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AIAA-93-2296

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Using Nonlinear Finite Element
Static Analysis**

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**AIAA/SAE/ASME/ASEE
29th Joint Propulsion
Conference and Exhibit**

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June 28-30, 1993 / Monterey, CA

NASA Technical Memorandum 106176
AIAA-93-2296

Army Research Laboratory
ARL-TR-158

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Prepared for the
29th Joint Propulsion Conference and Exhibit
cosponsored by the AIAA, SAE, ASME, and ASEE
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Abstract

A procedure is presented for performing three-dimensional stress analysis of spiral bevel gears in mesh using the finite element method. The procedure involves generating a finite element model by solving equations that identify tooth surface coordinates. Coordinate transformations are used to orientate the gear and pinion for gear meshing. Contact boundary conditions are simulated with gap elements. A solution technique for correct orientation of the gap elements is given. Example models and results are presented.

Introduction

Spiral bevel gears are used to transmit power between intersecting shafts. One such application is in helicopter transmission systems. In this critical application, the gears operate at relatively high rotational speed and transmit substantial power (i. e. 1500 HP at 21,000 rpm).

Prior research has focused on spiral bevel gear geometry¹⁻⁴ to reduce vibration and kinematic error, improve manufacturability and improve inspection. Stress analysis is another important area of ongoing research. Accurate prediction of contact stresses and tooth root/fillet stresses are important to increase reliability and reduce weight.

Much effort has focused on predicting stresses in gears with the finite element method. Most of this work has involved parallel axis gears with two dimensional models. Only a few researchers have investigated finite element analysis of spiral bevel gears^{5,6}.

Finite element analysis has been done on a single spiral bevel tooth using an assumed contact stress distribution⁷. The research reported here will utilize the numerical solution for spiral bevel surface geometry to study gear meshing. Pinion tooth and gear tooth surfaces will be developed based on the gear manufacturing kinematics. The individual teeth are then rotated in space to create a multi-tooth model (4 gear and 3 pinion teeth). The tooth pair contact zones are modeled with gap elements. The model development procedure and finite element results are presented.

Equations for Tooth Surface Coordinates

The system of equations, briefly summarized here, required to define the coordinates of a face-milled spiral bevel gear surface were developed by Handschuh and Litvin⁸.

A conical cutting head, attached to a rotating cradle, swings through the work piece. Parameters U and θ locate a point on the cutting head in coordinate system S_c attached to the cutting head as shown in figure 1 and

described by the following equations.

$$r_c = \begin{cases} r \cot \psi - U \cos \psi \\ U \sin \psi \sin \theta \\ U \sin \psi \cos \theta \\ 1 \end{cases} \quad (1)$$

The roll angle of the cradle ϕ_c is used to locate the rotating cradle with respect to the fixed machine coordinate system S_m . Parameters U , θ and ϕ_c , along with various machine tool settings can be used to completely define the location of a point on the cutting head in space.

Since the kinematic motion of cutting a gear is equivalent to the cutting head meshing with a simulated crown gear, an equation of meshing can be written in terms of a point on the cutting head (i.e., in terms of U , θ and ϕ_c). The equation of meshing for straight-sided cutters with a constant ratio of roll between the cutter and work piece is given in^{5,6} as:

$$\begin{aligned} (U - r \cot \psi \cos \psi) \cos \gamma \sin \tau + S(m_w - \sin \gamma) \\ \cos \psi \sin \theta \mp \cos \gamma \sin \gamma \sin (\theta - \phi_c) \\ \pm E_m (\cos \gamma \sin \psi + \sin \gamma \cos \psi \cos \tau) \\ - L_m \sin \gamma \cos \psi \sin \tau = 0 \end{aligned} \quad (2)$$

The upper and lower signs are for left and right hand gears respectively. The following machine tool settings are defined^{5,6,9}.

ψ	cutting blade angle
θ	$(\theta \mp \alpha \pm \phi_c)$
τ	cradle angle
γ	root angle of work piece
E_m	machining offset
L_m	vector sum of change of machine center to back and the sliding base
m_w	ϕ_c/ϕ_w , the relationship between the cradle and work piece for a constant ratio of roll
U	generating cone surface coordinate
S	radial location of cutting head in coordinate system S_m
r	radius of generating cone surface

This is equivalent to:

$$f_1(U, \theta, \phi_c) = 0 \quad (3)$$

Because there are 3 unknowns U , θ , and ϕ_c ; three equations must be developed to solve for the surface

coordinates of a spiral bevel gear. The 3 parameters U , θ and ϕ_c are defined relative to the cutting head and cradle coordinate systems (S_c and S_j) respectively. These parameters can be transformed through a series of coordinate transformations to a coordinate system attached to the work piece. Or U , θ , ϕ_c can be mapped into X_w , Y_w , Z_w in coordinate system S_w attached to the work piece. These transformations, used in conjunction with two other geometric requirements, give the two additional equations.

