

Mechanical Properties of G-10 Glass–Epoxy Composite

K. Ravi-Chandar and S. Satapathy

Institute for Advanced Technology
The University of Texas at Austin

IAT.R 0466

Approved for public release; distribution unlimited.

REPORT DOCUMENTATION PAGE

Form Approved
OMB NO. 0704-0188

Public reporting burden for this collection of information is estimated to average 1 hour per response, including the time for reviewing instructions, searching existing data sources, gathering and maintaining the data needed, and completing and reviewing the collection of information. Send comments regarding this burden estimate or any other aspect of this collection of information, including suggestions for reducing this burden, to Washington Headquarters Services, Directorate for Information Operations and Reports, 1215 Jefferson Davis Highway, Suite 1204, Arlington, VA 22202-4302, and to the Office of Management and Budget, Paperwork Reduction Project (0704-0188), Washington, DC 20503.

1. AGENCY USE ONLY (Leave blank)		2. REPORT DATE August 2007	3. REPORT TYPE AND DATES COVERED Technical Report (Sept. 2005-Sept. 2006)	
4. TITLE AND SUBTITLE Mechanical Properties of G-10 Glass-Epoxy Composite			5. FUNDING NUMBERS Contract # DAAD17-01-D-0001 DO 0012	
6. AUTHOR(S) K. Ravi-Chandar and S. Satapathy				
7. PERFORMING ORGANIZATION NAME(S) AND ADDRESS(ES) Institute for Advanced Technology The University of Texas at Austin 3925 W. Braker Lane, Suite 400 Austin, TX 78759-5316			8. PERFORMING ORGANIZATION REPORT NUMBER IAT.R 0466	
9. SPONSORING / MONITORING AGENCY NAME(S) AND ADDRESS(ES) US Army Research Laboratory ATTN: AMSRL-WM-B Aberdeen Proving Ground, MD 21005-5066			10. SPONSORING / MONITORING AGENCY REPORT NUMBER	
11. SUPPLEMENTARY NOTES The views, opinions, and/or findings contained in this report are those of the author(s) and should not be considered as an official Department of the Army position, policy, or decision, unless so designated by other documentation.				
12a. DISTRIBUTION / AVAILABILITY STATEMENT Approved for public release; distribution unlimited.			12b. DISTRIBUTION CODE A	
13. ABSTRACT (Maximum 200 words) The mechanical properties of G10 glass-epoxy composites were determined in compression and tension tests.				
14. SUBJECT TERMS G10, glass-epoxy composite			15. NUMBER OF PAGES 12	
			16. PRICE CODE	
17. SECURITY CLASSIFICATION OF REPORT Unclassified	18. SECURITY CLASSIFICATION OF THIS PAGE Unclassified	19. SECURITY CLASSIFICATION OF ABSTRACT Unclassified	20. LIMITATION OF ABSTRACT UL	

MECHANICAL PROPERTIES OF G10 GLASS-EPOXY COMPOSITE

K. Ravi-Chandar¹ and S. Satapathy²

*The University of Texas at Austin
College of Engineering¹ and Institute for Advanced Technology²*

The mechanical properties of G10 glass-epoxy composites were determined in compression and tension tests. From weight and volume measurements, the volume fraction of glass was estimated to be about 56 percent. The mechanical property results are summarized in Tables I and II. The samples were cut from sheet stock with z-axis along the thickness direction. Appendix A shows the sample dimensions. The modulus values were determined by fitting a straight line to the initial linear portion of the stress-strain curves shown in Figs. 1–5. For the compression test, the strain was calculated from the displacement of the crosshead divided by its gage length; these are plotted in Figs. 1–3. The modulus was calculated after determining that the compliance of the loading system was 1.57×10^{-8} m/N. However, in the tension tests, a clip gage extensometer with a gage length of 1 inch was used. Therefore, the modulus measurements from the tension test are the correct values, while the modulus from the compression tests may be biased with the compliance of the loading system. Photographs of failed compression specimens are shown in Figs. 6 and 7. A similar characteristic failure pattern with a crack at a 45 degree angle is observed. The only difference between the X, Y, and Z was the peak stress at which it cracked.

