

ASPECTS OF USE OF CFD FOR UAV CONFIGURATION DESIGN

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Report Documentation Page

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UCAV DESIGN PROBLEM

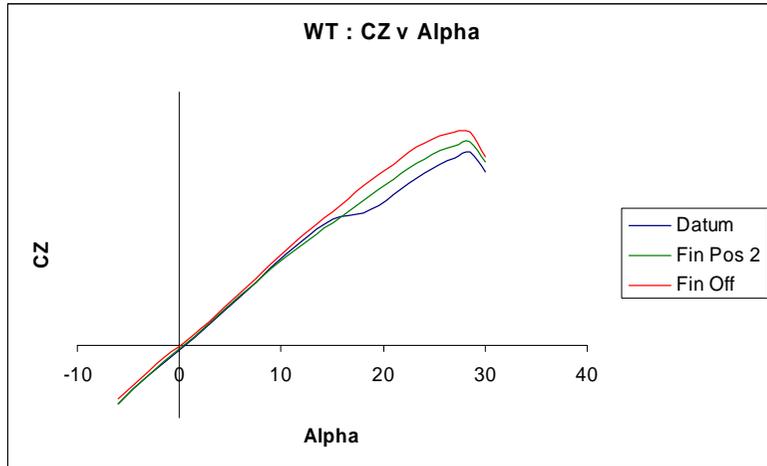
- Problem (for Aerodynamics) is as much due to novel planforms as Unmanned
- Novel planforms negate traditional Aerodynamic ground rules (sweep, span, AR etc)
- S&C is a significant challenge
- Requirement for rapid proto-typing for planform/basic layout studies and control surface optimisation
- Fast-response WT - small scale, stereo-lithography, PSP
- Fast-response CFD -Euler, High RE turb models RANS

TYPICAL EXAMPLE

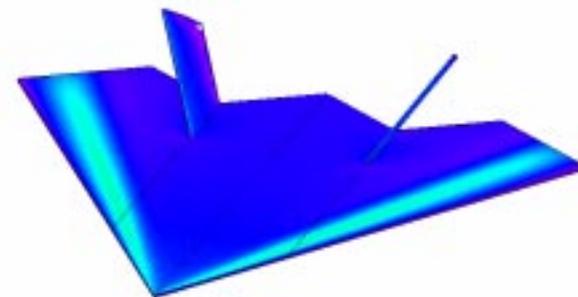
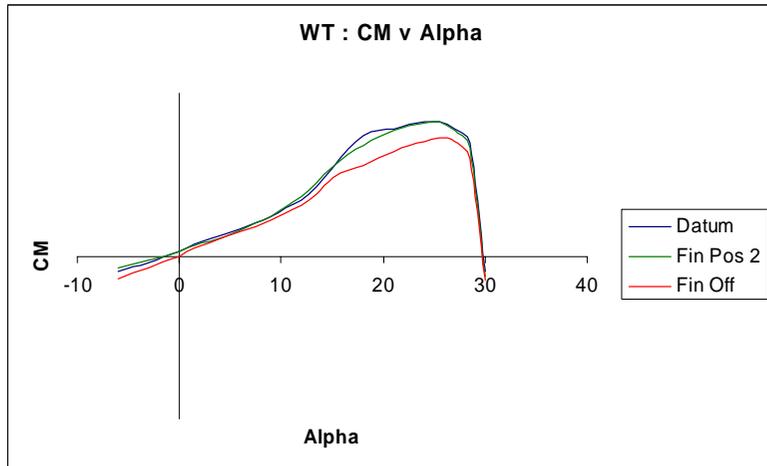
Investigation of Fin Position on a typical Novel Planform

- Establish credibility of CFD for prediction of general flow trends at low speed, high incidence for novel planforms
- Assist in interpretation of 'small-scale' wind tunnel testing

TYPICAL EXAMPLE



Small Scale WT testing -
Effect of fin position



Datum (flat-plate) model

NOTES ON CFD CALCULATIONS

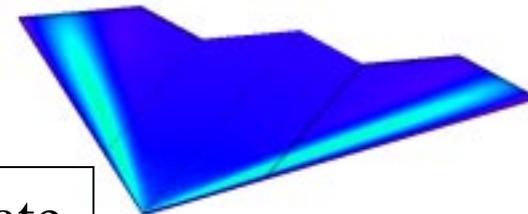
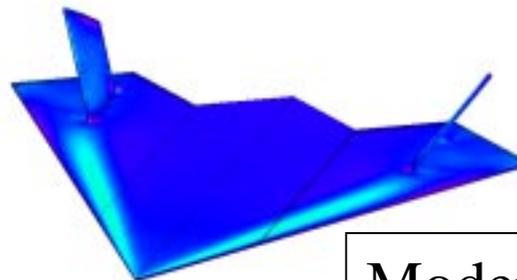
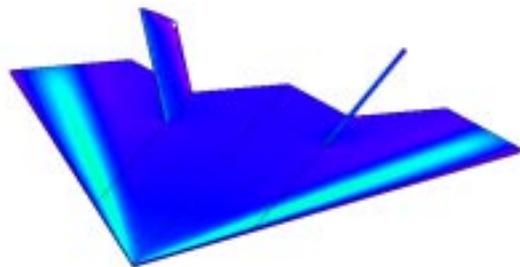
- 6-8million unstructured grid cells required for credible vortex capture from Euler, with particular emphasis on field resolution
- 2-3million 'BAE Systems Autogrid' cells required for equivalent capture from RANS
- $k\epsilon$ RNG turbulence model (wall function) suitable
- Euler solution turnaround 4hrs on 8 Origin processors, RANS 2 days