The correct U , θ and ϕ_c that solves the equation of meshing, must also, upon transformation to the work piece coordinate system S_w , result in an axial coordinate Z_w that matches with a preselected axial position Z . (See figure 2)

$$Z_w - Z = 0 \quad (4)$$

This equation along with the correct coordinate transformations (see equation 11) result in a second equation of the form

$$f_2(U, \theta, \phi_c) = 0 \quad (5)$$

A similar requirements for the radial location of a point on the work piece results in (See figure 2).

$$r - (X_w^2 + Y_w^2)^{1/2} = 0 \quad (6)$$

The appropriate coordinate transformations (see equation 11) will convert equation (6) into a function of U , θ , and ϕ_c .

$$f_3(U, \theta, \phi_c) = 0 \quad (7)$$

Equations (3), (5) and (7) form the system of nonlinear equations necessary to define a point on the tooth surface.

Solution Technique

An initial guess U^0 , θ^0 , ϕ_c^0 is used to start iterative solution procedures. Newton's method is used to determine subsequent values of the updated vector $(U^k, \theta^k, \phi_c^k)^T$.

$$\begin{Bmatrix} U^k \\ \theta^k \\ \phi_c^k \end{Bmatrix} = \begin{Bmatrix} U^{k-1} \\ \theta^{k-1} \\ \phi_c^{k-1} \end{Bmatrix} + \begin{Bmatrix} \gamma_1^{k-1} \\ \gamma_2^{k-1} \\ \gamma_3^{k-1} \end{Bmatrix} \quad (8)$$

Where the vector Y is the solution of

$$\begin{pmatrix} \frac{\partial f_1(U^{k-1})}{\partial U} & \frac{\partial f_1(\theta^{k-1})}{\partial U} & \frac{\partial f_1(\phi_c^{k-1})}{\partial \phi_c} \\ \frac{\partial f_2(U^{k-1})}{\partial U} & \frac{\partial f_2(\theta^{k-1})}{\partial U} & \frac{\partial f_2(\phi_c^{k-1})}{\partial \phi_c} \\ \frac{\partial f_3(U^{k-1})}{\partial U} & \frac{\partial f_3(\theta^{k-1})}{\partial U} & \frac{\partial f_3(\phi_c^{k-1})}{\partial \phi_c} \end{pmatrix} \cdot \begin{pmatrix} Y_1^{k-1} \\ Y_2^{k-1} \\ Y_3^{k-1} \end{pmatrix} = \begin{pmatrix} f_1(U^{k-1}, \theta^{k-1}, \phi_c^{k-1}) \\ f_2(U^{k-1}, \theta^{k-1}, \phi_c^{k-1}) \\ f_3(U^{k-1}, \theta^{k-1}, \phi_c^{k-1}) \end{pmatrix} \quad (9)$$

The 3x3 matrix in the preceding equation is the Jacobian matrix and must be inverted each iteration to solve for the Y vector. The equation of meshing, function f_1 , is numerically differentiated directly to find the terms for the Jacobian matrix. Function f_2 and f_3 can not be directly differentiated with respect to U, θ and ϕ_c . After each iteration U^{k-1} , θ^{k-1} , ϕ_c^{k-1} (in the cutting head coordinate system Sc) are transformed into the work piece coordinate system, S_w , with the series of coordinate transformations as given in equation 11.

$$\begin{pmatrix} X_w \\ Y_w \\ Z_w \\ 1 \end{pmatrix} = [M_{wc}] \begin{pmatrix} r \cot \psi - U \cos \psi \\ U \sin \psi \sin \theta \\ U \sin \psi \cos \theta \end{pmatrix} \quad (10)$$

where:

$$[M_{wc}] = [M_{wc} f(\phi_c)] [M_{cp}] [M_{ps}] [M_{ps} f(\phi_c)] [M_{pc}] \quad (11)$$

Each matrix [M] above represents a transformation from one coordinate system to another. (See Appendix I for the specific matrices.)