Table I. Compressive Properties

	Young's Modulus	Peak Stress	Average Modulus	Average Peak Stress	Standard Deviation of Modulus	Standard Deviation of Peak Stress
	GPa	MPa	GPa	MPa	GPa	MPa
X-1	17.47	368	18.8	365	1.26	4
X-2	18.89	368				
X-3	19.97	360				
Y-1	19.90	302	18.9	300	0.89	5
Y-2	18.33	294				
Y-3	18.39	303				
Z-1	7.32	437	7.83	440	0.87	7
Z-2	7.33	436				
Z-3	8.83	448				

Table II. Tension Properties

	Young's Modulus	Peak Stress	Average Modulus	Average Peak Stress	Standard Deviation of Modulus	Standard Deviation of Peak Stress
	GPa	MPa	GPa	MPa	GPa	MPa
X-t-1	18.63	223	18.83	233	0.22	10
X-t-2	18.80	244				
X-t-3	19.06	232				
Y-t-1	19.02	319	19.26	310	1.15	9
Y-t-2	18.25	301				
Y-t-3	20.50	310				

Results from the high strain-rate tests in a split-Hopkinson compression experiment are summarized in Table III. Figs. 11–13 show the results of stress-strain obtained from the Hopkinson bar experiments with aluminum bars. Reliable reflected signals were not always obtained since the specimen began to crumble. Therefore, estimates of both the strain rates and strains are difficult to obtain; it is accurate to indicate that the strain levels were in the order of 10^3 s^{-1} . Dynamic modulus estimates from these tests are also not likely to be accurate and hence these calculations were not performed. The peak stress at breaking is rather well defined since this depends only on the intensity of the strain signal in the output bar of the Hopkinson arrangement. For the X and Y orientations, the compressive strength increases by nearly a factor of two and may be attributed to the strain rate dependence of the polymer matrix. For the Z orientation, the specimen crumbled into a powder suggesting that dilation of the specimen occurred during the nonlinear increasing part of the stress strain curve shown in Fig. 13. Proper interpretation of the data beyond a strain level of about 3 percent is not easily accomplished and the values of peak stress indicated is likely to be a significant overestimate of the actual peak stress.

Table III. Hopkinson-Bar Properties

	Peak Stress
	MPa
X-H-3	-677
X-H-4	-617
Y-H-2	-528
Y-H-3	-528
Z-H-1	-901
Z-H-2	-856

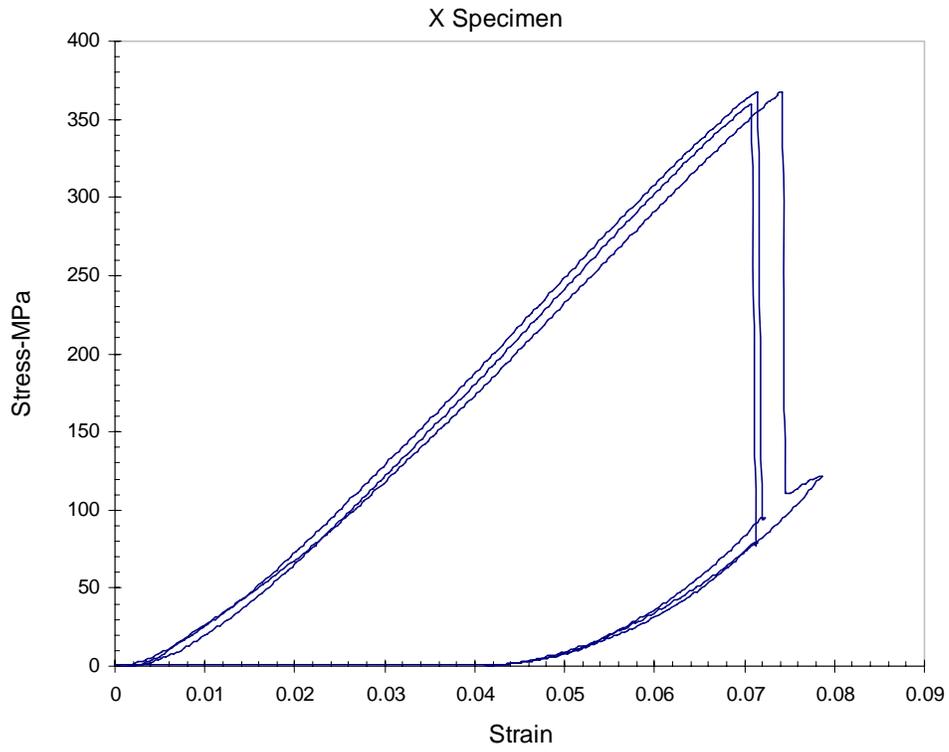


Figure 1. Stress-strain curve for G10-Orientation X; compression.