EFFECT OF FIN POSITION

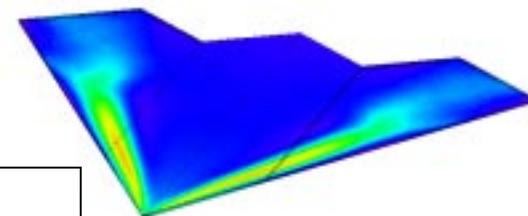
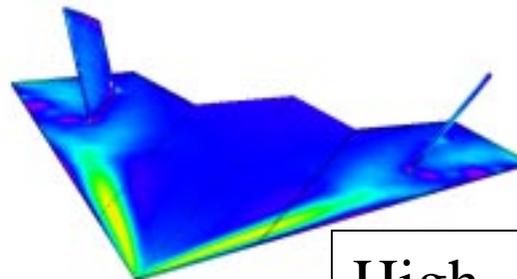
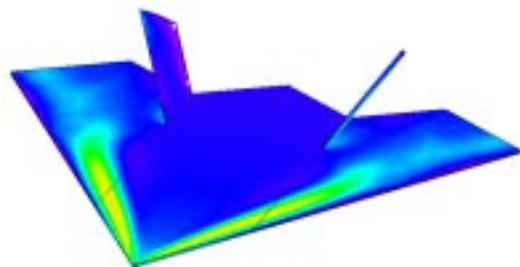
Datum

Fin Pos2

Fin Off



Moderate Incidence

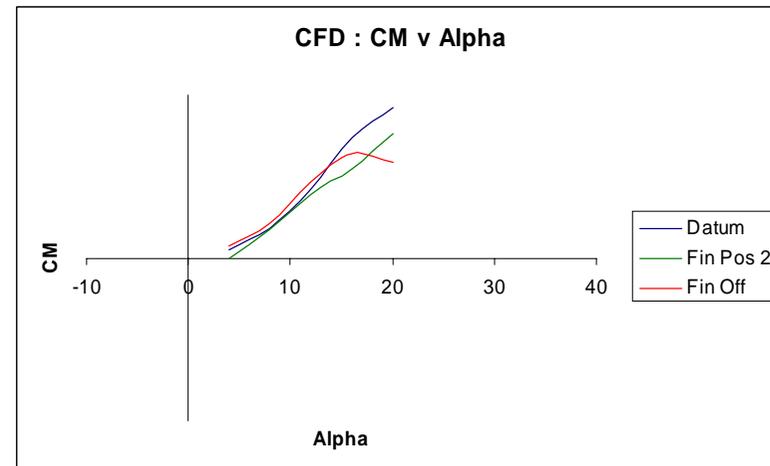
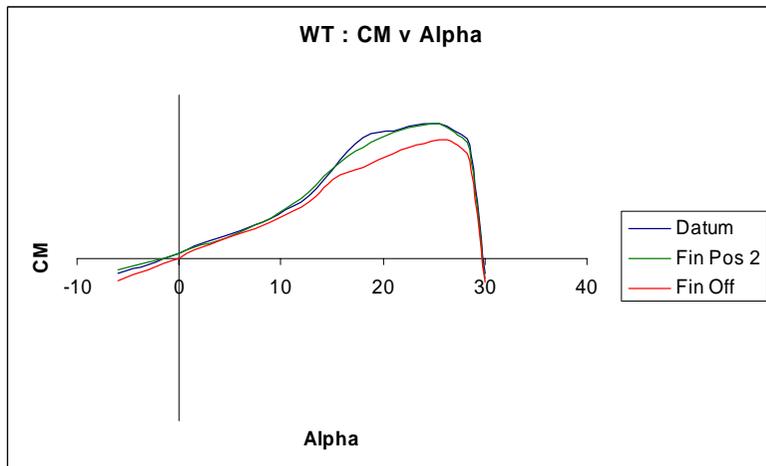
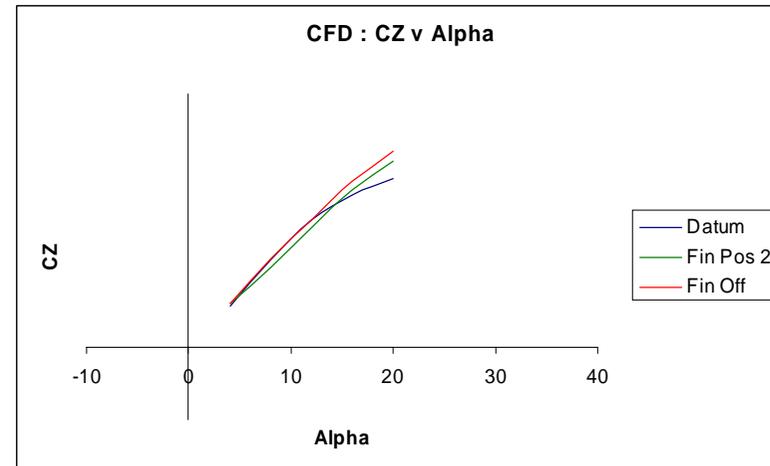
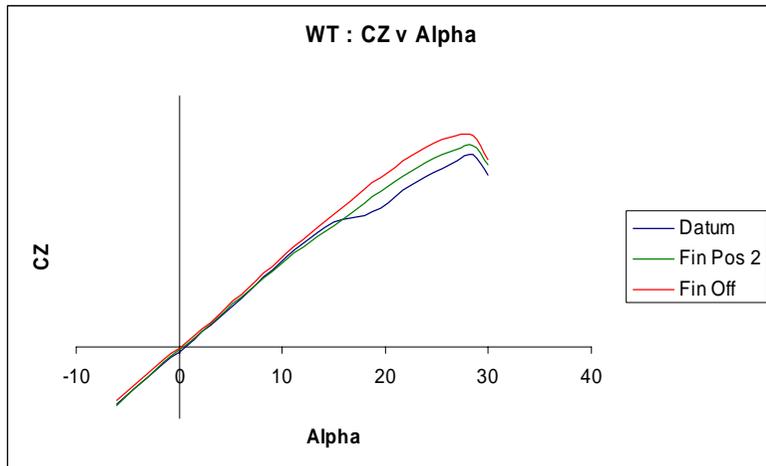


