to be placed on all control signals and phased in as appropriate. The applied approaches are summarized in Table 11.

Table 11. Asynchronous Design Approaches Implemented

| Step | Action | Benefit |
|------|---------------------------------|--|
| 1 | Remove global clock | Overhead of CTS eliminated, power reduced |
| 2 | Add phased latching sequence | Flattens power spectrum |
| 3 | Add delays within registers | Further flattens power spectrum |
| 4 | Improve MIPS control | Eliminates redundant latching, power reduced |
| 5 | Add clock gating | Power reduced |
| 6 | Remove unused inverting outputs | Power and area reduced |

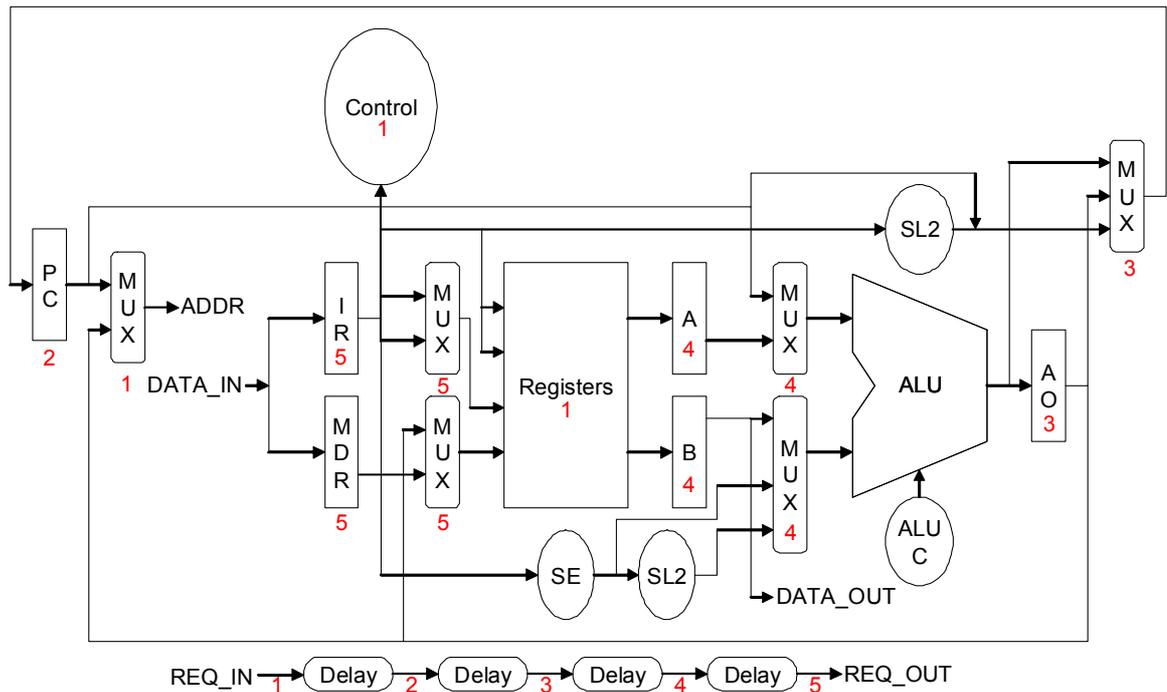


Figure 28. Phase-Latched Asynchronous Approach

The custom re-design of most elements in the MIPS architecture just discussed affects all steps in Table 10. Most notable, CTS and optimization are prevented in step 17. The asynchronous/RHBD variant was fabricated on AMS S35 run 1791. The final layout and fabricated die micrograph is shown in Figure 29.

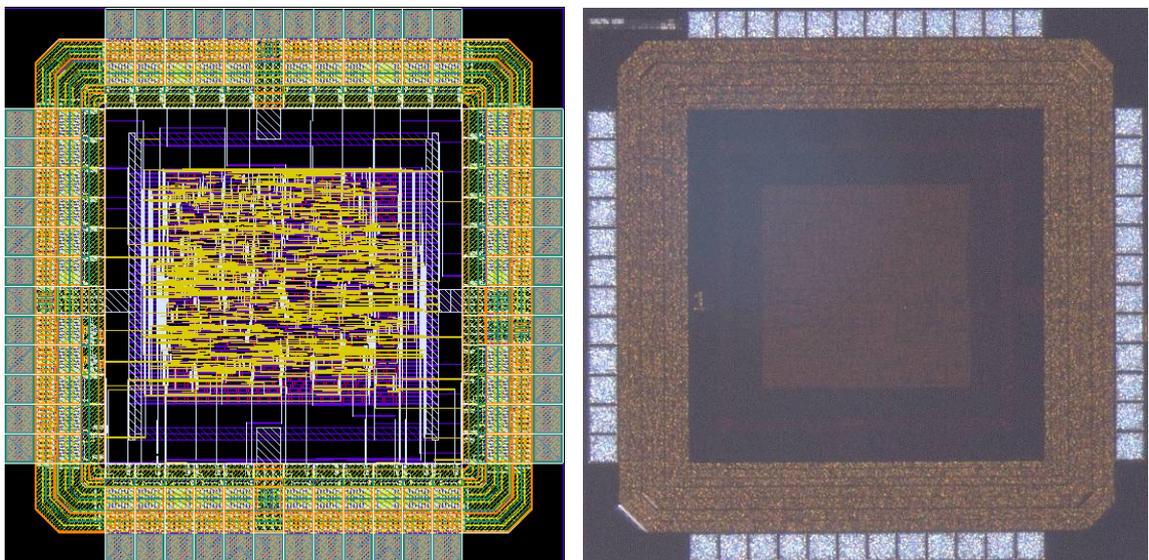


Figure 29. Asynchronous/RHBD Layout and Micrograph (720x720µm Core)