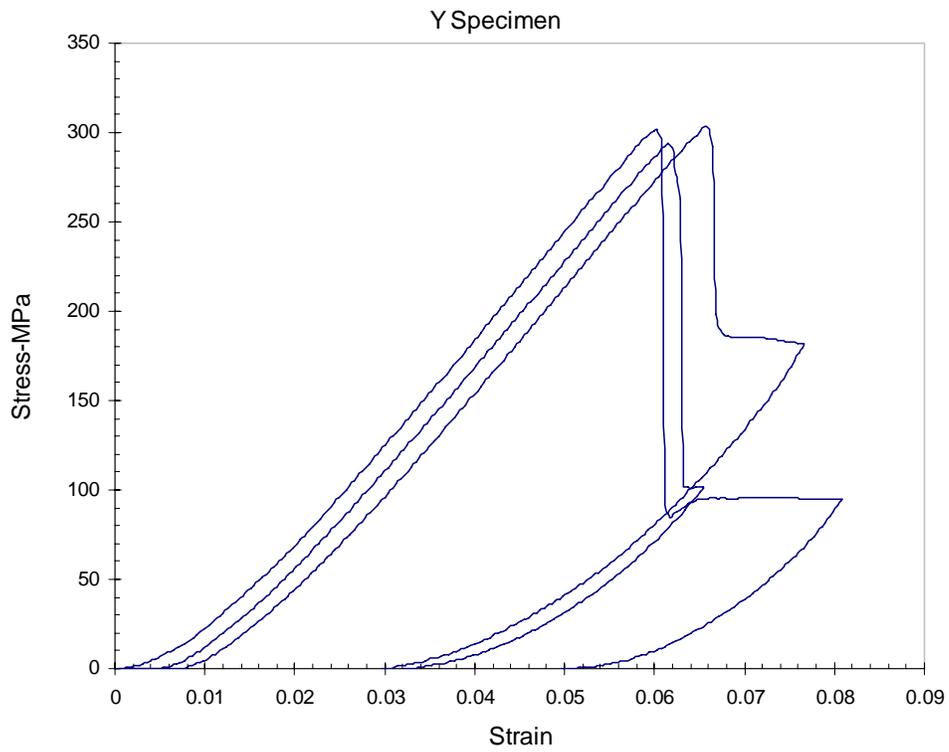


Figure 2. Stress-Strain curve for G10-Orientation Y; compression.

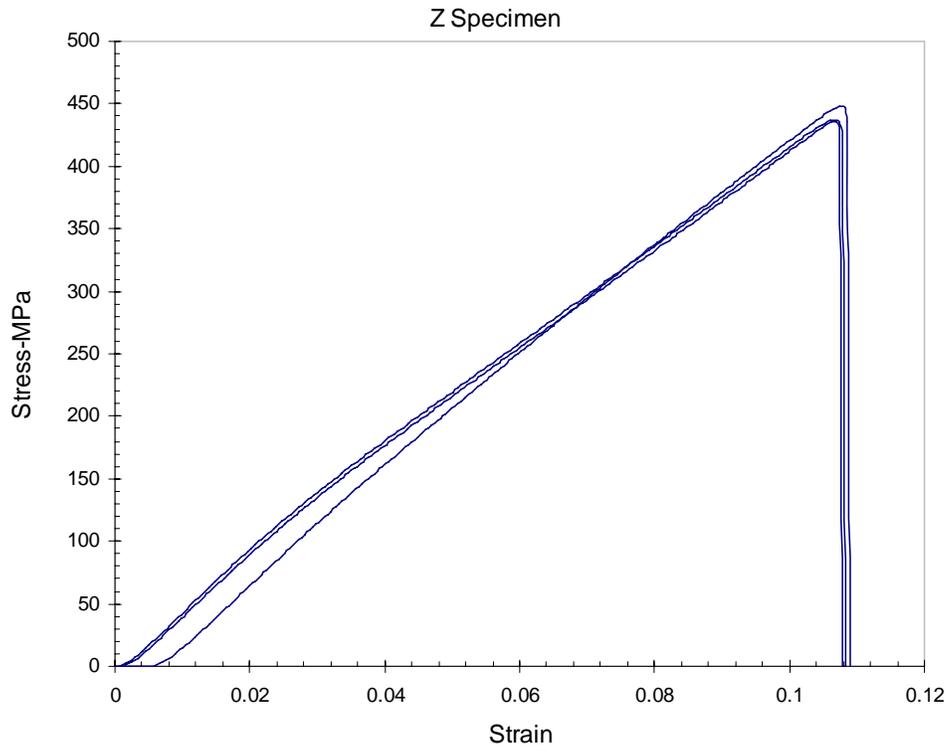


Figure 3. Stress-Strain curve for G10-Orientation Z; compression.

