Overload Effect on Fatigue of AerMet 100 Steel

E. U. Lee* and K. Sadananda**

Overloads have been reported to perturb the steady state fatigue crack growth condition and retard the crack growth. The retardation effect is known to depend on material plasticity, slip planarity, microstructure, overload ratio, and background stress intensity range $\Delta K$ and load ratio $R$. Extensive studies have been conducted to identify the effect of a single overload (or spike load) in terms of the number of delayed cycles.

This study was initiated to understand the effect of a single overload on the threshold fatigue crack growth in AerMet 100 steel as a function of $R$-ratio and environment. A single 100% overload (or overload ratio 2) was applied after the fatigue crack growth rate $da/dN$ reached the threshold one ($5 \times 10^{-10}$ m/cycle) in the $K$-decreasing test of load shedding. The background load and the corresponding overload were stepped up until the crack began to grow after each loading of $5 \times 10^{-5}$ cycles. Subsequently, the fatigue test was continued under constant amplitude loading at various $R$-ratios, ranging from $-1.1$ to 0.9, and frequency of 10 Hz in air and vacuum of $5 \times 10^{-5}$ torr. The overload-induced increase in the threshold stress intensity range $\Delta K_{th}$ was taken as the measure of overload retardation effect. Furthermore, the overall results were analyzed in terms of two parameters, $\Delta K_{th}$ and maximum stress intensity at threshold $K_{max}$.

Under background load in air, with increasing $R$-ratio, the $\Delta K_{th}$ changed little for $R < 0$, but it decreased linearly for $R = 0$ to 0.5 and then leveled off for $R > 0.5$. The corresponding $K_{max}$ changed a little for $R < 0$ but increased steeply for $R > 0$ with increasing $R$-ratio. After a 100% overload in air, the $\Delta K_{th}$ increased linearly for $R < 0$ but decreased for $R > 0$ with increasing $R$-ratio. The feature of the corresponding $K_{max}$ change with $R$-ratio was similar to that for the background load, except the $K_{max}$ was increased by overload.

Under background and after a single overload in vacuum, $\Delta K_{th}$ decreased linearly with increasing $R$-ratio for $R > 0$. The overload-induced increase in $\Delta K_{th}$ was less in vacuum than in air.

The fundamental threshold curve of $\Delta K_{th}$ vs $K_{max}$ indicates that (a) the resistance to fatigue crack growth is greater in vacuum than in air under the background load and after a single overload, but (b) the overload retardation effect is greater in air than in vacuum.

The related fractography, crack path and crack tip geometry were examined for the fatigue tests.

* Naval Air Warfare Center Aircraft Division, Patuxent River, MD
** Naval Research Laboratory, Washington, DC
**REPORT DOCUMENTATION PAGE**

<table>
<thead>
<tr>
<th>1. REPORT DATE</th>
<th>2. REPORT TYPE</th>
<th>3. DATES COVERED</th>
</tr>
</thead>
<tbody>
<tr>
<td></td>
<td>Abstract</td>
<td></td>
</tr>
</tbody>
</table>

4. TITLE AND SUBTITLE

Overload Effect on Fatigue of AerMet 100 Steel

5a. CONTRACT NUMBER

5b. GRANT NUMBER

5c. PROGRAM ELEMENT NUMBER

6. AUTHOR(S)

E. Lee  K. Sadananda

5d. PROJECT NUMBER

5e. TASK NUMBER

5f. WORK UNIT NUMBER

7. PERFORMING ORGANIZATION NAME(S) AND ADDRESS(ES)

Naval Air Warfare Center Aircraft Division
22347 Cedar Point Road, Unit #6
Patuxent River, Maryland 20670-1161

8. PERFORMING ORGANIZATION REPORT NUMBER

9. SPONSORING/MONITORING AGENCY NAME(S) AND ADDRESS(ES)

Naval Air Systems Command
47123 Buse Road Unit IPT
Patuxent River, Maryland 20670-1547

10. SPONSOR/MONITOR’S ACRONYM(S)

11. SPONSOR/MONITOR’S REPORT NUMBER(S)

12. DISTRIBUTION/AVAILABILITY STATEMENT

Approved for public release; distribution is unlimited.

13. SUPPLEMENTARY NOTES

14. ABSTRACT

Overloads have been reported to perturb the steady state fatigue crack growth condition and retard the crack growth. The retardation effect is known to depend on material plasticity, slip planarity, microstructure, overload ratio, and background stress intensity range ∆K and load ratio R. Extensive studies have been conducted to identify the effect of a single overload (or spike load) in terms of the number of delayed cycles.

This study was initiated to understand the effect of a single overload on the threshold fatigue crack growth in AerMet 100 steel as a function of R-ratio and environment. A single 100% overload (or overload ratio 2) was applied after the fatigue crack growth rate dω/dN reached the threshold one (5 x 10^-6 m/cycle) in the K-decreasing test of load shedding. The background load and the corresponding overload were stepped up until the crack began to grow after each loading of 5 x 10^4 cycles. Subsequently, the fatigue test was continued under constant amplitude loading at various R-ratios, ranging from -11 to 0.9, and frequency of 10 Hz in air and vacuum of 5 x 10^-4 torr. The overload-induced increase in the threshold stress intensity range ∆Kth was taken as the measure of overload retardation effect. Furthermore, the overall results were analyzed in terms of two parameters, ∆Kth and maximum stress intensity at threshold Kth.

15. SUBJECT TERMS

AerMet 100 steel

16. SECURITY CLASSIFICATION OF:

<table>
<thead>
<tr>
<th>a. REPORT</th>
<th>b. ABSTRACT</th>
<th>c. THIS PAGE</th>
</tr>
</thead>
<tbody>
<tr>
<td>Unclassified</td>
<td>Unclassified</td>
<td>Unclassified</td>
</tr>
</tbody>
</table>

17. LIMITATION OF ABSTRACT

18. NUMBER OF PAGES

19a. NAME OF RESPONSIBLE PERSON

E. U. Lee

19b. TELEPHONE NUMBER (include area code)

(301) 342-8069

Standard Form 298 (Rev. 8-98)
Prescribed by ANSI Std. Z39-18