4.3 Comparison, Simulation, and Test Results

A common test bench is used for NC-Verilog simulation, UltraSim simulation, and hardware testing using National Instruments Digital Waveform Editor and LabView. NC-Verilog is a functional simulator that uses back-annotated timing information for each element. Simulation results are available immediately. UltraSim is based on Spice, as it uses extracted parameters for a more accurate simulation, but uses a proprietary algorithm to allow for full-chip simulations in a reasonable amount of time. For example, most of the full-chip simulations take around one hour to run, versus days for this size of design on Spice or HSpice. The UltraSim results are advertised to be within 5% of Spice. The test bench is shown in Table 12, indicating expected output data (DATA_OUT) and expected address (ADDR) based on the instruction and data mix given to the microcontroller (DATA_IN).

Table 12. Common Test Bench Including Expected Results

| DATA_IN | Expected DATA_OUT | Expected ADDR |
|-----------------------------|-------------------|---------------|
| load R1 from address 0x0001 | | 0x0000 |
| 0xFFFF | | 0x0001 |
| load R2 from address 0x0002 | | 0x0004 |
| 0x0001 | | 0x0002 |
| R3 = R1 + R2 | | 0x0008 |
| store R3 to address 0x0000 | 0x0000 | 0x000C |
| R3 = R1 - R2 | | 0x0010 |
| store R3 to address 0x0000 | 0xFFFE | 0x0014 |
| R3 = R1 (bitwise and) R2 | | 0x0018 |
| store R3 to address 0x0000 | 0x0001 | 0x001C |
| R3 = R1 (bitwise or) R2 | | 0x0020 |
| store R3 to address 0x0000 | 0xFFFF | 0x0024 |
| R3 = R1 < R2 | | 0x0028 |
| store R3 to address 0x0000 | 0x0001 | 0x002C |
| branch if R1 = R2 | | 0x0030 |
| load R2 from address 0x0002 | | 0x0034 |
| 0xFFFF | | 0x0002 |
| R3 = R1 < R2 | | 0x0038 |
| store R3 to address 0x0000 | 0x0000 | 0x003C |
| branch if R1 = R2 | | 0xFE0C |
| jump to 0 | | 0xC000 |

Figure 30 illustrates an example of the complete NC-Verilog simulation testbench. Figure 31 is the result of the UltraSim simulation, which matches with the expected results and the NC-Verilog simulation. Figure 32 is the results of the LabView hardware results. Either the inputs or outputs can be shown in LabView at one time, so the output values are shown, indicating that the fabricated test chip performs functionally as expected. The simulation and hardware outputs for the synchronous/commercial gate, synchronous/RHBD, and asynchronous/RHBD designs are all very similar and are not repeated. The maximum frequency of all designs is 16.67 MHz in simulation, but unfortunately, the hardware test platform only operates up to 12.5 MHz.