High Incidence

Flat Plate CFD Euler, local velocity contours

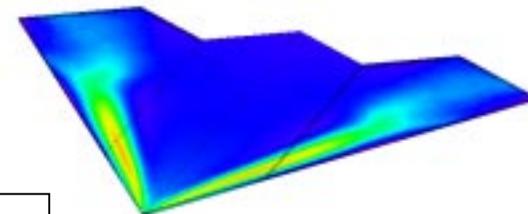
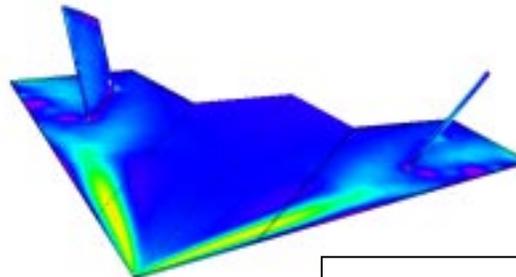
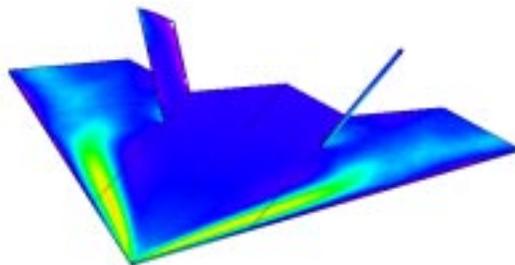
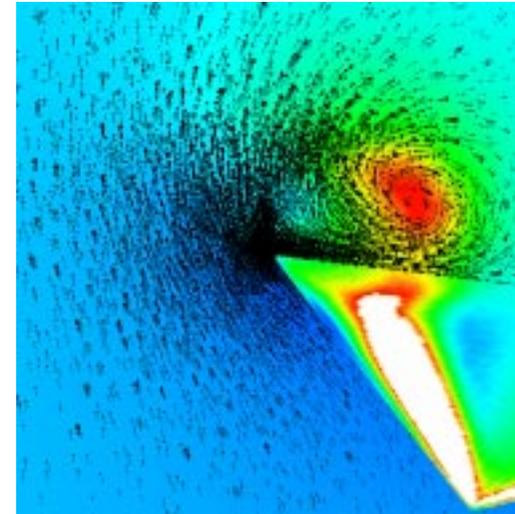
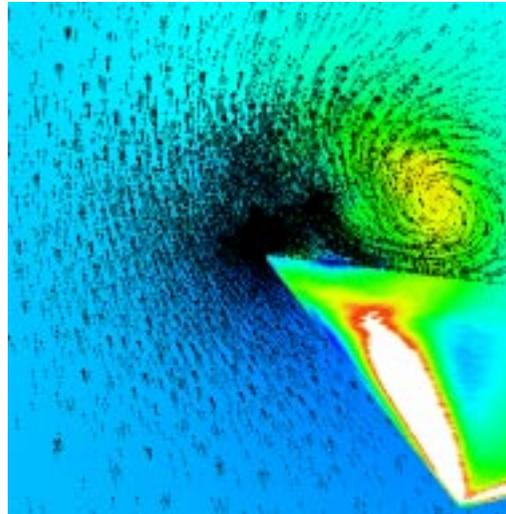
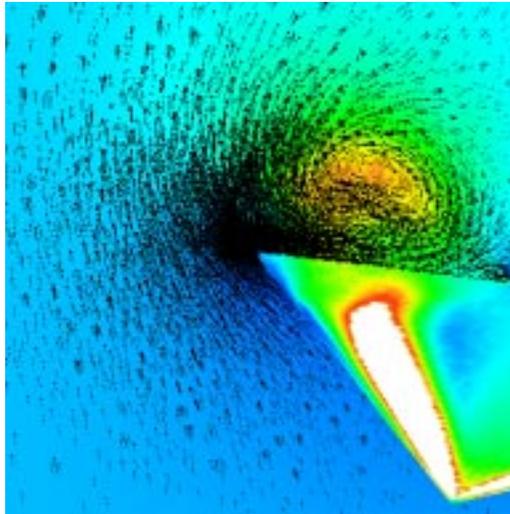
EFFECT OF FIN ON FORCES