Functions f_2 and f_3 are evaluated by starting with an initial U^k , θ^k and ϕ_c^k , performing the transformation in equation 11 and evaluating equations (4) and (6). The numerical differentiation of f_2 and f_3 is performed by transforming $U^k + inc$, $\theta^k + inc$, $\phi_c^k - inc$ (where inc is a small increment appropriate for

numerical differentiation into $X_w + inc$, $Y_w + inc$, $Z_w + inc$. Equations (4) and (6) are then used to evaluate the numerical differentiation. Function f_1 , f_2 , f_3 and the partial derivatives of f_1 , f_2 , f_3 required to the Jacobian matrix are updated each iteration. The iteration continues until the Y vector is less than a predetermined tolerance. This completes the solution techniques for a single point on the spiral bevel gear surface.

The four corners of the active profile are identified from the tooth geometry plane as shown in figure 2. Point 1 on the surface is chosen to be the lowest point of the active profile on the toe end. The initial guess to start the procedure has to be sufficiently close to the correct solution for convergence to occur. The solution for the first point proves to be an adequate initial guess for any subsequent points on the surface.

Subsequent interior surface points are found by incrementing $r = (X^2 + Y^2)^{1/2}$ and Z. By adjusting the increments used, a surface mesh of any density can be calculated. The process is repeated four times for each of four surfaces, gear convex, gear concave, pinion convex and pinion concave.

Since all four surfaces are generated independently, additional matrix transformations are required to obtain the appropriate orientation for meshing. The proper convex and concave surface orientation (for both the gear and pinion) is found by fixing the concave surface and rotating the convex surface until the correct tooth thickness is obtained. The correct angle of rotation is obtained by matching the tooth top land thickness with the desired value.

Gear and Pinion Orientations Required For Meshing

After generating the pinion and gear surface as described above, the pinion cone and gear cone apex will meet at the same point as shown in figure 3. This point is the origin of the fixed coordinate system attached to the work piece being generated. To place the gear and pinion in mesh with each other rotations described in the following example are required:

1. The 19th tooth is selected for meshing. This corresponds to rotating the generated tooth 190° CW about the Z_w axis. (For this example, $N_t = 36$ total teeth on the gear.) For the general case the gear tooth is rotated $360/N_t + 180$ CW about its axis of rotation (Z_w). This corresponds to selecting the i th tooth of the gear to be in mesh where $i = N_t/2 + 1$.
2. The pinion is rotated by 90° CCW about the Y axis.

Note: rotation (1) corresponds to selecting a different tooth on the gear to be in mesh; however, rotation (2) corresponds to physically rotating the entire pinion until it meshes with the gear.

3. Because the four surfaces are derived independently, their orientation is random with respect to meshing. The physical location of the gear and pinion after rotations (1) and (2) correspond to the gear and pinion in mesh with severe interference. To correct the interference the pinion is rotated CW about its axis of rotation (Z_w) until surface contact occurs. For this example the rotation was 3.56° . Figure 4 shows an example of a simulated gear pair meshing. The generated pinion tooth was copied and rotated 12 times and the generated gear tooth was copied and rotated 36 times.

Contact Simulation

The tooth pair mesh contact point can be located by a method described by Litvin⁶ or by a search technique. Pairs of finite element node points (one on each tooth surface) are evaluated until the pair with smallest separation distance is obtained. (A finer finite element mesh would yield greater resolution.) Once the contact point is established, a vector normal to the surface at the contact point is calculated.

The intersection of the normal vector on the pinion at the contact point with the gear surface identifies the second point required to define the gap element. Additional gap elements are obtained by taking additional vectors from other pinion surface finite element nodal points (parallel to the contact point normal vector), and calculating where they intersect the mating gear surface. Finite element nodal points on the gear surface are located to the intersection points of the normal vectors and the gear tooth surface.

The vector normal to the cutting surface is given in⁷ as:

$$\{n_m\} = \begin{Bmatrix} \sin \psi_c \\ \cos \psi_c \sin \tau \\ \cos \psi_c \cos \tau \end{Bmatrix} \quad (12)$$

This vector is written relative to S_w , a coordinate system fixed to the cutting head. To obtain the vector n_m relative to the coordinate system fixed to the work piece, the following transformation must be performed.

$$\{n_m\} = [L_w]\{n_m\} \quad (13)$$

Where $[L_w]$ is found by removing the 4th row and column from $[M_w]$.