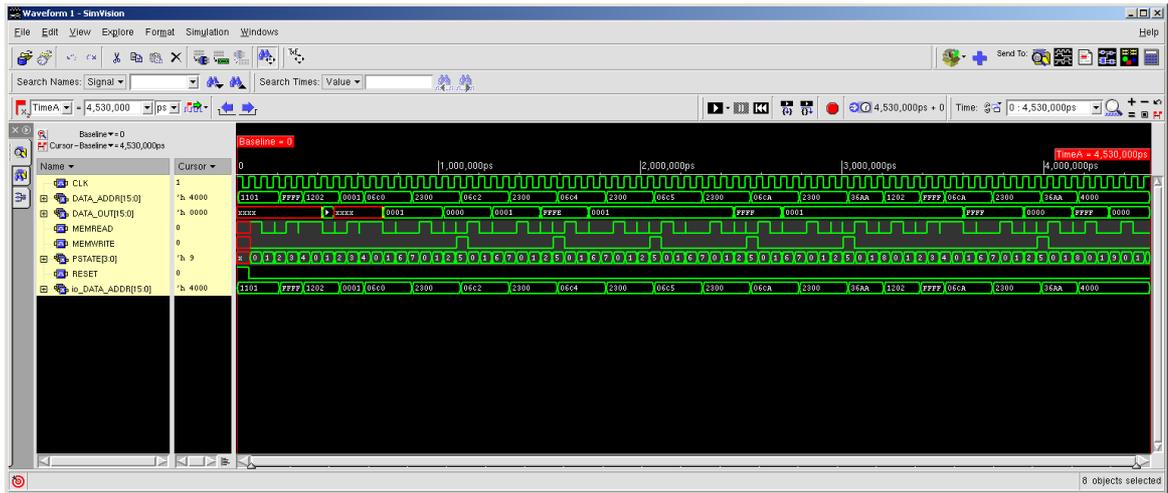


Figure 30. Example NC-Verilog Simulation Testbench Output

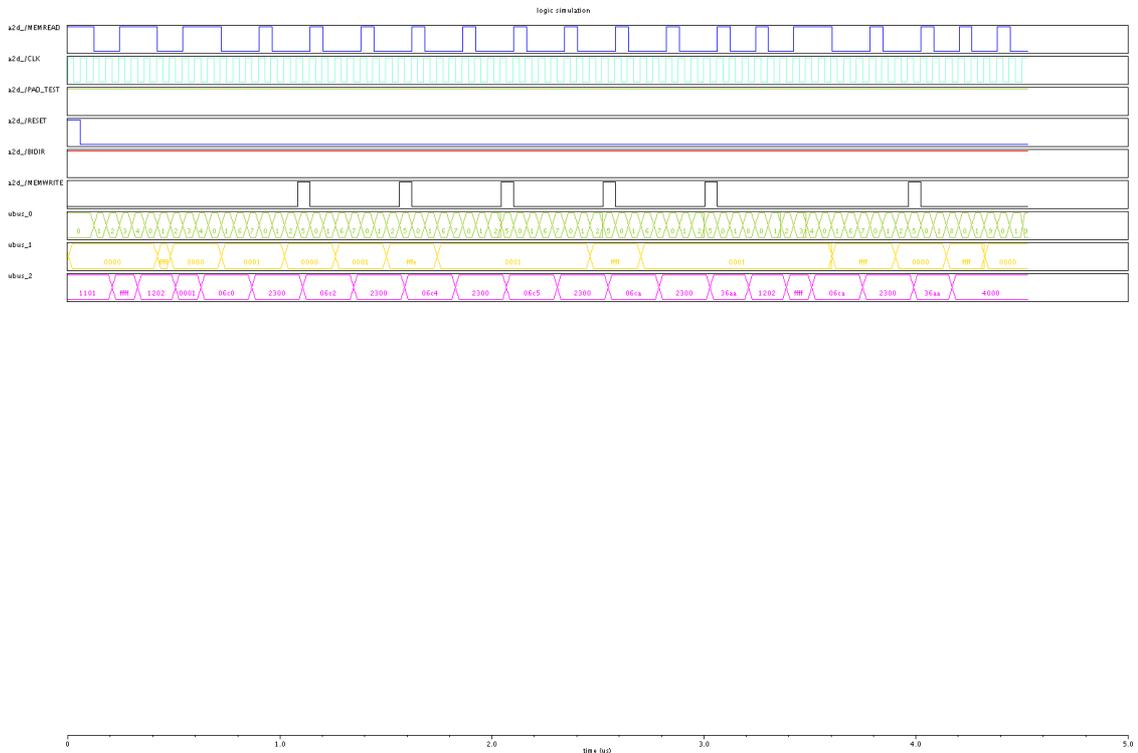


Figure 31. Example UltraSim Testbench Output



Figure 32. Example LabView Hardware Testbench Output

Although correct functionality is important to verify, the most important aspect of this comparison is the power performance. NC-Verilog is not able to report on power consumption. Figure 33 illustrates an example power output for the entire testbench and a closeup of an example single clock cycle in UltraSim. Figure 34 illustrates the power spectrum of the three designs, left to right. Note that the asynchronous design on the right has smoothest profile and lowest peaks.

