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ADIABATIC SHOCK BANDS IN ONE DIMENSION

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MAY 1989

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# 1 EQUATIONS OF MOTION

A simple one dimensional model, which may be thought of as simulating a torsional Kolsky bar test on a thin walled tube, has been studied extensively by both numerical and analytic techniques. For a body in simple shear

$$x = X + u(Y, t), \quad y = Y, \quad z = Z, \quad (1)$$

the nondimensional equations of thermoviscoplasticity may be written as

$$\begin{aligned} \text{Momentum:} \quad v_t &= s_y / \rho, \\ \text{Energy:} \quad \theta_t &= k\theta_{yy} + s\dot{\gamma}_p, \\ \text{Elasticity:} \quad s_t &= \mu(v_y - \dot{\gamma}_p), \\ \text{Flow Law:} \quad s &= \text{sgn}(\dot{\gamma}_p)\kappa g(\theta)|\dot{\gamma}_p|^m, \\ \text{Work hardening:} \quad \kappa_t &= M(\kappa, \theta)s\dot{\gamma}_p. \end{aligned} \quad (2)$$

where  $v$  is the velocity,  $\theta$  is the temperature,  $s$  is the shear stress,  $\kappa$  is the work hardening parameter,  $\dot{\gamma}_p$  is the plastic strain rate, and  $\text{sgn}$  indicates the algebraic sign of its argument.

Nondimensionalization follows the same scheme as used by [1], except that a dynamic value  $s_0 = \kappa_0(b\dot{\gamma}_0)^m$  is used for the characteristic stress rather than the static value  $\kappa_0$ . Here  $\dot{\gamma}_0$  is the imposed nominal strain rate (the velocity difference across the slab divided by the thickness) and  $\kappa_0$  is the initial yield strength. The flow stress has been chosen to depend multiplicatively on functions of the temperature, the plastic strain rate, and a work hardening parameter, so that at its most general the material model includes work hardening and rate hardening, opposed by thermal softening. This combination, as is well known, can lead to strain softening, which in turn offers the opportunity for localization. Elasticity, inertia, and heat conduction all modulate the response in ways that are becoming increasingly clear. Of these three modulating effects, heat conduction appears to be the most important in that it is thermal conductivity that regularizes the governing equations and that ultimately limits the intensity of the localization and prevents formation of a singularity. No explicit failure criterion has been included in the model.

Only insulated boundaries with constant prescribed velocity are considered. That is,  $\theta_y(\pm 1, t) = 0$  and  $v(\pm 1, t) = \pm 1$ . The initial strain rate is assumed to be close to unity everywhere, and the initial stress is assumed to be constant. To trigger a shear band, only perturbations in the initial temperature will be considered,  $\theta(y, 0) = \theta_0(y)$ , where  $\theta_0$  is small. Mechanical perturbations can also initiate a shear band, but for brevity we will omit them here; see [2].

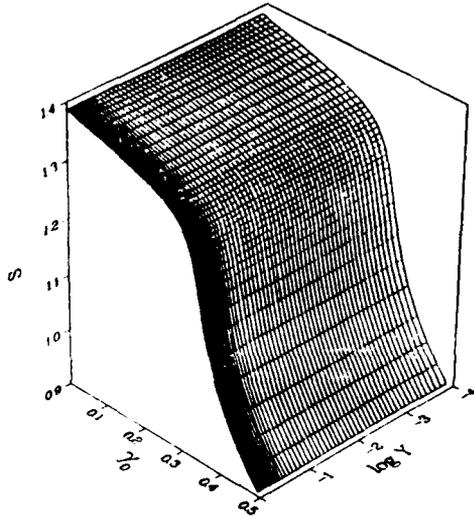


Figure 1: Stress response for a relatively low applied nominal strain rate,  $\dot{\gamma}_0 = 50/s$ .

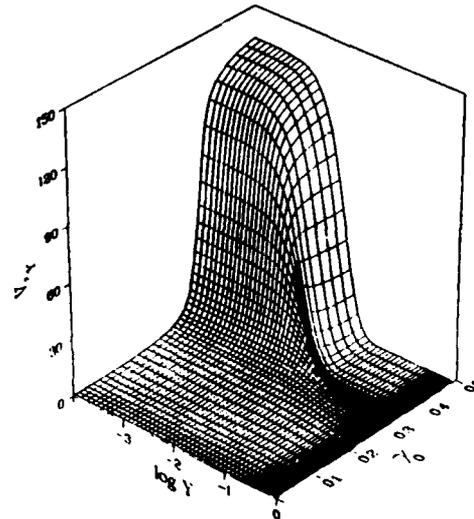


Figure 2: Velocity gradient for the same nominal strain rate.

