

**DTIC FILE COPY**

47

**AFWAL-TR-86-2105**  
**Supersedes AFWAL-TR-86-2105 dated February 1987**

**APPLICATION OF SYSTEM  
IDENTIFICATION  
TECHNIQUES TO  
COMBUSTOR POSTSTALL  
DYNAMICS**



**AD-A187 898**

**L.L. Steele  
J.R. Grant, Jr.  
D.P. Harrold  
J.J. Erhart**

**United Technologies Corporation  
Pratt & Whitney  
Government Products Division  
P.O. Box 109600  
West Palm Beach, FL 33410-9600**

**Subcontractor**

**R.P. Anex  
S.M. Rock**

**Systems Control Technology, Inc.  
1801 Page Mill Road  
Palo Alto, CA 94304**

**SEPTEMBER 1987**

**Final Report for Period December 1984 to September 1986  
Approved for public release; distribution unlimited.**

**Aero Propulsion Laboratory  
Air Force Wright Aeronautical Laboratories  
Air Force Systems Command  
Wright-Patterson Air Force Base, Ohio 45433-6563**

**DTIC  
ELECTE  
NOV 27 1987  
S A D**

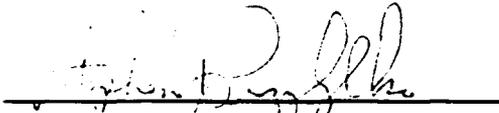
**7 11 17 0 59**

**NOTICE**

When Government drawings, specifications, or other data are used for any purpose other than in connection with a definitely related Government procurement operation, the United States Government thereby incurs no responsibility nor any obligation whatsoever; and the fact that the government may have formulated, furnished, or in any way supplied the said drawings, specifications, or other data, is not to be regarded by implication or otherwise as in any manner licensing the holder or any other person or corporation, or conveying any rights or permission to manufacture, use, or sell any patented invention that may in any way be related thereto.

This report has been reviewed by the Office of Public Affairs (ASD/PA) and is releasable to the National Technical Information Service (NTIS). At NTIS, it will be available to the general public, including foreign nations.

This technical report has been reviewed and is approved for publication.



STEPHEN J. PRZYBYLKO  
Project Engineer  
Components Branch  
Turbine Engine Division  
Aero Propulsion Laboratory



JACK D. MATTINGLY, Lt Col, USAF  
Chief, Components Branch  
Turbine Engine Division  
Aero Propulsion Laboratory

FOR THE COMMANDER:



MICHAEL E. STEFKOVICH, Maj, USAF  
Deputy Director  
Turbine Engine Division  
Aero Propulsion Laboratory

"If your address has changed, if you wish to be removed from our mailing list, or if the addressee is no longer employed by your organization please notify AFWAL/POTC, W-PAFB, OH 45433-6563 to help us maintain a current mailing list."

Copies of this report should not be returned unless required by security considerations, contractual obligations, or notice on a specific document.

UNCLASSIFIED

SECURITY CLASSIFICATION OF THIS PAGE

ADA187898

## REPORT DOCUMENTATION PAGE

1a. REPORT SECURITY CLASSIFICATION Unclassified		1b. RESTRICTIVE MARKINGS N/A	
2a. SECURITY CLASSIFICATION AUTHORITY N/A		3. DISTRIBUTION/AVAILABILITY OF REPORT Approved for public release. Distribution is unlimited.	
2b. DECLASSIFICATION/DOWNGRADING SCHEDULE N/A		4. PERFORMING ORGANIZATION REPORT NUMBER(S) P&W/GPD FR-18829	
4. PERFORMING ORGANIZATION REPORT NUMBER(S) P&W/GPD FR-18829		5. MONITORING ORGANIZATION REPORT NUMBER(S) AFWAL-TR-86-2105	
6a. NAME OF PERFORMING ORGANIZATION Pratt & Whitney Government Products Division	6b. OFFICE SYMBOL (If applicable)	7a. NAME OF MONITORING ORGANIZATION Air Force Wright Aeronautical Laboratories Aero Propulsion Laboratory (AFWAL/POTC)	
6c. ADDRESS (City, State and ZIP Code) P.O. Box 109600 West Palm Beach, FL 33410-9600		7b. ADDRESS (City, State and ZIP Code) Wright-Patterson Air Force Base Ohio 45433-6563	
8a. NAME OF FUNDING/SPONSORING ORGANIZATION Air Force Wright Aeronautical Laboratories	8b. OFFICE SYMBOL (If applicable) AFWAL/POTC	9. PROCUREMENT INSTRUMENT IDENTIFICATION NUMBER F33615-84-C-2432	
8c. ADDRESS (City, State and ZIP Code) Aero Propulsion Laboratory (AFWAL/POTC) Wright-Patterson AFB, Ohio 45433-6563		10. SOURCE OF FUNDING NOS.	
11. TITLE (Include Security Classification) Application of System Id. Techniques to Combustor Poststall Dynamics		PROGRAM ELEMENT NO. 61101F	PROJECT NO. 1L1R
12. PERSONAL AUTHOR(S) L.L. Steele, J.R. Grant, Jr., D.P. Harrold, and J.J. Erhart, Pratt & Whitney; R.P. Anex and S.M. Rock, SCT (Subcontractor)		TASK NO. P4	WORK UNIT NO. 06
13a. TYPE OF REPORT Final	13b. TIME COVERED FROM 12/84 TO 09/86	14. DATE OF REPORT (Yr., Mo., Day) September 1987	15. PAGE COUNT 139
16. SUPPLEMENTARY NOTATION			
17. COSATI CODES		18. SUBJECT TERMS (Continue on reverse if necessary and identify by block number)	
FIELD	GROUP	SUB. GR.	turbine engines, combustion, ignition, stability, blowout, stall, surge, system identification, poststall, gas turbine engines
21	02		
19. ABSTRACT (Continue on reverse if necessary and identify by block number)			
<p>The combustor (main burner) of gas turbine engines has been identified <sup>as</sup> to be a significant contributor to the degree of recoverability of engine surge events. Whether or not the combustor blows out and if so, whether it rapidly relights, can determine whether or not rotating (nonrecoverable) stall will occur. Component stall investigations have traditionally concentrated on the compression system and experience has shown that traditional methods of obtaining data are generally inadequate in identifying poststall characteristics. The use of system identification to obtain poststall compressor characteristics has been pursued with promising results. The objective of the present program is to apply system identification techniques to design experiments which will be suitable for extracting combustor poststall characteristics.</p>			
(Continued)			
20. DISTRIBUTION/AVAILABILITY OF ABSTRACT UNCLASSIFIED/UNLIMITED <input checked="" type="checkbox"/> SAME AS RPT. <input type="checkbox"/> DTIC USERS <input type="checkbox"/>		21. ABSTRACT SECURITY CLASSIFICATION Unclassified	
22a. NAME OF RESPONSIBLE INDIVIDUAL Stephen J. Przybylko		22b. TELEPHONE NUMBER (Include Area Code) (513) 255-4830	22c. OFFICE SYMBOL AFWAL/POTC

→ The program was entirely analytical in nature, and divided into five technical tasks:

1. Combustor Model Survey
2. Test Configuration Definition
3. System Identification
4. Test Evaluation
5. Test Methodology

↳ During the

During task 1, many models, modeling approaches, and combustion physics were examined. The objective was to provide an improved combustor dynamic model. The model derived under the Air Force Nonrecoverable Stall Investigation (F33615-79-C-2087) was selected as a starting point since it represented state of the art. Physics improvements were incorporated, including characteristic time approaches for ignition and stability. *During the test configuration definition,*

During task 2, the planned United Technologies Research Center Transient Combustion Facility was selected as the only cost-effective, flexible test vehicle of all candidates screened. ← The facility will be operational in 1987.

The system identification expertise of Systems Control Technology, Inc. was employed to perform the majority of the analysis in tasks 3 through 5 with Pratt & Whitney supporting. The results of these analyses indicate that with currently available instrumentation and model confidence, valuable testing allowing the identification of combustor poststall characteristics can be performed. The results of this contract will also provide the guidance to design such tests to obtain the most meaningful results.



## TABLE OF CONTENTS

<i>Section</i>	<i>Page</i>
I INTRODUCTION .....	1
II BACKGROUND DISCUSSION .....	2
1. Combustor Contributions to Engine Stall Behavior .....	2
2. Program Outline .....	6
3. Summary of Findings .....	7
III TEST CONFIGURATION DEFINITION .....	16
1. Test Inputs .....	16
2. Instrumentation .....	16
3. Data Acquisition .....	17
IV COMBUSTOR MODELING .....	21
1. Model Description .....	21
2. Limitations of the Model .....	28
V SYSTEM IDENTIFICATION BACKGROUND .....	29
1. Introduction .....	29
2. System Identification Procedure .....	29
3. Parameter Estimation Theory .....	29
4. Summary .....	35
VI APPLICATION OF SYSTEM IDENTIFICATION PROCEDURES .....	37
1. Introduction .....	37
2. Loading the Model into SCIDNT .....	37
3. Model Parameterization .....	39
4. Combustor Modes .....	39
5. Parameter Classification .....	41
6. Summary .....	42
VII TEST EVALUATION .....	43
1. Introduction .....	43
2. Test Evaluation Procedure .....	43
3. Parameter and Sensor Bias Determination .....	45
4. Analysis Tools .....	46
VIII TEST EVALUATION RESULTS .....	50
1. Introduction .....	50
2. Parameter Search Strategy .....	50
3. Example Test Evaluation .....	51
4. Summary of Evaluations .....	57

## TABLE OF CONTENTS (Continued)

<i>Section</i>		<i>Page</i>
IX	TEST METHODOLOGY .....	62
X	SUMMARY AND FUTURE EFFORTS .....	63
	1. Summary .....	63
	2. Future Efforts (Noncontractual) .....	63
	REFERENCES .....	64
	APPENDIX A — Sample Computer Model Prediction .....	67
	APPENDIX B — Samples of STATE, STATIC, and MEAS Subroutines .....	99
	APPENDIX C — Sample PARSEL Output .....	103
	APPENDIX D — SENSIT Output .....	111
	APPENDIX E — Sensor Noise and Bias Levels .....	117

## LIST OF ILLUSTRATIONS

<i>Figure</i>		<i>Page</i>
1	Combustors Are Typically Designed for Normal Flow Conditions .....	3
2	Combustor Reignition Following Surge Can Be Crucial to Engine Stall Recovery .....	5
3	Combustor Blowout/Relight Cycles Can Lead Normally Recoverable Engine Stalls to Stagnation .....	6
4	Existing Combustor Model Represents Outside and Inside Liner Regions as Separate Lumped Volumes .....	8
5	Main Program of Existing Poststall Combustor Model .....	9
6	Model Simulates Masses of Air and Fuel Separately for Transient Fuel/Air Ratio Calculation .....	10
7	Heat Release Calculation for Nonrecoverable Stall Investigation Combustor Dynamic Model .....	11
8	UTRC Transient Combustion Facility .....	14
9	Combustor Transients .....	15
10	High Response Mach Probe .....	18
11	Extensive Transient Instrumentation Is Required for Dynamic Model Validation Tests .....	19
12	Revised Routine STATIC Including Improved Heat Release Subroutine BURN02 .....	23
13	Simplified Flow Diagram for Heat Release Routine .....	23
14	Integrated System Identification Procedure .....	30
15	Poststall Combustor Simulation Outputs for 20 Hz, 250 psia $P_{T3}$ Sinusoid .....	40
16	Overplot of SCIDNT and Stand-Alone Combustor Simulation $T_{T4}$ Output .....	41
17	Combustor Test Evaluation Procedure .....	44
18	Burner Ignition State and Input $P_{T3}$ for Example Case .....	51
19	Output Sensitivity Matrix .....	52
20	Typical Plot of Output Error Vs Number of Estimation Parameters ...	53

**LIST OF ILLUSTRATIONS (Continued)**

<i>Figure</i>		<i>Page</i>
21	Output Cost Function Vs Number of Estimated Parameters .....	56
22	Combustor Efficiency During Mode Change .....	61

## LIST OF TABLES

<i>Table</i>		<i>Page</i>
1	Limiting Conditions for Transient Combustion Facility .....	13
2	Summary of Array Parameters .....	21
3	Poststall Combustor Model Candidate Parameter Set Divisions .....	22
4	Inputs and Outputs of the Required SCIDNT Subroutines .....	37
5	A Priori Parameter Bias Estimates .....	46
6	A Priori Sensor Model Bias Estimates .....	47
7	Sensed Combustor Outputs/Inputs .....	51
8	Summary of Parameter Selection Runs .....	55
9	Summary of Test Evaluations .....	57
10	Estimation Error Variation With Measurement Noise .....	58

## GLOSSARY OF TERMS

ALP	Air loading parameter
B	Mass transfer number
Ce <sub>1-12</sub>	Combustor model diddle factors
C <sub>p</sub>	Specific heat, constant pressure
d <sub>q</sub>	Spark kernal diameter
E	Ignitor spark energy
E <sub>G</sub>	Global activation energy for fuel type
FPLX	Fuel loading parameter
H	Heat of combustion
K	Thermal conductivity
L <sub>s</sub>	Latent heat of vaporization at droplet surface temperature
L <sub>o</sub>	Latent heat of vaporization at T <sub>o</sub>
M <sub>ox</sub>	Fractional mass concentration of O <sub>2</sub>
PC <sub>B</sub>	Percent of combustor annulus burning
Pr	Prandtl number = C <sub>p</sub> μ/k
Q <sub>av</sub>	Energy available to vaporize the fuel
Q <sub>r</sub>	Energy required to vaporize the fuel
Q <sub>burn</sub>	Burner heat release
r	Stoichiometric ratio
R	Universal gas constant
Re	Reynolds number
S <sub>L</sub>	Laminar flame speed
S <sub>T</sub>	Turbulent flame speed
SMD	Sauter mean diameter = D <sub>32</sub>
T	Temperature
T <sub>o</sub>	Initial liquid fuel temperature
T <sub>s</sub>	Initial fuel boiling temperature
ΔT	Temperature rise = function of (fuel, local f/a, T, P)
Δt	Transient time step
V	Velocity
Wa	Airflow
u'	RMS turbulent intensity = 0.75 $\bar{u}$
VOL	Burner volume
α	Thermal diffusivity = k/(ρC <sub>p</sub> )
η	Combustion efficiency
μ	Viscosity
φ	Equivalence ratio
ρ	Density
τ	Characteristic time

### *Subscripts*

a	Air
b	Burner
c	Reaction
e	Evaporation
f	Fuel
id	Ignition delay
ideal	Calculated from ideal temperature rise
L	Liquid
lag	Lagging in time during transients
pz	Primary zone

*Subscripts*

q	Quench
st	Stoichiometric
V	Vapor
vhr	Volumetric hot residence

## SUMMARY

This report describes a computer model of combustor transient performance, emphasizing recent improvements in the model's descriptions of heat release rate, autoignition, stability, and fuel injector performance. Also included is an independent analysis of the model identifying five parameters: combustor volume, liner pressure drop, combustion efficiency, turbine inlet flow parameter, and prediffuser exit area, which strongly affect the model's calculations of combustor exit temperature and flowrate. Finally, a unique, nearly complete, transient combustion facility is described along with the instrumentation which will be used in the first test, planned for 1987. Data from this test will be used to validate and further refine the model. Ultimately, the transient combustion model will become a tool used to promote stall recovery in future gas turbine engines.

## SECTION I

### INTRODUCTION

In current aircraft gas turbines, compressors must be designed to meet the contradictory demands of high pressure ratio (to optimize engine performance) and stall-free operation (to minimize aircraft vulnerability). In practice, compressor stall margin may be consumed by inlet distortion, pilot requested transients, and engine-to-engine manufacturing variations. Most stalls are self-recovering, with only a momentary loss of engine power. However, some stalls are nonrecoverable; they cause the engine to operate at low power and high turbine inlet temperature, and they can only be cleared by an in-flight shutdown and restart. While there has been considerable effort to simulate and control compressor stalls, little attention has been paid to the combustor's role in the engine post-stall scenario, even though combustor response to a compressor stall can determine whether or not the engine recovers.

This report summarizes Pratt & Whitney's efforts to model and understand transient operating conditions in a gas turbine combustor, with an ultimate goal of designing future combustors to promote stall recovery. Toward this end, Pratt & Whitney (P&W) has created a lumped-parameter computer model of a gas turbine combustor, and has subjected the model to independent analysis designed to identify model parameters which have the greatest impact on model predictions of combustor exit temperature and flowrate. This analysis has also resulted in a prediction of model accuracy when test sensor (e.g., pressure, temperature) accuracies are within specified limits. In 1987, these efforts will culminate in noncontractual validation of the model with test data from a transient combustion facility under construction at United Technologies Research Center (UTRC).

## SECTION II

### BACKGROUND DISCUSSION

#### 1. COMBUSTOR CONTRIBUTIONS TO ENGINE STALL BEHAVIOR

##### a. Normal Combustor Operation

The combustor is located at the heart of the gas turbine engine. From a thermodynamic cycle standpoint, the role of the combustor is simply to convert the chemical potential energy in the fuel to thermal energy by combustion. This thermal energy is then converted to kinetic energy in the turbines to drive the compression system and create thrust. In the combustion process, high efficiency at low pressure drop is desired. The design and performance of a combustor is much more complex than the blackbox thermodynamic description suggests.

A typical gas turbine combustor during normal (i.e., unstalled) operation is shown schematically in Figure 1. The compressor exit air Mach number is gradually reduced in the diffuser in a manner which limits total pressure loss. A dump section is provided for consistent flow behavior at the diffuser exit.

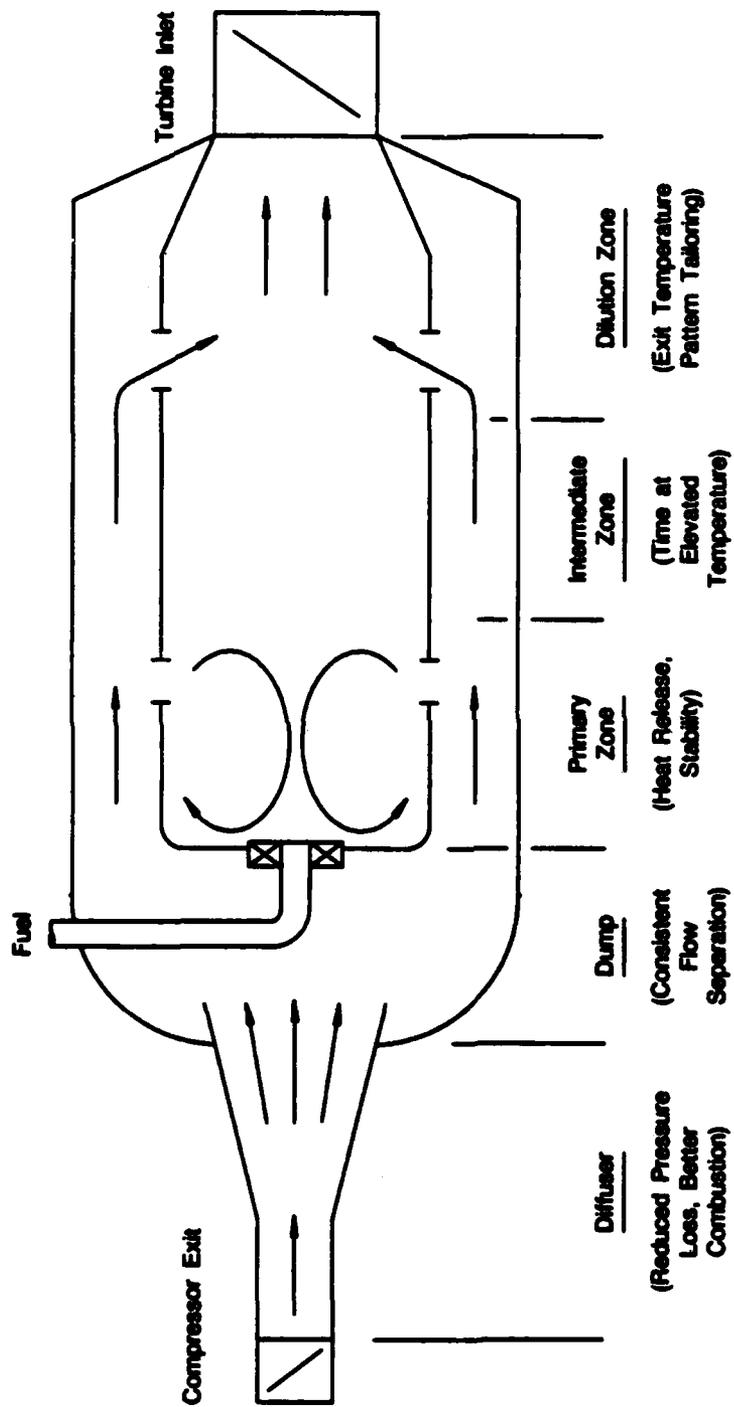
Combustor stability during normal operation is typically considered to be defined by conditions in the primary zone. Stoichiometry and swirler/fuel nozzle induced swirl strength are prime contributors in this regard.

Primary zone stoichiometry is controlled by the combustor fuel and front end/primary air flowrates and distributions. Fuel distribution is determined by the spray cone structure and the degree of atomization, and can vary significantly between nozzles of different designs, such as pressure atomizing or airblast. A pressure atomizing fuel nozzle design relies upon large fuel pressure drops to atomize. A pure airblast design uses the shearing effect of high velocity air on a fuel film for atomization.

Reverse flow in the primary zone occurs when a nondimensional ratio of tangential momentum to axial momentum (called the swirl number) exceeds a value of 0.6. The static pressure in the central core of the swirler then becomes sufficiently low and recirculation results. With increased swirl number, the recirculated mass flow is increased and, for very strong swirl levels, it can actually exceed the swirler flow.

Primary zone reverse flow creates stable combustion by recirculating hot reacting gases and fresh fuel and air which perpetuates combustion. Since only fuel in the vapor phase can burn, some of the heat release is used to help vaporize the injected fuel. The resulting overall fuel/air ratio in this zone must be within lean and rich limits and, in general, exceeds stoichiometric level.

The gases exiting the primary zone enter the intermediate zone. The intermediate zone provides time, at elevated temperature, for the recombination of the dissociated products, such as CO and H<sub>2</sub>. This recombination process helps to reduce engine smoke emissions and prevent low combustion efficiency at reduced power conditions.



FDA 276374

Figure 1. — Combustors Are Typically Designed for Normal Flow Conditions

The gases exiting the intermediate zone enter the dilution zone. In the dilution zone, the hot gases are mixed with cooler compressor exit air to reduce temperature levels and produce temperature patterns compatible with the turbine. Typically at high power, most of the combustion is complete before the hot gases enter the dilution zone.

Combustors are presently designed for these conditions of normal operation and are required to provide the following at acceptable levels:

- Durability
- Exit temperature patterns (hot spot and average profile)
- Starting capability at sea level and altitude
- Smoke and emissions (including efficiency).

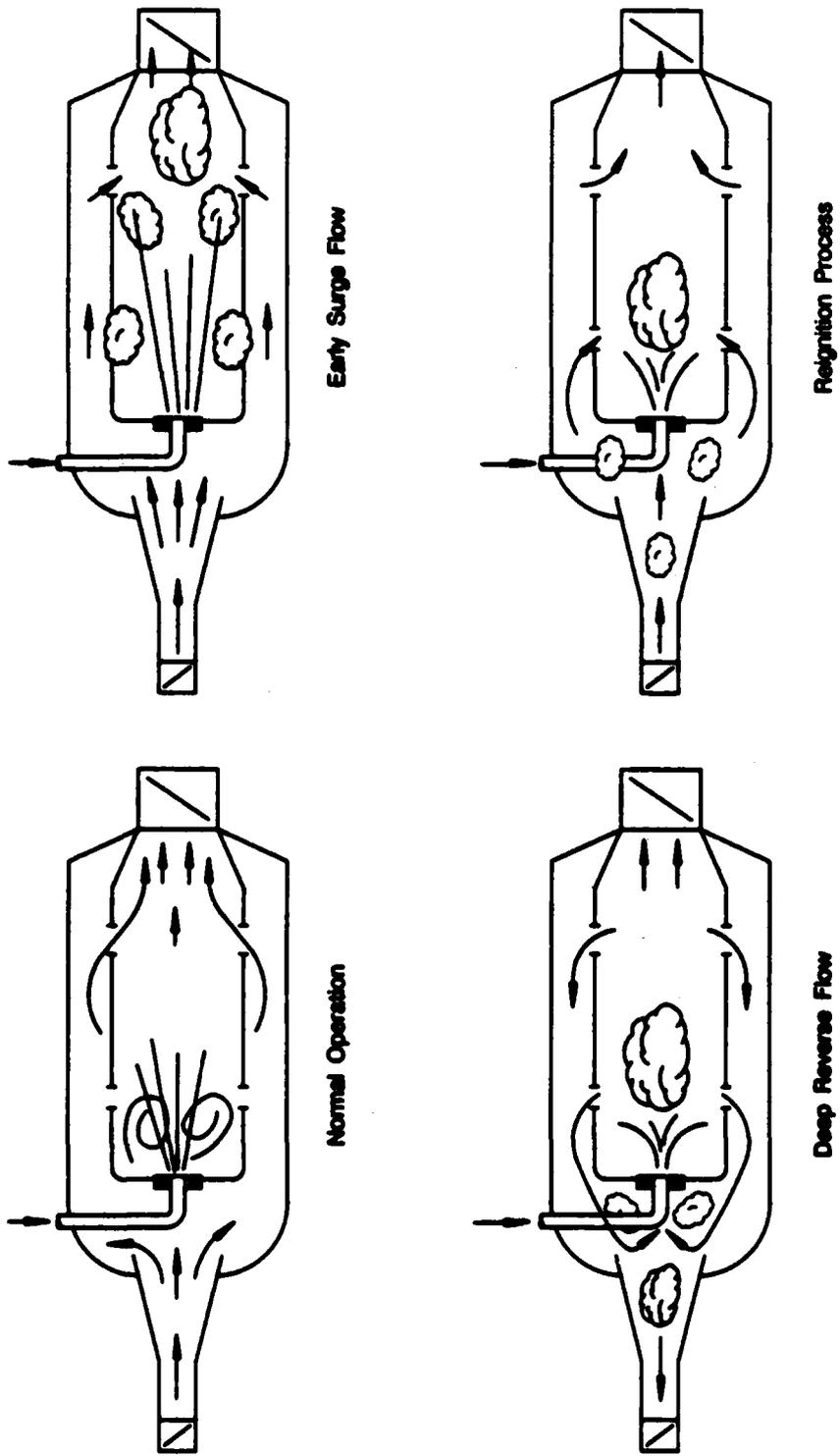
There are presently very few accepted guidelines for the design of "stall resistant combustor systems." Two of these are elevated fuel nozzle passage pressures to limit boiling potential, and increased combustor section volume. Conventional "normal operation" design guidelines must not be severely impacted by stall resistance design considerations, however.