BAE SYSTEMS



Flat Plate Wind Tunnel v CFD (Euler)

EFFECT OF FIN ON FLOWFIELD



High Inc

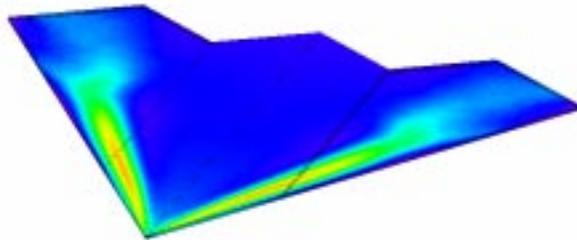
Datum

Fin Pos2

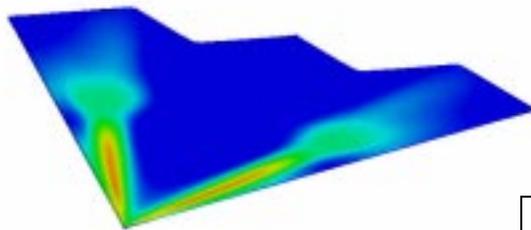
Fin Off

Flat Plate CFD Euler, velocity vectors, local vel contours

INVISCID v VISCOUS



EULER



RANS

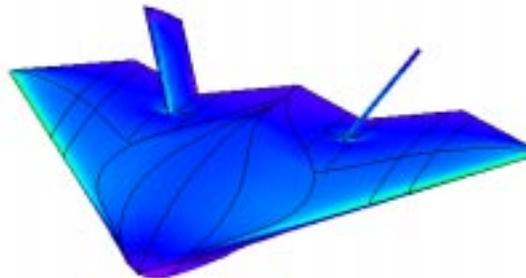
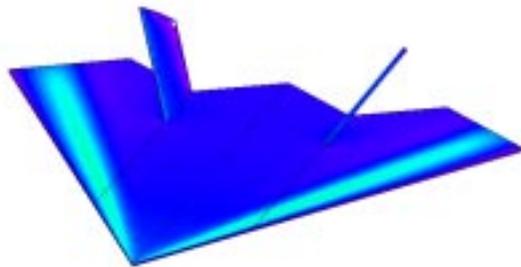
High Inc

Flat Plate CFD Euler v RANS

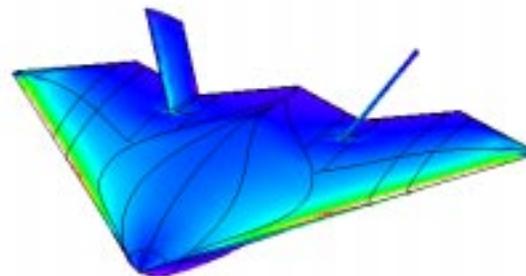
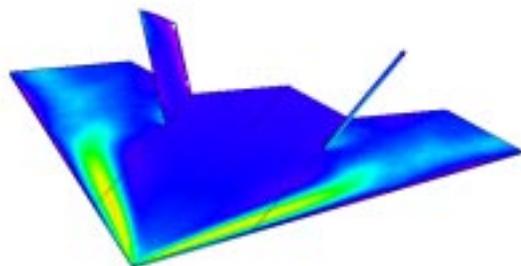
EFFECT OF THICKNESS

Flat Plate

10% t/c



Moderate Incidence



High Incidence

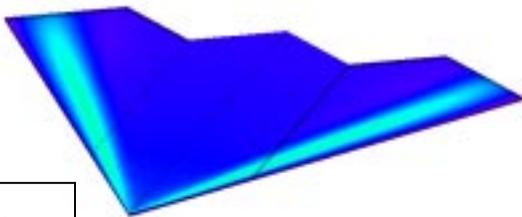
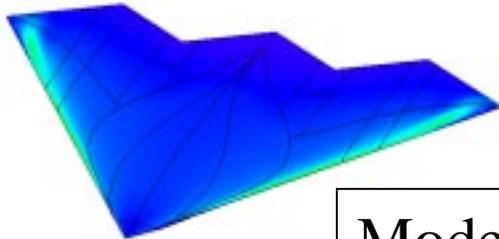
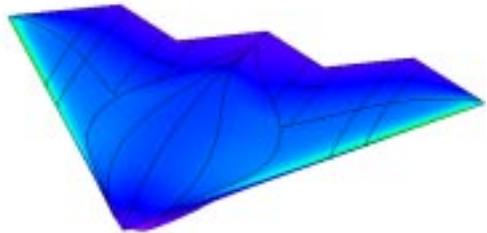
Flat Plate v Symmetric airfoil, CFD Euler, local vel contours