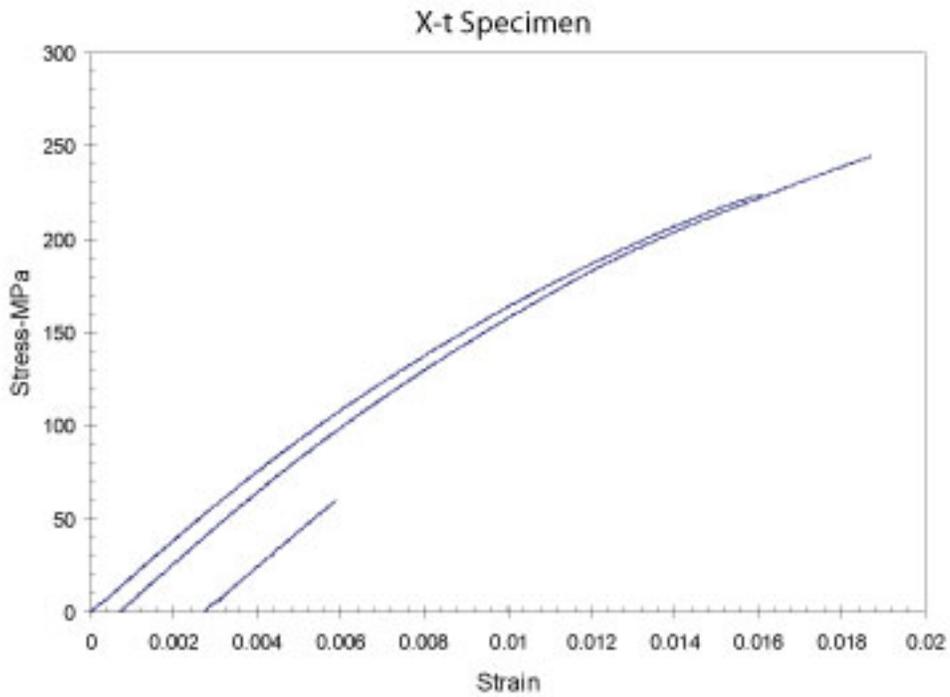


Figure 4. Stress-Strain curve for G10-Orientation X; tension.