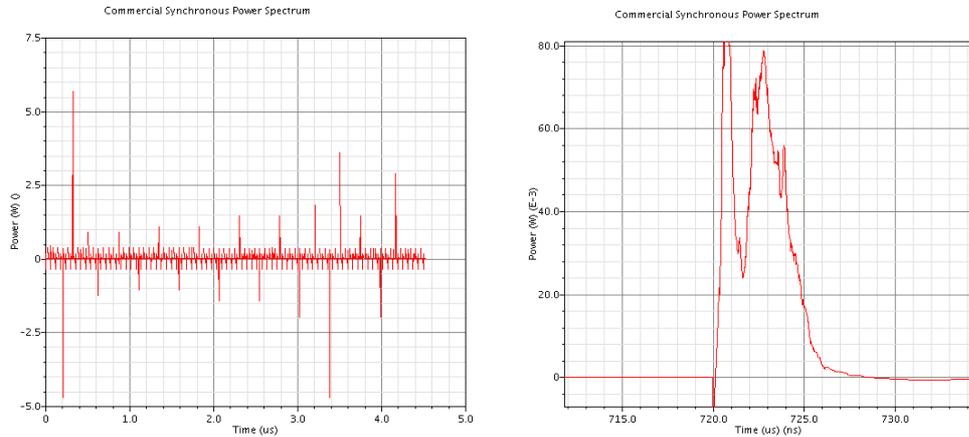


Figure 33. Example Simulation Testbench Power Spectrum in UltraSim

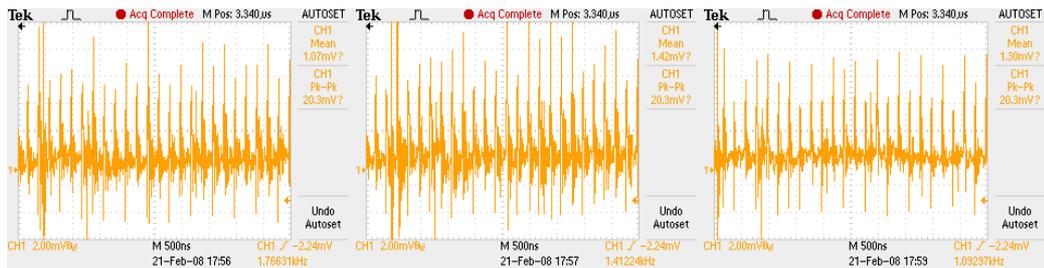


Figure 34. Example Hardware Testbench Power Spectrum

A comparison of results is given in Table 13. In this case study with this particular design, the application of RHBD resulted in a 200% core area increase from the baseline design and required 160% more energy for the same testbench at any frequency, as determined through UltraSim simulations. Figure 34 and Figure 35 clearly illustrate that all the asynchronous approaches taken to reduce the power and smooth the power spectrum are indeed effective. Figure 36 verifies that the final hardware results correlate nicely with the predicted simulation results, across the 1.25 to 12.5 MHz test points. The most significant result is that the asynchronous approach reduced the energy penalty to 85% (from 150%) for a 6% area increase with no performance impact.

Table 13. Comparison of Three Design Approaches

| Test Chip | Total Transistor Width (μm) | Core Area (μm) | Energy (nJ) (UltraSim) |
|-----------------------------|--|-----------------------------|------------------------|
| synchronous/commercial (sc) | 16,088 | 400×400 | 28 |
| synchronous/RHBD (sr) | 60,450 | 700×700 | 71 |
| asynchronous/RHBD (ar) | 55,973 | 720×720 | 51 |