EFFECT OF THICKNESS (FIN OFF) BAE SYSTEMS

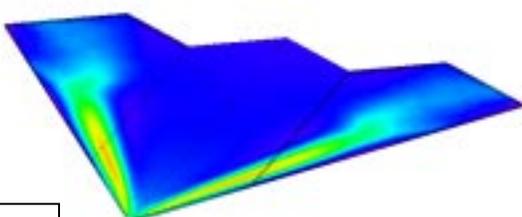
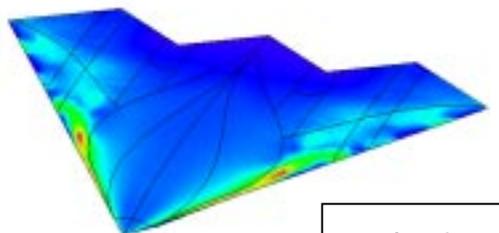
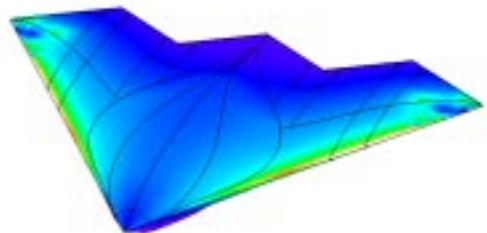
10% t/c

5% t/c

Flat Plate



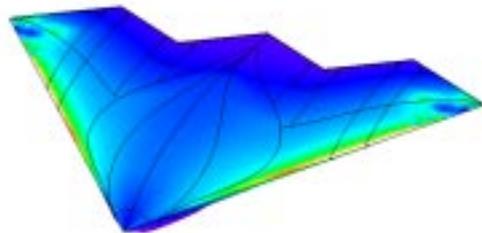
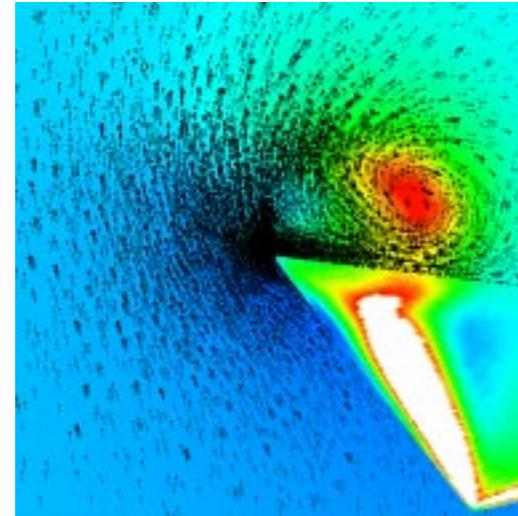
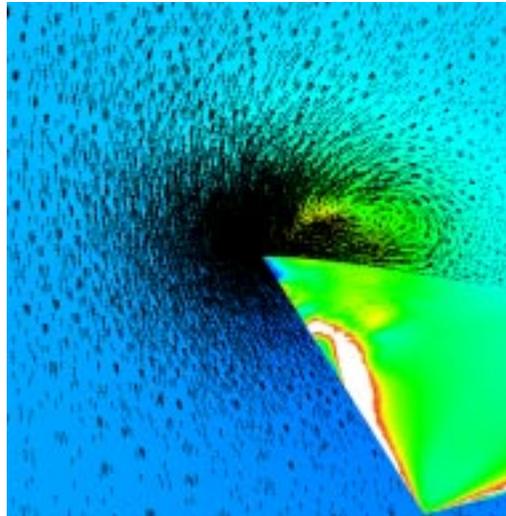
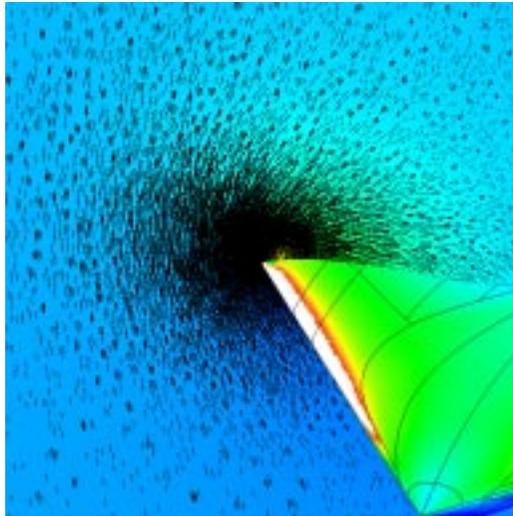
Moderate incidence