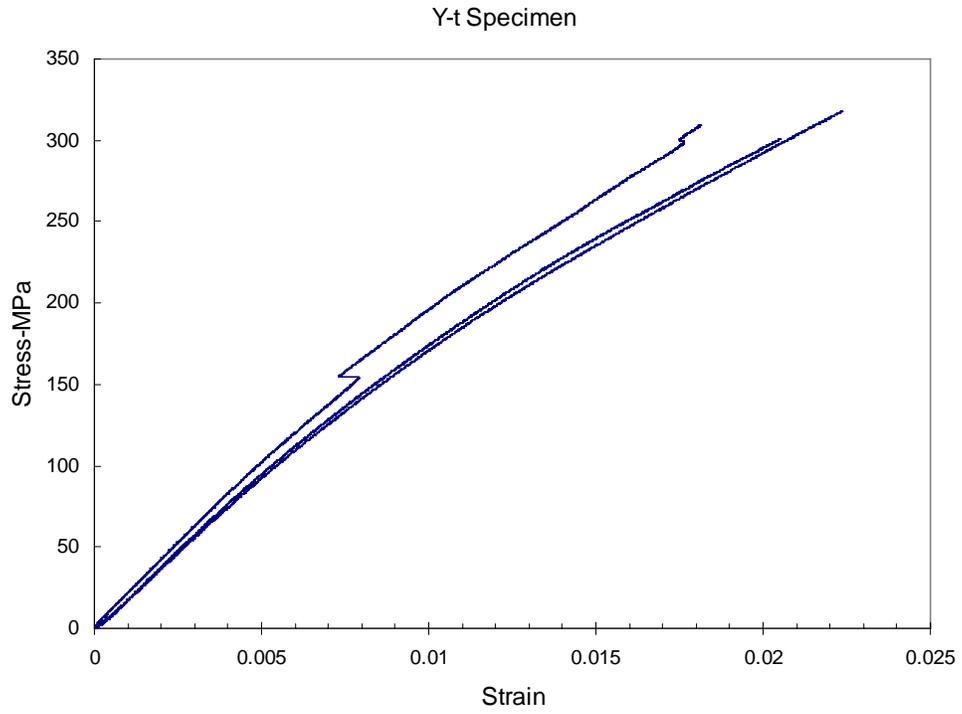


Figure 5. Stress-Strain curve for G10-Orientation Y; tension.



Figure 6. Photograph showing failure mode; Specimen X-3.



Figure 7. Photograph showing failure mode; Specimen Z-2.

The failure patterns in the tension tests are shown in Figs. 8, 9, and 10. While many of them splintered (Figs. 9 and 10), others had a cleaner break. However, all specimens broke on the gage section between the clip gage extensometer, indicating a valid peak stress measurement.

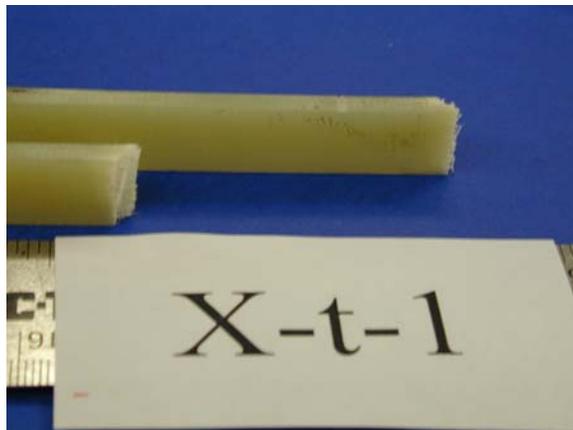


Figure 8. Photograph showing failure mode; Specimen X-t-1.

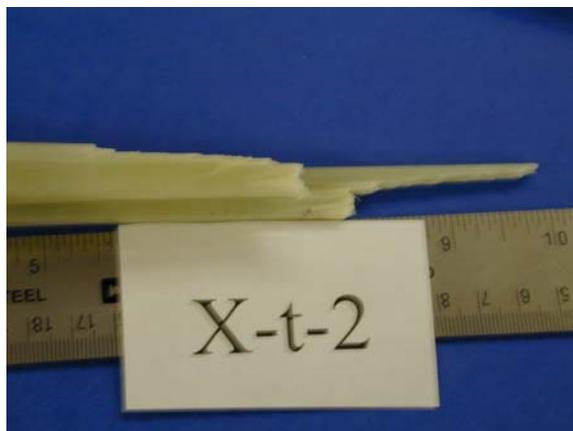


Figure 9. Photograph showing failure mode; Specimen X-t-2.

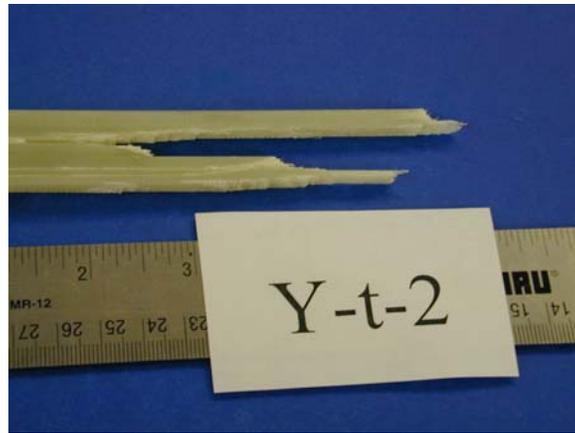


Figure 10. Photograph showing failure mode; Specimen Y-t-2.