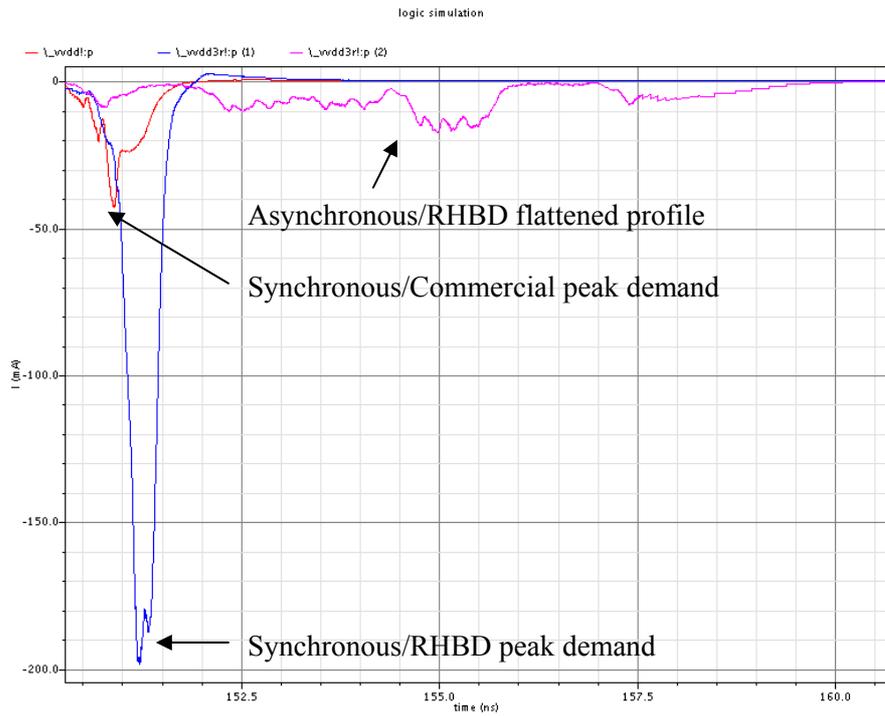


Figure 35. Single Clock Cycle Comparison in UltraSim

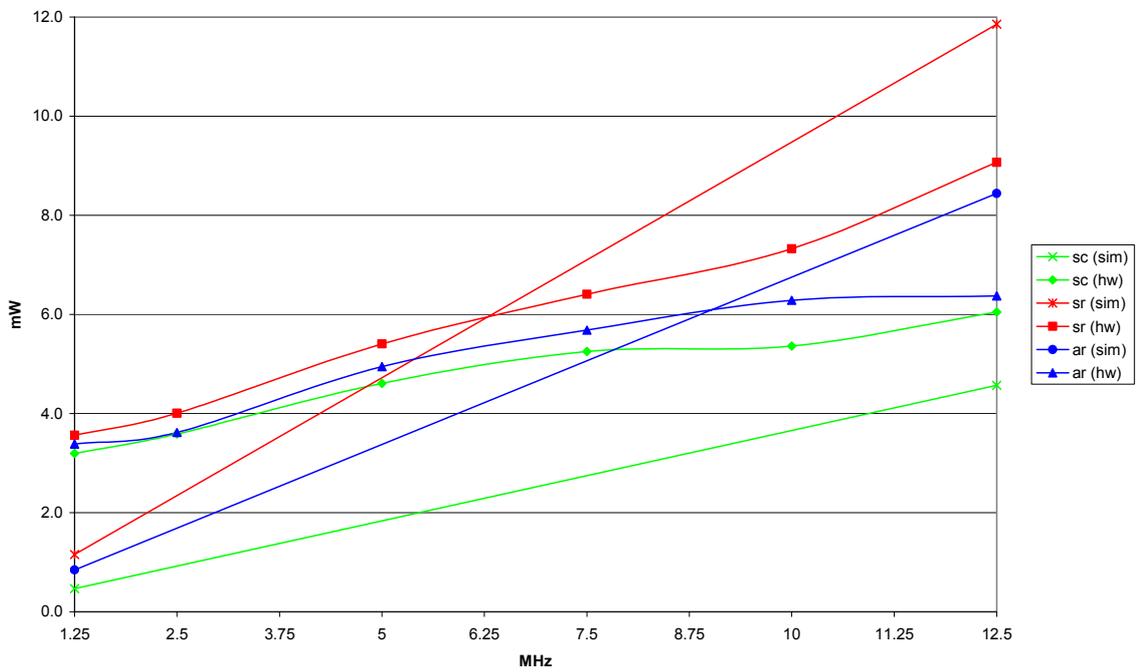


Figure 36. Comparison of Simulation and Hardware Power Consumption

5 Conclusions

5.1 Summary of Results

A cost-effective monolithic system-on-a-chip approach is under investigation to fabricate large numbers of wireless sensor nodes for hostile environments including space. Two essential building blocks have been developed and reported on: integrated solar cells in CMOS and radiation hardening by design of asynchronous logic.

A first-ever design for integrated solar cells in commercial SiGe BiCMOS is presented. Two prototype designs have been designed, fabricated, and tested. The average efficiency of the first prototype is 2.4%, compared to an estimated, but unverified 1% from previous work. The actual efficiency of the junction is 8.3%, without considering the metallization overhead. An improved design demonstrates 3.44% efficiency, a 40% improvement. The junction efficiency alone is 11.3%. However, power from these first two prototypes cannot be harnessed properly in the current implementation. A final design, overcoming this limitation, has been submitted for fabrication and will be reported in a later publication. This novel development has potential widespread application to a rapidly growing number of SoC designs.

The application of radiation hardening by design to asynchronous logic is suggested as a unique approach for bare die SoC implementations in hostile environments. A case study is presented using a common design. Starting with a common synchronous microcontroller design implemented with commercial logic gates, the application of RHBD results in an expected 200% core area increase and requires 160% more energy. The most significant result is that the application of asynchronous design reduced the energy penalty to 85% (from 160%) for a 6% area increase with no performance impact. Additionally, electromagnetic interference is greatly reduced. This approach provides environmental tolerance to radiation and temperature extremes.

5.2 Future Research Directions

A suggested next step would be the monolithic integration of the developed solar cells and microcontroller with a single-chip radio design and simple sensor. The focus of this work would be to minimize or eliminate the traditional external components and establish self-powered wireless interconnectivity. To date this complete monolithic approach has not been demonstrated in the literature and would make a great impact on a number of technology applications.

Due to lack of time and resources this project could not address two other important aspects of the SoC design detailed below, which should be undertaken by a follow-up project:

System Configuration—A simple configuration, such as two die sandwiched together, could help meet power and thermal requirements. An investigation is required to determine the material composition and minimal packaging. A preliminary design has been reported in [46].

Stand-alone Transceiver—All SoC transceivers to date require external passive devices, precision frequency oscillators, and antennas. Research is needed to determine if a very simple transceiver, perhaps using on-off keying (OOK) modulation, could be implemented on CMOS without any external components. However, it has been clearly demonstrated that an external antenna will be required to achieve any meaningful range.

Bibliography

- [1] Helvajian, H. and E. Y. Robinson, "Micro- and Nanotechnology for Space Systems," The Aerospace Corporation, Rept. ATR-93(8349)-1, 31 March 1993.
- [2] S.W. Janson, H. Helvajian, and E. Y. Robinson, "The Concept of 'Nanosatellite' for revolutionary low-cost space systems," *Proceedings of the 44th Congress of the International Astronautics Federation*, Graz, Austria, Oct. 1993, Paper IAF-93-U.5.575
- [3] H. Helvajian, *Microengineering Aerospace Systems*, Reston, VA, AIAA Press, 1999.
- [4] S. W. Janson, "Mass-Produced Silicon Spacecraft for 21st Century Missions," in *Proc. AIAA Space 1999*, Albuquerque, NM, 1999, Paper AIAA-1999-4458.
- [5] J. Keller, "Startup to Develop Satellite-On-A-Chip," *Military & Aerospace Electronics*, Vol. 5, No. 2, Feb. 1994, p. 1
- [6] D. J. Barnhart, T. Vladimirova, and M. N. Sweeting, "Satellite-on-a-Chip: A Feasibility Study," in *Proc. Fifth Round Table on Micro/Nano Technologies for Space Workshop*, Noordwijk, The Netherlands, 2005, pp. 728-735.
- [7] D. J. Barnhart, T. Vladimirova, and M. N. Sweeting, "Satellite-on-a-Chip Development for Future Distributed Space Missions," in *Proc. CANEUS Micro-Nano Technologies for Aerospace Applications Conf.*, Toulouse, France, 2006, Paper CANEUS 2006-11045.
- [8] S. Janson, A. Huang, W. Hansen, L. Steffeney, and H. Helvajian, "Development of an Inspector Satellite Using Photostructurable Glass/Ceramic Materials," in *Proc. AIAA Space 2005*, Long Beach, CA, 2005, Paper AIAA-2005-6802.
- [9] D. J. Barnhart, T. Vladimirova, M. N. Sweeting, R. L. Balthazor, L. C. Enloe, L. H. Krause, T. J. Lawrence, M. G. Mcharg, J. C. Lyke, J. J. White, A. M. Baker, "Enabling Space Sensor Networks with PCBSat," in *Proc. USU/AIAA Small Satellite Conf.*, Logan, UT, 2007, Paper SSC07-IV-4.
- [10] K. Romer and F. Mattern, "The Design Space of Wireless Sensor Networks," *IEEE Wireless Communications*, vol. 11, no. 6, Dec. 2004, pp. 54-61.
- [11] K. Bult, A. Burstein, D. Chang, M. Dong, M. Fielding, E. Kruglick, J. Ho, F. Lin, T. H. Lin, W. J. Kaiser, H. Marcy, R. Mukai, P. Nelson, F. Newberg, K. S. J. Pister, G. Pottie, H. Sanchez, O. M. Stafsudd, K. B. Tan, C. M. Ward, S. Xue, and J. Yao, "Low Power Systems