## 2 FINITE ELEMENT SOLUTIONS

Finite element calculations, [3] and [4] are able to follow the development of the band through successive stages of formation, including early unstable growth, followed somewhat later by rapid localization and loss of stress carrying capacity, and culminating in a highly localized, late stage morphology. In the earlier calculations elasticity and work hardening were ignored, and linear thermal softening was assumed; the material parameters used were for a moderately high strength steel. Typical results from these calculations are shown in Figures 1 and 2. Note the slowly varying (post localization) late stage which is completely independent of the details of the initial conditions. Instead, it is determined solely by the functional form of the flow law, the various nondimensional parameters, the boundary conditions, and the applied nominal strain rate,  $\dot{\gamma}_0$ , with larger values of the latter corresponding to more intense localization.

These calculations, which treat the spatially discretized equations as a system of stiff ordinary differential equations for the temporal evolution of

the nodal values of  $v, \theta, s, \psi$ , appear to be highly robust and to resolve the spatial and temporal features of the solution with complete fidelity. In many instances it is possible to exceed the Courant condition (based on the elastic wave speed) by several orders of magnitude so that, even though the required grid refinement may be on the order of 1/10000 of the spatial range or less, the computational time is often only a few minutes.

### 3 APPROXIMATE SOLUTIONS

Mathematical analysis yields several types of specialized solutions that illuminate the various regions of the overall numerical solution. Homogeneous solutions that vary in time, but not in space, reveal whether or not strain softening occurs for the particular constitutive equation in use. While these solutions usually can not be written down explicitly, the system of partial differential equations degenerates to a low order system of ordinary differential equations which can be solved numerically. For example, with  $\theta_{yy} = 0$ , (2.2) and (2.5) may be combined to give an ODE with solution  $\kappa = \hat{\kappa}(\theta)$ , and then (2.4) may be written  $\dot{\gamma}_p = F(s, \theta)$ . Finally, with  $v_y = 1$ , (2.2) and (2.3) may be combined and reduced to the ODE

$$\frac{1}{\mu} \frac{ds}{d\theta} = \frac{1 - F(s, \theta)}{sF(s, \theta)}, \quad \text{with } s(0) = 1. \quad (3)$$

Inspection of this reduced system reveals that since the nondimensional elastic modulus  $\mu = \bar{\mu}/s_0$  is generally rather large (in the range of 50 to 100) elasticity contributes only a boundary layer that modulates the stress as it decays to a characteristic function of temperature only,  $F(s, \theta) = 1$ , or reverting to (2.4),  $s = \hat{\kappa}(\theta)g(\theta)$ . This represents essentially rigid-plastic response. In inhomogeneous problems elasticity plays the same minor role up to the time of intense localization, and so will be ignored for the time being. Moreover, since the stress remains essentially constant in  $y$  prior to localization except at very high applied strain rates (e.g.,  $\dot{\gamma}_p > 10^4/s$ , in the cases discussed here) it is sufficient to consider the quasi-static approximation in order to estimate the time to intense localization. Thus, for analytical purposes equations (2) may be replaced by the following simpler set

$$\begin{aligned} s_y &= 0, \\ \theta_t &= k\theta_{yy} + sv_y, \\ s &= \text{sgn}(v_y)\kappa g(\theta)|v_y|^m, \\ \kappa_t &= M(\kappa, \theta)sv_y. \end{aligned} \quad (4)$$

Small perturbations about the homogeneous solution yield a system of linear partial differential equations, which may be inspected for stability of the underlying homogeneous solution. The difficulty here is that the perturbation equations have time varying coefficients since the underlying motion is unsteady. An explicit solution does not seem possible in general, but for the special case of a rigid-perfectly plastic material with power law rate hardening and linear thermal softening, an exact solution may be written down for a temperature perturbation, see Wright and Walter (1987). For example, the strain rate is given by

$$v_y = 1 + (\epsilon a/m) \exp(at/m)(\psi - 1), \quad (5)$$

where  $\psi_t = k\psi_{yy}$ ,  $\psi_y = 0$  at the boundaries,  $\epsilon\psi_0 = \theta_0$  with  $\int_0^1 \psi_0 dy = 1$ , and the subscript 0 denotes initial conditions.