#### **b. Combustor During Stall**

During normal combustor operation, the distribution of airflow is bookkept in the interests of providing desired exit temperature patterns, durability, starting, and smoke and emissions. These airflow distributions are determined by measured steady-state pressures. Figure 2 illustrates in a simple manner what is believed to happen to combustor air flow during an engine surge. For pre-surge conditions, the airflow distribution is normal. During the early stage of the surge cycle, the airflow into the combustor is retarded as the compressor discharge flow is reduced. As the engine approaches the trough of the surge cycle, it is speculated that deep reverse flow within the combustor section occurs. Hot reactants and products are drawn forward out of the liner. Primary zone recirculation patterns and, therefore, stability in the conventional sense may vanish from the primary zone. On the repressurization side of the surge cycle, forward flow is reestablished and some of the reactants and products are drawn back through the combustor liner. During these last two steps, it is considered crucial whether or not the combustor blows out and, if so, does it relight (if normal terminology can be applied).

In the case of a rotating stall, the amplitude of the combustor pressure variations is decreased while the frequency is increased. Local combustor response to a passing stalled cell might be visualized as being somewhat similar to surge response, but of reduced severity. The passing stall cell induces a local reverse flow of reduced strength relative to a surge induced reverse flow. The cell must affect local combustion to some extent as it passes, but complete blowout is not likely.

Combustor poststall dynamic effects have a significant impact on engine system stall recovery. Possibly the most important of these is the tendency for repeated combustor blowout/relight during surge cycling to reduce engine power level far into the sub-idle region. Once in the far sub-idle (or starting) region, it can be easy for the engine to fall into rotating stall (stagnation). As shown in Figure 3, repeated combustor relight/blowouts on the surge cycles drive average combustor pressure well below idle level. The end result of this transient was rotating stall.



FDA 276375

Figure 2. — Combustor Reignition Following Surge Can Be Crucial to Engine Stall Recovery

A study was conducted to determine the feasibility of using either existing rigs or engine configurations to conduct timely yet relatively inexpensive transient tests. No suitable facility presently exists at P&W to accomplish this goal. However, the United Technologies Research Center Transient Combustion Facility (planned to be operational in 1987) has this objective and was selected under this study.

The combustor model was integrated into the Systems Control Technologies (SCT) parameter estimation code and subjected to system identification analysis. This analysis determined the feasibility of identifying combustor model characteristics from rig test data with the goal of including the identified model in a full-scale engine poststall model.

SCT and Pratt & Whitney defined sensor sets which are currently practical for poststall combustor testing and sensors which are not currently available but could be developed in the near future. Accuracy levels for all sensors were then defined, and estimates of combustor model parameter uncertainties were made.

Using these uncertainty estimates, the optimum set of parameters to identify was defined for each combination of input, sensor set, and uncertainties. These estimation sets are optimum in the sense that they will minimize combustor model output error due to estimation error.

Estimation accuracy analysis was then performed in an iterative fashion for each estimation set. The final result of this analysis is a specification of the test instrumentation, inputs, and model uncertainty required to determine combustor model characteristics from test data.

A test input which drives the combustor into its different modes of operation, is desirable and has been selected as an input to the transient rig. It is essential that the state of the burner be determined through proper instrumentation of the facility and rig, and that the computer model be flexible enough to react to and simulate that change. The JETAS computer system installed in the transient facility's control room is capable of accessing data at as high a rate as 500 cuts/sec. This is more than adequate for the type of instrumentation and resolution desired.

### **3. SUMMARY OF FINDINGS**

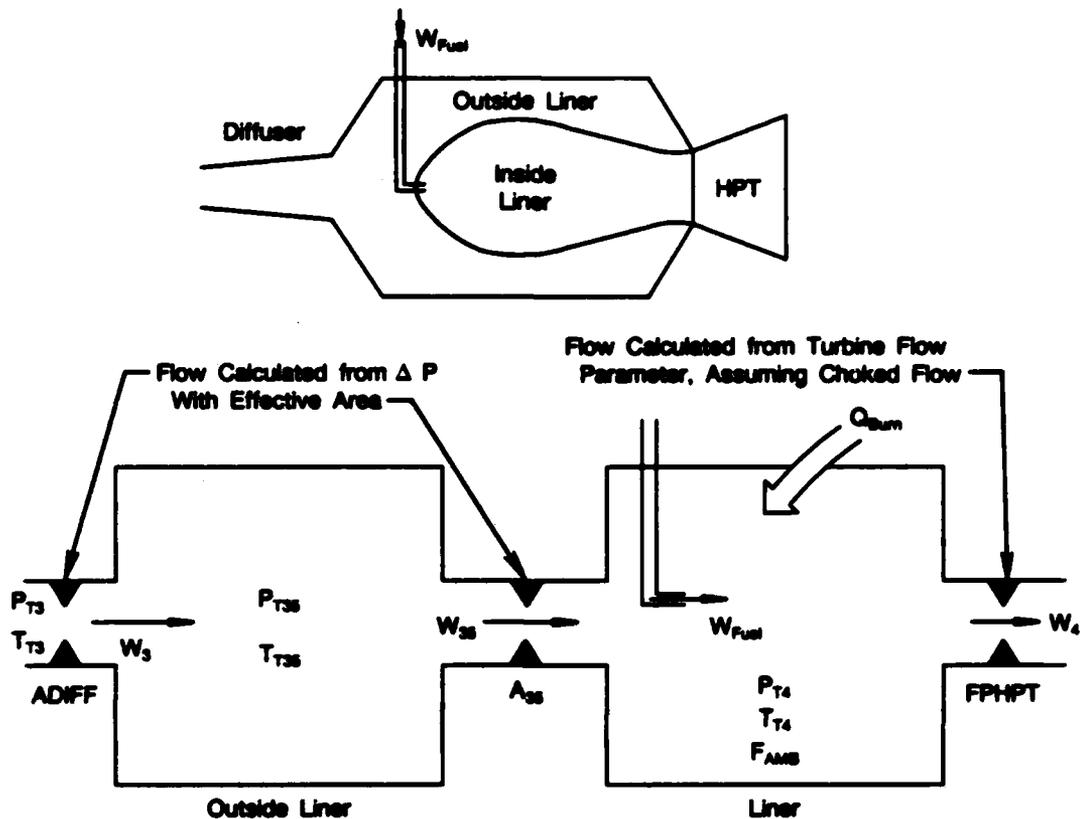
#### **a. Combustor Model Survey**

During the literature search, most of the technical papers on combustor modeling reviewed were concerned with steady-state operation, but the lumped volume approaches taken, lends them to transient analysis as well. No papers were identified which specifically presented poststall combustor models. However, many papers were assembled which concentrate on the important physical mechanisms which occur during poststall combustor operation, namely mixing, evaporation and chemical reaction. These mechanisms along with gas dynamic effects such as residence time and turbulence intensity help to control combustor ignition, stability, flame propagation and efficiency. A bibliography of the important publications which were used in the formulation of the poststall combustor model under this program are listed in the reference section.

##### **(1) Existing Digital Computer Model**

A review and critique of the existing state-of-the-art P&W poststall combustor dynamic computer model developed under the AF contract F33615-79-C-2087 was also performed. The model is shown schematically in Figure 4 and in flow chart form in Figure 5. It consists of two lumped-parameter volumes, one representing the region surrounding the combustor liner and the other representing the region inside the liner. Flow into the outside liner is calculated based on

the "dump loss" associated with sudden expansion at the diffuser exit and likewise, flow through the liner is based on the liner pressure drop. Combustor exit flow is calculated by assuming a choked high turbine since the turbine is not modeled.



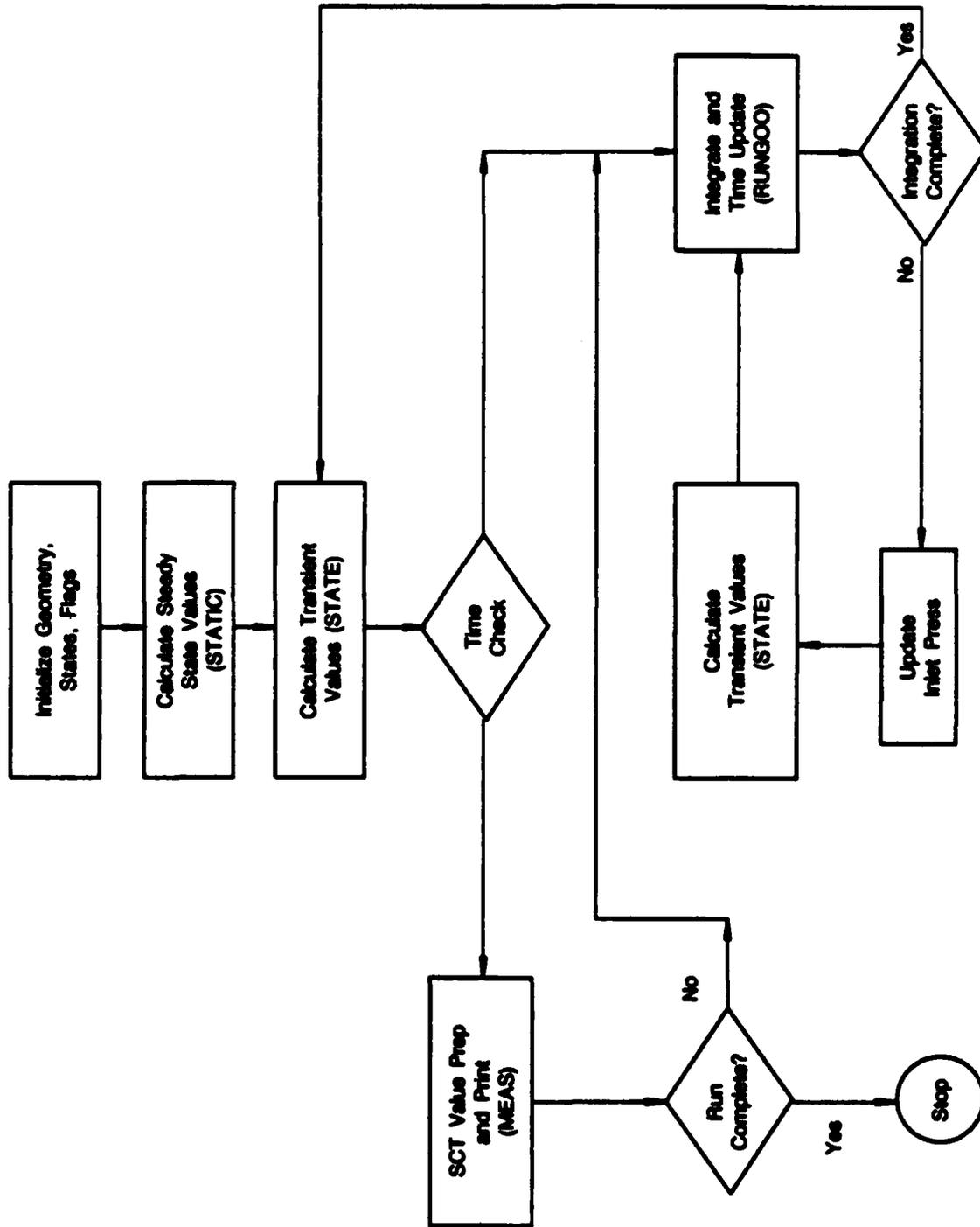
FDA 276378

Figure 4. — Existing Combustor Model Represents Outside and Inside Liner Regions as Separate Lumped Volumes

Each of the two volumes incorporates dynamic equations for conservation of mass and conservation of energy. The liner volume also includes conservation of fuel so that fuel/air ratio can be taken as the ratio of the mass of fuel within the volume to the mass of air as shown in Figure 6.

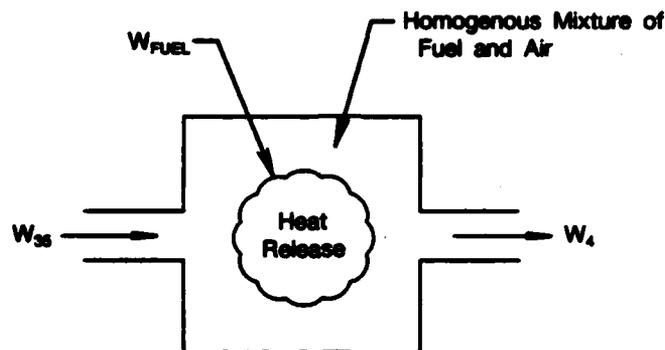
The heat release due to combustion is obtained by reading steady-state temperature rise curves with fuel/air ratio and applying a first-order lag intended to account for transient residence time effects as shown in Figure 7. This provides a somewhat realistic transient temperature rise with minimal computational complexity.

The model also includes other important physical processes using empirical correlations. Blowout limits are checked and, if exceeded, the temperature rise is zero. Following blowout, the combustor does not immediately relight once stability limits are no longer violated. Rather, some time elapses before the relight. This autoignition delay time has been correlated with entering air temperature and pressure. Once the relight is initiated, the flame must propagate around the annulus; it is assumed that the propagation is a linear function of time.



FDA 331055

Figure 5. — Main Program of Existing Poststall Combustor Model



- Continuity EQN for Air ~ Mass of Air
- Continuity EQN for Fuel ~ Mass of Fuel

$$f/a = \frac{\text{Mass of Fuel}}{\text{Mass of Air}}$$

FDA 276405

*Figure 6. — Model Simulates Masses of Air and Fuel Separately for Transient Fuel/Air Ratio Calculation*

## (2) Shortcomings

An analysis of the function of the various subroutines and how they interact as well as the overall methodology of the model was undertaken in the review. The weaknesses in the model and the methods of dealing with the weaknesses will be discussed in this section.

### (a) Heat Release

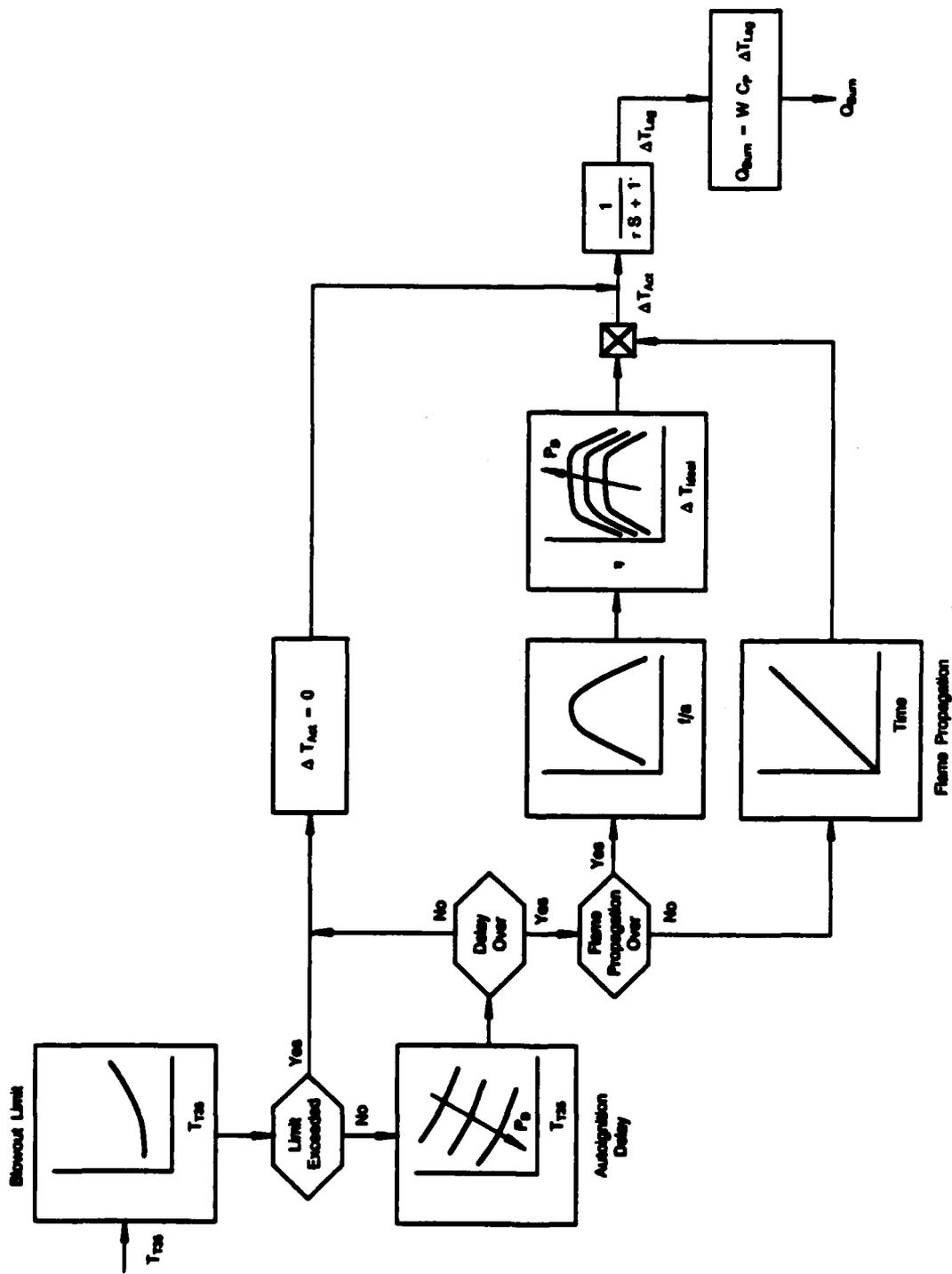
The efficiency values which are applied to the ideal temperature rise are based upon a simple functional form involving only two variables. The literature search indicated that steady-state combustor efficiencies are dependent on mixing, evaporation, and reaction effects which involve many variables. Also, the application of a first order time lag may be appropriate, but the time constant involves more physical effects than only residence time.

### (b) Combustion Zones

The "inside liner" volume was previously one lumped volume. This was divided into a primary and a secondary zone because the primary zone is most influential in terms of ignition and stability and is separable from the secondary zone in this regard.

### (c) Ignition Model

Autoignition delay time was previously the only criterion used for a combustor "on/off" decision. Using the characteristic time approach, a spark ignition criterion was added for better modeling and completeness.



FDA 276379

Figure 7. — Heat Release Calculation for Nonrecoverable Stall Investigation Combustor Dynamic Model

**(d) Stability**

The original blowout characteristic did not allow the inclusion of all the important variables revealed in the literature search, and was based upon steady state rig data. A stability calculation, better than the original, was required to determine if blowout limits are exceeded. A characteristic time approach was adopted.

**(e) Fuel injector**

An improved fuel injector model was incorporated to allow calculation of droplet size. Fuel nozzle design and fuel type are strong drivers on spray droplet size (typically identified by the Sauter mean diameter or SMD) which is involved in the physics of evaporation, flame propagation, and efficiency.

**(3) Approach**

A combustor modeling approach which has been given much attention in the literature is that of characteristic times. Characteristic times can be formulated for mixing, evaporation, chemical reaction, residence, and quenching. Relationships between these characteristic times can be developed to form criteria for spark induced ignition and stability. The previous model used a first order lag approach but the time constant is now represented as the sum of the residence, evaporation, and reaction characteristic times.

**b. Test Configuration Definition**

A survey of possible test configurations indicated that no appropriate rigs existed for simulating combustor transient operation. It was also apparent that engine testing was far too late in the development cycle and a very expensive operability test vehicle. Early in this program effort, a capital appropriation for the construction of the Transient Combustion Facility at the United Technologies Research Center in East Hartford was approved. The purpose of this facility is to provide a less expensive approach to operability testing which can be performed much earlier in the engine development cycle. It will also serve as an excellent poststall combustor dynamic test vehicle for the refinement of the poststall combustor computer model. The facility will be capable of simulating compressor surge and rotating stall, spooldown relight conditions and lean deceleration conditions. Flow capabilities for this facility are summarized in Table 1 and a schematic of the facility itself is shown in Figure 8. The different possible airflow transients are illustrated in Figure 9.

**c. System Identification**

Optimal parameter estimation sets have been identified for a range of possible instrumentation sets and levels of model confidence. These results indicate that, with current instrumentation and model confidence, estimation of four model parameters will produce the best results. These parameters include a constant bias on combustor efficiency, the resistive areas, and volumes. Parameter estimation techniques require the coefficients in the combustor stability equation to be identified separately, after other model parameters have been identified. These terms may be estimated accurately from test data after initial identification efforts. As with all estimation results, identified model accuracy is dependent on the model structure.

TABLE 1.

LIMITING CONDITIONS FOR TRANSIENT COMBUSTION FACILITY

Airflow

- Steady State Operation — Continuous
  - Flowrate 10 pps
  - Pressure 230 psia
  - Temperature 1000°F
- Transient or Short Duration Steady State (30 sec) Operation
  - Flowrate 0-20 pps
  - Pressure 280 psia max
  - Temperature 1000°F max
  - Cycling 10 Hz
  - Frequency
  - Rise/Decay 1-5 sec
  - Time

Fuel Flow

- Steady State or Transient Operation
  - Flowrate 0-0.5 pps
  - Pressure 1000 psia max
  - Temperature Ambient — 320°F
  - Cycling 30 Hz
  - Frequency
  - Rise/Decay Instantaneous
  - Time

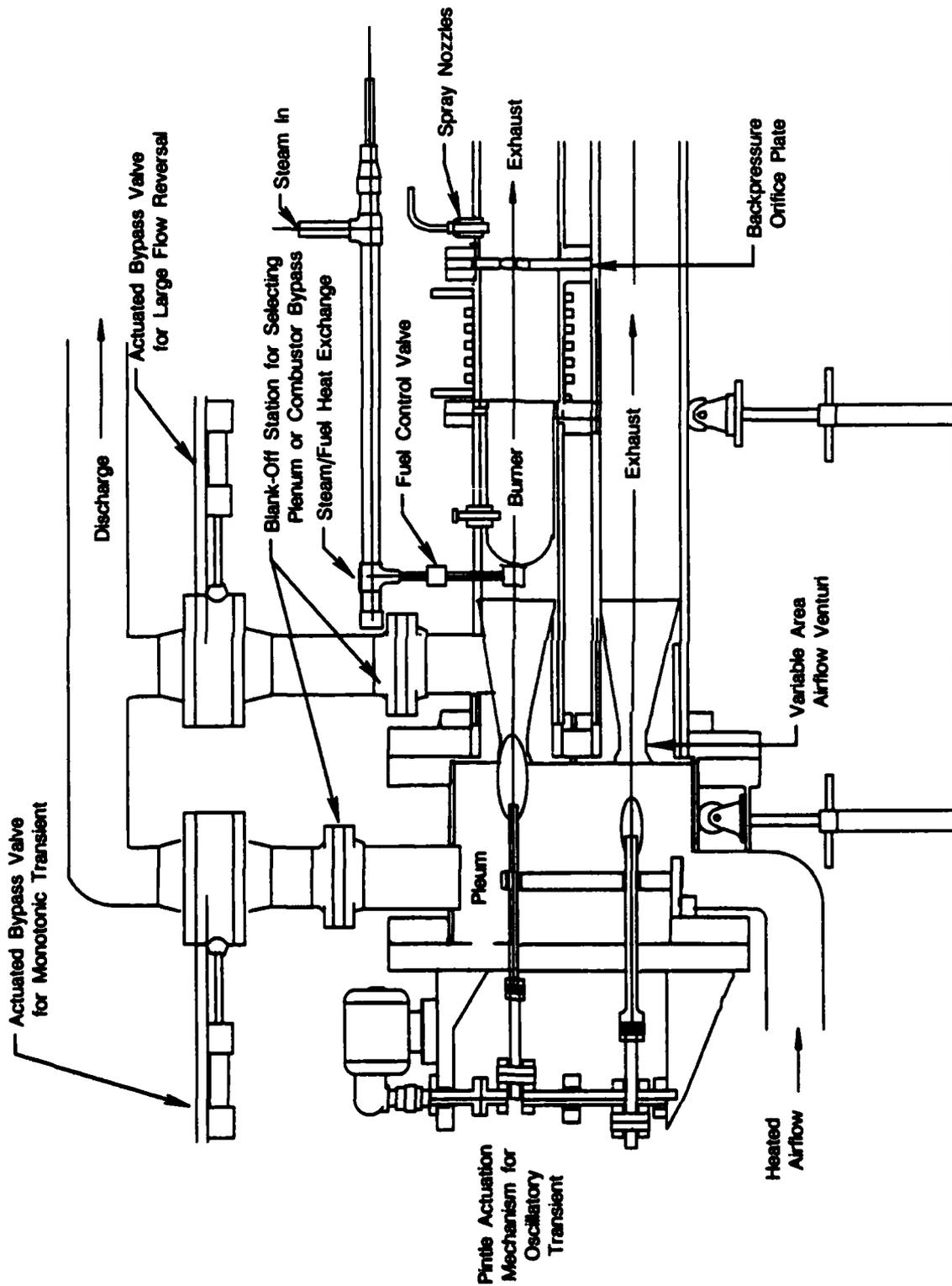
Data Acquisition — Fast Response (>1000 Hz)

- 24 Channels
- Expect UTRC Capacity to Increase

mscc

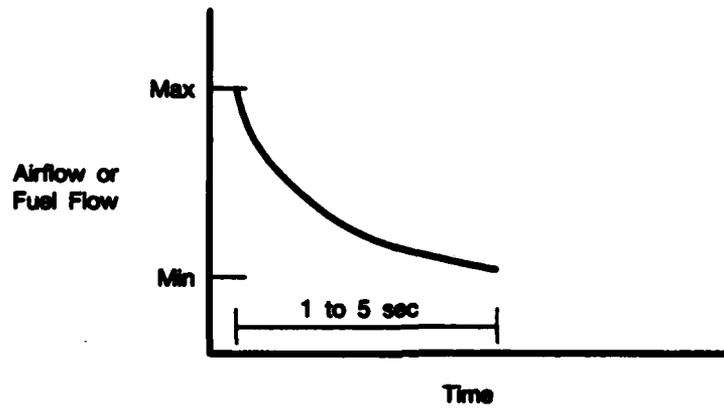
d. Test Evaluation

The result of the test evaluations can be summarized as follows: Non-estimated model parameters must be known on the average to within 5 percent of the true values. Systematic measurement errors must be known extremely well (sensor bias within  $\pm 0.5$  percent and sensor time constraints to within  $\pm 0.25$  percent). With this knowledge and currently available instrumentation, a poststall combustor model could be identified such that predicted combustor exit temperature and flow would be accurate to within 5 percent of the true values (combined total error).