- for Wireless Microsensors”, in *Proc. 1996 International Symposium on Low Power Electronics and Design*, Monterey, CA, 1996, pp. 17-21.
- [12] V. S. Hsu, J. M. Kahn, and K. S. J. Pister, “Wireless Communications for Smart Dust,” Electronics Research Laboratory Memorandum, University of California, Berkeley, no. M98/2, 1998.
- [13] B. Warneke, M. Last, B. Liebowitz, and K. S. J. Pister, “Smart Dust: Communicating with a Cubic-Millimeter Computer,” *IEEE Computer*, vol. 34, no. 1, Jan. 2001, pp. 44-51.
- [14] B. A. Warneke, M. D. Scott, B. S. Leibowitz, Z. Lixia, C. L. Bellew, J. A. Chediak, J. M. Kahn, B. E. Boser, and K. S. J. Pister, “An Autonomous 16 mm³ Solar-powered Node for Distributed Wireless Sensor Networks,” in *Proc. IEEE Sensors 2002*, vol. 2, 2002, pp. 1510-1515.
- [15] S. E. Hollar, “COTS Dust,” M.S. Thesis, Dept. Mech. Eng., Univ. of California, Berkeley, CA, 2000.
- [16] G. Asada, M. Dong, T. S. Lin, F. Newberg, G. Pottie, and W. J. Kaiser, “Wireless Integrated Network Sensors: Low-Power Systems on a Chip,” in *Proc. 24th European Solid-State Circuits Conf.*, Editions Frontieres, Paris, 1998, pp. 9-16.
- [17] C. C. Enz, A. El-Hoiydi, J.-D. Decotignie, and V. Peiris, “WiseNET: An Ultralow-power Wireless Sensor Network Solution,” *IEEE Computer*, vol. 37, no. 8, Aug. 2004, pp. 62-70.
- [18] B. W. Cook, S. Lanzisera, and K. S. J. Pister, “SoC Issues for RF Smart Dust,” *Proceedings of the IEEE*, vol. 94, no. 6, Jun. 2006, pp. 1177-1196.
- [19] D. J. Barnhart, T. Vladimirova, M. N. Sweeting, R. L. Balthazor, L. C. Enloe, L. H. Krause, T. J. Lawrence, M. G. Mcharg, J. C. Lyke, J. J. White, A. M. Baker, “Enabling Space Sensor Networks with PCBSat,” in *Proc. USU/AIAA Small Satellite Conf.*, Logan, UT, 2007, Paper SSC07-IV-4.
- [20] B. J. Hosticka, “CMOS Sensor Systems,” in *Proc. International Conf. on Solid State Sensors and Actuators*, vol. 2, Chicago, 1997, pp. 991-994..
- [21] O. Yadid-Pecht and R. Etienne-Cummings, *CMOS Imagers: From Phototransduction to Image Processing*, Boston, MA, Kluwer Academic Publishers, 2004.
- [22] M. He, X. Yuan, J. K. Moh, J. Bu, and X.-J. Yi, “Monolithically Integrated Refractive Microlens Array to Improve Imaging Quality of an Infrared Focal Plane Array,” *Optical Engineering*, vol. 43(11), 2004, pp. 2589-2594.
- [23] S. Ghosh and M. Bayoumi, “On Integrated CMOS-MEMS System-on-Chip,” in *Proc. 3rd International IEEE-NEWCAS Conf.*, Quebec, 2005, pp. 31-34.