From such a solution a criterion for absolute stability of the homogeneous solution can be obtained, as well as the early growth rates for the field variables. Clearly, stability depends on the competition between the exponential factor and the diffusion factor. By inspection of (5) with  $\psi$  expressed by its Fourier components, it may be seen that for this solution to decay in time, a certain combination of material parameters, which we will call the "shear band susceptibility", must be less than the nondimensional thermal conductivity times  $\pi$  squared,  $\chi_{SB} \equiv a/m < k\pi^2$ . In dimensional terms this is

$$\chi_{SB} \equiv \frac{\bar{a}\kappa_0(b\dot{\gamma}_0)^m}{\rho cm} < \frac{k\pi^2}{\rho c\dot{\gamma}_0 H^2}, \quad (6)$$

where  $\bar{a}$  is the negative of the initial slope of  $g(\theta)$ ,  $\kappa_0$  is the strength parameter,  $\dot{\gamma}_0$  is the nominal strain rate,  $H$  is the half thickness in the slab,  $c$  is the heat capacity, and  $b$  is the characteristic time of the viscoplastic response.

If the homogeneous solution is not initially stable, then it is essentially the susceptibility that determines the initial growth rate. Of course, the long time behavior of the linearized system is meaningless when the underlying homogeneous flow is not stable. At present the perturbation equations that occur in the more general cases with work hardening can only be treated by special approximate techniques, but a different perturbation technique yields further insights when work hardening is absent.

By transforming variables, it is possible to find an approximate solution that for many cases appears to track the full finite element solution with great accuracy until immediately before intense localization occurs. Full details will be given by [2] but a sketch of the method as applied to a rigid-perfectly plastic material follows. Let  $H(\theta) = \int_0^\theta g^{1/m} d\theta$  and let a new time scale  $T(t)$  be defined by  $dT/dt = s^{(1+m)/m}$ . Now the energy equation may be

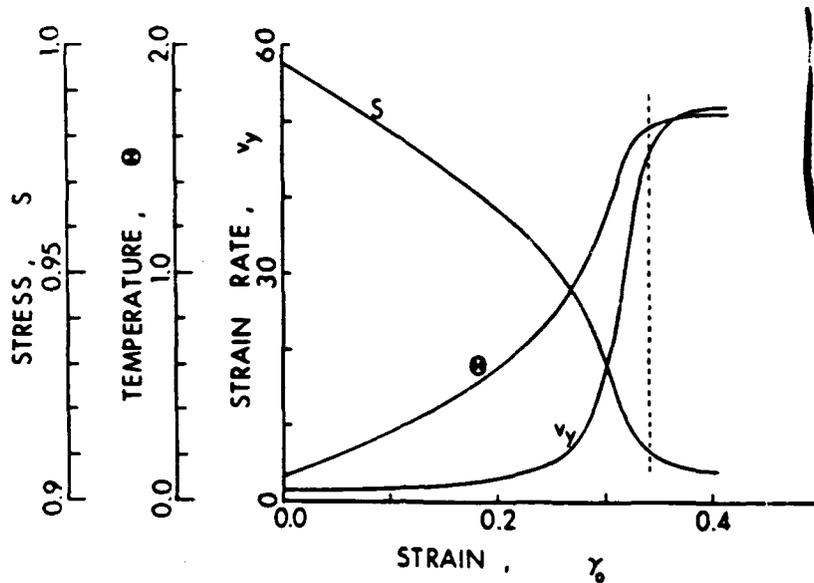


Figure 3: Shear band response by parametric solution for a typical case.

written as

$$H_T = 1 + k \frac{dt}{dT} \left( H_{vv} - \frac{H_{\theta\theta}}{H_\theta^2} H_v^2 \right). \quad (7)$$

Upon dropping the last term on the right hand side, an approximate solution may be obtained in the form

$$H = T + \psi, \quad s^{-1/m} = \int_0^1 g^{-1/m} dy, \quad dt/dT = s^{-(1+m)/m}, \quad (8)$$

in which  $\psi_t = k\psi_{vv}$  and  $\psi_0 = H(\theta_0)$ . Equation (8.1) gives an implicit relationship among  $\theta$ ,  $t$ ,  $T$  and  $y$ ; equation (8.2) gives  $s$  as a function of both time scales and follows from use of (4.3) in the relation  $\int_0^1 v_y dy = 1$ ; (8.3) is an ODE from which the dependence on  $t$  may be recovered.