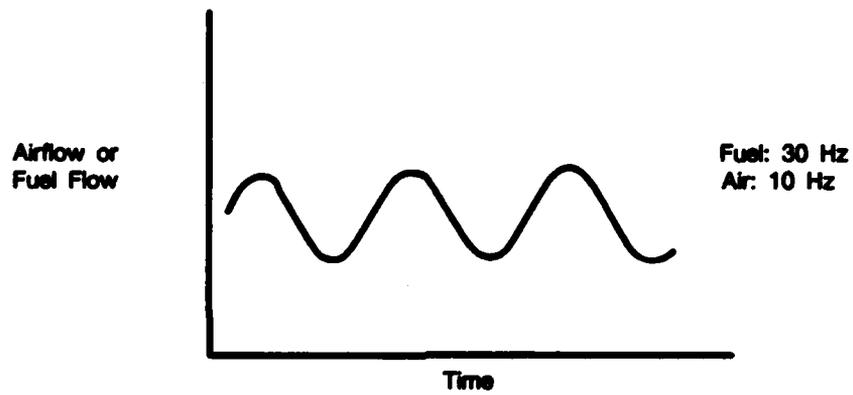


FDA 307906

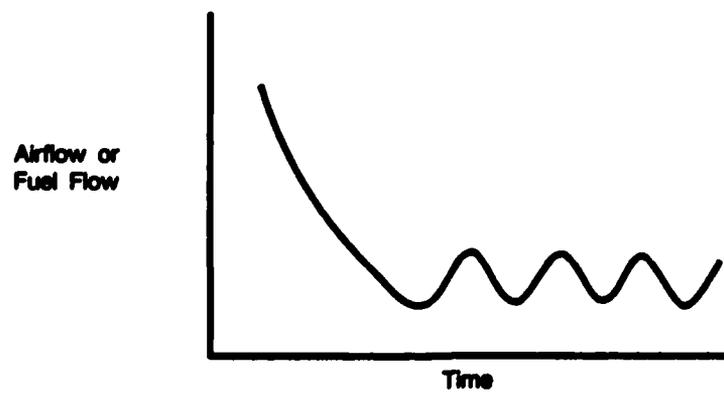
Figure 8. — UTRC Transient Combustion Facility



a) Monotonic Transient



b) Oscillating Transient



c) Combined Transient

FDA 307909

Figure 9. — Combustor Transients

## SECTION III

### TEST CONFIGURATION DEFINITION

#### 1. TEST INPUTS

The combustor transient model was exercised at SCT using two input test cases. The first test case simulates the compressor exit pressure during multiple planar surge. This consists of a 20 Hz signal imposed on an inlet pressure varying from 50 to 250 psia. The second test case is a slow increase in fuel flow, from 0.0 to 2.0 pps with combustor inlet pressure constant at 14.7 psia.

#### 2. INSTRUMENTATION

It is important to state at this time that the rig and facility instrumentation must be capable of indicating at what time during the transient the burner is in the lit and unlit modes.

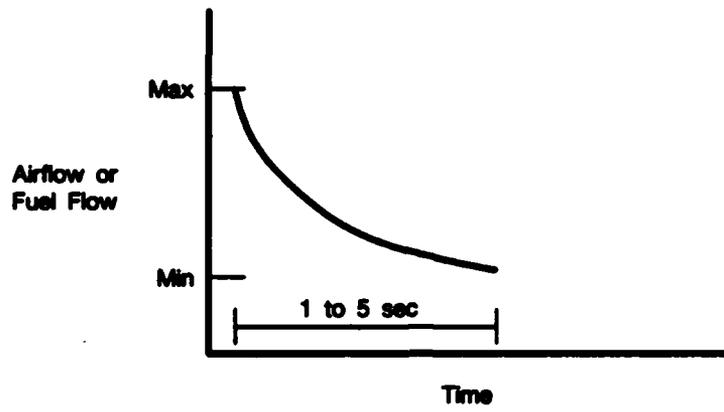
The selection of combustor and rig instrumentation will be determined by such factors as their frequency response, accuracy, durability, and nonobtrusiveness. Adequate but not excessive redundancy is important to allow for instrumentation failure without incurring excessive cost or influencing the flowpath because of large instrumentation bundles. Selection of instrumentation based upon frequency response, however, is likely to be the most important factor for a dynamic combustor test. A typical rule of thumb for the measurement of transient phenomena is that data should be sampled at twice (2X) the event frequency for frequency information and 10X for amplitude and phase information.

Facility transient capabilities must be considered as well as the dynamics of engine stall events when selecting instrumentation based upon frequency response. For the measurement of pressure at high frequencies there are at least three possibilities: Statham, Kulite, or Kistler sensors. Statham pressure sensors can sample at rates up to approximately 1000 Hz, but Kulites or Kistlers have much higher maximum sampling rates at nearly 20,000 and 50,000 Hz, respectively. These sampling rates assume flush mounting with no leads which detract from the response levels. If leads of any substantial length are required, Kulites and Kistlers might be left as the only adequate pressure sensors, since Stathams display limited response margin. Accuracy levels for these devices are typically about 1 percent of full scale.

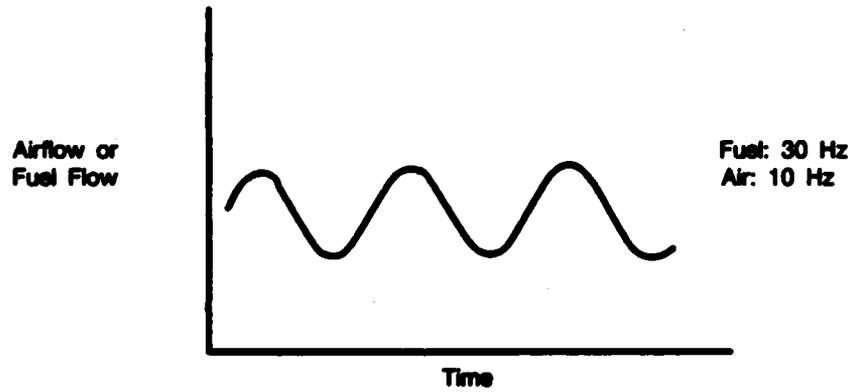
Given the importance of burner state, an optical viewing probe will be used as a data gathering device and the signal will be recorded on video tape. This video recording will have a time signal superimposed on the signal from the host computer so that during data analysis, plots may be compared to the video transmission.

With a test input of 20 Hz, it is necessary to have a minimum of 40 Hz frequency response for frequency definition, and 200 Hz for amplitude information. For the transient combustor test rig, the minimum line length for burner pressure instrumentation will be about 6 inches. With these line lengths required to route the instrumentation to an accessible location the best Statham sensor frequency response would be 23 Hz. The more capable Kulite sensor for the same conditions will provide 65 Hz frequency response. Given SCT's conclusion that 50 Hz is adequate, the pressure data should allow accurate examination.

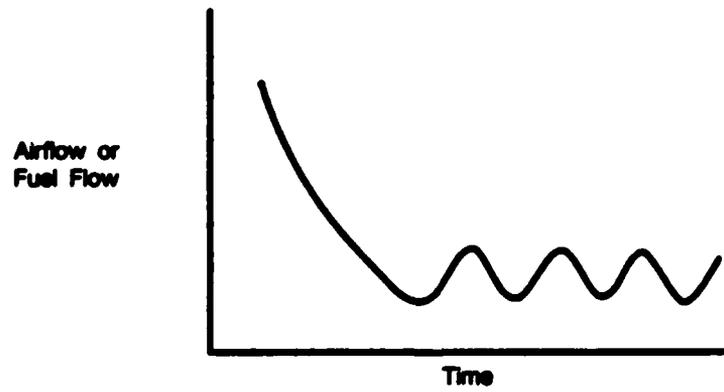
Air temperature response will be limited to 5 Hz at best, and for the majority of the transient testing will be on the order of 1 Hz. Conventional platinum-iridium thermocouples with wire diameters ranging from 15 mils to 3 mils will provide durability during the checkout phase (15 mils) and somewhat better response time (3 mils at 5 Hz) during crucial transient data gathering.



a) Monotonic Transient



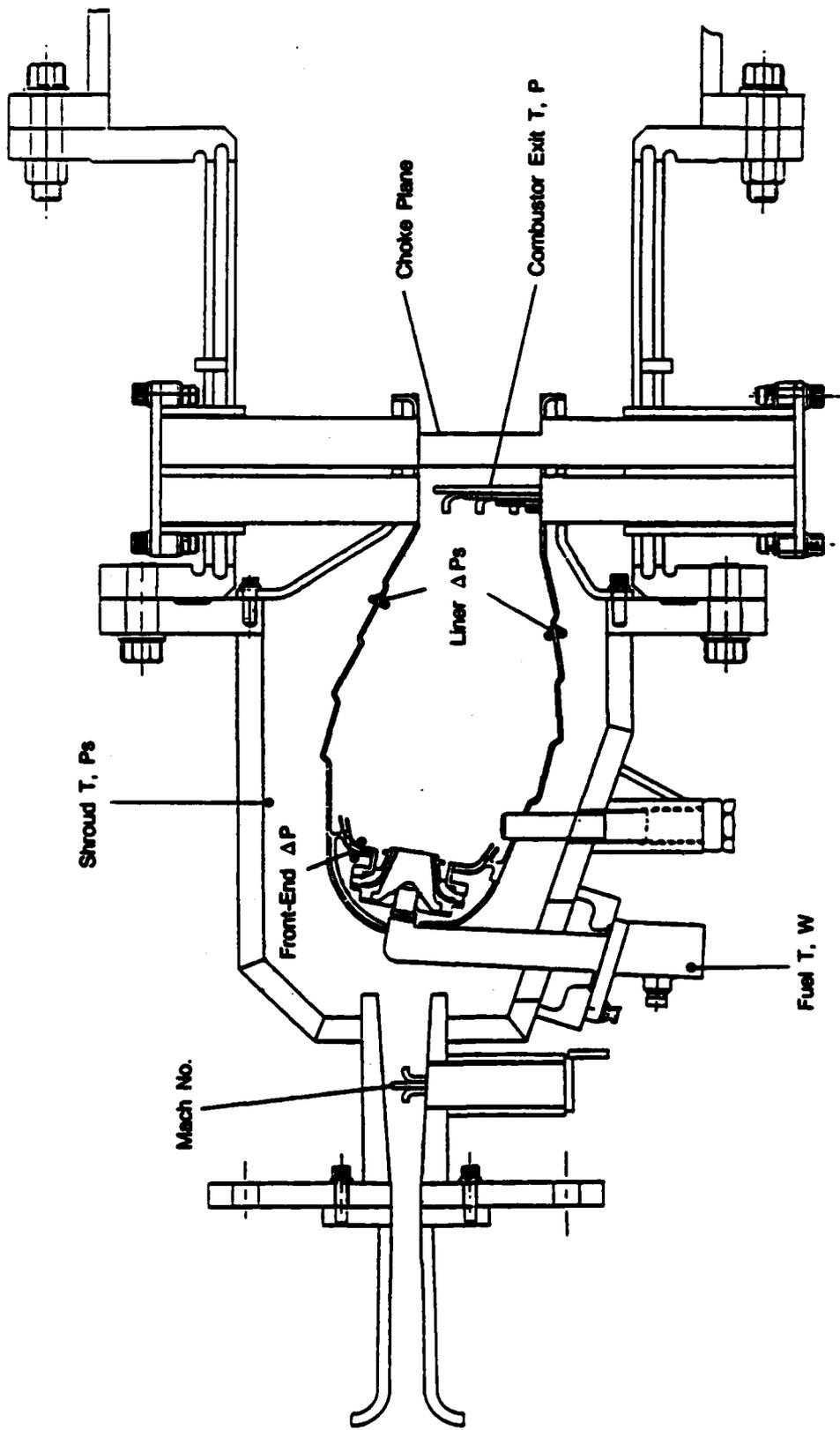
b) Oscillating Transient



c) Combined Transient

FDA 307909

Figure 9. — Combustor Transients



FD 331051

Figure 11. — Extensive Transient Instrumentation Is Required for Dynamic Model Validation Tests

Two 14 track FM tape systems can be used to record up to 26 critical, high response pressures (one each for time). These FM recorders are capable of recording frequencies well above the range of Kulite sensor response expected from this data.

The use of the video recorder is made necessary by the distinct modes of combustor operation (on or off). The optical viewing probe is capable of sending a signal to the computer as a digital representation of the optical data, but the ability to see with the eye the front end of the burner, will be irreplaceable.

With a data acquisition system as capable as what has been described here, and the knowledge that the instrumentation is adequate to perform its required task, the data accumulated should provide the proper information to verify the digital computer model.

**SECTION IV**  
**COMBUSTOR MODELING**

**1. MODEL DESCRIPTION**

The burner physics coded into the original digital computer combustor model inadequately described the burner process for the scope of this program. As listed in the Summary of Findings there were several weaknesses that were recognized and more accurately modeled in the improved computer code. This section will describe the improvements and explain how each was incorporated into the computer code.

Flexibility has been designed into the computer program so that data may be matched by changing constants in equations or by modifying the physical makeup of the burner itself. The input, output, and burner geometry parameters are listed in Table 2, and a description of these parameters is given in Table 3. An example of combustor model predictions for specified inlet conditions are provided in Appendix A.

**TABLE 2.**  
**SUMMARY OF ARRAY PARAMETERS**

<i>States</i>	<i>Output</i>	<i>Diddle Factors</i>
X(1) = P <sub>T35</sub>	Y(1) = P <sub>T3</sub>	P(1) = FPHPT
X(2) = T <sub>T35</sub>	Y(2) = T <sub>T3</sub>	P(2) = ADIFF
X(3) = P <sub>T4</sub>	Y(3) = W <sub>3</sub>	P(3) = ALINR
X(4) = T <sub>T4</sub>	Y(4) = P <sub>T3M</sub>	P(4) = VOL35
X(5) = F <sub>AMB</sub>	Y(5) = T <sub>T3M</sub>	P(5) = VOL4
	Y(6) = W <sub>3M</sub>	P(6) = AP3
	Y(7) = P <sub>T35</sub>	P(7) = BP3
	Y(8) = T <sub>T35</sub>	P(8) = AP35
	Y(9) = W <sub>35</sub>	P(9) = BP35
	Y(10) = P <sub>T35M</sub>	P(10) = AP4
	Y(11) = T <sub>T35M</sub>	P(11) = BP4
	Y(12) = W <sub>T35M</sub>	P(12) = AT3
	Y(13) = P <sub>T4</sub>	P(13) = BT3
	Y(14) = T <sub>T4</sub>	P(14) = $\epsilon_{T3}$
	Y(15) = W <sub>4</sub>	P(15) = A <sub>T35</sub>
	Y(16) = P <sub>T4M</sub>	P(16) = B <sub>T35</sub>
	Y(17) = T <sub>T4M</sub>	P(17) = $\epsilon_{T35}$
	Y(18) = W <sub>4M</sub>	P(18) = A <sub>T4</sub>
	Y(19) = W <sub>FMB</sub>	P(19) = B <sub>T4</sub>
	Y(20) = W <sub>FMBM</sub>	P(20) = $\epsilon_{T4}$
	Y(21) = Q <sub>BURN</sub>	P(21) = AWF
	Y(22) = Q <sub>BURNM</sub>	P(22) = BWF

228C

**a. Airflow Management**

The improved heat release calculation "BURN O2" more accurately models the burner physics. The burner airflow is split into several components, illustrated in Figure 1. These include swirler (or fuel nozzle) airflow, primary (or recirculation) zone airflow and intermediate and dilution zone airflows. The swirler airflow is primarily responsible for fuel atomization which, in turn, impacts combustor ignition and stability. The recirculation airflow, a portion of the air entering through the primary holes, is specified to calculate primary zone stoichiometry, which impacts combustor stability, ignition, chemical efficiency, and circumferential flame propagation.

TABLE 3.

POSTSTALL COMBUSTOR MODEL CANDIDATE PARAMETER SET DIVISIONS

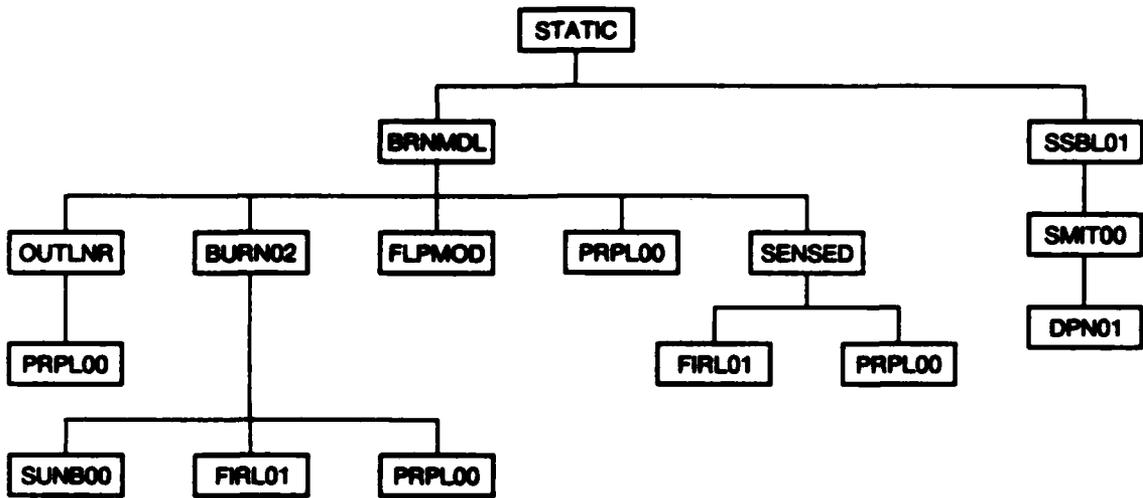
<i>Parameter Name</i>	<i>Description</i>
<b>Mode Change Parameters</b>	
CSTAB	Bias on Characteristic Fuel Evaporations/Reaction Times in Stability Equation
CIG	Bias on Steady State Fuel Evaporation/Reaction Times in Ignition Criterion Equation
CIGE	Scale Factor on Evaporation Time in Ignition Criterion
CIGC	Scale Factor on Reaction Time in Ignition Criterion
<b>Combustor Physical Parameters</b>	
FPHPT	Resistive Area at Exit
ADIFF	Resistive Area at Inlet (Diffuser Dump Loss)
ALINR	Burner Liner Loss
VOL 35	Outer Case Volume
VOL 4	Liner Volume
CPROP	Scale Factor on Flame Propagation (Percent Burning)
CIGDLY	Ignition Delay for Autoignition
CSTABE	Scale Factor on Fuel Evaporation Time (Heat Release)
CSTABC	Scale Factor on Fuel Reaction Time (Heat Release)
CTAU	Bias on Heat Release Lag
CTAUM	Scale Factor on Mining Time (Heat Release)
CEFFA	Quadratic Combustion Efficiency Correlation Term
CEFFB	Linear Combustion Efficiency Correlation Term
DELEFF	Bias on Combustor Efficiency
BOTIME	Time of Blowout
<b>Sensor Model Parameters</b>	
AP3	Diffuser Total Pressure Measurement Scale Factor
BP3	Diffuser Total Pressure Measurement Bias Factor
TAUP3	Diffuser Total Pressure Measurement Time Constant
AP35	Outer case Total Pressure Measurement Scale Factor
BP35	Outer case Total Pressure Measurement Bias Factor
TAUP35	Outer case Total Pressure Measurement Time Constant
AP4	Linear Total Pressure Measurement Scale Factor
BP4	Linear Total Pressure Measurement Bias Factor
TAUP4	Linear Total Pressure Measurement Time Constant
AT3	Diffuser Total Temperature Measurement Scale Factor
BT3	Diffuser Total Temperature Measurement Bias Factor
TAUT3	Diffuser Total Temperature Measurement Time Constant
AT35	Outer case Total Temperature Measurement Scale Factor
BT35	Outer case Total Temperature Measurement Bias Factor
TAU35	Outer case Total Temperature Measurement Time Constant
AT4	Linear Total Temperature Measurement Scale Factor
BT4	Linear Total Temperature Measurement Bias Factor
TAUT4	Linear Total Temperature Measurement Time Constant
AWF	Fuel Flow Measurement Scale Factor
BWF	Fuel Flow Measurement Bias Factor

2281C

**b. Heat Release Routine Logic**

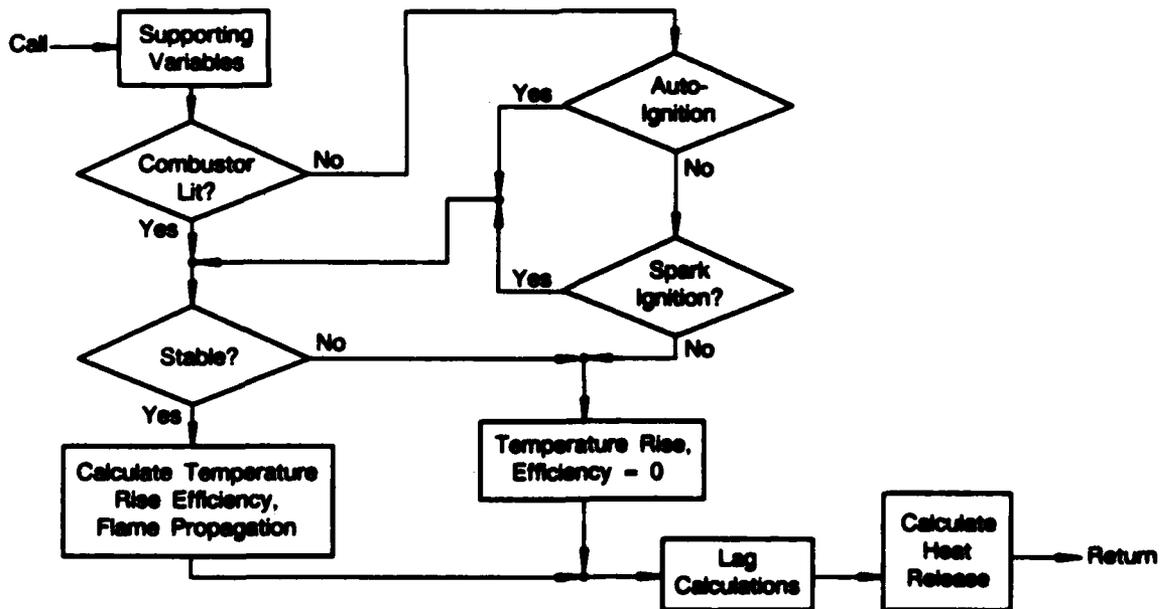
The basic combustor physical processes that are modeled are contained in the program's heat release routine. Figure 12 shows how the improved heat release calculation fits into routine "STATIC" of the main program. STATIC is used to calculate steady-state combustor parameters. The heat release routine is employed to evaluate the state of the combustor (lit or

unlit) and combustor heat release at each time step. Referring to Figure 13, supporting variables for the various combustor correlations are first calculated, after which the state of the combustor at the previous time step is checked.



FDA 331052

Figure 12. — Revised Routine STATIC Including Improved Heat Release Subroutine BURN02



FDA 307813

Figure 13. — Simplified Flow Diagram for Heat Release Routine

If the burner is not lit, conditions in the primary zone are examined to determine if they are favorable for autoignition. If not, the potential for spark ignition is checked. Since the ignitor does not operate continuously, an evaluation of the occurrence of spark ignition includes an

assessment of whether or not the ignitor has fired between the previous and present time steps, as well as an examination of primary zone conditions. Should it be determined that autoignition and spark ignition do not occur, temperature rise and efficiency are equated to zero, and time lag calculations are made.

If the combustor is found to be lit at the previous time step or if autoignition or spark ignition occurs, conditions in the primary zone are examined to determine if stable operation is possible at the present time step. If not, combustor blowout occurs, efficiency and temperature rise are equated to zero, and time lag is calculated. If stable combustion is possible, chemical efficiency, combustor temperature rise, and the extent of flame propagation around the combustor annulus are determined prior to the time lag/heat release calculations. The flame propagation calculation does not apply when the combustor has been lit as a result of autoignition. In this case, it is assumed that conditions are favorable for autoignition at multiple locations around the circumference of the burner, and propagation is instantaneous. If the combustor lights via spark ignition, however, the time required for the flame front to propagate around the annulus is taken into account.

A description of the correlations used to evaluate the various physical processes referred to above is provided in the following section.

#### **c. Heat Release Routine Correlations**

The heat release routine first calculates a number of supporting variables which include:

- Inlet air properties
- Atomization characteristics (Sauter mean diameter)
- Turbulent intensities within the combustor
- Fuel droplet Reynolds number
- Mass transfer numbers
- Effective primary zone fuel-air ratio (based on the fraction of vaporized fuel).

These parameters are necessary to evaluate the various combustor physical processes discussed in the following paragraphs.