Each gap element is connected between a node on the pinion surface, hereafter designated node 1, and the corresponding intersection node point on the gear surface, designated node 2. The intersection point on the gear surface is found as follows:

Consider node 1, a point on the pinion with coordinates P_x, P_y, P_z in S_w . Let Q_x, Q_y, Q_z be any point in space such that $(Q_x, Q_y, Q_z)^T = b(n_x, n_y, n_z)^T$ where b is a scale factor, and n_x, n_y and n_z are the components of the normal vector at the contact point. The intersection of the normal vector from node 1 with the mating gear surface defines node 2 and has to satisfy the following three equations:

$$\begin{aligned} Q_x &= G_x = X_g + b n_x \\ Q_y &= G_y = Y_g + b n_y \\ Q_z &= G_z = Z_g + b n_z \end{aligned} \quad (14)$$

Where G_x, G_y, G_z is a point on the gear surface. A point on the gear surface must also satisfy

$$\begin{Bmatrix} G_x \\ G_y \\ G_z \end{Bmatrix} = [M_{gc}] \begin{Bmatrix} r \cos \psi_c - U \cos \psi_c \\ U \sin \psi_c \sin \theta \\ U \sin \psi_c \cos \theta \\ i \end{Bmatrix} \quad (15)$$

Equation 15 leads to a system of 3 nonlinear algebraic equations. These three equations, along with the equation of meshing for the gear surface provide a system of 4 equations and 4 unknowns (u, θ, ϕ_c, b). These equations are solved with Newton's method described earlier. The intersection of the normal from node 1 on the pinion with the gear surface is now obtained.

This procedure is used to locate the intersection of normals from all points on the pinion surface (in the contact zone) with the gear surface. The gear tooth surface is remeshed utilizing the intersection points as shown in figure 5. Gap elements are connected between corresponding nodal points on the pinion and the intersection points on the gear surface.

Finite Element Model

An example analysis was performed using gears from the NASA Lewis Spiral Bevel Gear Test Facility. In this case, the left hand pinion mates with the right hand gear. Counter clockwise rotation of the pinion results in the concave surface on the pinion mating with the convex gear surface.

The design data for the pinion and gear are given in Table I. The design data are used with methods given in⁶ to determine the machine tool settings for the straight sided cutter data given in Table II.

The finite element gear pair model, shown in figure 6, contains 4 gear teeth and three pinion teeth. The model had 10,101 nodes (30,303 degrees of freedom) and 7596 eight noded 3D brick elements.

The three pinion teeth are fixed in space with zero displacements and rotations. This is done by setting x, y and z displacements equal to zero on the four corner nodes of each rim section. The gear is constrained to rotation about its axis of rotation. The gear is loaded with a torque of 9450 in lbs on the gear by applying a 125 lb force located 2.0976" from the gear axis of rotation.

At the orientation chosen between the pinion and gear two pairs of teeth were in contact. One pair had contact near the middle region of the tooth and another pair had contact near the toe (i.e., about to go out of mesh). Initially the model started with a total of twenty one gap elements. For the tooth that is approximately midway through mesh, fifteen gap elements were used. For the tooth about to leave mesh, four gap elements were used. The analysis starts with one gap element closed in each contact zone. Within the finite element code an iterative process is used to determine how many gap elements must close to reach static equilibrium. The solution iterated four times before reaching equilibrium with four gap element closed in the main contact region and one closed in the edge contact region. Stress contours for the two pinion teeth with contact are shown in Figures 7 and 8.

The average nodal minimum principal stresses in the main contact zone average -204,000 psi with a maximum of -299,900 psi and a minimum of -103,574 psi. The corresponding elemental stresses average -103,500 psi with a maximum and minimum of -123,900 psi and -79,500 psi, respectively. Nodal stresses are higher because of load

concentration from the gap elements. These stress ranges bracket the estimated hertzian stresses for the gear set under the same load conditions. Contact with only 4 gap elements, along with large stress gradient among adjacent nodes indicate the need for a finer finite element mesh for improved stress prediction.

Conclusions

A multi tooth finite element model was used to perform three-dimensional contact analysis of spiral bevel gears in mesh. Four gear teeth and three pinion teeth are generated by solving the equations, based on gear manufacturing kinematics, that identify tooth surface coordinates. The gear and pinion are orientated for meshing with coordinate transformations.

Surface stresses are evaluated with gap elements. The gear surface is remeshed with nodal points relocated to identify the correct gap element orientation. Initial FEA stress results compare favorably with calculated hertzian contact stresses. However, large stress gradients between adjacent nodes in the contact zones indicate a need for greater finite element mesh refinement.