This solution compares extremely well with the full finite element solution until the time of intense localization, [2], and is far easier to compute. Figure 3 shows the stress, temperature and strain rate in the center of the band as calculated from (8) for the same initial perturbation as used by [3]. Furthermore, from consideration of the simple defect structure  $\theta_0(y) = \epsilon \cos \pi y$ , it can easily be shown from (8) that the same shear band susceptibility mentioned previously is the most important physical quantity in determining the time or nominal strain from instability to intense local-

ization, [2]. The critical strain,  $\gamma_{cr}$ , in this case obeys the inequality

$$\gamma_{cr} \geq \frac{1}{\chi_{SB}} \ln \frac{2}{\epsilon \chi_{SB}}. \quad (9)$$

Heat conduction, which is primarily responsible for the inequality in (9), can have a substantial delaying effect on localization, especially at lower nominal strain rates. In fact, it is probably more accurate to say that localization occurs at a critical temperature, rather than at a critical strain. Heat conduction delays localization by reducing the temperature in the center of the band at a given nominal strain and spreading the thermal energy to the surrounding cooler material.

At higher rates inertial effects, which are not included in the parametric solution outlined here, become important and can also have a delaying effect on localization, [3]. A simple analysis and explanation for this effect is not yet available.

From the definition of  $H$ , expanded by use of Laplace's method, it also can be shown that to within a simple linear scaling, all rigid-perfectly plastic materials with fairly general thermal softening are equivalent to one that softens exponentially with temperature, [2], at least with respect to the critical localization strain for small perturbations. When there is only a temperature perturbation, it turns out that the critical strain may be estimated by solving (8.3), which now reads

$$\frac{dt}{d\hat{T}} = \frac{mC}{a} \left[ \int_0^1 \frac{dy}{1 - \hat{H}} \right]^{1+m}, \quad (10)$$

where  $\hat{H} = 1 - g^{1/m} = \hat{T} + \psi(y, t)$  and  $\psi$  satisfies a diffusion equation as in (8). The constant  $C = 1 + O(m)$  may be calculated from knowledge of the actual softening function. For softening functions with the same initial slope  $-a$ ,  $C$  is exactly 1 for exponential softening, less than 1 for more rapid softening, and greater than 1 for weaker softening, but in no case does it depend on the softening behavior at large temperatures. This correspondence makes it vividly clear that the critical strain to localization for small perturbations is dominated by the thermal softening behavior near ambient temperature, and *not* by the behavior at large temperatures. In addition, more extensive finite element computations, to be reported by [4], indicate that the terms associated with  $C$  in the solution of (10) provide a correction of the proper sign and magnitude to the value of  $\gamma_{cr}$  as given by (9).

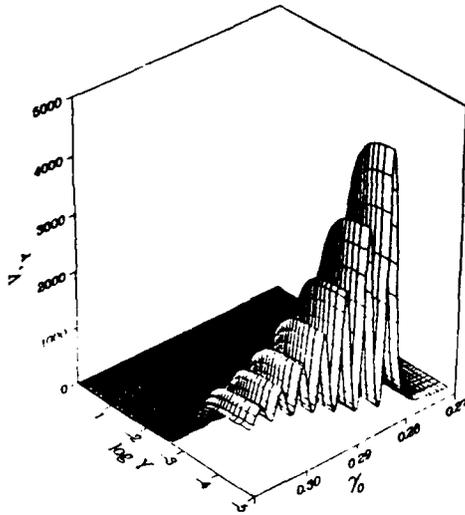


Figure 4: Velocity gradient for elastic perfectly-plastic case with  $\dot{\gamma}_0 = 750/s$ .

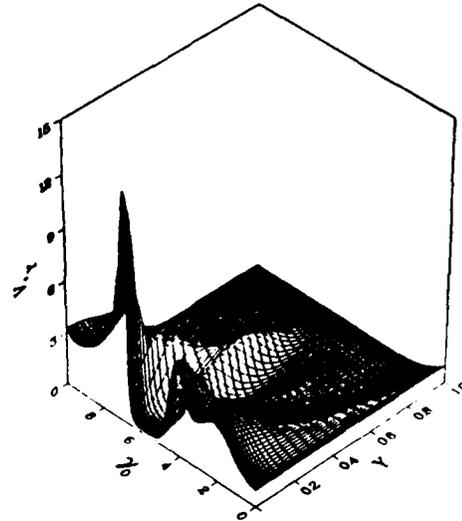


Figure 5: Velocity gradient for full material model (copper data) at  $\dot{\gamma}_0 = 330/s$ .