##### **(1) Spark Ignition**

The ignition and stability models use the characteristic time approach presented by Ballal and Lefebvre and correlated to many P&W commercial and military aircraft gas turbine engine data by Andreadis. This approach was viewed as being the simplest to incorporate while exhibiting the important physics with flexibility. The spark ignition model states that for ignition to occur, the passage of the spark must create a small, roughly spherical volume of air (spark kernel) with a sufficiently high temperature to initiate evaporation and chemical reaction. If the rate of heat release by combustion exceeds the rate of heat loss by thermal conduction and turbulent diffusion, then the kernel grows to a size capable of flame propagation. The ignition is successful when the time required for the fuel to evaporate ( $\tau_e$ ) and chemically react ( $\tau_c$ ) is less than (on an order basis) the time required for the cold mixture to quench the spark kernel ( $\tau_q$ ) by thermal conduction turbulent diffusion. In equation form:

$$\tau_q > C_1 + C_2\tau_c + C_3\tau_e \text{ (msec)} \quad (1)$$

where  $C_1$ ,  $C_2$ , and  $C_3$  are constants of proportionality or "free variables" which were determined by Andreadis to be:

$$\begin{aligned} C_1 &= 0.0 \\ C_2 &= 0.1 \\ C_3 &= 1.0 \end{aligned}$$

The spark kernel quench time ( $\tau_q$ ) was defined as the ratio of the heat capacity of the spark kernel divided by the average rate of heat loss from the kernel by thermal conduction and turbulent diffusion, i.e.:

For moderate turbulence:

$$\tau_q = \frac{d_q^2}{8(\alpha + 0.08 u' d_q)} \text{ (msec)} \quad (2)$$

For high turbulence, this reduces to:

$$\tau_q = \frac{d_q}{0.64 u'} \text{ (msec)} \quad (3)$$

where:

$$d_q = \left[ \frac{6E}{\pi C_p \rho_s \Delta T_s} \right]^{1/3} \quad (4)$$

and

$$u' = 0.75 \bar{u} \quad (5)$$

The droplet evaporation time ( $\tau_e$ ) is defined as the ratio of the mass of fuel present to the average rate of fuel evaporation for polydisperse spray. Thus according to Ballal and Lefebvre, the droplet evaporation time is given:

$$\tau_e = \frac{C_e^2 \rho_f C_p (SMD)^2}{8 C_e K_s \phi \log(1 + B)(1 + 0.25 C_e^{0.25} Re^{0.5})} \text{ (msec)} \quad (6)$$

where the mass transfer number ( $B$ ) is defined as the available vaporization energy divided by the energy required to vaporize the fuel. In equation form:

$$B = \frac{Q_{av}}{Q_r} \quad (7)$$

The energy required to vaporize the fuel is stated as:

$$Q_r = L_v + C_{pL}(T_3 - T_0) = L_0 + C_{pL}(T_3 - T_0) \quad (8)$$

The energy available to vaporize the fuel is calculated by:

$$Q_{av} = (M_{av} H/r) + C_{pL}(T_3 - T_0) \quad (9)$$

The constants used in the evaporation equation,  $Ce_1$ ,  $Ce_2$ , and  $Ce_3$  are:

$$\begin{aligned} Ce_1 &= 0.31 \\ Ce_2 &= 0.21 \\ Ce_3 &= 0.46 \end{aligned}$$

The characteristic reaction time calculation is dependent upon the relative value of the root mean square of the fluctuation velocity and the laminar and turbulent flame speeds.

$$\tau_c = \frac{12.5\alpha}{(S_L - 0.16u')^2} \quad \text{for } u' < 2S_L \quad (10)$$

or

$$= \frac{15.6\alpha}{u'(S_T - 0.63u')} \quad \text{for } u' > 2S_L \quad (11)$$

where the laminar flame speed is determined from a P&W relationship:

$$S_L = S_{L0} \left( \frac{T_3}{T_{30}} \right)^{1.4} \left( \frac{T_a - T_3}{T_{a0} - T_{30}} \right)^{-1.4} \exp(-2 \times 10^4/T_a + 2 \times 10^4/T_{a0}) \quad (12)$$

From propane and ethane data

$$\begin{aligned} S_{L0} &= 38 \text{ cm/sec} \\ T_{a0} &= 4114^\circ\text{R} \\ T_{30} &= 540^\circ\text{R} \end{aligned}$$

and the flame speed correlated for combustor turbulent intensity using the Shchelkim model as follows:

$$S_T = S_L \left[ 1 + \left( \frac{u'}{S_L} \right)^{1.85} \right] \quad (13)$$

## (2) Stability

For the combustion stability model, the stabilization process is viewed as occurring in the shear layer surrounding the combustor recirculation zone by way of turbulent mixing of fresh reactants and hot products and partially oxidized fuel. The stability criterion is met when the volumetric residence time ( $\tau_{vhr}$ ) of the hot turbulent eddies present in the shear-layer region is greater than the evaporation time ( $\tau_e$ ) plus chemical reaction time ( $\tau_c$ ) (on an order basis). In equation form:

$$\tau_{vhr} > C_4 + C_5\tau_c + C_6\tau_e(\text{msec}) \quad (14)$$

where  $C_4$ ,  $C_5$ , and  $C_6$  were found experimentally by Andreadis to be:

$$\begin{aligned} C_4 &= 0.0 \\ C_5 &= 5.0 \\ C_6 &= 5.0 \end{aligned}$$

The volumetric hot residence time ( $\tau_{vhr}$ ) is defined as the ratio of the primary zone volume divided by the airflow through the primary zone times the gas density. In equation form:

$$\tau_{vhr} = \frac{V_{pz} \rho_g}{W_{a_{pz}}} \text{ (msec)} \quad (15)$$

### (3) Autoignition

The autoignition correlation employed in the current model is based on the work of Spadaccini and TeVelde. In this correlation, the residence time of the unreacting fuel-air mixture ( $\tau_{vhr}$ ) is compared with the ignition delay time ( $\tau_{id}$ ). When ( $\tau_{vhr}$ ) is equal to or greater than ( $\tau_{id}$ ), autoignition occurs. In equation form:

$$\tau_{vhr} > \tau_{id} \text{ (msec)} \quad (16)$$

Ignition delay time is defined as:

$$\tau_{id} = \frac{A}{p^2} \exp\left(\frac{E_G}{RT}\right) \text{ (msec)} \quad (17)$$

where:

$$A = 2.527 \text{ E} - 10 \text{ for JP-4}$$

$$\frac{E_G}{R} = 3.9\text{E} + 4$$

### (4) Combustion Efficiency

The correlation used for calculating combustion efficiency was developed by P&W and has been successfully applied to several commercial engine combustors. In this correlation, combustion or chemical efficiency ( $\eta$ ) is dependent on air loading parameter (ALP) as follows:

$$1/\eta = 0.9943 + C_7(\text{FLPX}) + C_8(\text{FLPX})^2 \quad (18)$$

$$\text{FLPX} = \text{ALP} \phi_{pz} 10^{(-0.00079 \eta \Delta T_p)} \quad (19)$$

$$\text{ALP} = \frac{3.72 W_{a_b}}{P_{T3}^{1.8} (\text{VOL}) 10^{(0.000794 T_{T3})}} \quad (20)$$

where the constants  $C_7$  and  $C_8$  are for the PW2037 and were adopted here because the F100(3) system is being modeled and also has an airblast fuel nozzle system.

$$C_7 = -1.963$$

$$C_8 = 2.501$$

### (5) Flame Propagation

The time required for flame propagation around the combustor annulus following spark ignition is calculated from combustor geometry and turbulent flame speed. The percent of the annulus lit is a linear function of time and flame speed.

## (6) Combustor Heat Release

The current model calculates heat release due to combustion ( $Q_{\text{burn}}$ ) by employing steady-state ideal temperature rise data and correcting for chemical efficiency and the extent of flame propagation around the annulus. Furthermore, a first order lag is applied to account for transient effects. This method provides a realistic transient temperature rise with minimal computational complexity. An approach requiring a transient solution of the chemical composition equations would be too complex for a system type model. Heat release is calculated as follows:

$$Q_{\text{burn}} = W_g C_{pg} PC_B \Delta T_{\text{lag}} \quad (20)$$

where:

$$\Delta T_{\text{lag}} = \eta \Delta T_{\text{ideal}} \left( 1 - \exp \frac{-t}{\tau_v + \tau_c + \tau_q} \right) (PC_B) \quad (21)$$

The lag time ( $\tau_{\text{lag}}$ ) is a function of evaporation, chemical reaction, and combustor residence times as follows:

$$\tau_{\text{lag}} = C_9 + C_{10} \tau_{\text{res}} + C_{11} \tau_v + C_{12} \tau_c \quad (22)$$

where:

$$\begin{aligned} C_9 &= 0.0 \\ C_{10} &= 5.0 \\ C_{11} &= 5.0 \\ C_{12} &= 1.0 \end{aligned}$$

## 2. LIMITATIONS OF THE MODEL

While much progress is being made regarding the development of the P&W combustor dynamic model, there are a number of model aspects which require further efforts. First, the model does not include the required physics for assessing the impact of fuel boiling on combustor transient behavior. The model is currently configured only for constant (liquid) fuel flowrates. Second, the model does not have the capability of handling multiple fuels. Although most of the correlations employed are dependent upon fuel properties, the model fuel property library contains only a single propellant (JP-4) at present. Third, the model does not have the capability to simulate multiple combustor configurations. With the exception of the efficiency correlation, the model uses variables rather than empirical constants to account for the influence of combustor geometry. However, the combustor/fuel nozzle swirler geometry, inside/outside liner volumes, and airflow splits libraries are currently limited to one configuration — the F100(3).

## SECTION V

### SYSTEM IDENTIFICATION BACKGROUND

#### 1. INTRODUCTION

System identification is a powerful technique used to validate and enhance computerized models of dynamic systems. Techniques have been developed which allow definition of system test procedures, identification of model structures and parameters which simulate the dynamic system, and evaluation of the uncertainty and sensitivity of the identified model. The system identification procedure represents an integrated approach to the problem of modeling and understanding uncertain dynamic phenomena. State-of-the-art system identification procedures and theory are discussed in this section.

#### 2. SYSTEM IDENTIFICATION PROCEDURE

Figure 14 illustrates the integrated system identification procedure which is applicable to a wide variety of dynamic systems. The system identification process consists of an iterative loop of test planning, actual testing, and data processing. There are two feedback loops in the process which may be vital to the success of the system identification. An inner loop, closed during data collection, checks the quality of the data produced in the tests. This checking is done in real or nearly real time. If the quality of the data is determined to be poor, then corrections to the testing procedure can be made while the test crew and facilities are still available.

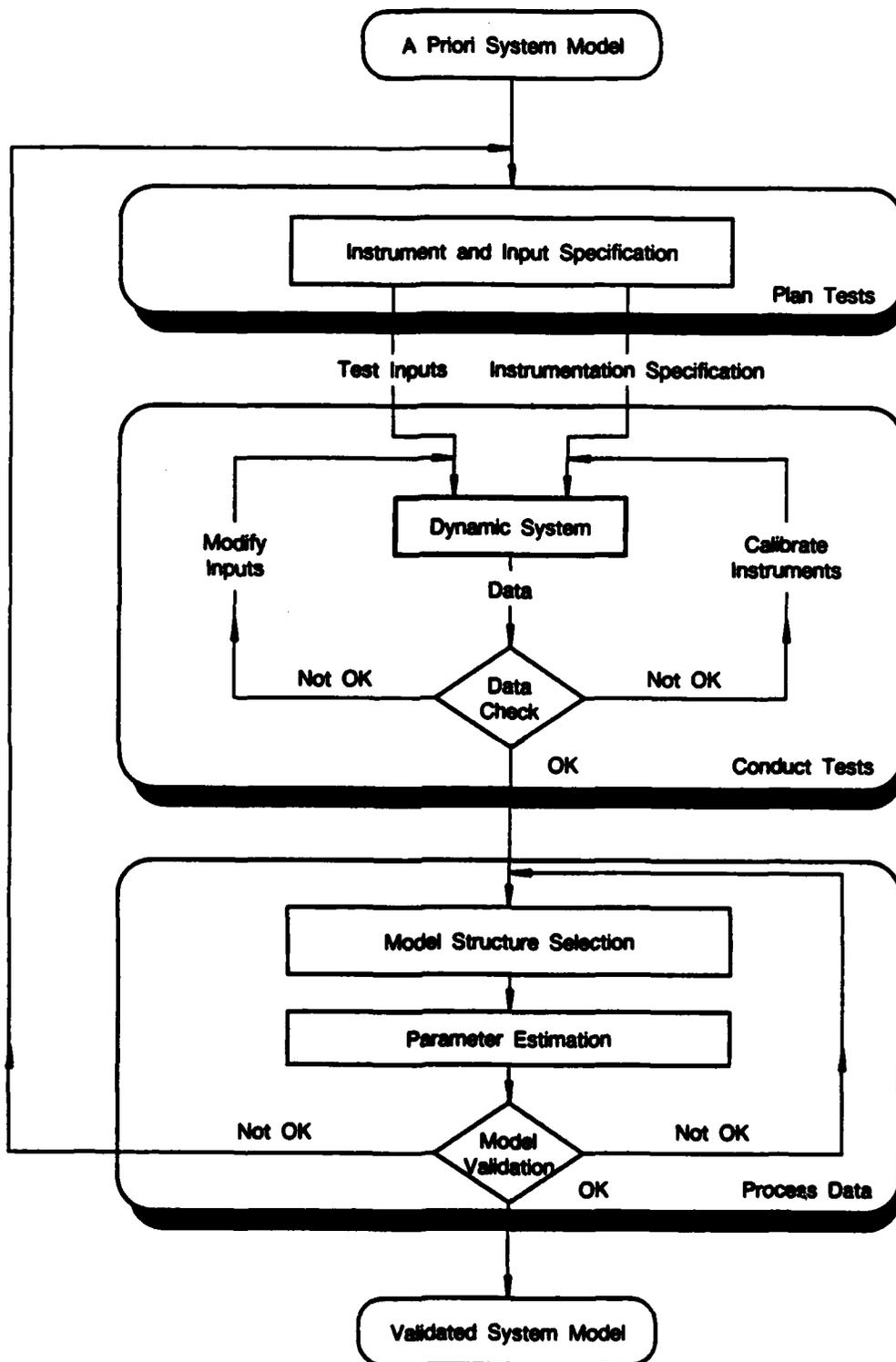
An outer loop around the entire identification process is also closed. For the first pass through the identification process, a model structure is chosen. Given an initial model structure, test input and sensor specifications may be generated. As the model structure and parameter values are iterated upon, the characteristics of the model which result may indicate that different input signals or sensors are required for complete model identification. If this is the case, then additional tests are required.

A key component of the identification process is the estimation of system parameters. The theory of parameter estimation is presented in the following section.

#### 3. PARAMETER ESTIMATION THEORY

The techniques used to identify model parameters from transient data are drawn directly from parameter estimation theory developed at SCT. This section briefly summarizes the concepts of parameter estimation and identification sensitivity. Sensitivity tools are used to determine the effects of instrumentation (e.g., noise, sample rates, lags) and model structure (e.g., tri-modal models, nuisance parameters) on the identification process.

The basic idea in system parameter identification is to find model parameters that make the model results match the actual data as closely as possible. Parameters identified by SCT's parameter estimation algorithm, SCIDNT, represent maximum likelihood values of the estimated parameters. That is, the identified parameters that make the model best follow the unknown system are, most likely, the actual parameters of the plant.



FDA 276398

Figure 14. — Integrated System Identification Procedure

Stated mathematically, the problem is equivalent to minimizing a performance index, the error between measured plant outputs and corresponding model outputs:

$$L(\theta) = \sum_1^N (y - \hat{y}) \quad (23)$$

where  $\dot{x} = f(x, u, \omega, t, \theta)$

and  $\hat{y} = h(x, u, t, \theta) + v$

$N$  = Number of data points

$x$  =  $n_x$  states

$y$  =  $n_y$  outputs

$u$  =  $n_u$  inputs

$\theta$  =  $n_p \times 1$  parameter vector

$t$  = Time

$\omega$  =  $n_x \times 1$  process noise vector

$v$  =  $n_y \times 1$  output noise vector

$$E(\omega) = 0 \quad E(\omega \omega') = Q$$

$$E(v) = 0 \quad E(v v') = R$$

where  $E$  is the expected value

and  $E[x x']$  is the covariance matrix.

The estimated plant outputs are determined by using both state equations and output measurements.

If  $\omega$  and  $v$  are Gaussian, then the performance index in Equation 23, can be expressed as a likelihood function (Equation 24). This likelihood function is Gaussian and indicates the likelihood (probability) that a certain parameter set,  $\theta$ , has produced the measured data,  $y$ . Since  $\theta$  is the random variable here, the likelihood function covers  $n_p$ -space, and instead of a single normal curve, the function is actually an  $n_p$ -space surface.

$$L(\theta) = N \frac{\exp(-v^2/2\sigma^2)}{\sqrt{2\pi}\sigma} \quad (24)$$

To find an optimal set of parameters, the likelihood function in Equation 24 must be maximized, i.e., a  $\theta$  must be found which gives the largest value of  $L$ . Equation 24 can be maximized as is, but a simpler approach considers the negative log of Equation 24. Finding the minimum of the negative log likelihood function is equivalent to finding the maximum of the likelihood function:

$$J(\theta) = - \ln L(\theta)$$

$$\min J(\theta) = \max L(\theta)$$

where:

$$J(\theta) = \sum_1^N (v v' R^{-1} + 2 \ln R^{1/2}) \quad (25)$$

- J = Negative log likelihood (cost function)
- v = Y - Y(θ)
- Y(θ) = Model outputs
- θ = Model parameters
- Y = Plant measured outputs
- R = E(v v') = Covariance of measurement noise.

Minimizing the negative log likelihood is a least squares problem, very similar in form to Equation 23, the general optimization problem. Because v and ω are Gaussian, the relationship in Equation 25 still produces maximum likelihood parameter estimates.

The maximum likelihood parameters occur where J is at a minimum. Finding this minimum requires the use of the gradient of J since the minimum of J occurs where the gradient (slope) of J is zero. The parameter identification algorithm is based upon finding where the gradient of J is zero:

$$\frac{\partial J}{\partial \theta_{\text{final}}} = 0 \quad (26)$$

SCIDNT, the SCT algorithm, uses a Gauss-Newton method to converge to gradient zeros. The convergence follows a first-order iterative algorithm. Consider some initial parameter guess, θ<sub>i</sub>, the gradient of J at θ<sub>i</sub>, and the Hessian (i.e., the gradient of the gradient, or second partial of J) at θ<sub>i</sub>. A first-order guess of where the gradient is zero is

$$\theta_{i+1} = \theta_i - pM^{-1} \times g \quad (27)$$

- where:
- θ = np × 1 parameter vector
  - g = Gradient of J
  - M = Gradient of g and second partial of J
  - p = a user defined scalar (< 1) used to control the rate of convergence.

Convergence occurs by iterating through the algorithm until the gradient drops below a minimal threshold (e.g., 0.001 percent). The critical problem at this point is, how are the gradient and Hessian determined for a nonlinear model. Differentiating Equation 25 with respect to θ yields the following relationships for the gradient and Hessian (written in matrix form for the jth parameter):

$$g(j) = \sum_{i=0}^N \left( v' R^{-1} \frac{\partial v}{\partial \theta(j)} - (1/2) v' R^{-1} \frac{\partial R}{\partial \theta(j)} R^{-1} v + 1/2 \text{Tr} R^{-1} \frac{\partial R}{\partial \theta(j)} \right) \quad (28)$$

$$M = \sum_{i=0}^N \left( \frac{\partial v}{\partial \theta} R^{-1} \frac{\partial v}{\partial \theta} + \text{Tr} \left( R^{-1} \frac{\partial R}{\partial \theta} R^{-1} \frac{\partial R}{\partial \theta} \right) \right) \quad (29)$$

$$M = \sum_{i=0}^N \left( \frac{\partial v}{\partial \theta} R^{-1} \frac{\partial v}{\partial \theta} \right) \quad (30)$$

SCIDNT approximates the gradient and Hessian values by forming estimates of the  $v$  and  $R$  partials using a perturbational technique for each parameter  $\theta(j)$ :

$$\frac{\partial v}{\partial \theta(j)} = \frac{(y(\theta + \Delta\theta) - y) - (y(\theta) - y)}{\Delta\theta(j)} \quad (31)$$

$$= \frac{(y(\theta + \Delta\theta)) - y(\theta)}{\Delta\theta(j)} \quad (32)$$

In effect, SCIDNT propagates two output models, one at  $\theta + \Delta\theta$  and one at  $\theta$ . The partial of  $R$  is found in a manner similar to that used above. From these approximations and the stepping algorithm, SCIDNT produces maximum likelihood parameter estimates.

The key points regarding identifiability and sensitivity lie in the Hessian. The Hessian is also known as the Fisher information matrix. Its name derives from a unique property where its inverse is the covariance matrix of the estimated parameters. As a result, the square root of  $M$  inverse is a measure of the uncertainty of the parameter estimates.

$$M^{-1} = E(\theta \theta') \quad (33)$$

$$\text{diag } M^{-1/2} = E(\theta) \quad (34)$$

This property is the basis for the identifiability formulae developed in the following subsections.

The uncertainty relationship in Equation 33 is an ideal, i.e., the uncertainty predicted by the information matrix is a minimal value based upon exact model and plant agreement. Any disagreement between model and plant in form or parameter values will be reflected as an increase in uncertainty. However, once the model and plant disagree, the inverse square root of  $M$  is no longer a mathematically correct prediction of uncertainty and the values predicted by  $M$  will be less than actual uncertainties. More information and background on parameter estimation and applicable algorithms can be found in References 24 through 31.

#### a. Identifiability

This discussion focuses on the use of the Hessian for developing identification relationships. These relationships relate uncertainty to sample rates, noise levels, and limited sensor sets.

Identifiability is defined as a measure of the likelihood that any estimated parameter is within a specific range of the actual parameter value. Identifiability can also be thought of as a measure of confidence (or uncertainty) in a parameter estimate. This is a more quantitative view of identifiability. Mathematically, uncertainty is expressed as a standard deviation, where a single standard deviation width indicates a 68 percent probability that the actual parameter value is within that width.

Factors affecting  $M$  (and uncertainty) may be understood by expressing  $M$  in a different form. The information matrix relationship in Equation 30 equates  $M$  to a sum of partial innovation vector products normalized by the measurement noise covariance matrix,  $R$ . If  $R$  is considered to be diagonal, then Equation 30 can be expanded to a scalar relation where each measurement (or innovation) contribution is seen separately:

$$M = \sum_{j=1}^m \frac{1}{\sigma(y)^2} \left( \sum_{i=1}^N \left( \frac{\partial v}{\partial \theta} \right)^2 \right) \quad (35)$$

In this form, the relationships between parameter uncertainty, number of data points ( $N$ ), number of measurements ( $m$ ), and measurement noise level ( $\sigma(y)$ ) are more clearly visible. Generally speaking, parameter uncertainty decreases with  $1/(N)$ , i.e., the ability of the parameter estimation algorithm to identify a parameter accurately will improve with the square root of  $N$ . One simple way to increase the number of data points is to use a higher sample rate. The number of points can also be increased by making several runs under the same conditions. In this approach, even though sample rate has not changed, the number of data points for the same type of information has increased. Other conditions impinge upon the selection of sample rate minimums, but generally, identification can be improved without increasing sample rates by using multiple identification runs (i.e., multiple maneuvers).

Equation 34 indicates that uncertainty decreases as the number of outputs is increased — not a very surprising result. Equally predictable is the fact that uncertainty increases with measurement noise. In fact, uncertainty increases in direct proportion to overall increases in measurement noise levels.

The term within the highest level brackets in Equation 34 has a unique characteristic. From Equations 31 and 34 it is apparent that this term is only dependent on the model output and the number of data points. Thus, if  $\theta$  is near the actual  $\theta$ , then the values for each output  $j$  are constant. This is illustrated by rewriting Equation 34:

$$\sigma(\theta) = \sum_{j=1}^m \frac{m(v, \theta)}{\sigma(y)} \quad (36)$$

where  $m(v, \theta)$  is the contribution of the  $j$ th innovation over the entire measurement period ( $N$  points) and  $\sigma(y)$  is the noise level associated with the  $j$ th measurement.

Since  $m(v, \theta)$  is constant near  $\theta$  actual, it is possible to relate uncertainty changes to various combinations of sensor noise levels. In fact by making  $\sigma(y)$  very large, it is possible to see how uncertainty will change without the contributions of a sensor, i.e., what happens to uncertainty under limited sensor sets.

This completes the development of identifiability formulae. These results are used later in this report to explain how uncertainty is quantified. The following section uses the information matrix to show how modeling errors affect identification. Together, these identifiability and sensitivity tools explain a great deal about how successful an identification will be and how identification can be improved.

#### b. Sensitivity

The effect of an error in a parameter that is assumed known on the identification of another is called parameter sensitivity. For in-install engine parameters, the effects of biased or incorrect parameters are substantial. Parameters being identified (affected) are customarily called "estimate parameters"  $\theta$ ; remaining (affecting) parameters are called "nuisance parameters" ( $\phi$ ). As with identifiability, answers to the sensitivity problem are found within the Fisher information matrix.

The information matrix defined in Equation 30 is an  $np \times np$  matrix, for  $np$  model parameters. If only  $ne$  parameters are being estimated,  $M$  can be rearranged and partitioned as follows:

$$M_{12} = \begin{matrix} & \begin{matrix} ne & nn \end{matrix} \\ \begin{matrix} ne \\ nn \end{matrix} & \begin{bmatrix} M_{11} & M_{12} \\ M_{21} & M_{22} \end{bmatrix} \end{matrix} \quad (37)$$

where:  $ne$  = Number of estimate parameters  
 $nn$  = Number of nuisance parameters  
 $np$  =  $ne + nn$

$M_{11}$  is the Fisher information matrix that relates the changes in  $J$  due to changes in only the estimate parameters,  $\theta$ . During parameter identification, only  $M_{11}$  (the portion of the information matrix which pertains to the parameters being identified) is used.

A relationship between  $\theta$  and  $\sigma$  can be found by using other portions of the information matrix (namely  $M_{12}$ ) in conjunction with  $M_{11}$ .  $M_{12}$  can be expressed as:

$$M_{12} = E \left( \frac{\partial J}{\partial \theta} \frac{\partial J'}{\partial \phi} \right) \quad (38)$$

Combining the information in  $M_{11}$  and  $M_{12}$  yields

$$\begin{aligned} X_b &= M_{11}^{-1} M_{12} = E \left( \frac{\partial \theta}{\partial J} \frac{\partial \theta}{\partial J} \frac{\partial J}{\partial \theta} \frac{\partial J'}{\partial \phi} \right) \\ &= E \left( \frac{\partial \theta}{\partial \phi} \right) \end{aligned} \quad (39)$$

where  $X_b$  = bias matrix.

The bias matrix indicates how much the estimate parameters change for a unit change in the nuisance parameters. Equation 38 is the exact solution for the scalar case ( $J$ ,  $\theta$ , and  $\sigma$  scalars) and has similar interpretations in the general matrix case. Equation 38 can now be used to estimate the effects of biased parameters on the estimation of other parameters. For the worst case results, the absolute value of each element in the product of  $X_b$  and  $\sigma$  is taken. The worst case estimate parameter ( $\theta$ ) errors are:

$$\text{Worst Error} = |M_{11}^{-1} M_{12}| |\Delta \phi| \quad (40)$$

where  $\Delta \phi$  is the change in off nominal.

Equation 39 is the primary means used in analyzing potential effects of nuisance parameter errors. A typical application would be the effect of a change in a sensor time constant. For example, assume  $T_{T4}$  has a modeled time lag constant of 100 milliseconds, but the actual sensor time constant is 120 milliseconds. How might this nuisance parameter error bias estimates of poststall combustor parameters? Questions of this type are addressed in a subsequent section of this report.

#### 4. SUMMARY

In the past ten years, system analysis tools have been developed and implemented which are directed toward system identification. The tools include algorithms for mathematical model

structure determination, parameter estimation, and test planning. A unified approach to the problem of understanding uncertain dynamic phenomena has been formulated using these analysis techniques. Application of system identification to combustor poststall operation is discussed in the remainder of this report.

## SECTION VI

### APPLICATION OF SYSTEM IDENTIFICATION PROCEDURES

#### 1. INTRODUCTION

The application of system identification techniques to the poststall combustor model required extensions of the system identification procedure. Because the combustor can operate in different discrete modes such as blown out or ignited from spark or autoignition, special care is required to ensure the success and validity of system identification analysis. The system identification techniques developed in this program to accommodate the complicated nature of the combustor model are detailed in this chapter. Included are discussions of model parameterization, separation of the discrete combustor modes of operation, and classification of model parameters by function in the model. Inclusion of the combustor model into SCIDNT is the first topic presented.

#### 2. LOADING THE MODEL INTO SCIDNT

Prior to inserting the Generic Poststall Combustor model into SCIDNT, the simulation was run on the VAX 11/785 to verify correct operation and to generate test cases to verify the output of the SCIDNT resident simulation. This initial evaluation was completely successful. The simulation matched test case results provided by P&W exactly.

Following simulation installation and testing on the VAX 11/785, the combustor model was loaded into SCIDNT. Out of foresight and necessity, SCIDNT was designed with modularity in mind. SCIDNT subroutines are modular and generic so that dynamic simulations such as the combustor poststall model are easily incorporated.