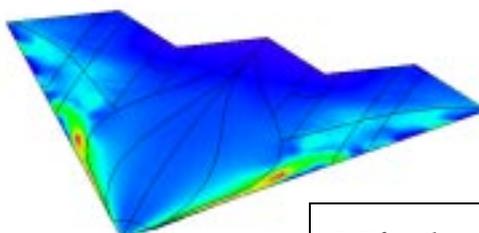
High Incidence

Flat Plate v Symmetric airfoil, CFD Euler, local vel contours

EFFECT OF T/C ON FLOWFIELD

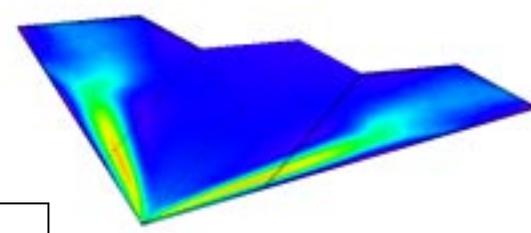


10% t/c



5% t/c

High Inc



Flat plate

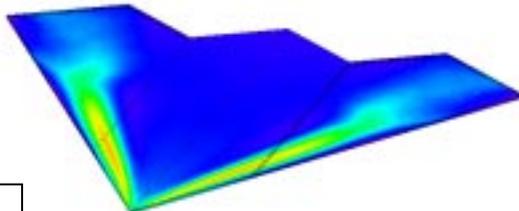
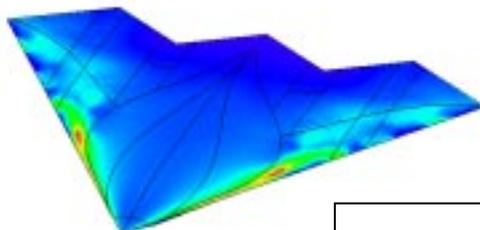
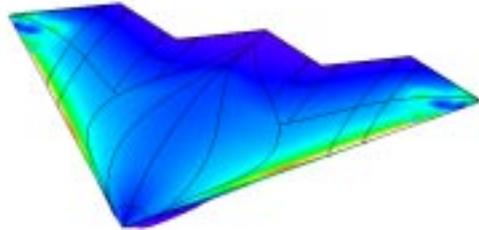
CFD Euler, velocity vectors and local vel contours

EFFECT OF T/C, EULER v RANS

10% t/c

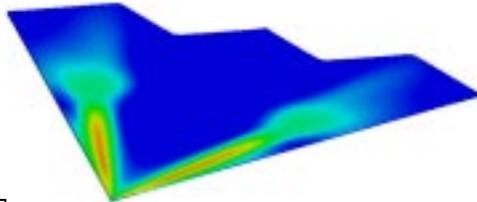
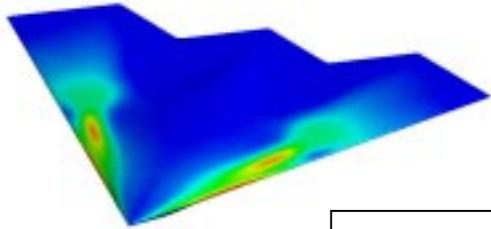
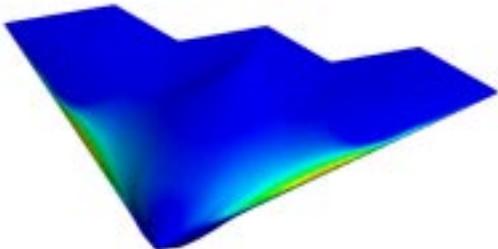
5% t/c

Flat Plate



Euler

High Inc

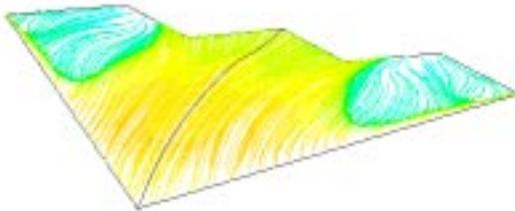


RANS

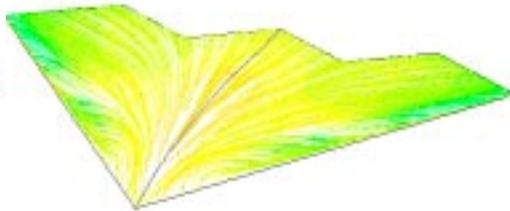
Flat Plate v Symmetric airfoil, Euler v RANS