References

1. Litvin, F.L., Theory of Gearing, NASA RP-1212, 1989.
2. Huston, R., and Coy, J., A Basic for Analysis of Surface Geometry of Spiral Bevel Gears. Advanced Power Transmission Technology, NASA CP-2210, AVRADCOM TR 82-C-16, G.K. Fischer, ed., 1981, pp. 317-334.
3. Tsai, Y.C., and Chiu, P.C., Surface Geometry of Straight and Spiral Bevel Gears, J. Mech. Trans. Automat. Des., vol. 109, no.4, Dec. 1987 pp. 443-449.
4. Clewley, L., and Gosselin, C., Kinematic Analysis of Bevel Gears. ASME Paper 84-DET-177, Oct. 1984.
5. Litvin, F.L., Tsung, W.J., and Lee, H.T., Generation of Spiral Bevel Gears with Conjugate Tooth Surfaces and Tooth Contact Analysis. NASA CR-4088, AVSCOM TR 87-C-22, 1987.
6. Litvin, F.L., and Lee, H.T., Generation and Tooth Contact Analysis of Spiral Bevel Gears with Predesigned Parabolic Functions of Transmission Errors. NASA CR-4259, AVSCOM TR 89-C-014, 1989.
7. Chao, H.C., Baxter, M., and Cheng, H.S., A Computer Solution for the Dynamic Load, Lubricant Film Thickness, and Surface

Temperatures in Spiral Bevel Gears. Advanced Power Transmission Technology, NASA CP-2210, AVRADCOM TR-82-C-16, G. K. Fischer, ed., 1981, pp. 345-364.

8. Drago, R.J., and Uppaluri, B.R. Large Rotorcraft Transmission Technology Development Program, Vol. 1, Technical Report. (D210-11972-1-VOL-1, Boeing Vertol Co., NASA Contract NAS3-22143) NASA CR-168116, 1983.
9. Handschuh, R.F., and Litvin, F.L., A Method of Determining Spiral-Bevel Gear Tooth Geometry for Finite Element Analysis, NASA TP-3096m AVSCOM TR 91-C-020, 1991.
10. Burden, R.L., and Faires, J.D., Numerical Analysis, 3rd edition, Prindle, Weber & Schmidt, 1985.

Appendix I

Coordinate transformations involving both rotation and translation require mixed matrix operations of multiplication and addition. Matrix representation of coordinate transformations will need only multiplication of matrices if position vectors are represented by homogeneous coordinates. The following 4 x 4 matrices are required to transform the homogeneous coordinates of a point on the cutting head to a point on the work piece.

Matrix $[M_{21}]$ transforms coordinate system S_c , attached to the cutting head, into system S_1 , rigidly connected to the cradle.

$$[M_{21}] = \begin{bmatrix} 1 & 0 & 0 & 0 \\ 0 & \cos q & =\sin q & \mp\sin q \\ 0 & \pm\sin q & \cos q & \mp\cos q \\ 0 & 0 & 0 & 1 \end{bmatrix} \quad (16)$$

Matrix $[M_{22}]$ transforms coordinate system S_1 into system S_m attached to the frame.

$$[M_{22}] = \begin{bmatrix} 1 & 0 & 0 & 0 \\ 0 & \cos \phi_c & \mp\sin \phi_c & 0 \\ 0 & \pm\sin \phi_c & \cos \phi_c & 0 \\ 0 & 0 & 0 & 1 \end{bmatrix} \quad (17)$$

Matrix $[M_{23}]$ transforms coordinate system S_m into system S_p , which orientates the apex of the gear being manufactured.

$$[M_{23}] = \begin{bmatrix} \cos \delta & 0 & -\sin \delta & -L_m \sin \delta \\ 0 & 1 & 0 & \pm E_m \\ \sin \delta & 0 & \cos \delta & L_m \cos \delta \\ 0 & 0 & 0 & 1 \end{bmatrix} \quad (18)$$

Matrix $[M_{24}]$ transforms coordinate system S_p into system S_a , which located the apex of the gear being manufacturer with respect to S_m .

$$[M_{24}] = \begin{bmatrix} \cos \mu & 0 & \sin \mu & 0 \\ \mp\sin \phi_w & \cos \phi_w & 0 & 0 \\ -\sin \mu & 0 & \cos \mu & 0 \\ 0 & 0 & 0 & 1 \end{bmatrix} \quad (19)$$