SCIDNT requires three subroutines as inputs:

- (1) STATE
- (2) STATIC
- (3) MEAS.

Inputs and outputs of these three required SCIDNT subroutines are shown in Table 4. STATE calculates the state derivatives (XDOT). Inputs to STATE include the past states (X), the vector of model parameters (P), and the input control vector (U). STATIC is the SCIDNT initial condition module called at the beginning of each propagation to determine the system initial states. Inputs to STATIC are the input control vector and the model parameter vector. MEAS, the third required SCIDNT module, takes the current state, state derivatives (XDOT), the control input vector, and the vector of model parameters as inputs. The measurement vector (Y) is the MEAS output. Note that any parameter in the P vector can be identified, and SCIDNT is capable of identifying up to 50 parameters simultaneously.

TABLE 4.  
INPUTS AND OUTPUTS OF THE REQUIRED SCIDNT SUBROUTINES

<i>Subroutine</i>	<i>Input</i>	<i>Output</i>
STATE	X, P, U	XDOT
STATIC	U, P	XINITIAL
MEAS	X, XDOT, U, P	Y

2261C

A sample of the STATE, STATIC, and MEAS subroutines from the combustor simulation can be found in Appendix B of this report. These subroutines demonstrate the organizational complexity resulting from dividing the highly modular and flexible code across its many designed boundaries.

The Generic Poststall Combustor model was created expressly for use in the SCIDNT identification code. The model was written with the STATE, STATIC, and MEAS routines in the code. This greatly simplified the installation of the combustor model in SCIDNT. There were, however, a number of changes which were required to ensure proper operation of simulation within the parameter estimation code.

The modifications made to the combustor simulation are conceptually simple but required a good deal of effort to implement and test. The simplest change was to convert all real variables and FORTRAN functions to double precision. This is done as a matter of course in parameter estimation work at SCT to allow maximum numerical accuracy.

The major area of modification of the combustor simulation was to eliminate all reliance of the code on memory between subroutine calls. That is, any code which assumes that on the next pass through that piece of code the variables will still have the values they had on the previous pass. This is characteristic of first order lag approximations and flags which control discrete mode changes. The reason that this is required is that SCIDNT calls the simulation  $N\theta$  times at each time step. Once for each perturbed model corresponding to each perturbed parameter to be identified. SCIDNT was written in this way to conserve memory usage. The SCIDNT code computes the  $N\theta \times NX$  state derivatives at each time step and integrates  $N\theta$  models together. Thus, the code must have all dependence on "memory" in variables (that values will not change between time steps) replaced by vector variables which are indexed by the perturbed model number.

As an example, the combustor ignition condition is stored in the flag MBLITE. If MBLITE equals 0, the combustor is blown out. If MBLITE equals 1, the combustor was ignited by a spark, and if equal to 2, the combustor was ignited by autoignition. SCIDNT will call the simulation with the nominal parameter values and then call it repeatedly using the parameter set with one value perturbed at a time. Then one time step will be integrated over. If on a given call the perturbed parameter value makes the combustor blow out, then MBLITE equals 0. If there isn't a separate MBLITE for each model, on the next call the combustor will be out by virtue of the previous call, which is not correct for the next model.

The combustor model was carefully examined to determine all occurrences of this type of assumed "memory" and, where found, scalar variables were replaced by vectors indexed by the propagation number. The total number of such terms was 18. The SCIDNT code was modified to pass a flag (IPROP), corresponding to the current propagation number, to the STATE, STATIC, and MEAS routines. This flag is used to index the vectors of memory variables.

A related change was required to assure that the correct outputs would be produced for each perturbed model. The simulation was written so that outputs were not computed at the end of a time step, but rather the subroutines were called just to pass the output values to the output routine. This is equivalent to saying that all outputs of the simulation require memory. Model states were already stored for each propagation by SCIDNT, but all non-state outputs were made vectors indexed by propagation number.

In addition to modifying the combustor simulation to account for "memory" variables, the SCIDNT code was modified to use the fourth order Runge-Kutta integration routine which came in the simulation. This was done because the simulation was written so that first order lag approximation would not be updated during intermediate integration time steps based on flags

from this integration routine. By using the simulation integration routine, this work was not repeated and debugged for the SCIDNT integration routine. Also, the performance of the simulation using this type of integration and the corresponding integration step size were known to be good.

The operation of the combustor simulation was verified against the stand-alone simulation which was mounted and tested on SCT's VAX 11/785. The outputs of the stand-alone and SCIDNT simulations were found to correspond exactly. SCIDNT simulation response to a 20 Hz, 250 lb/in.<sup>2</sup> amplitude sinusoid in inlet total pressure, with inlet total temperature and fuel flow fixed at 1000°R and 2.78 lbm/sec respectively is shown in Figure 15. Comparison of exit total temperature,  $T_{T_4}$ , between the stand-alone and SCIDNT simulations is presented in Figure 16. The outputs overplot each other exactly.

### 3. MODEL PARAMETERIZATION

In order to be able to perform identification, the model of interest must be described in a set of constant parameters which can be identified from test data. The combustor simulation was written so that most combustor dynamic characteristics are computed as a function of constant parameters which are usable in parameter estimation.

The choice of which model terms to make the candidates for parameter estimation is made at the highest level based on the purpose of the model and a knowledge of the physics of the system. In this program, the initial choice of parameters was made by Pratt & Whitney after consultation with SCT. This is a large set of all likely parameters of interest.

The process of choosing which model parameters to make the identified parameters in the estimation effort is discussed in Section VII.4a of this report.

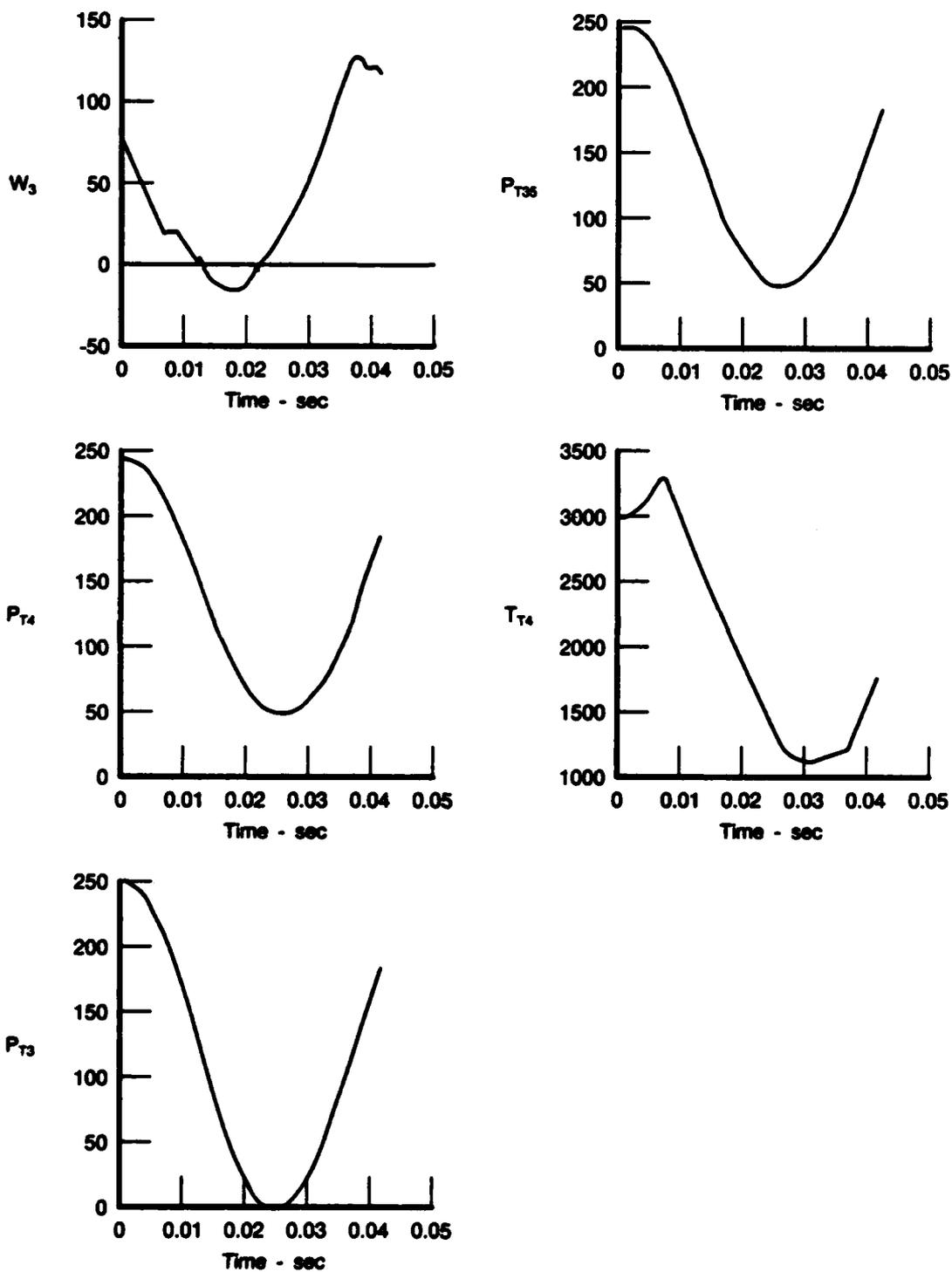
Many parameters which comprise a dynamic engine simulation, especially one formed from test data, are scheduled or stored for data table lookup. (An example in the poststall combustor model is the fuel to air ratio for a given operating condition.) These characteristics must be described by a fitted equation in constant parameters to be identifiable through parameter estimation. Characteristics of this type would probably be generated from combustor rig testing and would require this type of formulation to allow identification.

### 4. COMBUSTOR MODES

Based on experience with identification of poststall compressor models and from observed poststall test data, the discrete modes of combustor operation have been separated for analysis. The model parameter identifiabilities have been evaluated in each mode of operation separately, assuming that the combustor could be tested in this manner. The discrete modes of operation are: blown out, ignited by spark, and ignited by autoignition.

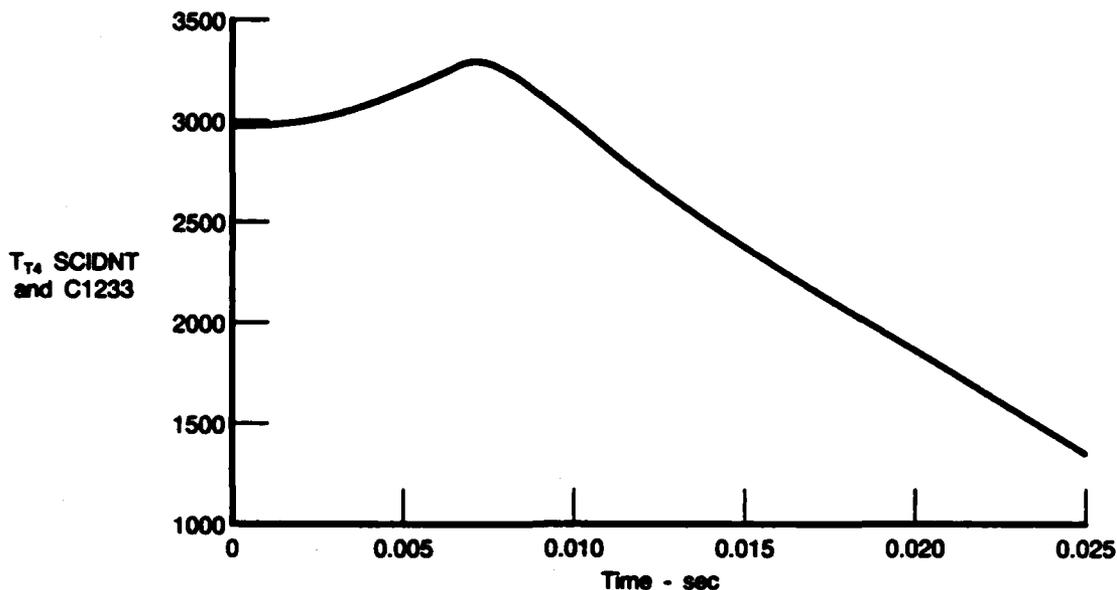
The change of operating mode in the combustor, e.g., blowout, has very dramatic effects on virtually all outputs. This is very similar to the effect of stall occurring in the compressor. SCT has found that a parameter which describes a discrete and dramatic event such as combustor blowout or compressor stall must be identified after the parameters which describe the continuous, nominal system operation are identified. This is because a small error in a parameter which controls the discrete change can create very large output errors. If the outputs are very much in error, the partial derivatives used for parameter estimation will be in error and the parameter estimation algorithm will be unable to converge to the correct parameter values. In addition, if there is any significant model structure error in the equations which control the discrete mode change, the true parameter values for each mode change will be different. The

parameter estimation algorithm will be unable to hit this "moving target," and each incorrect mode change will disrupt the estimation process of the parameters.



FDA 331053

Figure 15. — Poststall Combustor Simulation Outputs for 20 Hz, 250 psia P<sub>73</sub> Sinusoid



FDA 331054

Figure 16. — Overplot of SCIDNT and Stand-Alone Combustor Simulation  $T_{T4}$  Output

The poststall combustor model parameters which affect only off mode change equations are few. These are CSTAB, a constant bias term in the test for combustor stability, and CIG, CIGE, and CIGC constants in the spark ignition criteria equation.

For this program, it was assumed that the combustor could be tested separately in blownout, ignited by spark, and ignited by autoignition modes, or, alternately, that mode could be detected through use of an UV probe and then fixed in the model to match test data during estimation. All parameters but those listed above could be identified in the appropriate modes, and then the "pre-tuned" model could be used to estimate the mode change parameters separately. This is justifiable because no error in the mode change variables will bias the remaining parameter estimates. This is evident if no mode change occurs during the test, but, even if a mode change does occur, it will be forced in the model at the same time. This requires a measurement of the time of mode change. An accuracy requirement for this measurement is discussed in Section VII. It is important to realize, however, that this is really unnecessary since estimation of all but the mode controlling parameters can be done without a mode change occurring during testing.

## 5. PARAMETER CLASSIFICATION

One further subdivision is made in parameter types for this program. This is to divide the non-mode controlling parameters into the physical combustor parameters such as volumes, areas, and reaction times and into sensor model parameters such as sensor bias, scale factor, and time constant. This is done for clarity since these three divisions of parameters are treated differently in the test evaluation process. A summary of the poststall combustor model candidate parameter set division is presented in Table 3.

A comment should be made regarding the importance of modeling the sensors used in testing. Sensors are subject to a variety of errors that degrade both state and parameter estimation accuracies. Reference 35 presents an analytical technique to determine the effect of sensor errors on estimation accuracies. Both random (e.g., additive uncorrelated noise in

measurements) and systematic (e.g., instrument bias or scale factor) errors are treated. One important conclusion is that systematic errors of relatively small magnitude in comparison with random errors can cause significant parameter estimation bias. If such systematic errors are unavoidable, then parameters modeling them can be added to the set of total parameters to be estimated. This technique can reduce overall parameter estimation uncertainty. For this reason, detailed sensor models have been included in the poststall combustor model.

Sensor lags, particularly in temperature probes, are an inevitable nuisance in parameter identification. Unless sensor dynamics are explicitly modeled, the error between estimated and measured combustor response will hamper identification efforts. The type of sensor model used depends on the measurement dynamics in question. First-order lags, scale, and bias errors are modeled for all of the sensors in the combustor model used in this program.

## **6. SUMMARY**

This program required several procedural extensions to accommodate the combustor model in the system identification process. The topics discussed exemplify those procedures required to apply SCIDNT and the overall system identification process to highly complicated nonlinear system models. These modifications also require a very good understanding of the parameter estimation process and the physics and model structure of the poststall combustor — and therefore close teamwork between the engine modeling, parameter estimation, and testing personnel.

## SECTION VII

### TEST EVALUATION

#### 1. INTRODUCTION

In this chapter the method of test evaluation procedure used in this program is presented. The final result of the procedure is a specification of the inputs, sensor accuracy, and nonidentified model parameter value accuracy required in order to identify a model with a given accuracy of combustor output response prediction. In summary, this testing is of value with current instrumentation and can be very useful if modest improvements can be made in the state of the art of combustor instrumentation.

#### 2. TEST EVALUATION PROCEDURE

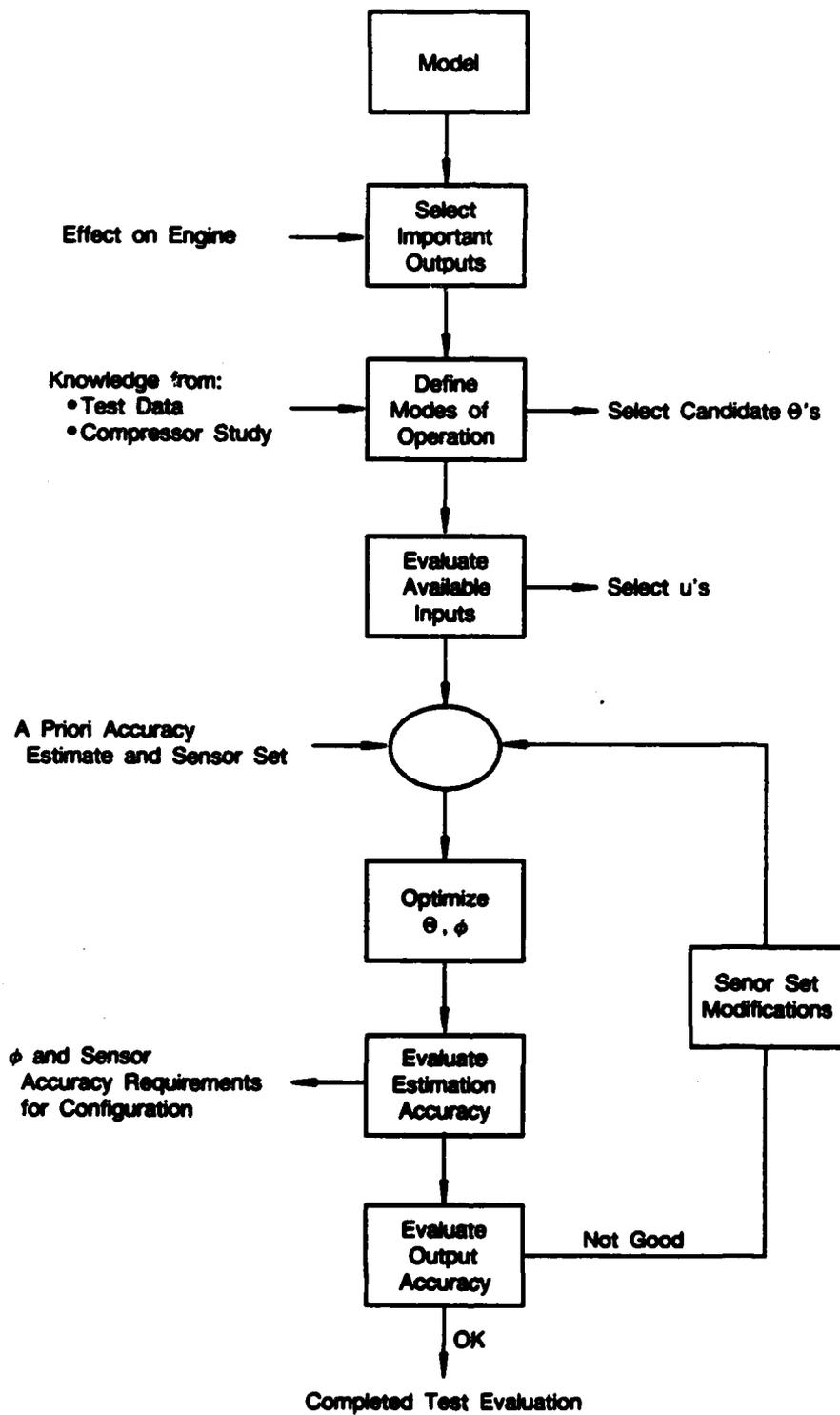
Given a model of a physical system and the purpose for which the model is intended, the test evaluation procedure is used to analyze possible testing methods and to evaluate achievable model output accuracy. Used in an iterative fashion, the procedure can specify the a priori model parameter accuracy and sensor calibration required to achieve desired model fidelity. The test evaluation procedure is a many step process which requires a good deal of insight into the physics of the system and the practical considerations of the testing procedure.

In this program, the poststall combustor rig under consideration does not actually exist at this time. Because of this fact, a large number of assumptions have been required as to possible inputs, sensors, sensor quality, etc., in order to perform this test evaluation. Obviously, all combinations of sensors, inputs, and other test variables could not be analyzed in this effort. A select set of most probable test configurations have been evaluated. If a poststall combustor rig became available and facility capabilities and instrumentation were more definite, additional analysis would provide more specific and definite prediction of testing quality.

The test evaluation procedure applied in this program is shown in Figure 17. The evaluation is begun by selecting the important model outputs. Usually a parameter identification study is undertaken with a goal of increasing model fidelity for certain model outputs of interest so these outputs are predefined. In this program, the goal is to model combustor outputs which affect poststall engine operation, so the outputs of interest are those which drive the upstream and downstream engine components.

The next step was to define the discrete modes of operation of the combustor. This was done as an initial step in selecting the set of model parameters to identify. The reasoning behind this mode distinction and the selection of the candidate model parameters is discussed in Section VI.4 of this report. The result was a set of candidate model parameters from which a specific subset will be chosen for parameter estimation. A distinction is made between the parameters in this set which control change of combustor operating mode and those that do not. This distinction is made because the groups of parameters will be identified separately.

Step four in the procedure is to evaluate the inputs to the combustor. In this program, inputs which are characteristic of engine poststall operation and realizable in rig test were provided by Pratt & Whitney. SCT evaluated these inputs and selected those best suited for parameter estimation.



FDA 331056

Figure 17. — Combustor Test Evaluation Procedure

The next stage in the procedure enters an iterative evaluation process. The a priori values of the model parameter values and sensor characteristics were estimated. Using this information, the candidate estimation set was systematically evaluated to determine the optimum set of parameters to identify.

The assumed sensor and parameter accuracies were then used to determine how accurately the optimum set of parameters could be estimated. This accuracy was then used to evaluate how accurately the identified model could predict the desired outputs. Iteration was then performed, if necessary, to determine how much better the sensor set and knowledge of the nonestimated parameter values needed to be to achieve acceptable model output accuracy.

If acceptable model fidelity could not be achieved with reasonable sensor and model parameter knowledge, additional sensors were added to the sensor set. When this was necessary or if an alternate sensor set was to be investigated, the procedure was reentered at the stage of defining the sensor accuracy and reoptimizing the split between identified and nonidentified parameters.

### 3. PARAMETER AND SENSOR BIAS DETERMINATION

In order to perform test evaluation, a priori estimates of model parameter accuracy and sensor quality must be made. These estimates are used to select the parameters to identify and to begin the estimation accuracy iteration which produces the final specification of required sensor quality and parameter bias levels. Pratt & Whitney and SCT worked together to identify a reasonable set of these values, representative of realizable instrumentation and reflecting current modeling technique. The development of these values is discussed in this section.

#### a. Physical Parameters

Through discussion with Pratt & Whitney, allowable bias levels were estimated for the candidate model parameters. The bias levels are the probable errors in the model parameters. For example, the resistive exit area is 18 in.<sup>2</sup>, and its bias is estimated to be  $\pm 1.8$  in.<sup>2</sup>. Parameters for which good bias estimates were not available were assumed to be accurate to 10 percent. For example, CTAU, a direct bias on the heat release time constant was assumed to be correct to within  $\pm 10$  percent of the nominal time constant value, or 0.0006 second. A list of the a priori bias level estimates for the model physical parameters is presented in Table 5.

#### b. Sensor Model Parameter Biases

The overall sensor accuracies were estimated by Pratt & Whitney test engineers. The sensor inaccuracy was then divided evenly among the sources of sensor error. The sources of sensor error were modeled as shown in Equation 1.

$$P_{T3M} = AP3 \times P_{T3L} + BP3 \quad (41)$$

where  $P_{T3M}$  is the measured output,  $P_{T3L}$  is the model variable after a first order lag and AP3 and BP3 are scale and bias factors. The three sources of sensor error are then the scale, bias, and lag time constant errors.

TABLE 5.

## A PRIORI PARAMETER BIAS ESTIMATES

<i>Parameter Name*</i>	<i>Bias Level (<math>\pm</math>)</i>
FPHPT	1.8
ADIFF	7.0
ALINR	10.0
VOL 35	250.0
VOL 4	260.0
CPROP	0.1
CIGDLY	0.1
CSTABE	0.1
CSTABC	0.1
CTAU	0.0006
CTAUM	0.1
CEFFA	$4 \times 10^{-6}$
CEFFB	$4 \times 10^{-4}$
DELEFF	2.0
BOTIME	0.001

\*Parameters Are Described in Table 3

221C

As an example, Pratt & Whitney estimates that PT43, the diffuser total pressure, can be measured to within  $\pm 1.5$  psia. Dividing this accuracy between three error sources means that each can contribute  $\pm 0.5$  psia error. This then is the estimated error for the bias factor BP3. The scale factor can contribute  $\pm 0.5$  psia also, so if  $P_{T3}$  is nominally 250 psia, AP3 is found to be 0.002. If  $P_{T3}$  is a 50 Hz signal, a 50 msec lag time constant error will produce  $\pm 0.5$  psia error in  $P_{T3}$ . This type of analysis was done for all of the sensor models. The biases generated are shown in Table 6.

#### 4. ANALYSIS TOOLS

##### a. Choosing the Parameters to Identify

The poststall combustor model used in this program is far more complicated than a standard combustor component model. The combustor simulation contains a very large number of physical parameters which could be identified. Selection of the set of parameters to identify is performed on two levels in the test evaluation procedure as shown in Figure 17. A number of analysis methods drawn from the theory described in Section IV.3 and a knowledge of combustor physics allow a useful set of identification parameters to be chosen. This section describes these methods and their application to the combustor simulation.

TABLE 6.  
A PRIORI SENSOR MODEL BIAS ESTIMATES

<i>Parameter Name*</i>	<i>Bias Level (<math>\pm</math>)</i>
AP3	$2 \times 10^{-3}$
BP3	$5 \times 10^{-1}$
TAUP3	$5 \times 10^{-5}$
AP35	$2 \times 10^{-3}$
BP35	$5 \times 10^{-1}$
TAUP35	$5 \times 10^{-5}$
AP4	$2 \times 10^{-3}$
BP4	$5 \times 10^{-1}$
TAUP4	$5 \times 10^{-5}$
AT3	$3 \times 10^{-3}$
BT3	3.0
TAUT3	$5 \times 10^{-5}$
AT35	$3 \times 10^{-3}$
BT35	3.0
TAU35	$5 \times 10^{-5}$
AT4	$3 \times 10^{-3}$
BT4	3.0
TAUT4	$5 \times 10^{-5}$
AWF	$2.5 \times 10^{-2}$
BWF	$7 \times 10^{-2}$

\*Parameters Are Described in Table 3

221C

The first stage at which parameters are selected is in choosing a set of candidate parameters to subject to the quantitative analysis tools. This is done by applying available knowledge of the physical system. Parameters which are known to a high degree of certainty are not included in the list of candidate parameters (e.g., accurately measured geometry, gas density). Additionally, knowledge of the phenomena of interest can be used to select all parameters known to be important. This step is aided by a simple sensitivity analysis where parameters are perturbed and the model propagated so that a qualitative measure of the parameters' influence is found (e.g., which parameters directly affect combustor blowout). The goal of this process is to choose a large set of candidate parameters including all parameters which may be of importance.