EFFECT OF THICKNESS

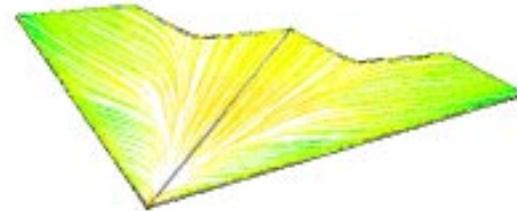
10% t/c



5% t/c



Flat Plate



High Inc

RANS

CFD RANS, surface flow patterns

SUMMARY

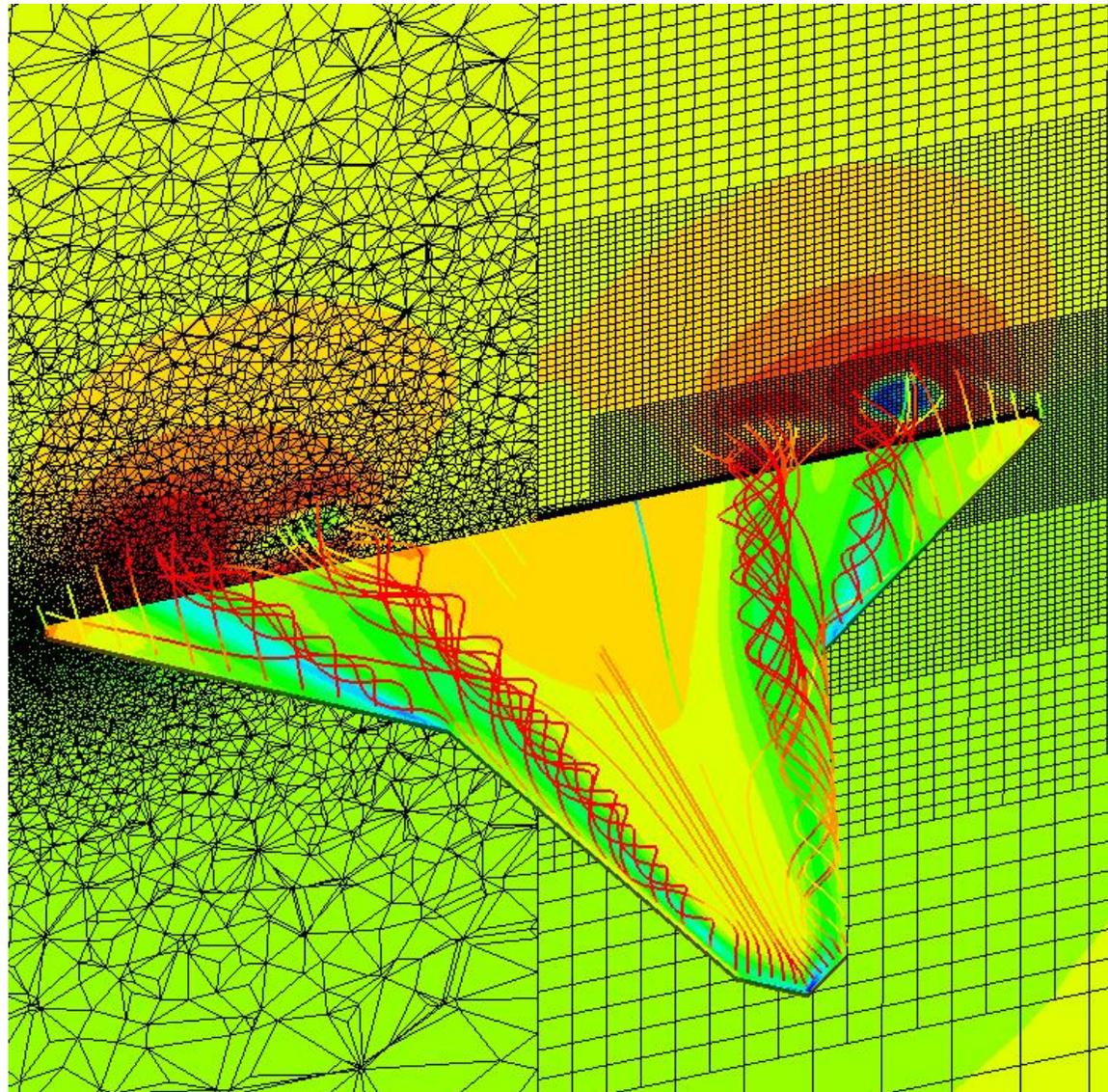
- Euler showing good prediction of flat plate
- Absolute values of pitching moment poor at high incidence, though engineering decisions can be made by interpretation
- RANS improves absolute predictions, though at too great an overhead in CPU time to be practical for design optimisation
- Difference in flow behaviour between thin and thick airfoils defines limit of applicability of flat plate wind tunnel models

FURTHER ASSESSMENT OF CFD

- RANS assessed on BAE Systems Autogrid meshes for a vortical flow case and a mixed attached/separated flow case
- kg results poor for both cases in terms of comparison with limited WT data, RANS ($k\epsilon$ RNG) and engineering judgement
- $k\epsilon$ RNG results good for both cases