Matrix $[M_{25}]$ transforms coordinate S_a into system S_w attached to the work piece.

$$[M_{25}] = \begin{bmatrix} \cos \phi_w & \pm\sin \phi_w & 0 & 0 \\ \mp\sin \phi_w & \cos \phi_w & 0 & 0 \\ 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 1 \end{bmatrix} \quad (20)$$

TABLE I - PINION AND GEAR DESIGN DATA

	PINION	GEAR
Number of teeth pinion	12	35
Dedendum angle, deg	1.5666	3.8833
Addendum angle, deg	3.8833	1.5666
Pitch angle, deg	18.4333	71.5666
Shaft angle, deg	90.0	90.0
Mean spiral angle, deg	35.0	35.0
Face width, mm (in)	25.4(1.0)	25.4(1.0)
Mean cone distance, mm (in)	81.05(3.191)	81.05(3.191)
Inside radius of gear blank, mm (in)	5.3(0.6094)	3.0(.3449)
Top land thickness, mm (in)	2.07(0.082)	2.489(.098)
Clearance, mm (in)	0.762(0.030)	0.92954(0.0366)

TABLE II - GENERATION MACHINE SETTINGS

	PINION		GEAR	
	Concave	Convex	Concave	Convex
Radius of cutter, r , in	2.9656	3.0713	3.0325	2.9675
Blade angle, ψ , deg	161.9543	24.33741	58.0	22.0
Vector sum, L_w	0.038499	-0.051814	0.0	0.0
Machine offset, E_m , in	0.154576	-0.1742616	0.0	0.0
Cradle to cutter distance, s , in	2.947802	2.8010495	2.85995	2.85995
Cradle angle q , deg	63.94	53.926	59.234203	59.234203
Ratio of roll, M_w	0.30838513	0.32204285	0.950864	0.950864
Initial cutter length, u , in	9.59703	7.42534	8.12602	7.89156
Initial cutter orientation, θ , deg	126.83544	124.43689	233.9899	234.9545
Initial cradle orientation, ϕ_c , deg	-0.85813	-11.38663	-0.35063	-12.3384

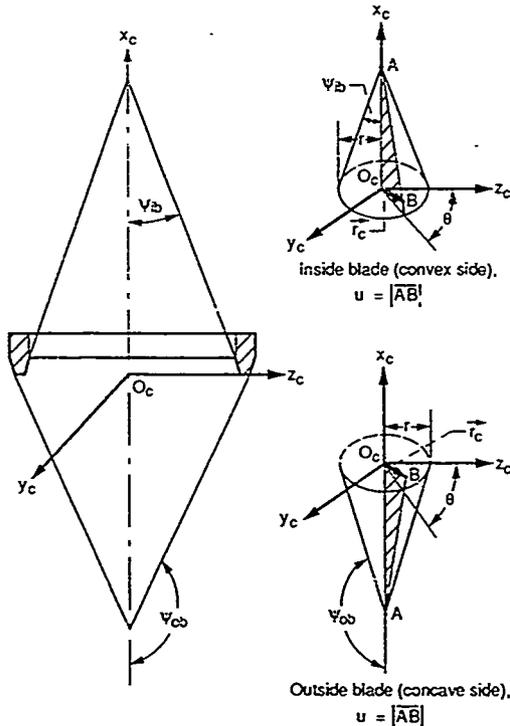


Figure 1. Cutting head cone surfaces and attached coordinate system.

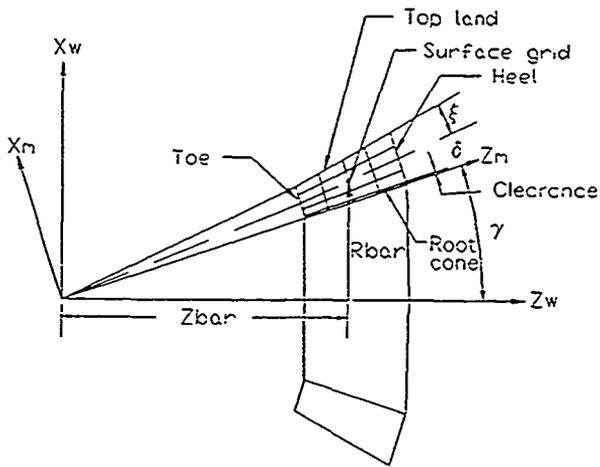


Figure 2. Projection of gear tooth with XZ plane.