In this program, Pratt & Whitney and SCT worked together to select the set of candidate model parameters shown on Table 5.

The purpose of the combustor model in this program is to correctly model performance in poststall engine operation. Because the phenomena of interest are outputs of the simulation, an output sensitivity study is used to choose the optimal parameter set to be identified. Parameter selection through output sensitivity is described in the following section.

#### b. Output Sensitivity

SCIDNT identifies selected model parameters to minimize the model output error. The model parameters to be identified are chosen so as to maximize the fidelity of the model outputs of interest in a given study. This section describes the process of selecting the optimum set of parameters to be identified. A computer code has been written at SCT which automates the parameter-selection process.

When a parameter identification study is undertaken, there is usually a primary goal of increasing model fidelity for certain model outputs of interest (e.g., output flow, component efficiency). Therefore, the parameters to be identified are chosen to minimize error in a weighted sum of these outputs. Specifically, the output error to be minimized can be expressed as

$$\Delta Y_{\text{pred}} = \frac{\partial y}{\partial \theta} \Delta \theta - \frac{\partial y}{\partial \phi} \Delta \phi \quad (42)$$

where  $\theta$  is a vector of identified parameters and  $\phi$  is a vector of nonidentified parameters, also known as nuisance parameters. If too many parameters are identified, the overall parameter identifiability decreases, and  $\Delta Y_{\text{pred}}$  becomes large due to the  $\Delta \theta$  term. Conversely, if too few parameters are selected for identification,  $\Delta Y_{\text{pred}}$  increases due to the biasing effects of the  $\Delta \phi$  (errors in important parameters which are not identified). An optimal choice of parameters to be identified (set of  $\theta$ s and  $\phi$ s) will minimize the output error due to the parameter estimate errors and the errors in the nonestimated parameter values.

Selection of the correct parameter set is a very difficult step in the identification process and one which is very difficult to perform by intuition. For instance, parameters which have relatively certain values and do not directly affect the outputs of interest appear to be likely candidates for nuisance parameters ( $\phi$ s) but may in fact bias the parameter estimates and so require identification. Since,

$$\Delta \theta = M_{11}^{-1} + M_{11}^{-1} M_{12} \Delta \phi \quad (43)$$

as discussed in Section V.3a, a nuisance parameter error may indirectly affect the output of interest by biasing the parameter values estimated.

The computer code developed by SCT to select the identified parameters requires the user to define a priori estimates of the uncertainty in all parameters ( $\Delta \phi$  values). The code then works through all combinations of  $\theta$ s and  $\phi$ s, evaluating for each set the weighted sum of model output errors defined by the user as the performance parameter ( $\Delta Y_{\text{pred}}$ ). The code then returns the optimal choice of  $\theta$  parameters to minimize uncertainty for the outputs selected (subject to the weighting and  $\Delta \phi$  values chosen).

### c. Prediction of Estimation Accuracy

Once the set of parameters to be identified has been chosen, the task is to determine how accurately those parameters' values can be estimated. Alternately, the task is to determine through iteration how good the sensors and knowledge of the plant must be in order for the model to adequately predict combustor response. This is done using a combination of identifiability and sensitivity analyses. The theoretical basis for, and generic application of, these analysis methods are described in Sections V.3a and V.3b of this report.

As described in Section V, the error in estimated parameters comes from two sources. The first is estimation inaccuracy due to sensor noise; the second is estimation bias due to incorrect nuisance parameter values. Given an estimate of the sensor quality and the accuracy of model parameter values, estimation accuracy, can be predicted prior to test. Working backwards in an iterative fashion, the needed sensor set and nuisance bias levels can be found for a desired level of estimation accuracy.

In this program, a desired model output accuracy has been chosen (i.e., 5 percent combined output error in exit total temperature and flow,  $T_{T4}$  and  $W_4$ ). From the output sensitivity matrix,

the estimated parameter accuracies required to achieve this are determined. Working backward from these values in the identifiability/sensitivity analysis, a possible sensor quality and nuisance bias level mix is found to meet these requirements.

## SECTION VIII

### TEST EVALUATION RESULTS

#### 1. INTRODUCTION

This section reports on the results of the 63 test evaluations performed in this program. The result of each evaluation is a specification of sensor quality and model parameter knowledge required in order for a certain fidelity of model to be produced through testing and identification. Each result is dependent upon the sensor set evaluated, the input used, and the set of parameters chosen for identification. In addition, there are an infinite number of tradeoffs possible between model parameter uncertainty and sensor quality. In other words, for a given input, parameter, and sensor set, if the model is known better the sensors can be worse. Because a huge volume of output has been generated in this program, this section will describe in detail the evaluation of one input, sensor, parameter set, and then summarize the results of the other evaluations.

There are two extremes of possible test evaluation results: 1) If the plant model parameters are completely unknown, then testing will require many sensors of extreme accuracy, and 2) If the plant model is completely known, then testing is practically unnecessary, and evaluations will show the sensor set can be small and of low quality. Reality lies somewhere between the two extremes. The test evaluation results follow this general trend but provide a quantitative measure of just how good the model knowledge and sensor quality must be to achieve required model identification accuracy.

The sensor and parameter bias levels which were used to begin the evaluation process were chosen to be reasonable combustor rig instrumentation and model parameter uncertainties. As iteration toward a final uncertainty specification progresses, there are an immense number of possible combinations of bias levels. In this program, a very large number of these possible combinations have been evaluated. While details vary, valuable generalizations about the results can be made. These results are summarized at the end of this section.

#### 2. PARAMETER SEARCH STRATEGY

The variable terms in a test evaluation include combustor operating mode, sensor set, parameter set, sensor noise level, and sensor and nuisance parameter bias levels. In order to produce meaningful results, a logical strategy for selecting sets of these variables is required. The strategy used to select a search path through this variable space is described in this section.

As described in Section VI.4, the combustor operation has been divided into three modes: blown-out, lit by ignitor, and lit by autoignition. The normal mode of operation is the latter, and effort has been concentrated on this mode in this program. In the blown-out mode the combustor is simply flow resistance and volume and, as such, is a subset of normal operation. The lit by ignition case is important because it allows the three parameters which determine if the combustor will light from spark to be identified. However, once lit, this mode is identical to the autoignition mode in terms of model parameter identifiability. Analysis of all modes has been performed; however, effort has concentrated on the autoignition mode of operation.

The strategy used to choose both sensor sets, sensor quality, and bias levels for evaluation has been to define the best of a realizable, probable system description and to make small excursions from this system. For example, the sensor set defined by P&W for a combustor rig test included the first four sensors listed in Table 7. This was the nominal sensor set from which SCT began. Because  $W_3$  may not be measurable in the reverse direction, the set of sensors 2 through 4 was evaluated. Also, if turbine flow were measurable and a choked turbine is assumed,

a measurement of  $W_4$  could be created. Therefore, a sensor set including all of the sensors in Table 7 was evaluated.

TABLE 7.

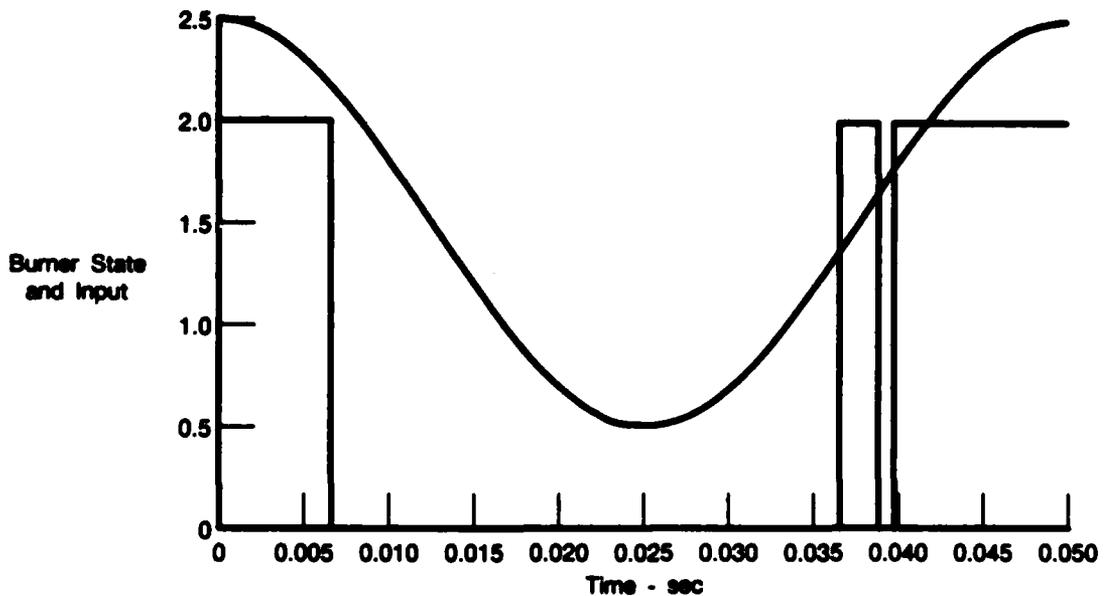
SENSED COMBUSTOR OUTPUTS/INPUTS

<i>Output</i>	<i>Description</i>
1. $W_3$	Mass Flowrate Into Outside Liner
2. $P_{T3}$	Total Pressure at Burner Shroud
3. $P_{T4}$	Combustor Total Pressure
4. $T_{T4}$	Combustor Total Temperature
5. $W_4$	Combustor Exit Mass Flowrate
<i>Input</i>	
1. $P_{T3}$	Diffuser Total Pressure
2. $T_{T3}$	Diffuser Total Temperature
3. $W_{FMS}$	Fuel Mass Flowrate

2281C

3. EXAMPLE TEST EVALUATION

This section presents the detailed result and interpretation of one test evaluation analysis. The sensor set used includes sensors 1 through 5 on Table 7. The input is a 250 psia amplitude, 20 Hz sinusoid in diffuser total pressure with diffuser total temperature and fuel mass flowrate constant at 1000°R and 2.78 lbm/sec respectively. (Sample poststall combustor model responses to this input were presented in Figure 16.) This input forces the combustor to blowout and relight repeatedly. Figure 18 is a plot of  $P_{T3}$  and burner condition through one cycle of the  $P_{T3}$  input sinusoid. A burner state of 2 indicates ignition by autoignition; 0 indicates blowout. Only the output data where burner state equals 2 is used for this evaluation.



FDA 331057

Figure 18. — Burner Ignition State and Input  $P_{T3}$  for Example Case

The sensor bias and parameter bias levels used for this evaluation are those on Tables 5 and 6. Note that a number of these bias terms need not be considered for this input. TAUT3 is not of interest since the  $T_{T3}$  input is a constant value. AT35, BT35, and TAUT35 are not needed because  $T_{T35}$  is not a measurement. CPROP and CIGDLY are used only if the burner ignites from an ignitor spark, which it does not for this input.

The important outputs of this model were selected based on the goal of predicting poststall combustor performance as it influences poststall engine operation. Because a combustor is normally modeled as a gas temperature rise and flow resistance, the outputs selected as measures of model performance were  $T_{T4}$  and  $W_4$ , the combustor exit temperature and flow. Figure 19 contains the output sensitivity matrix for all outputs with respect to the model physical parameters. The matrix is not normalized, so interpretation requires some care. Initial evaluation of Figure 19 indicates that the parameters FPHPT, CTAU, DELEFF, and BOTIME have the largest influence on the outputs  $T_{T4}$  and  $W_4$ . This does not mean that these are the best parameters to identify, however. As discussed in Section VII.4, the influences of the sensor noise and nuisance parameter bias must be considered also.

OUTPUT SENSITIVITY MATRIX =

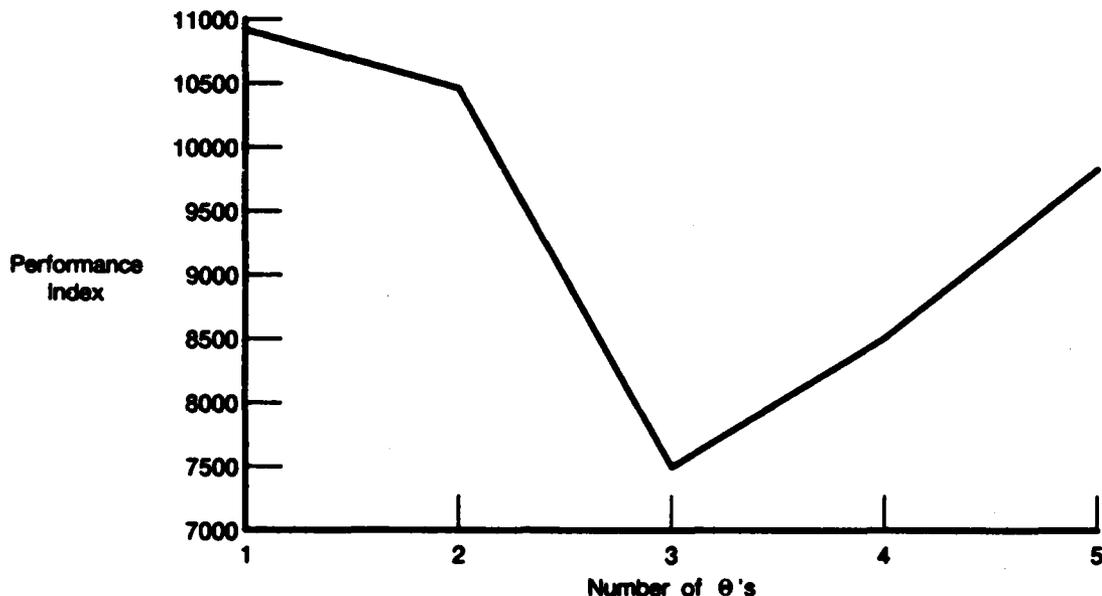
Starting at row 1 columns 1 thru 6						
	<u>FPHPT</u>	<u>ADIFF</u>	<u>ALINR</u>	<u>VOL35</u>	<u>VOL4</u>	<u>CSTABE</u>
W3	1.41700+00	-3.06080-02	-3.85940-02	-2.22290-01	-2.50150-01	-2.57560-02
PT35	-3.04650-02	1.45280-02	6.74060-04	3.90300-03	4.52610-03	4.04680-04
PT4	-5.30280-02	1.48030-02	1.41700-02	3.82290-03	8.61480-03	8.71940-04
TT4	-8.18550-01	-9.78810-03	1.62750-02	-3.36080-03	1.24560-01	9.92950-03
W4	1.23980+00	1.93980-02	7.13230-03	5.33410-03	-4.62830-02	-3.50530-03
Starting at row 1 columns 7 thru 12						
	<u>CSTABC</u>	<u>CTAU</u>	<u>CTAUM</u>	<u>CEFFA</u>	<u>CEFFB</u>	<u>DELEFF</u>
W3	-4.53160-04	-3.55430+00	-3.14770-02	-7.61620-05	8.76800-03	-3.52060-01
PT35	7.88930-06	4.56680-02	4.87500-04	3.12430-06	-1.63490-04	8.04000-03
PT4	1.79710-05	1.04410-01	1.06280-03	5.10080-06	-2.95980-04	1.35890-02
TT4	1.73970-04	1.35650+00	1.11590-02	1.04080-04	-1.84980-02	9.09440-01
W4	-5.88300-05	-4.59800-01	-3.85870-03	-4.14770-05	8.10620-03	-3.99160-01
Starting at row 1 columns 13 thru 13						
	<u>BOTIME</u>					
W3	-2.32760-01					
PT35	3.93230-03					
PT4	7.66980-03					
TT4	1.71020-01					
W4	-6.66540-02					

FD 331058

Figure 19. — Output Sensitivity Matrix

Estimation parameters are selected by use of the SCT developed code PARSEL. This code evaluates output error for all combinations of  $\theta$ s and  $\phi$ s to determine the optimal set of parameters to identify. This is a very large task which is well suited to computer automation. In general, if very few parameters are estimated, error will be high due to the bias effects. Conversely, if too many parameters are estimated, insufficient information will be available from the sensors and errors will again be large. A typical plot of the output performance index versus

the number of parameters estimated as determined by the PARSEL code for this program is shown in Figure 20.



FDA 331059

Figure 20. — Typical Plot of Output Error Vs Number of Estimation Parameters

#### a. Parameter Selection Output

The parameter selection code was run for this case to determine the optimum mix of  $\theta$ s and  $\phi$ s. The program output is included in Appendix C of this report. The first output page lists the a priori uncertainty and bias levels used for all parameters. If any parameters are declared to always be nuisance terms, they are listed with their bias levels, as are the measurements and their assumed noise levels. For this run, CTAU and CTUAM are defined to be nuisance parameters.

The second page describes the performance index to be evaluated and minimized. The description lists J, the performance index, as a sum of terms, while in fact it is a root sum square of the terms listed.

Page 3 is the start of the actual code output. The code examines all combinations of N parameters where N goes from 1 to a user specified limit. For example, page three describes the three best sets of 7 parameters to identify. The code has formed, evaluated, and ordered all possible combinations and lists the p best sets, where p is specified by the user.

In this case, page 3 shows that, if only one parameter is identified, the best is parameter 1, FPHPT. Only slightly worse, the second choice is 41, DELEFF. The root sum square of the estimation and bias error terms for estimated FPHPT is 349.22 while for DELEFF is 350.48. For each, the nuisance parameter set is listed.

The PARSEL code output continues in this manner for the best three combinations of two estimated parameters up to the best three sets of five parameters. The best set of parameters to identify is that which minimizes the performance index overall. Figure 21 is a plot of the lowest

The parameter estimation accuracy is evaluated through the use of an SCT developed computer code called SENSIT. This code evaluates the parameter estimation error for a given set of sensor noise levels and nuisance parameter biases. These two error sources are termed parameter identifiability and sensitivity. The theory used to evaluate these is presented separately in Sections V.3a and V.3b of this report.

The parameter estimation accuracy predicted by SENSIT will produce an error in the model outputs. The output error is computed by multiplying the output sensitivity matrix with the predicted parameter estimation errors. If the resulting output errors are within acceptable limits, then the sensor quality and bias levels assumed will produce sufficiently good test and estimation results, which in turn translate into the required quality of model.

#### (1) *SENSIT Output Discussion*

Appendix D contains the SENSIT output for the first estimation accuracy iteration in the example test evaluation. This run evaluates the estimation accuracy which can be achieved with the a priori sensor set and model knowledge. The first page lists all of the parameters for which sensitivity data were generated and the nominal parameter values to be used in normalization. Also, the measurements used when the sensitivity data were generated are listed.

The second page presents the estimation parameter set and corresponding a priori uncertainty, the nuisance parameter set, and the measurement set to be evaluated with the assumed measurement noise levels. In this output, pages 3 and 4 contain optional output of the  $M_{11}$  and  $M_{12}$  portions of the information matrix and the bias matrix. (A detailed description of these matrices is presented in Section V.) Page 5 of the SENSIT output contains the measurement contributions to the second derivative of the cost function. This matrix shows which measurements are important in the identification of the various parameters. For example, only Y04 and Y05 which are  $T_{T4}$  and  $W_4$  provide information about AT4, BT4, and TAUT4, the sensor model parameters for the  $T_{T4}$  measurement. This is quite logical since these parameters directly scale the  $T_{T4}$  measurement which is in turn used to compute  $W_4$ . Similarly, AP4 and BP4 are only reflected in the  $P_{T4}$  and  $W_4$  measurements. The reasoning here is identical. From these values, it can also be seen that the flow measurements are very important for the estimation of all parameters. The next most critical measurement is the exit temperature, followed by the pressures. Note that for estimation of the combustor efficiency bias, the temperature measurement provides more information than the flow measurements.

The final page of the output contains the run summary information. Each parameter in the estimation set is presented along with the computed standard estimation errors and estimation bias errors. In this case, the estimation bias on the second and third parameters, ADIFF and VOL 35, are exceedingly large. The two error values for each parameter are combined and multiplied by the output sensitivity matrix described in Section VIII.3. This gives the output error created by the estimation error, if too large iteration or sensor quality and bias levels are required. For this run, it is evident from the bias errors in ADIFF and VOL 35 and the output sensitivity matrix of Table 8 that iteration is required.

The nuisance bias values to use for the next iteration are a function of how the investigator has chosen to search the sensor/bias space. In this case, the goal is to minimize excursions from the a priori estimates which reflect current rig test capabilities and modeling confidence.

TABLE 8.

## SUMMARY OF PARAMETER SELECTION RUNS

<i>Run No.</i>	<i>Combustor Mode</i>	<i>Sensor Set</i>	<i>Noise Level</i>	<i>Bias Level</i>	<i>Optimum Estimation Set</i>
1	2	5	Nom	Nom	1,4,41
2	2	5	Low	Nom	1,2,4,41
3	2	5	Nom	Low	4
4	2	5	Low	Low	4
5	2	5	Nom	High	1,4,41
6	2	5	Low	High	1,2,4,41
7	2	4	Nom	Nom	1,41
8	2	4	Low	Nom	1,41
9	2	4	Nom	Low	4
10	2	4	Low	Low	4
11	2	4	Nom	High	1,41
12	2	4	Low	High	1,3,41
13	2	3	Nom	Nom	41
14	2	3	Low	Nom	1,41
15	2	3	Nom	High	1,3
16	2	3	Low	High	1,3
17	1	5	Nom	Nom	1
18	1	4	Nom	Nom	1
19	1	3	Nom	Nom	1
20	0	5	Nom	Nom	1
21	0	4	Nom	Nom	1
22	0	3	Nom	Nom	1

2281C

From the bias matrix on pages 3 and 4, the biases which have the largest effect on each parameter estimation bias can be determined. Lowering each of these parameters' bias levels slightly forms a first iteration. The bias level assumed must be included in this process. If the bias matrix has a large entry corresponding to a nuisance parameter but its bias level is small, obviously this parameter is not a large contributor to the estimation bias.

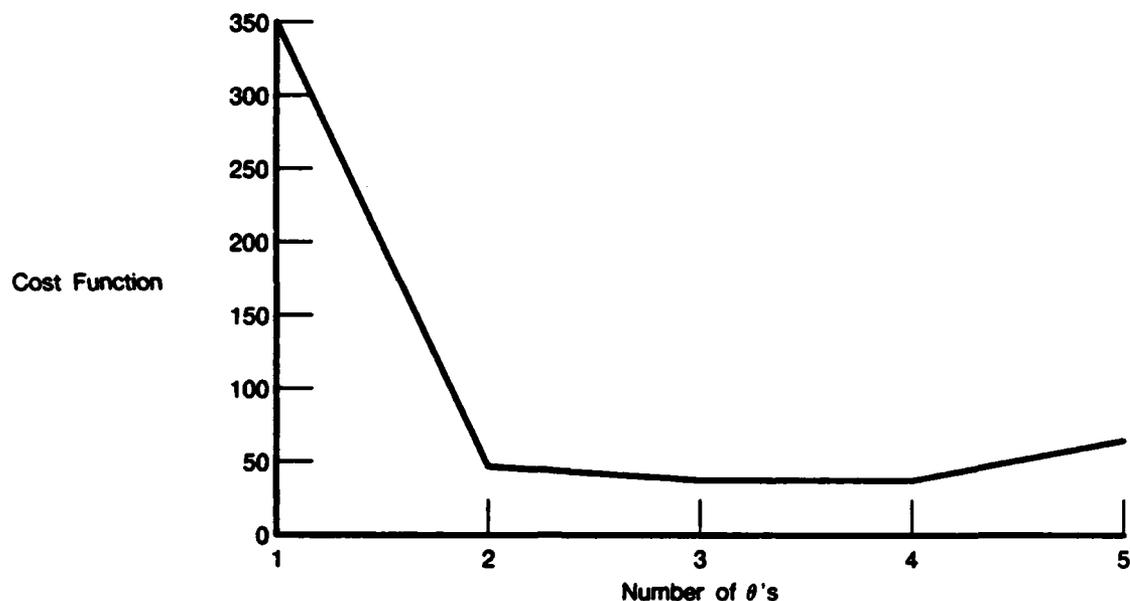
The nuisance bias levels assumed for the run are listed on the final page of the SENSIT output.

Iteration on the bias terms continues in the above manner until acceptable output error levels are achieved. In Appendices D and E, an intermediate run and the final SENSIT run for this evaluation are presented. The final run estimation errors result in errors of 3.62 percent in  $T_{T4}$  and 1.39 percent in  $W_4$ . These levels are considered to be good enough. (For this study, a combined error of 5 percent was an arbitrarily assumed acceptable output error limit.)

The acceptable nuisance bias levels found for this case are small but achievable. The sensor models must be known very well, particularly sensor time constants and any scale factors. This result comes as no surprise, since these terms have a very large effect on the outputs. The required model physical parameter biases range from 0.1 percent to 3 percent of the nominal values. These levels can certainly be achieved for the physically measurable parameters such as the combustor volumes.

The final bias (or uncertainty) levels found for the nuisance parameters in this example are only one of many possible sets which will produce an acceptable level of output error. Many more

performance function for each page of output (best sets of N parameters). The result for this case is that the best set of 3 parameters is optimal; however, the best sets of 2 and 4 parameters are not significantly worse.



FDA 331080

Figure 21. — Output Cost Function Vs Number of Estimated Parameters

The PARSEL output shows that the optimal set contains parameters 1, 4, and 41. However, the second and third best sets of three are nearly equal to the first choice. If the best three sets for each number of parameters are examined, it can be seen that the difference between the three best sets is always the exchange of one variable. For example, the difference between the first and second best set of five parameters is whether parameter 4 or 5 is estimated. The cost function values for these sets are nearly equal. This shows that parameters 4 and 5 have a significant effect on the system. However, the two are not identifiable at the same time. This indicates that parameters 4 and 5 probably have the same effect on the system. This is quite logical since these parameters are the two volumes of the combustor.

The other parameters selected for estimation are the resistive area at the exit and DELEFF, a bias on combustor efficiency. These are quite logical since the outputs in which error is to be minimized are exit flow and exit total temperature. If a fourth parameter is added to the estimation set, the PARSEL code indicates that it should be the inlet resistive area, ADIFF.

This output shows that for the assumed sensor set and bias levels there are a few sets of estimate parameters which will yield approximately the same result in terms of model fidelity in the selected outputs. For this example, we will examine the set containing parameter numbers 1, 2, 4, and 41.