- [24] International Technology Roadmap for Semiconductors (ITRS) [online database]. <http://www.itrs.net/>
- [25] R. L. Petritz, "Current Status of Large Scale Integration Technology," *IEEE Journal of Solid-State Circuits*, Vol. 2, No. 4, Dec. 1967, pp. 130-147.
- [26] E. E. Swartzlander, "VLSI, MCM, and WSI: A Design Comparison," *IEEE Design and Test of Computers*, vol. 15, no. 3, July 1998, pp. 28-34.
- [27] L. Alkalai and W-C Fang, "An Integrated Microspacecraft Avionics Architecture Using 3D Multichip Module Building Blocks," in *Proc. 1996 IEEE International Conference on Computer Design: VLSI in Computers and Processors*, Austin, TX, 1996, pp. 141-144.
- [28] J. Spoto, "Looking Beyond Monolithic Myopia," *IEE Electronic Systems and Software*, vol. 1, no. 4, August/September 2003, pp. 12-15.
- [29] M. M. Mojarradi, E. Brandon, R. Bugga, E. Wesseling, U. Lieneweg, H. Li, and B. Blalock, "Power Management and Distribution for System On a Chip for Space Applications," in *Proc. AIAA Space 1999*, Albuquerque, NM, 1999, Paper AIAA-1999-4686.
- [30] S. Roundy, D. Steingart, L. Frechette, P. Wright, and J. Rabaey, "Power Sources for Wireless Sensor Networks," in *Proc. 1st European Workshop on Wireless Sensor Networks*, vol. 29, 2004, pp. 1-17.
- [31] C. L. Bellew, S. Hollar, and K. S. J. Pister, "An SOI Process for Fabrication of Solar Cells, Transistors, and Electrostatic Actuators," in *Proc. 12th International Conf. on Solid State Sensors, Actuators and Microsystems*, vol. 2, Boston, MA, 2003, pp. 1075-1078.
- [32] L. Nazhandali, B. Zhai, J. Olson, A. Reeves, M. Minuth, R. Helfand, S. Pant, T. Austin, and D. Blaauw, "Energy Optimization of Subthreshold-Voltage Sensor Network Processors," in *Proc. 32nd International Symposium on Computer Architecture*, Madison, WI, 2005, pp. 197-207.
- [33] C.-Y. Wu, F. Cheng, C.-T. Chiang, and P.-K. Lin, "A Low-power Implantable Pseudo-BJT-based Silicon Retina With Solar Cells for Artificial Retinal Prostheses," in *Proc. 2004 International Symposium on Circuits and Systems*, Vancouver, Canada, vol. 4, 2004, pp. IV-37-40.
- [34] C. Wang and F. Devos, "Design and Implementation of Electrical-Supply-Free VLSI Circuits," *IEE Proceedings of Circuits, Devices and Systems*, vol. 152, no. 3, Jun. 2005, pp. 272-278.

- [35] S. Bermejo, P. Ortega, and L. Castañer, "Fabrication of Monolithic Photovoltaic Arrays on Crystalline Silicon by Wafer Bonding and Deep Etching Techniques," *Prog. Photovolt.: Res. Appl.*, vol. 13, 2005, pp. 1-9.
- [36] G. Erdler, M. Frank, M. Lehmannb, H. Reinecke, and C. Müller, "Chip Integrated Fuel Cell," *Sensors and Actuators*, vol. 132, 2006, pp. 331-336.
- [37] R. Wartena, A. E. Curtright, C. B. Arnold, A. Pique, K. E. Swider-Lyons, "Li-ion Microbatteries Generated by a Laser Direct-write Method," *Journal of Power Sources*, vol. 126, no. 1, Feb 2004, pp. 193-202.
- [38] T. T. Le, J. Han, A. von Jouanne, K. Mayaram, and T. S. Fiez, "Piezoelectric Micro-power Generation Interface Circuits," *IEEE Journal of Solid-State Circuits*, vol. 41, no. 6, Jun. 2006, pp. 1411-1420.
- [39] C. Sauer, M. Stanacevic, G. Cauwenberghs, and N. Thakor, "Power Harvesting and Telemetry in CMOS for Implanted Devices," *IEEE Trans. Circuits and Systems*, vol. 52, no. 12, Dec. 2005, pp. 2605-2613.
- [40] T. Vladimirova and M. N. Sweeting, "System-on-a-Chip Development for Small Satellite Onboard Data Handling," *AIAA Journal of Aerospace Computing, Information, and Communication 2004*, vol. 1, no.1, 2004, pp. 36-43.
- [41] C. Chien, *Digital Radio Systems on a Chip: A Systems Approach*, Boston, MA, Kluwer Academic Publishers, 2001.
- [42] J. Lin, X. Guo, R. Li, J. Branch, and J. E. Brewer, "10 Times Improvement of Power Transmission Over Free Space Using Integrated Antennas on Silicon Substrates," in *Proc. IEEE Custom Integrated Circuits Conf.*, Orlando, FL, 2004, pp. 697-700.
- [43] K. K. O, K. Kim, B. A. Floyd, J. L. Mehta, H. Yoon, C.-M. Hung, D. Bravo, T. O. Dickson, X. Guo, R. Li, N. Trichy, J. Caserta, W. R. Bomstad, J. Branch, D.-J. Yang, J. Bohorquez, E. Seok, L. Gao, A. Sugavanam, J.-J. Lin, J. Chen, and J. E. Brewer, "On-Chip Antennas in Silicon ICs and Their Application," *IEEE Transactions on Electron Devices*, vol. 52, no. 7, Jul. 2005, pp. 1312-1323.
- [44] H. Stockman, "Communication by Means of Reflected Power," *Proceedings of the IRE*, Oct. 1948, pp. 1196-1204.
- [45] P. H. R. Popplewell, V. Karam, A. Shamim, A. J. Rogers, and C. Plett, "An Injection-Locked 5.2 GHz SoC Transceiver with On-Chip Antenna for Self-Powered RFID and Medical Sensor Applications," in *Proc. 2007 IEEE Symposium on VLSI Circuits*, Jun. 2007, pp. 88 - 89.