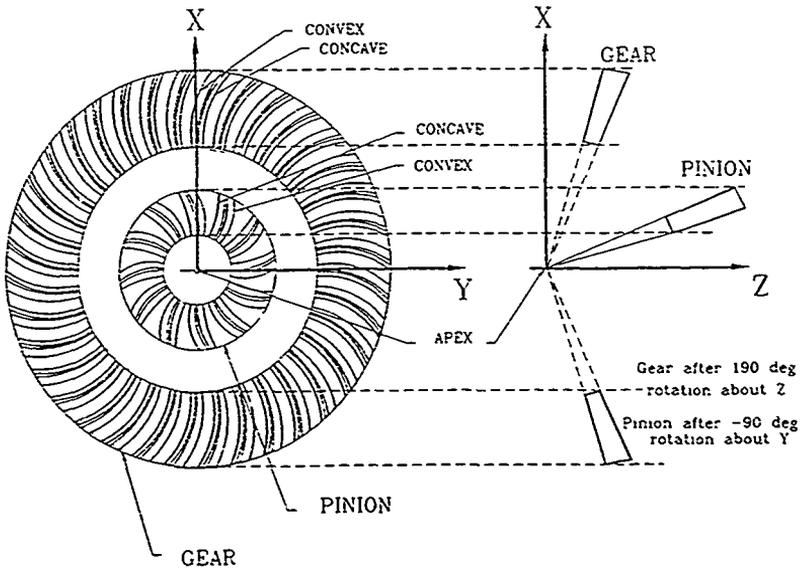


Figure 3. Orientation of gear and pinion based on solution of the system of equations, and after rotations required for mesh.

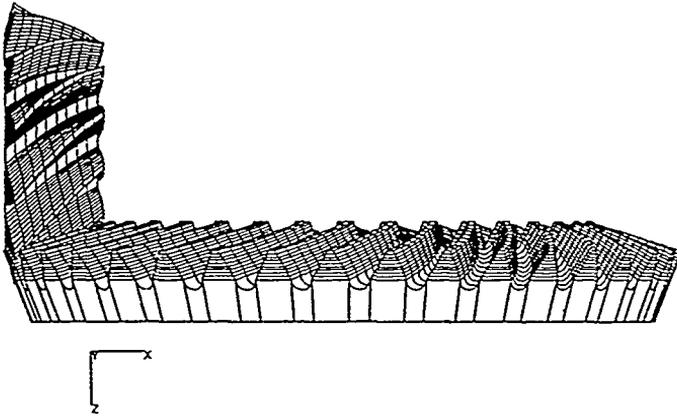


Figure 4. Complete 3D model of gear and pinion in mesh.

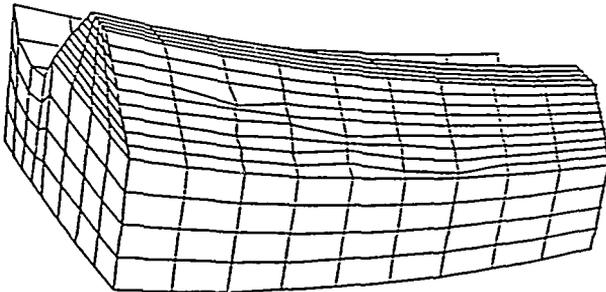


Figure 5. Distorted gear after connecting gap elements.

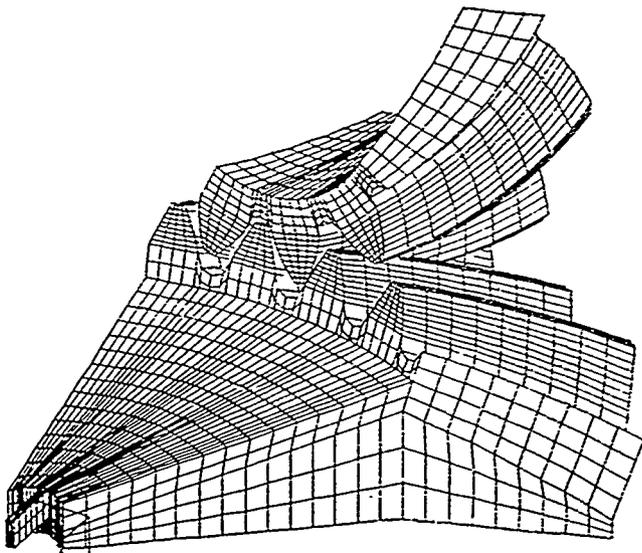


Figure 6. Seven tooth model used for finite element analysis of mesh.