#### b. Estimation Accuracy

Once the estimation set has been chosen, an estimate of the achievable identification accuracy for this set is produced. More importantly, a specification of the sensor quality and plant model knowledge required to achieve a desired fidelity of identified model can be produced.

iterations must be made, studying tradeoffs between various parameter bias levels. This process is performed repeatedly for various sensor sets, estimation sets, and combustor operating modes to complete the test evaluation process.

#### 4. SUMMARY OF EVALUATIONS

This section describes the test evaluations that were performed in this program. The parameter selection analyses made are summarized in Table 9. The sensor set numbers correspond to the number of sensors used in the analysis; the specific sensors are described in Section VIII.2. Standard noise and bias levels were defined. These are described as High, Low, or Nominal, where high is twice the nominal and low is one half of nominal bias levels defined in Tables 5 and 6.

TABLE 9.  
SUMMARY OF TEST EVALUATIONS

<i>Evaluation Set No.</i>	<i>Combustor Mode</i>	<i>Sensor Set</i>	<i>Noise Level</i>	<i>Estimation Set</i>	<i>Approximate No. of SENSIT Estimation Accuracy Evaluations</i>
1	2	5	Nom	1,4,41	4
2	2	5	Nom	1,2,4,41	11
3	2	4	Nom	1,2,4,41	4
4	2	4	Nom	1,41	5
5	2	4	Nom	1,41	4
6	2	4	Nom	1,3,41	6
7	2	3	Nom	1,2,4,41	1
8	2	3	Nom	1,41	5
9	2	3	Low	1,3	2
10	1	4	Nom	1	3
11	0	4	Nom	1	3

2281C

Table 9 reflects the general trends which are expected in parameter selection: when sensor noise is low, more parameters may be identified because more information is available about the system. Similarly, when fewer sensors are available, fewer model parameters can be identified. When bias levels are high (the model is less certain), more parameters are identified to minimize bias errors.

The parameters which have been selected as the optimal estimation sets are consistent and logical. When the five measurements:  $W_3$ ,  $P_{T35}$ ,  $P_{T4}$ ,  $T_{T4}$ , and  $W_4$  are available, the largest estimation set contains parameters 1, 2, 4, and 41, the exit resistive area, inlet resistive area, outer case volume, and efficiency bias respectively. Physical intuition supports this set as those most likely to minimize errors in exit flow and temperature. As was discussed in Section VIII.3.a, the volumes of the combustor are both important parameters; however, because they have the same effects on the available measurements, they cannot be identified at the same time.

If exit flow is dropped from the measurement set, less information is available for estimation. Table 9 indicates that, with nominal bias levels, the inlet resistive area and outer case volume should be dropped from the estimation set. If model confidence is high (low bias), then the volume is the only parameter to be identified.

If both of the flow measurements are dropped, no volumes are selected for estimation, and parameter 3, the burner liner loss, appears in the estimation sets. This is physically

understandable since the remaining sensors (the combustor pressure and temperature and the pressure at the burner shroud) are in the center of the combustor, so parameters which can best be identified are in this area, directly driving these measurements. For example, the inlet resistive area a priori value is more accurate than the area can be identified with these measurements, so to minimize model error, this area is not estimated. The parameters which are estimated are the exit resistive area, delta efficiency, and liner loss.

From the results of the parameter selection analyses, test evaluation sets were formed. The test evaluation sets are the sensors, noise levels, and estimation sets to be carefully analyzed in the estimation accuracy analyses. The evaluation sets formed and processed in this program are listed on Table 10, along with the approximate number of estimation accuracy iterations which were performed for each.

TABLE 10.  
ESTIMATION ERROR VARIATION WITH MEASUREMENT NOISE

Parameter	Percent Increase in Estimation Error			
	FPHPT	ADIFF	VOL 35	DELEFF
10 Percent Noise Increase in Measurement				
$W_3$ and $W_4$	8.39	7.12	8.98	4.93
$P_{T35}$ and $P_{T4}$	0.13	1.82	0.21	0.19
$T_{T4}$	1.47	0.50	0.61	5.18

2281C

#### a. Generalization of Evaluation Results

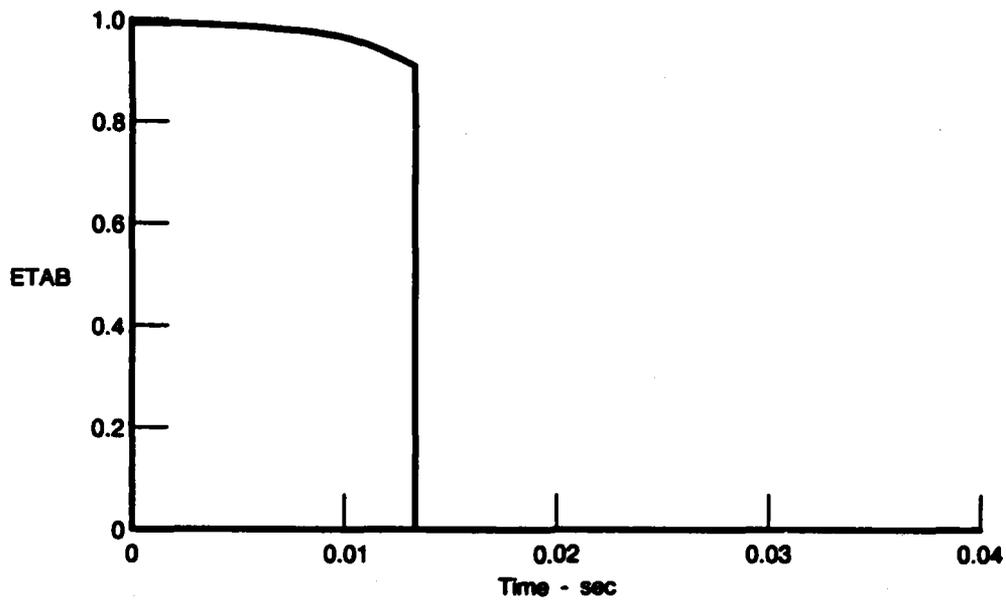
In this section, selected results from the broad range of test evaluations are presented.

- Combustor identification must be performed in two stages. In the first stage the combustor operating mode (e.g., blown out, lit) is fixed in the model and parameters affecting gross transient behavior are identified. In the second stage the mode change functions are identified. This program has addressed Stage 1.
  - Parameters which affect the stability functions are very important in that they drastically effect the combustor transient, however their effect is limited to a very small portion of the total transient time. From a practical point of view the parameters are divided into those which control and do not control mode change based on frequency separation.
  - The important combustor model parameters which are only identifiable during combustor mode transition are the time constants in the stability equation and the coefficients in the combustor efficiency correlation equation.
- Change of combustor operating mode (e.g., blowout) must be measurable in order to perform accurate parameter estimation.

- If identification must be performed using data from a transient during which the combustor mode changes, the mode change must be forced in the model so that model and data are consistent.
- If operating mode is sensed and forced in the model, the time of mode change must be sensed to within  $\pm 0.001$  sec.
- Efficiency is a very important term in combustor models. The delta combustor efficiency used in this program is indistinguishable from the heat value of fuel in steady state, but is separable from transient data. The importance of DELEFF identified simply means that efficiency must be computed very accurately.
- Overall combustor efficiency was found to be very important while the coefficients of the combustor efficiency correlation equation were not. This is due to the very brief drop in efficiency prior to blowout. As shown in Figure 22, the combustor efficiency is virtually constant until blowout. Since the efficiency correlation coefficients contribute almost nothing to the overall efficiency value until blowout (and then the terms are relatively small), the coefficients were not of primary importance in predicting transient behavior which has few or no mode changes.
- Estimation accuracy decreases rapidly if flow measurements are not available, however acceptable results can still be obtained using only pressure and temperature probes.
  - If sensors which measure  $W_3$ ,  $P_{T35}$ ,  $P_{T4}$ ,  $T_{T4}$ , and  $W_4$  are available and the RMS noise levels on these sensors are 0.5, 1.5, 1.5, 10.0, and 0.5 respectively, then four model parameters can be successfully identified. The optimal set of these to minimize model error in outputs  $T_{T4}$  and  $W_4$  is FPHPT, ADIFF, VOL 35, and DELEFF. (Once this identification has been performed, other important parameters such as VOL 4 can be substituted into the estimation set, based on the knowledge that certain parameters such as VOL 4 and VOL 35 cannot be identified at the same time.) To perform this identification, the model nuisance parameters must be known within an average of 6 percent of their true values. Additionally, any sensor scale factors must be known to within  $\pm 4 \times 10^{-3}$  of their true values, on the average. Sensor bias errors must be below  $\pm 1/2$  percent on the average, and sensor time constants must be known with  $\pm 1/4$  percent of their true values, on the average.
  - If the sensor noise and bias levels reported in Appendix E (which meet the above requirements) are used to estimate FPHPT, ADIFF, VOL 35, and DELEFF, the resulting output accuracy is 3.62 percent error in  $T_{T4}$  and 1.39 percent in  $W_4$ .
  - If the above noise and bias levels are used to identify the same four parameters but only the  $W_3$ ,  $P_{T35}$ ,  $P_{T4}$ , and  $T_{T4}$  measurements are available, the achievable output accuracy is 4.21 percent error in  $T_{T4}$  and 2.23 percent error in  $W_4$ . This

error results from a 44.7 percent average increase in the parameter estimation errors.

- If the above noise and bias levels are used to identify the same four parameters but only the  $P_{T35}$ ,  $P_{T4}$  and  $T_{T4}$  measurements are available, the achievable output accuracy is 5.97 percent error in  $T_{T4}$  and 4.16 percent error in  $W_4$ . This error results from a 76.5 percent average increase in the parameter estimation errors.
- If a  $W_4$  sensor is not available, with the nominal sensor and model uncertainties, the optimal set of three estimation parameters is FPHPT, ALINR, and DELEFF. Similarly, if the  $W_3$  sensor is not present in the measurement set, only two parameters should be identified, the optimal set of which is FPHPT and DELEFF.
- Measurement noise is not the major cause of estimation error. Systematic measurement errors such as sensor biases are much more significant sources of estimation error than is measurement noise.
  - Estimation accuracy variation with sensor noise. Using the nominal sensor and parameter bias values, estimation accuracy of parameters 1, 2, 4, and 41 has been evaluated for a 10 percent increase in sensor noise levels. Table 10 summarizes these evaluations. It is evident that flow measurements are very important, particularly in estimation of the volume. The temperature measurement is the second most important and is particularly valuable in the estimation of combustor efficiency. The estimated combustor measurement noise levels have been shown to be acceptable, and Table 10 indicates that estimation accuracy is not overly sensitive to random measurement noise. Obviously, measurement noise should be minimized, particularly in the flow measurements; but, in comparison to systematic errors such as sensor bias and scale, measurement noise is not a critical source of estimation error for poststall combustor model identification.
- Inputs which represent poststall engine behavior and drive the combustor into its operating modes are sufficient for initial parameter estimation efforts. Specifically, inputs which are characteristic of engine surge sufficiently excite all modeled combustor dynamics.
- Estimation results are dependent upon the operating point where estimation is performed, and limited by the model structure used. For example, if a combustor characteristic is strongly correlated with flow, however this dependency is unmodeled, estimation results will vary significantly with operating point, and cannot explicitly reflect the flow dependence.



FDA 331061

*Figure 22. — Combustor Efficiency During Mode Change*

## SECTION IX

### TEST METHODOLOGY

The data reduction will consist primarily of processing and compiling data into tabular, graphical, and video form for eventual comparison to model predictions. However, some intermediate computations will be required to determine pressure losses and combustion efficiency.

Combustion system percent pressure losses will be calculated in the usual manner for the system (diffuser inlet to combustor exit), combustor dome, and combustor liner:

$$\Delta P_{\text{loss}} = \frac{P_{\text{up}} - P_{\text{down}}}{P_{\text{up}}} \times 100 \quad (44)$$

Upstream pressure ( $P_{\text{up}}$ ) and downstream pressure ( $P_{\text{down}}$ ) will be defined according to the normal flow direction.

A temperature-based combustion efficiency will be calculated from the data using the following equation:

$$\eta = \frac{\Delta T_{\text{act}}}{\Delta T_{\text{ideal}}} \quad (45)$$

The actual temperature rise can be calculated from the diffuser inlet temperature and either a measured or a calculated combustor exit temperature. The measured combustor exit temperature will be determined using the combustor exit temperature instrumentation. The calculated exit temperature will be based on the choked flow conditions at the simulated turbine using the flow parameter (FP). Flow parameter is defined as:

$$FP = \frac{W_a(T_3)^{0.5}}{PA} = f(Mn, \gamma) \quad (46)$$

Since Mach number, flowrate, pressure, and flow area will be known at the choke plane, this equation can be solved iteratively for temperature and gamma.

## SECTION X

### SUMMARY AND FUTURE EFFORTS

#### 1. SUMMARY

A lumped-parameter computer model for combustor transient behavior during compressor stall events has been described. Recent improvements to the model include a characteristic time approach to stability, the addition of spark ignition and flame propagation, droplet size calculation, and the division of the combustor into primary and secondary zones. This model has undergone a type of analysis called Systems Identification, which has subjected the model to various transient boundary conditions that highlight the input parameters which most affect the model's calculated results. Those parameters are: prediffuser exit area, combustor volume, liner flow area, combustion efficiency, and turbine inlet flow parameter. Systems Identification also specifies the accuracy of pressure and temperature data, from transient combustion experiments, needed to improve the model's accuracy. Finally, this report reviews a transient combustion facility being built at United Technologies Research Center (UTRC) and scheduled to become operational in early 1987.

#### 2. FUTURE EFFORTS (NONCONTRACTUAL)

In the fourth quarter of 1987, testing of a four-nozzle combustor sector, currently under construction, will begin at the new UTRC facility. Data from these transient combustion tests will be used to validate and upgrade the computer model. Eventually, the model will be incorporated into the design system, allowing combustors to be designed for stall recovery along with the more traditional features of operability, pattern factor, and combustion efficiency. The model will also be absorbed into the engine simulation computer program, complementing the existing compressor stall model, and increasing the realism with which engine cycles are calculated.

## REFERENCES

1. Lefebvre, A. H., *Gas Turbine Combustion*, McGraw Hill, New York, 1983.
2. Bruce, T. W., Mongia, H. C., and Reynolds, R. S., *Combustor Design Criteria Validation, Volume-1, Element Tests and Model Validation Final Report USARTL-TR-78-55A*, March 1979.
3. Ballal, D. R. and Lefebvre, A. H., "Flame Propagation in Heterogeneous Mixtures of Fuel Droplets, Fuel Vapor and Air," *Symposium (Int'l) on Combustion*, pp 31-328, the Combustion Institute Pittsburgh, 1980.
4. Ernst, R. C. and Andreadis, R., "Fuel Effects on Gas Turbine Engine Combustion," *Final Report, AFWAL-TR-83-2048, ESL-TR-83-65*, June 1983.
5. Al Dabbagh, N. A. and Andrews, G. E., "Weak Extinction and Velocity for Grid Plate Stabilized Premixed Flames," *Combustion and Flame* 55 : 31-52 (1984).
6. Ballal, D. R. and Lefebvre, A. H., "A General Model of Spa gaseous and Liquid Fuel-Air Mixtures," *18th Symposium (Int'l) on Combustion* pp 1737-1746, the Combustion Institute, Pittsburgh, 1981.
7. Zabetakis, M. G., Furno, A. L. and Jones, G. W., "Minimum Spontaneous Ignition Temperatures of Combustibles in Air," *Industrial and Engineering Chemistry*, V46 No. 10, pp 2173-2178, 1954.
8. Swithenbank, J., Turan, A., Felton, P. G. and Spaldi, D. B., "Modelling of Mixing, Evaporation, and Kinetics in Gas Turbine Combustors," *AGARD CP275*, Cologne, Germany, October 1979.
9. Peters, J. E. and Mellor, A. M., "Characteristic Time Ignition Model Extended to an Annular Gas Turbine Combustor," *The Combustion Institute Fall Meeting*, Tempe, Arizona, October 1981.
10. Hunter, S. C., Johnson, K. M., Mongia, H. C. and Woo, M. P., *Advanced Small, High-Temperature Rise Combustor Program, Volume I, Analytical Model Derivation and Combustor Element Rig Tests (Phase I and II)*, USAAMRDL Technical Report 74-3A, February 1974.
11. Browkaw, R. S. and Jackson, J. L., "Effect on Temperature, Pressure, and Composition on Ignition Delays for Propane Fuels," *5th Symposium (Int'l) on Combustion*, the Combustion Institute, Pittsburgh, pp 563-569, 1955.
12. Mellor, A. M., "Semi-Empirical Correlations for Gas Turbine Emissions, Ignition, and Flame Stabilization," *Prog. Energy Combustion Science, Turbine Combustor Modelling, Combustion Science and Technology*, 1970, Vol. 2, pp 67-80.
13. Odgers, J., "Combustion Modelling Within Gas Turbine Engines, Some Applications and Limitations," *AIAA 15th Aerospace Sciences Meeting*, Paper No. 77-52, Los Angeles, January 1977.
14. Ballal, D. R. and Lefebvre, A. H., "Ignition and Flame Quenching of Flowing Heterogeneous Fuel-Air Mixtures," *Combustion and Flame* 35: 155-168, 1979.

15. Ogerby, I. T., "A Literature Review on Turbine Combustor Modeling and Emissions," Arnold Engineering and Development Center AEDC-TR-73-163, November 1973.
16. Ballal, D. R. and Lefebvre, A. H., "Ignition and Flame Quenching of Quiescent Fuel Mists," Proc. Royal Soc. Land. A 364, 277-294, 1978.
17. Spadaccini, L. J., "Autoignition Characteristics of Hydrocarbon Fuels at Elevated Temperatures and Pressures," United Aircraft Research Laboratories Report UAR-N141, September 1974.
18. Swithenbank, J., Poll, I., Vincent, M. W., "Combustion Design Fundamentals," 14th Symposium (Int'l) on Combustion, pp 627-638, Penn. State University, August 1972.
19. Adelman, H. G., "A Time Dependent Theory of Spark Ignition," 18th Symposium (Int'l) on Combustion, the Combustion Institute, pp 1333-1342, 1981.
20. Odgers, J. and Carrier, C., "Modelling of Gas Turbine Combustors; Considerations for Power," pp 105-113, April 1973.
21. Andreadis, D., "Altitude Ignition/Lean Decel Study," Pratt & Whitney, Engineering Division, AFWAL-TR-85-2054, November 1985.
22. Westmore, W., "F100 Stagnation Stall Fixes to be Tested," *Aviation Week and Space Technology*, pp 45-48, November 27, 1978.
23. Allen, D. M., "Mean Square Error of Prediction as a Criterion for Selecting Variables," *Technometrics*, Vol. 13, No. 3, Aug. 1971, pp 469, 475.
24. Trankle, T. L., "Practical Aspects of System Identification," ASME Paper 79-WA/DSC-23, presented at Winter Annual Meeting, New York, December 3, 1979.
25. Lawson, C. L., and Hanson, R. J., *Solving Least Square Problems*, Prentice-Hall, 1974.
26. Marquardt, D. W., "An Algorithm for Least Squares Estimation of Nonlinear Parameters," *J. Soc. Indust. Appl. Math.*, Vol. 11, No. 2, 1963, pp 431-441.
27. Kalman, R. E., Bucy, R., "New Results in Linear Filtering and Prediction," *Trans. ASME*, Vol. 83D, 1961, p. 95.
28. Luenberger, D. G., "Observing the State of a Linear System," *IEEE Trans. Military Electronics*, Vol. MIL-8, 1964, pp 74-80.
29. Fisher, R. A., "Two New Properties of the Mathematical Likelihood," *Proc. Roy. Soc., London*, Vol. 144, 1934.
30. Trankle, T. L., Vincent, J. H., Franklin, S. N., "System Identification of Nonlinear Aerodynamic Models," prepared for NATO AGARDOGRAPH "The Techniques and Technology of Nonlinear Filtering and Kalman Filtering," February 1982.
31. Forsythe, G., Moler, C. B., *Computer Solution of Linear Algebraic Systems*, Prentice-Hall, 1967.
32. Bierman, G. J., *Factorization Methods for Discrete Sequential Estimation*, Academic Press, New York, 1977.

33. Maine, R. E. and LLiff, K. W., "User's Manual for MMLE3, A General FORTRAN Program for Maximum Likelihood Parameter Estimation," NASA TP-1563.
34. Gupta, N. K., Hall, Jr., W. E., "Design and Evaluation of Sensor Systems for State and Parameter Estimation," *J. Guidance and Control*, Vol. 1, No. 6, November/December 1978, pp. 397-403.

SLOW FUEL DECEL (10,000 PPM TO 0 PPM) IN TEN SECONDS AT CONSTANT INLET TEMPERATURE AND PRESSURE

SMITE HAS CONVERGED IN 25 ATTEMPTS  
 \*\* STATISTICS \*\* 31 TOTAL PASSES,  
 1 JACOBIAN EVALUATIONS, 16 BROYDEN UPDATES

\*\*\*\*\*

BROD1

1 TIME	.0	.1004965	.2009932	.3014819	.4019746	.5024673	.6029601
2 ADIFF	70.00000	70.00000	70.00000	70.00000	70.00000	70.00000	70.00000
3 ALINR	100.00000	100.00000	100.00000	100.00000	100.00000	100.00000	100.00000
4 DFAMB	-.6013601E-03	-3.257597	-3.221265	-3.185199	-3.149009	-3.113031	-3.076905
5 DPT35	-1.832967	-102043.5	-101718.9	-101395.4	-101073.2	-100753.0	-100431.7
6 DPT4	-1.778266	129299.2	128769.4	128239.2	127710.2	127181.2	126652.1
7 DTT35	-2.116904	-742109.3	-733254.7	-724457.6	-715740.6	-707122.6	-698595.2
8 DTT4	-71.06857	332406.1	329354.8	326295.4	323288.6	320292.6	317337.1
9 FAMB	.3584334E-01	.1563948E-01	.1547930E-01	.1531936E-01	.1515944E-01	.1499978E-01	.1484017E-01
10 FA3	.0	.0	.0	.0	.0	.0	.0
11 FA35	.0	.0	.0	.0	.0	.0	.0
12 FPHPT	18.00000	18.00000	18.00000	18.00000	18.00000	18.00000	18.00000
13 GAMB	1.400000	1.400000	1.400000	1.400000	1.400000	1.400000	1.400000
14 GAM35	1.400000	1.400000	1.400000	1.400000	1.400000	1.400000	1.400000
15 GAM4	1.400000	1.400000	1.400000	1.400000	1.400000	1.400000	1.400000
16 IUPDAT	.0	.0	.0	.0	.0	.0	.0
17 PRDIFF	1.016828	1.124823	1.124230	1.123631	1.123032	1.122427	1.121828
18 PRLINR	1.008378	1.336489	1.336885	1.337286	1.337697	1.338113	1.338534
19 PT3	250.0000	250.0000	250.0000	250.0000	250.0000	250.0000	250.0000
20 PT35	245.8627	222.2572	222.3742	222.4929	222.6116	222.7315	222.8505
21 PT4	243.8199	166.2993	166.5867	166.8755	167.1637	167.4530	167.7416
22 R3	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000
23 R35	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000
24 R4	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000
25 TT3	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000
26 TT35	999.9998	2050.314	2041.967	2033.617	2025.309	2017.008	2008.752
27 TT4	2988.941	2501.069	2489.303	2477.562	2465.833	2454.122	2442.442
28 MFB	2.777776	2.749859	2.721943	2.694026	2.666109	2.638192	2.610275
29 M3	77.49799	190.0780	189.7281	189.3754	189.0214	188.6614	188.3045
30 M35	77.50311	231.2261	231.5576	231.8920	232.2257	232.5613	232.8942
31 M4	80.27542	59.85492	60.09993	60.34659	60.59441	60.84395	61.09436

OUTLNR

32 CV35P	.1709999	.1709999	.1709999	.1709999	.1709999	.1709999	.1709999
33 DPT35	-1.832967	-102043.5	-101718.9	-101395.4	-101073.2	-100753.0	-100431.7
34 DTT35	-2.116904	-742109.3	-733254.7	-724457.6	-715740.6	-707122.6	-698595.2
35 FAMB	.3584334E-01	.1563948E-01	.1547930E-01	.1531936E-01	.1515944E-01	.1499978E-01	.1484017E-01
36 M3	240.0000	240.0000	240.0000	240.0000	240.0000	240.0000	240.0000
37 M35	239.9999	492.0752	490.0718	488.0681	486.0740	484.0818	482.1003
38 M4	717.3457	600.2566	597.4326	594.6147	591.7998	588.9890	586.1840
39 PT35	245.8627	222.2572	222.3742	222.4929	222.6116	222.7315	222.8505
40 QORNC3	.0	.0	.0	.0	.0	.0	.0

41	RH035	.3841606E-03	.1693774E-03	.1701594E-03	.1709492E-03	.1717421E-03	.1725419E-03	.1733434E-03
42	R35	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000
43	TT3	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000
44	TT35	999.9998	2050.314	2041.967	2033.617	2025.309	2017.008	2008.752
45	TT4	2988.941	2501.069	2489.303	2477.562	2465.833	2454.122	2442.442
46	U35	171.0000	350.6035	349.1763	347.7485	346.3276	344.9082	343.4966
47	M3	77.49799	190.0780	189.7281	189.3754	189.0214	188.6614	188.3045
48	M35	77.50311	231.2261	231.5576	231.8920	232.2257	232.5613	232.8942