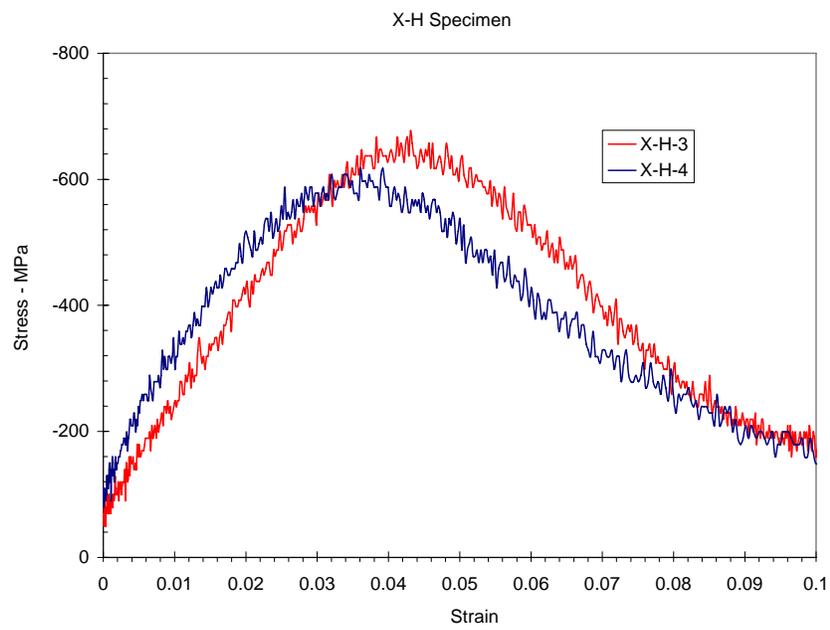


Figure 11. Stress-strain curve for G10-Orientation X; Hopkinson.

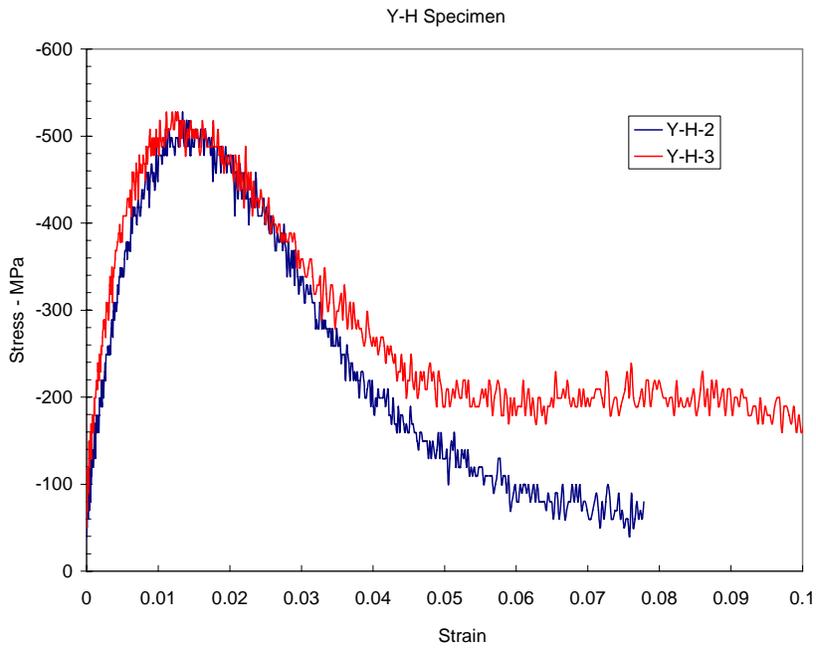


Figure 12. Stress-strain curve for G10-Orientation Y; Hopkinson.

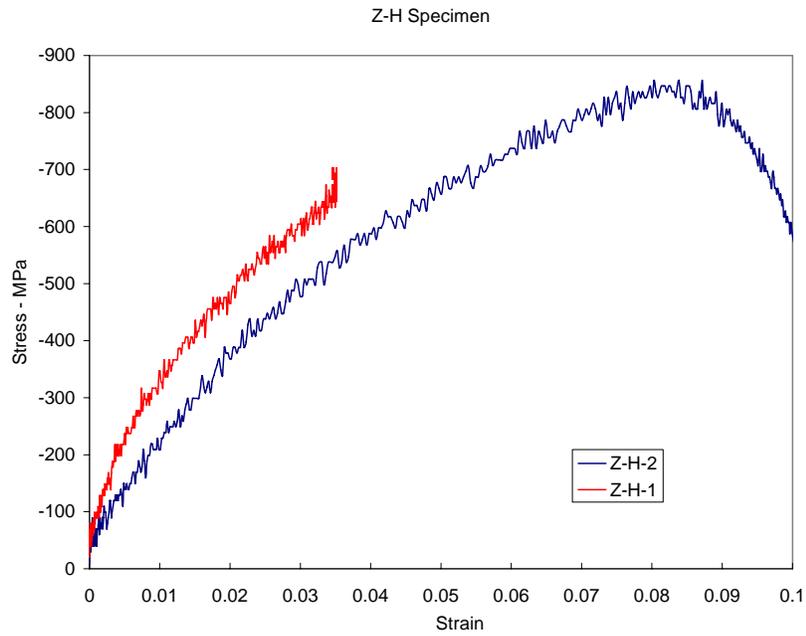


Figure 13. Stress-strain curve for G10-Orientation Z; Hopkinson.