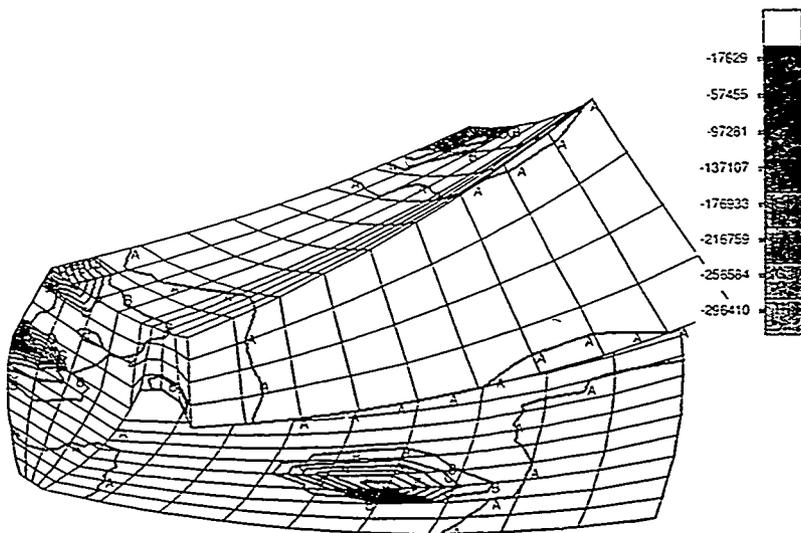


Figure 7. Stress contours in pinion tooth.

REPORT DOCUMENTATION PAGE

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1. AGENCY USE ONLY (Leave blank)		2. REPORT DATE May 1993	3. REPORT TYPE AND DATES COVERED Technical Memorandum	
4. TITLE AND SUBTITLE Contact Stress Analysis of Spiral Bevel Gears Using Nonlinear Finite Element Static Analysis			5. FUNDING NUMBERS WU-505-62-10 1L162211A47A	
6. AUTHOR(S) G.D. Bibel, A. Kumar, S. Reddy, and R. Handschuh				
7. PERFORMING ORGANIZATION NAME(S) AND ADDRESS(ES) NASA Lewis Research Center Cleveland, Ohio 44135-3191 and Vehicle Propulsion Directorate U.S. Army Research Laboratory Cleveland, Ohio 44135-3191			8. PERFORMING ORGANIZATION REPORT NUMBER E-7876	
9. SPONSORING/MONITORING AGENCY NAME(S) AND ADDRESS(ES) National Aeronautics and Space Administration Washington, D.C. 20546-0001 and U.S. Army Research Laboratory Adelphi, Maryland 20783-1145			10. SPONSORING/MONITORING AGENCY REPORT NUMBER NASA TM-106176 ARL-TR-158 AIAA-93-2296	
11. SUPPLEMENTARY NOTES Prepared for the 29th Joint Propulsion Conference and Exhibit cosponsored by the AIAA, SAE, ASME, and ASEE, Monterey, California, June 28-30, 1993. G. D. Bibel, University of North Dakota, Grand Forks, North Dakota 58201, A. Kumar and S. Reddy, University of Akron, Akron, Ohio 44325, and R. Handschuh, U.S. Army Vehicle Propulsion Directorate, NASA Lewis Research Center. Responsible person: R. Handschuh (216) 433-3969.				
12a. DISTRIBUTION STATEMENT Unclassified - Unlimited Subject Category 37			12b. DISTRIBUTION CODE	
13. ABSTRACT (Maximum 200 words) A procedure is presented for performing three-dimensional stress analysis of spiral bevel gears in mesh using the finite element method. The procedure involves generating a finite element model by solving equations that identify tooth surface coordinates. Coordinate transformations are used to orientate the gear and pinion for gear meshing. Contact boundary conditions are simulated with gap elements. A solution technique for correct orientation of the gap elements is given. Example models and results are presented.				
14. SUBJECT TERMS Gears, Gear teeth; Mechanical drives			15. NUMBER OF PAGES 12	
			16. PRICE CODE A03	
17. SECURITY CLASSIFICATION OF REPORT Unclassified	18. SECURITY CLASSIFICATION OF THIS PAGE Unclassified	19. SECURITY CLASSIFICATION OF ABSTRACT Unclassified	20. LIMITATION OF ABSTRACT	