- [46] D. J. Barnhart, T. Vladimirova, and M. N. Sweeting, "Very Small Satellite Design for Distributed Space Missions," *AIAA Journal of Spacecraft and Rockets*, vol. 44, no. 6, Nov-Dec 2007, pp. 1294-1306.
- [47] J. P. McKelvey, *Solid State Physics*, Malabar, FL, Krieger Publishing Company, 1993.
- [48] H. Shao, C-Y Tsui, and W-H Ki, "An Inductor-less Micro Solar Power Management System Design for Energy Harvesting Applications," in *Proc. 2007 International Symposium on Circuits and Systems*, New Orleans, LA, 2007, pp. 1353-1356.
- [49] A. Holmes-Siedle and L. Adams, *Handbook of Radiation Effects*, 2nd Ed., Oxford University Press, 2002.
- [50] C. P. Brothers and D. Alexander, "Radiation Hardening Techniques for Commercially Produced Microelectronics for Space Guidance and Control Applications," in *Proc. 20th Annual American Astronautical Society Guidance and Control Conf.*, Breckenridge, CO, 1997, pp. 169-180.
- [51] T. Ma and P. Dressendorfer, *Ionizing Effects in MOS Devices and Circuits*, New York: John Wiley and Sons, 1989.
- [52] M. Hartwell, C. Hafer, P. Milliken, and T. Farris, "Megarad Total Ionizing Dose and Single Event Effects Test Results of a Radhard-by-Design 0.25 Micron ASIC," *Radiation Effects Data Workshop*, Atlanta, GA, 2004, pp. 104-109.
- [53] A. J. Martin and M. Nystrom, "Asynchronous Techniques for System-on-Chip Design," *Proceedings of the IEEE*, vol. 94, no. 6, Jun. 2006, pp. 1089-1120.
- [54] S. H. Unger, *Asynchronous Sequential Switching Circuits*, New York: Wiley-Interscience, 1969.
- [55] I. Blunno, J. Cortadella, A. Kondratyev, L. Lavagno, K. Lwin, and C. Sotiriou, "Handshake Protocols for De-synchronization," in *Proc. International Symposium on Asynchronous Circuits and Systems*, Hersonissos, Crete, 2004.
- [56] S. Hauck, "Asynchronous Design Methodologies: An Overview," in *Proc. of the IEEE*, vol. 83, Jan. 1995, pp. 69-93.
- [57] C. Farnsworth, D. A. Edwards, and S. S. Sikand, "Utilising Dynamic Logic for Low Power Consumption in Asynchronous Circuits," in *Proc. International Symposium on Asynchronous Circuits and Systems*, Salt Lake City, UT, 1994, pp. 186-194.
- [58] K. Y. Yun, "Synthesis of Asynchronous Controllers for Heterogeneous Systems," Ph.D. dissertation, Stanford University, Stanford, CA, 1994.

- [59] J. A. Brzozowski and C-J. H. Seger. *Asynchronous Circuits*. Springer-Verlag, 1995.
- [60] B. W. Hunt, K. S. Stevens, B. W. Suter, and D. S. Gelosh, "A Single Chip Low Power Asynchronous Implementation of an FFT Algorithm for Space Applications," in *Proc. International Symposium on Asynchronous Circuits and Systems*, San Diego, CA, 1998, pp. 216-223.
- [61] D. J. Barnhart, "An Improved Asynchronous Implementation of a Fast Fourier Transform Architecture for Space Applications," M.S. thesis, Air Force Institute of Tech., Wright-Patterson AFB, OH, Mar. 1999.
- [62] D. J. Barnhart, P. Duggan, B. Suter, C. Brothers, and K. Stevens, "Total Ionizing Dose Characterization of a Commercially Fabricated Asynchronous FFT for Space Applications," *Government Microcircuit Applications Conf. Digest of Papers*, Anaheim, CA, March 2000.
- [63] D. F. Cox, "Asynchronous Logic Design with Subcells with an Application for Space," in *Proc. IEEE International Conference on Electronics, Circuits and Systems*, Paphos, Cyprus, 1999, pp. 1225- 1230.
- [64] W. Jang and A. J. Martin, "SEU-tolerant QDI Circuits," in *Proc. International Symposium on Asynchronous Circuits and Systems*, New York, 2005, pp. 156-165.
- [65] M. Renaudin and Y. Monnet, "Asynchronous Design: Fault Robustness and Security Characteristics," in *Proc. IEEE International On-Line Testing Symposium*, Como, Italy, 2006.
- [66] D. A. Patterson and J. L. Hennessy, *Computer Organization and Design, 3rd ed.*, Morgan Kaufmann, 2007, pp. 282-339.