ARL ACKNOWLEDGMENT

The research reported in this document was performed in connection with Contract number DAAD17-01-D-0001 with the US Army Research Laboratory. The views and conclusions contained in this document are those of the authors and should not be interpreted as presenting the official policies or position, either expressed or implied, of the US Army Research Laboratory or the US Government unless so designated by other authorized documents. Citation of manufacturers or trade names does not constitute an official endorsement or approval of the use thereof. The US Government is authorized to reproduce and distribute reprints for government purposes notwithstanding any copyright notation hereon.

APPENDIX A

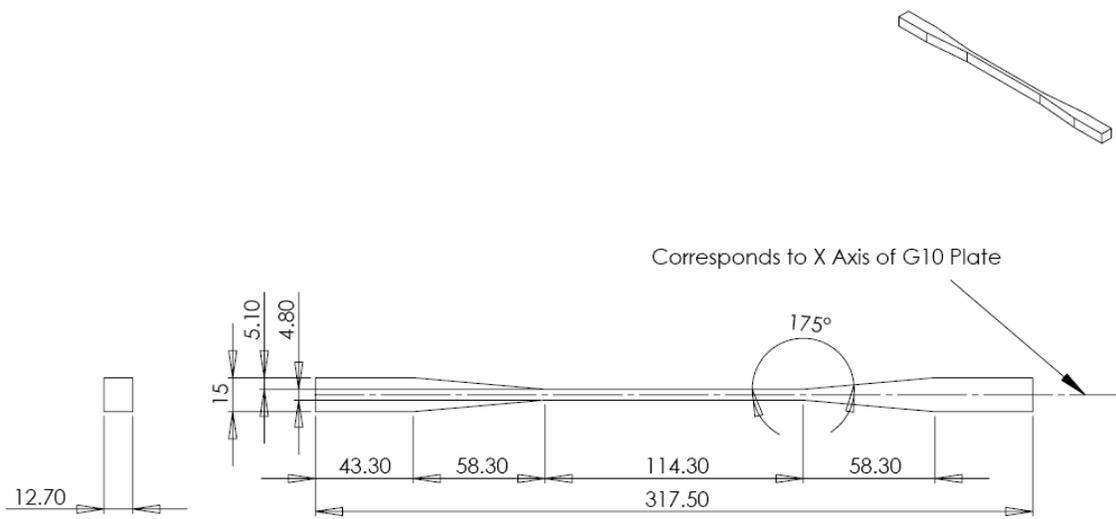


Figure A-1. Sample dimensions for tensile tests.

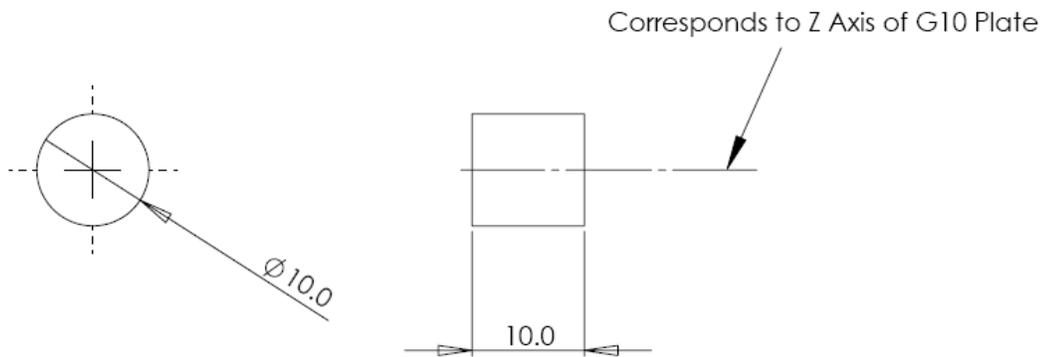


Figure A-2. Sample dimension for Split-Hopkinson bar experiment.