## 4 POST LOCALIZATION BEHAVIOR

In contrast to the results above, the late stage behavior within a shear band, after intense localization has occurred, must depend on high temperature response, but even then extreme temperatures do not necessarily occur. In the rigid-perfectly plastic case or if the work hardening saturates, the late stage morphology appears to be well represented by a rapidly shearing boundary layer in the center of the band and an exterior region where there is little or no further plastic work, and where temperature changes occur mainly by heat conduction from the actively shearing region. The solution in the core of the band is well represented by a steady solution, [2], which can be found from a simple quadrature, [5].

When work hardening and/or elasticity are restored, the full model is required once again. Numerical studies of (2) show that the localization and late stage response may become much more complex. Two notable effects will be reported in detail by [4] and are noted briefly here.

First of all with elasticity (but not work hardening) included, there can

be a considerable amount of elastic energy stored in the material outside the location at which the shear band will form. When intense localization occurs, this stored energy is released as the stress collapses in the band. In a narrow slab communication with the boundaries is rapid, and the net result is that the strain rate and velocity initially overshoot the values they would have attained in the rigid case followed by rapidly decaying oscillations, in which the stress and temperature also participate, before settling on the final late stage morphology as in Figure 4. The period of oscillation is longer (by a factor of about 10 in cases run to date) than that for an elastic wave communicating with the boundary, so these waves are clearly *not* simple damped elastic waves. Rather, they appear to be a consequence of the competing hardening and softening effects included in the model although they are, of course, elastically *driven* in the sense just outlined. The intensity of the overshoot and the severity of the oscillations seem to increase rapidly with increasing nominal strain rate. Both the initial overshoot and the oscillations are suppressed by an artificially high elastic modulus presumably because then there is less stored energy for a given stress level. Indeed, the (effectively) rigid cases illustrated in the first two figures were obtained by using a value for  $\mu$  equal to 100 times the physical value.

On the other hand, if work hardening continues throughout band formation, there may be repeated localizations on the same band whether or not elasticity is included. The first localization appears in the center of the original perturbation, but as the strain rate shoots up, so does the local rate of hardening. As  $\kappa$  increases the effect of thermal softening is overcome, and the strain rate at the slab center falls so that the peak rate occurs slightly to the sides. When these secondary peaks have experienced enough hardening in turn, the maximum strain rate returns to the center. Then the whole process can repeat as in Figure 5 where the material parameters used were for annealed copper. This gives the stress response a choppy, or irregularly oscillating appearance after the first localization, and can delay complete stress collapse to strains much larger than that at which the first localization occurs. It is also of interest to observe the very different time and length scales on which the localization occurs in Figure 4 and 5. The material model is the same in the two cases; only the parameter values are different.

## 5 SUMMARY

From a variety of analytical techniques it has been found that shear band formation is a multistep process, that instability and intense localization are not coincident in time, that a fully formed shear band often behaves like a boundary layer with a well defined, calculable, spatial distribution, and that a

theoretical "shear band susceptibility" can be identified and calculated from macroscopic laboratory measurements. The susceptibility, which appears naturally in simplified analyses, appears to be the key quantity in assessing stability, early growth rate, and time to intense localization. Elasticity has little effect until intense localization occurs, and then it may add rather elaborate structure to the late stage morphology. Work hardening, which is known to be stabilizing before localization, has a partially stabilizing effect after localization where it also adds structure to the late stage morphology.

## References

- [1] T. W. Wright and R. C. Batra. The interaction and growth of adiabatic shear bands. *Internat. J. Plast.*, 1:205-212, 1985.
- [2] T. W. Wright. Approximate analysis for the formation of adiabatic shear bands. Submitted to *J. Mech. Phys. Solids*, Jan. 1989.
- [3] T. W. Wright and J. W. Walter, Jr. On stress collapse in adiabatic shear bands. *J. Mech. Phys. Solids*, 35(6):701-720, 1987.
- [4] J. W. Walter, Jr. Numerical experiments in adiabatic shear bands. In preparation, 1989.
- [5] T. W. Wright. Steady shearing in a viscoplastic solid. *J. Mech. Phys. Solids*, 35(3):269-282, 1987.

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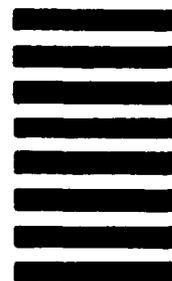


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