VORTICAL FLOW CASE

EULER

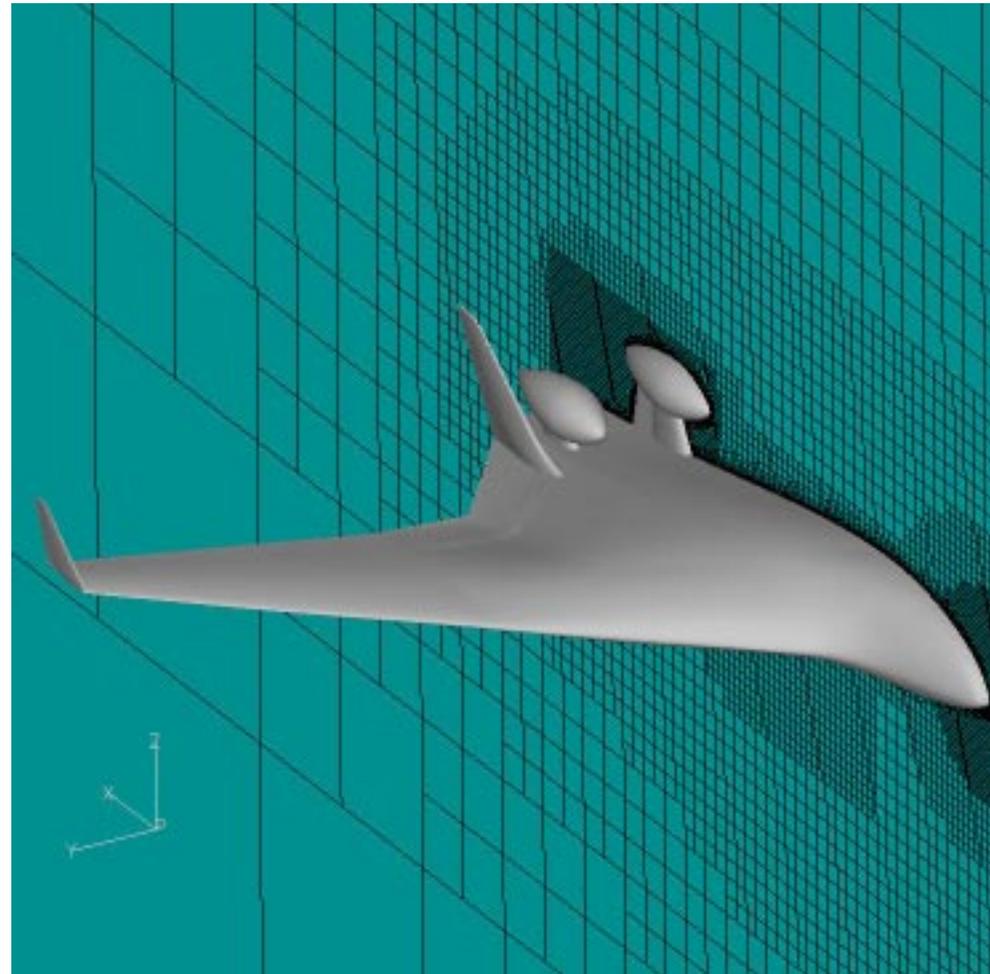


RANS

MIXED ATTACHED/SEPARATED FLOW CASE

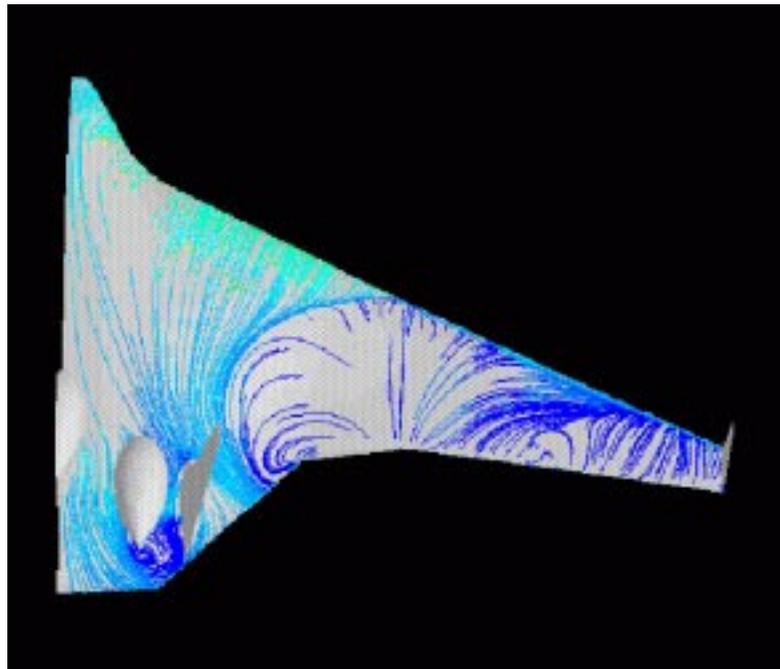
BAE SYSTEMS

BWB

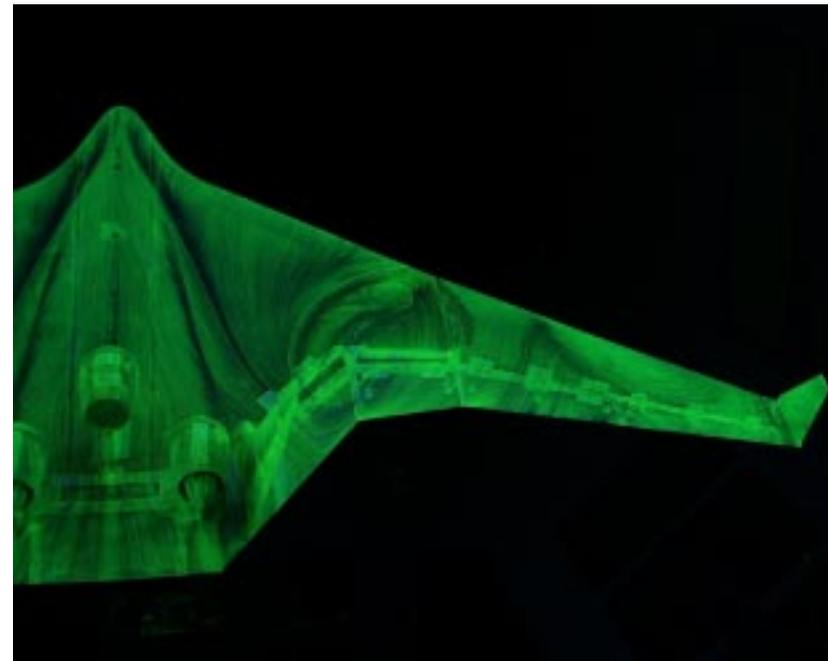


MIXED ATTACHED/SEPARATED FLOW CASE

BWB High Incidence



RANS KERNG



WIND TUNNEL

CONCLUSIONS

- Novel Planforms mean S+C Issues must be addressed early in the UCAV design cycle
- CFD and WT must work together here
- Requirement for rapid assessment
- Flat-plate and stereo-lith small-scale WT models, in conjunction with Euler and 'reduced-accuracy' RANS CFD can be applied here
- This approach requires engineering judgement and expertise to be fully effective