BURNER

49	CV4P	.1709999	.1709999	.1709999	.1709999	.1709999	.1709999	.1709999
50	DELAY	.0	.0	.0	.0	.0	.0	.0
51	DFAMB	-.6013601E-03	-3.257597	-3.221265	-3.185199	-3.149009	-3.113031	-3.076905
52	DLYP	.0	.0	.0	.0	.0	.0	.0
53	DLYPRP	.0	.0	.0	.0	.0	.0	.0
54	DHAIK	.5447360E-02	172.2928	172.3738	172.4559	172.5361	172.6165	172.6932
55	DMFUEL	.2861023E-05	1.828174	1.805819	1.783503	1.761248	1.739033	1.716882
56	DPT4	-1.778266	129299.2	128769.4	128239.2	127710.2	127181.2	126652.1
57	DTBIOL	2010.774	735.4402	727.5566	719.6875	711.8423	704.0149	696.2170
58	DTBLAG	1988.809	739.0916	733.2769	723.2769	715.3984	707.5430	699.7102
59	DTBURN	1988.809	739.0515	731.1375	723.2371	715.3608	707.5027	699.6736
60	DTT4	-71.04857	332406.1	329354.8	326295.4	323286.6	320292.6	317337.1
61	ETAB	.9890761	1.004910	1.004922	1.004932	1.004943	1.004954	1.004965
62	FAO	.3584085E-01	.1189251E-01	.1175493E-01	.1161759E-01	.1148068E-01	.1134407E-01	.1120799E-01
63	FAP	.1433634	.4757005E-01	.4701971E-01	.4647036E-01	.4592271E-01	.4537629E-01	.4483195E-01
64	FAPA	.1433634	.4757005E-01	.4701971E-01	.4647036E-01	.4592271E-01	.4537629E-01	.4483195E-01
65	M35	239.9999	492.0752	490.0718	488.0681	486.0740	484.0818	482.1003
66	M4	717.3457	600.2566	597.6147	594.6147	591.7998	588.9890	586.1860
67	M4SS	717.3140	669.4573	665.5540	661.6545	657.7695	653.8921	650.0308
68	IUPDAT	.0	.0	.0	.0	.0	.0	.0
69	MBLITE	2.000000	2.000000	2.000000	2.000000	2.000000	2.000000	2.000000
70	PCBURN	1.000000	1.000000	1.000000	1.000000	1.000000	1.000000	1.000000
71	PT4	243.8199	166.2993	166.5867	166.8755	167.1637	167.4530	167.7416
72	QBURN	38316.58	10617.19	10546.46	10475.35	10403.79	10331.93	10259.60
73	QDBNLP	.0	.0	.0	.0	.0	.0	.0
74	RHASS	.3313944	.2701208	.2718668	.2736286	.2754049	.2771983	.2790037
75	R4	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000
76	TEND	.0	.0	.0	.0	.0	.0	.0
77	TIME	.0	.004965	.2009892	.3014819	.4019746	.5024673	.6029601
78	TLITE	.0	.1004965	.2009892	.3014819	.4019746	.5024673	.6029601
79	TT35	999.9998	2050.314	2041.967	2033.617	2025.309	2017.008	2008.752
80	TT4	2988.941	2501.069	2489.303	2477.562	2465.833	2454.122	2442.442
81	U4	511.1089	427.6829	425.6707	423.6631	421.6575	419.6548	417.6575
82	MFB	2.777778	2.749859	2.721943	2.694026	2.666109	2.638192	2.610275
83	M35	77.50311	231.2261	231.5576	231.8920	232.2257	232.5613	232.8942
84	M4	80.27542	59.85492	60.09993	60.34659	60.59441	60.84395	61.09436

SENSED PARAMETERS

85	PT3H	250.0000	250.0000	250.0000	250.0000	250.0000	250.0000	250.0000
86	PT35H	245.8627	222.2572	222.3742	222.4929	222.6116	222.7315	222.8505
87	PT4H	243.8199	166.2993	166.5867	166.8755	167.1637	167.4530	167.7416
88	QBURNH	38320.61	-4961.098	-4770.422	-4797.633	-4826.730	-4855.977	-4885.977
89	TT3H	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000

90 TT35H	999.9998	2050.314	2041.967	2033.617	2025.309	2017.008	2008.752
91 TT4H	2988.941	2501.069	2489.303	2477.562	2465.833	2454.122	2442.442
92 MFRBH	2.77778	2.749859	2.721943	2.694026	2.666109	2.638192	2.610275
93 M3H	77.49799	190.0780	189.7281	189.3754	189.0214	188.6614	188.3045
94 M35H	77.50311	231.2261	231.5576	231.8920	232.2257	232.5613	232.8942
95 M4H	80.27542	59.85492	60.09993	60.34659	60.59441	60.84395	61.09436

\*\*\*\*\*

BRANDL

1 TIME	.7034528	.8039455	.9044382	1.004929	1.105373	1.205818	1.306263
2 ADIFF	70.00000	70.00000	70.00000	70.00000	70.00000	70.00000	70.00000
3 ALINR	100.0000	100.0000	100.0000	100.0000	100.0000	100.0000	100.0000
4 DFANB	-3.041101	-3.005198	-2.969367	-2.933508	-2.897903	-2.862359	-2.826691
5 DPT35	-100115.1	-99796.81	-99480.31	-99166.44	-98853.50	-98541.31	-98228.31
6 DPT4	126124.8	125594.0	125065.1	124537.4	124008.9	123481.0	122950.8
7 DTT35	-690125.5	-681710.1	-673389.5	-665191.3	-657024.6	-648928.6	-640899.5
8 DTT4	314394.5	311452.2	308556.6	305689.1	302814.9	299983.7	297171.9
9 FANB	.1468078E-01	.1452154E-01	.1436243E-01	.1420347E-01	.1404473E-01	.1388612E-01	.1372762E-01
10 FA3	.0	.0	.0	.0	.0	.0	.0
11 FA35	.0	.0	.0	.0	.0	.0	.0
12 FPHPT	18.00000	18.00000	18.00000	18.00000	18.00000	18.00000	18.00000
13 GAN3	1.400000	1.400000	1.400000	1.400000	1.400000	1.400000	1.400000
14 GAN35	1.400000	1.400000	1.400000	1.400000	1.400000	1.400000	1.400000
15 GAN4	1.400000	1.400000	1.400000	1.400000	1.400000	1.400000	1.400000
16 IUPDAT	.0	.0	.0	.0	.0	.0	.0
17 PRODIFF	1.121219	1.120609	1.119995	1.119386	1.118767	1.118148	1.117526
18 PRLINR	1.326965	1.325389	1.323829	1.322270	1.320724	1.319179	1.317632
19 PT3	250.0000	250.0000	250.0000	250.0000	250.0000	250.0000	250.0000
20 PT35	222.9716	223.0928	223.2151	223.3367	223.4602	223.5840	223.7083
21 PT4	168.0312	168.3225	168.6132	168.9039	169.1951	169.4873	169.7805
22 R3	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000
23 R35	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000
24 R4	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000
25 TT3	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000
26 TT35	2000.502	1992.256	1984.048	1975.909	1967.748	1959.598	1951.467
27 TT4	2430.777	2419.127	2407.495	2395.946	2384.392	2372.816	2361.258
28 MFRB	2.582358	2.554441	2.526525	2.498608	2.470704	2.442801	2.414897
29 M3	187.9398	187.5708	187.2012	186.8292	186.4519	186.0729	185.6901
30 M35	233.2319	233.5672	233.9033	234.2362	234.5735	234.9124	235.2486
31 M4	61.34650	61.60065	61.85593	62.11171	62.36930	62.62926	62.89102

OUTLNR

32 CV35P	.1709999	.1709999	.1709999	.1709999	.1709999	.1709999	.1709999
33 OPT35	-100115.1	-99796.81	-99480.31	-99166.44	-98853.50	-98541.31	-98228.31
34 DTT35	-690125.5	-681710.1	-673389.5	-665191.3	-657024.6	-648928.6	-640899.5
35 FANB	.1468078E-01	.1452154E-01	.1436243E-01	.1420347E-01	.1404473E-01	.1388612E-01	.1372762E-01
36 H3	240.0000	240.0000	240.0000	240.0000	240.0000	240.0000	240.0000
37 H35	480.1204	478.1414	476.1714	474.2183	472.2593	470.3032	468.3518
38 H4	583.3862	580.5906	577.7988	575.0271	572.2539	569.4761	566.7017
39 PT35	222.9716	223.0928	223.2151	223.3367	223.4602	223.5840	223.7083
40 QBANCS	.0	.0	.0	.0	.0	.0	.0
41 RH035	.1741529E-03	.1749689E-03	.1757890E-03	.1766091E-03	.1774397E-03	.1782765E-03	.1791186E-03



91 TT4M	2430.777	2419.127	2407.495	2395.946	2364.392	2372.818	2361.258
92 NFMFM	2.582358	2.554441	2.526525	2.498608	2.470704	2.442801	2.414897
93 M3M	187.9398	187.5708	187.2012	186.8292	186.4519	186.0729	185.6901
94 M35M	233.2319	233.5672	233.9033	234.2362	234.5735	234.9124	235.2486
95 M4M	61.34650	61.60065	61.85593	62.11171	62.36938	62.62926	62.89102

\*\*\*\*\*

BROWD

1 TIME	1.406708	1.507153	1.607597	1.708042	1.808487	1.908932	2.009377
2 ADIFF	70.00000	70.00000	70.00000	70.00000	70.00000	70.00000	70.00000
3 ALINR	100.0000	100.0000	100.0000	100.0000	100.0000	100.0000	100.0000
4 DFAMB	-2.791181	-2.755801	-2.720138	-2.684515	-2.649103	-2.613697	-2.578246
5 DPT35	-97916.56	-97610.94	-97277.12	-96946.69	-96615.12	-96285.31	-95956.25
6 DPT4	122420.3	121893.9	121317.9	120742.4	120167.1	119589.2	119011.2
7 DTT35	-632929.4	-625094.1	-616658.7	-608296.4	-600005.9	-591767.3	-583623.6
8 DTT4	294362.1	291599.9	288577.1	285600.8	282635.7	279681.3	276768.7
9 FAMB	.1356928E-01	.1341109E-01	.1325310E-01	.1309523E-01	.1293755E-01	.1278001E-01	.1262254E-01
10 FA3	.0	.0	.0	.0	.0	.0	.0
11 FA35	.0	.0	.0	.0	.0	.0	.0
12 FPPT	18.00000	18.00000	18.00000	18.00000	18.00000	18.00000	18.00000
13 GAMB	1.400000	1.400000	1.400000	1.400000	1.400000	1.400000	1.400000
14 GAMB5	1.400000	1.400000	1.400000	1.400000	1.400000	1.400000	1.400000
15 GAMB4	1.400000	1.400000	1.400000	1.400000	1.400000	1.400000	1.400000
16 IUPDAT	.0	.0	.0	.0	.0	.0	.0
17 PROIFF	1.116899	1.116272	1.115591	1.114905	1.114211	1.113511	1.112814
18 PRLINR	1.316088	1.314567	1.312903	1.311245	1.309597	1.307942	1.306292
19 PT3	250.0000	250.0000	250.0000	250.0000	250.0000	250.0000	250.0000
20 PT35	223.8339	223.9596	224.0963	224.2343	224.3740	224.5149	224.6557
21 PT4	170.0751	170.3676	170.6875	171.0086	171.3305	171.6550	171.9796
22 R3	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000
23 R35	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000
24 R4	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000
25 TT3	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000
26 TT35	1943.338	1935.299	1926.584	1917.880	1909.186	1900.487	1891.820
27 TT4	2349.715	2338.264	2325.841	2313.391	2300.964	2288.529	2276.104
28 NFMFB	2.386993	2.359090	2.331186	2.303283	2.275379	2.247476	2.219572
29 M3	185.3037	184.9130	184.4867	184.0565	183.6198	183.1765	182.7306
30 M35	235.5872	235.9244	236.2881	236.6538	237.0221	237.3910	237.7588
31 M4	63.15469	63.41800	63.70654	63.99789	64.29129	64.58781	64.88637

OUTLNR

32 CV35P	.1709999	.1709999	.1709999	.1709999	.1709999	.1709999	.1709999
33 DPT35	-97916.56	-97610.94	-97277.12	-96946.69	-96615.12	-96285.31	-95956.25
34 DTT35	-632929.4	-625094.1	-616658.7	-608296.4	-600005.9	-591767.3	-583623.6
35 FAMB	.1356928E-01	.1341109E-01	.1325310E-01	.1309523E-01	.1293755E-01	.1278001E-01	.1262254E-01
36 H3	240.0000	240.0000	240.0000	240.0000	240.0000	240.0000	240.0000
37 H35	466.4011	464.4717	462.3801	460.2910	458.2046	456.1167	454.0366
38 H4	563.9316	561.1833	558.2019	555.2139	552.2314	549.2468	546.2649
39 PT35	223.8339	223.9596	224.0963	224.2343	224.3740	224.5149	224.6557
40 QMBNC5	.0	.0	.0	.0	.0	.0	.0
41 RH035	.1799690E-03	.1808181E-03	.1817469E-03	.1826842E-03	.1836303E-03	.1845868E-03	.1855487E-03
42 R35	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000

43 TT3	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000
44 TT5	1943.338	1935.299	1926.584	1917.880	1909.186	1900.487	1891.820	1883.154	1874.488
45 TT4	2349.715	2338.264	2325.841	2313.391	2300.964	2288.529	2276.104	2263.679	2251.254
46 U35	332.3108	330.9360	329.4458	327.9573	326.4707	324.9832	323.5010	322.0188	320.5366
47 M3	185.3037	184.9130	184.4867	184.0565	183.6198	183.1765	182.7306	182.2867	181.8428
48 M35	235.5872	235.9244	236.2881	236.6538	237.0221	237.3910	237.7588	238.1266	238.4944
BURNER									
49 CV4P	.1709999	.1709999	.1709999	.1709999	.1709999	.1709999	.1709999	.1709999	.1709999
50 DELAY	.0	.0	.0	.0	.0	.0	.0	.0	.0
51 DFAPB	-2.791181	-2.755801	-2.720138	-2.684515	-2.649103	-2.613697	-2.578246	-2.542795	-2.507344
52 DLYP	.0	.0	.0	.0	.0	.0	.0	.0	.0
53 DLYPRP	.0	.0	.0	.0	.0	.0	.0	.0	.0
54 DPAIR	173.2780	173.3456	173.4148	173.4832	173.5520	173.6181	173.6812	173.7443	173.8074
55 DMFUEL	1.541502	1.519840	1.497919	1.476048	1.454231	1.432458	1.410749	1.389040	1.367291
56 DPT4	122420.3	121893.9	121317.9	120742.4	120167.1	119589.2	119011.2	118433.2	117855.2
57 DTBDL	634.5686	626.9583	618.5891	610.2402	601.9106	593.6055	585.3289	577.0748	568.8462
58 DTBLAG	637.8120	630.1704	621.7585	613.3662	604.9973	596.6504	588.3291	580.0280	571.7474
59 DTBURN	637.7717	630.1252	621.7139	613.3228	604.9507	596.6035	588.2849	580.0000	571.7474
60 DTT4	294362.1	291599.9	288577.1	285600.8	282651.3	279681.7	276687.7	273668.7	270625.7
61 ETAB	1.005048	1.005052	1.005052	1.005052	1.005051	1.005051	1.005051	1.005051	1.005051
62 FAO	.1013210E-01	.9999342E-02	.9865861E-02	.9732708E-02	.9599857E-02	.9467401E-02	.9335395E-02	.9203389E-02	.9071383E-02
63 FAP	.4052839E-01	.3999737E-01	.3946345E-01	.3893083E-01	.3839943E-01	.3786960E-01	.3734158E-01	.3681356E-01	.3628554E-01
64 FAPA	.4052839E-01	.3999737E-01	.3946345E-01	.3893083E-01	.3839943E-01	.3786960E-01	.3734158E-01	.3681356E-01	.3628554E-01
65 H35	466.4011	464.4717	462.3801	460.2910	458.2046	456.1167	454.0366	451.9566	449.8766
66 H4	563.9216	561.1833	558.2019	555.2139	552.2314	549.2468	546.2649	543.2829	540.3009
67 H4SS	619.4761	615.7124	611.6021	607.4990	603.4038	599.3127	595.2356	591.1625	587.0944
68 IUPDAT	.0	.0	.0	.0	.0	.0	.0	.0	.0
69 MBLITE	2.000000	2.000000	2.000000	2.000000	2.000000	2.000000	2.000000	2.000000	2.000000
70 PCBURN	1.000000	1.000000	1.000000	1.000000	1.000000	1.000000	1.000000	1.000000	1.000000
71 PT4	170.0751	170.3676	170.6875	171.0086	171.3305	171.6550	171.9796	172.3041	172.6286
72 QBURN	9667.398	9591.383	9506.410	9421.000	9335.043	9248.719	9161.887	9075.474	8988.561
73 QDBNLP	.0	.0	.0	.0	.0	.0	.0	.0	.0
74 RMASS	.2940485	.2959965	.2981364	.3003048	.3024949	.3047147	.3069575	.3091999	.3114423
75 R4	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000
76 TEND	.0	.0	.0	.0	.0	.0	.0	.0	.0
77 TIME	1.406708	1.507153	1.607597	1.708042	1.808487	1.908932	2.009377	2.109822	2.209377
78 TLITE	1.406708	1.507153	1.607597	1.708042	1.808487	1.908932	2.009377	2.109822	2.209377
79 TT5	1943.338	1935.299	1926.584	1917.880	1909.186	1900.487	1891.820	1883.154	1874.488
80 TT4	2349.715	2338.264	2325.841	2313.391	2300.964	2288.529	2276.104	2263.679	2251.254
81 U4	401.8013	399.8430	397.7187	395.5898	393.4648	391.3384	389.2136	387.0886	384.9636
82 MFVB	2.386993	2.359090	2.331186	2.303283	2.275379	2.247476	2.219572	2.191668	2.163764
83 M35	235.5872	235.9244	236.2881	236.6538	237.0221	237.3910	237.7588	238.1266	238.4944
84 M4	63.15469	63.41800	63.70654	63.99789	64.29129	64.58781	64.88637	65.18493	65.48349
SENSED PARAMETERS									
85 PT3M	250.0000	250.0000	250.0000	250.0000	250.0000	250.0000	250.0000	250.0000	250.0000
86 PT5M	223.8339	223.9596	224.0963	224.2343	224.3740	224.5149	224.6557	224.7966	224.9374
87 PT4M	170.0751	170.3676	170.6875	171.0086	171.3305	171.6550	171.9796	172.3041	172.6286
88 QBURNH	-5139.582	-5139.582	-5139.582	-5139.582	-5139.582	-5139.582	-5139.582	-5139.582	-5139.582
89 TT3M	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000
90 TT5M	1943.338	1935.299	1926.584	1917.880	1909.186	1900.487	1891.820	1883.154	1874.488
91 TT4M	2349.715	2338.264	2325.841	2313.391	2300.964	2288.529	2276.104	2263.679	2251.254

92 NFMHM	2.306993	2.359090	2.331186	2.303283	2.275379	2.247476	2.219572
93 N3M	185.3037	184.9130	184.4867	184.0565	183.6198	183.1765	182.7306
94 N3SM	235.5872	235.9244	236.2881	236.6538	237.0221	237.3910	237.7588
95 N4M	63.15469	63.41800	63.70654	63.99789	64.29129	64.58781	64.88637

\*\*\*\*\*

BRAND

1 TIME	2.109821	2.210266	2.310711	2.411156	2.511600	2.612045	2.712490
2 ADIFF	70.00000	70.00000	70.00000	70.00000	70.00000	70.00000	70.00000
3 ALINR	100.00000	100.00000	100.00000	100.00000	100.00000	100.00000	100.00000
4 DFAMB	-2.543099	-2.507834	-2.472662	-2.437656	-2.402614	-2.367666	-2.332793
5 DPT35	-95632.31	-95306.00	-94980.81	-94656.62	-94336.37	-94015.62	-93695.12
6 DPT4	118434.1	117855.0	117274.2	116693.1	116112.0	115528.9	114944.2
7 DTT35	-575535.1	-567530.1	-559594.0	-551696.9	-543905.5	-536167.2	-528489.1
8 DTT4	273865.4	270995.8	268133.7	265287.2	262485.1	259690.9	256908.4
9 FAMB	.1246531E-01	.1230815E-01	.1215115E-01	.1199432E-01	.1183755E-01	.1168095E-01	.1152448E-01
10 FA3	.0	.0	.0	.0	.0	.0	.0
11 FA35	.0	.0	.0	.0	.0	.0	.0
12 FPHPT	18.00000	18.00000	18.00000	18.00000	18.00000	18.00000	18.00000
13 GAMS	1.400000	1.400000	1.400000	1.400000	1.400000	1.400000	1.400000
14 GAMS35	1.400000	1.400000	1.400000	1.400000	1.400000	1.400000	1.400000
15 GAMA	1.400000	1.400000	1.400000	1.400000	1.400000	1.400000	1.400000
16 IUPDAT	.0	.0	.0	.0	.0	.0	.0
17 PRDIFF	1.112102	1.111391	1.110676	1.109949	1.109225	1.108493	1.107755
18 PRLINR	1.304652	1.303013	1.301366	1.299729	1.298093	1.296460	1.294820
19 PT3	250.0000	250.0000	250.0000	250.0000	250.0000	250.0000	250.0000
20 PT35	224.7995	224.9433	225.0881	225.2355	225.3824	225.5314	225.6817
21 PT4	172.3060	172.6331	172.9629	173.2941	173.6257	173.9593	174.2957
22 R3	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000
23 R35	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000
24 R4	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000
25 TT3	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000
26 TT4	1883.152	1874.506	1865.858	1857.212	1848.597	1839.981	1831.363
27 TT4	2263.688	2251.273	2238.859	2226.449	2214.040	2201.636	2189.225
28 NFMHM	2.191669	2.163765	2.135861	2.107958	2.080054	2.052151	2.024247
29 M3	182.2735	181.8166	181.3526	180.8794	180.4062	179.9234	179.4360
30 M35	238.1315	238.5017	238.8735	239.2478	239.6215	239.9960	240.3721
31 M4	65.18753	65.49113	65.79790	66.10735	66.41920	66.73405	67.05235

OUTLNR

32 CV35P	.1709999	.1709999	.1709998	.1709999	.1709999	.1709999	.1709999
33 DPT35	-95632.31	-95306.00	-94980.81	-94656.62	-94336.37	-94015.62	-93695.12
34 DTT35	-575535.1	-567530.1	-559594.0	-551696.9	-543905.5	-536167.2	-528489.1
35 FAMB	.1246531E-01	.1230815E-01	.1215115E-01	.1199432E-01	.1183755E-01	.1168095E-01	.1152448E-01
36 M3	240.0000	240.0000	240.0000	240.0000	240.0000	240.0000	240.0000
37 M35	451.9563	449.8813	447.8057	445.7307	443.6633	441.5955	439.5271
38 M4	543.2849	540.3054	537.3262	534.3477	531.3696	528.3926	525.4138
39 PT35	224.7995	224.9433	225.0881	225.2355	225.3824	225.5314	225.6817
40 ORGNCS	.0	.0	.0	.0	.0	.0	.0
41 RHO35	.1845220E-03	.1875023E-03	.1884926E-03	.1894941E-03	.1905012E-03	.1915198E-03	.1925493E-03
42 R35	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000
43 TT3	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000

44 TT35	1093.152	1674.506	1665.858	1657.212	1648.597	1639.981	1631.363
45 TT4	2263.688	2251.273	2236.859	2226.449	2214.040	2201.636	2189.225
46 US5	322.0188	320.5405	319.0615	317.5830	316.1101	314.6367	313.1631
47 M3	182.2735	181.8166	181.3526	180.8794	180.4062	179.9234	179.4360
48 M35	238.1315	238.5017	238.8735	239.2478	239.6215	239.9960	240.3721
BURNER							
49 CV4P	.1709999	.1709999	.1709999	.1709999	.1709999	.1709999	.1709999
50 DELAY	.0	.0	.0	.0	.0	.0	.0
51 DF4MB	-2.543099	-2.507834	-2.472662	-2.437656	-2.402614	-2.367666	-2.332793
52 DLYP	.0	.0	.0	.0	.0	.0	.0
53 DLYPAP	.0	.0	.0	.0	.0	.0	.0
54 DM4IR	173.7465	173.8068	173.8655	173.9239	173.9793	174.0325	174.0837
55 DMFLU	1.389090	1.367491	1.345940	1.324442	1.303011	1.281633	1.260307
56 DPT4	118434.1	117855.0	117274.2	116693.1	116112.0	115528.9	114944.2
57 DTBIDL	577.0657	568.8345	560.6248	552.4348	544.2720	536.1326	528.0151
58 DTBLAG	580.0300	571.7556	563.5039	555.2759	547.0708	538.8909	530.7317
59 DTBURN	579.9807	571.7078	563.4568	555.2258	547.0225	538.8423	530.6843
60 DTT4	273865.4	270995.8	268133.7	265287.2	262485.1	259690.9	256908.4
61 ETAB	1.005052	1.005052	1.005052	1.005053	1.005054	1.005054	1.005055
62 FAO	.9203605E-02	.9072326E-02	.8941390E-02	.8810770E-02	.8680582E-02	.8550767E-02	.8421302E-02
63 FAP	.3681442E-01	.3628930E-01	.3576556E-01	.3524308E-01	.3472233E-01	.3420307E-01	.3368521E-01
64 FAPA	.3681442E-01	.3628930E-01	.3576556E-01	.3524308E-01	.3472233E-01	.3420307E-01	.3368521E-01
65 M35	451.9563	449.8813	447.8057	445.7307	443.6633	441.5955	439.5271
66 M4	543.2849	540.3054	537.3262	534.3477	531.3696	528.3926	525.4138
67 M4SS	591.1636	587.1028	583.0466	578.9971	574.9602	570.9292	566.9026
68 IUPDAT	.0	.0	.0	.0	.0	.0	.0
69 PBLITE	2.000000	2.000000	2.000000	2.000000	2.000000	2.000000	2.000000
70 PCBURN	1.000000	1.000000	1.000000	1.000000	1.000000	1.000000	1.000000
71 PT4	172.3060	172.6331	172.9629	173.2941	173.6257	173.9593	174.2957
72 QBURN	9074.578	8986.785	8898.570	8809.883	8720.633	8630.965	8540.824
73 QBURNLP	.0	.0	.0	.0	.0	.0	.0
74 RM4SS	.3092268	.3115225	.3138483	.3162018	.3185825	.3209932	.3234373
75 R4	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000
76 TEND	.0	.0	.0	.0	.0	.0	.0
77 TIME	2.109821	2.210266	2.310711	2.411156	2.511600	2.612045	2.712490
78 TLITE	2.109821	2.210266	2.310711	2.411156	2.511600	2.612045	2.712490
79 TT35	1893.152	1874.506	1865.858	1857.212	1848.597	1839.981	1831.363
80 TT4	2263.688	2251.273	2236.859	2226.449	2214.040	2201.636	2189.225
81 U4	387.0906	384.9675	382.8447	380.7227	378.6008	376.4795	374.3574
82 MFMB	2.191669	2.163765	2.135861	2.107958	2.080054	2.052151	2.024247
83 M35	238.1315	238.5017	238.8735	239.2478	239.6215	239.9960	240.3721
84 M4	65.18753	65.49113	65.79790	66.10735	66.41920	66.73405	67.05235
SENSED PARAMETERS							
85 PT3M	250.0000	250.0000	250.0000	250.0000	250.0000	250.0000	250.0000
86 PT35M	224.7995	224.9433	225.0881	225.2355	225.3824	225.5314	225.6817
87 PT4M	172.3060	172.6331	172.9629	173.2941	173.6257	173.9593	174.2957
88 QBURNM	-5413.348	-5455.836	-5498.266	-5541.004	-5585.652	-5630.199	-5675.039
89 TT3M	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000
90 TT35M	1893.152	1874.506	1865.858	1857.212	1848.597	1839.981	1831.363
91 TT4M	2263.688	2251.273	2238.859	2226.449	2214.040	2201.636	2189.225
92 MFMBM	2.191669	2.163765	2.135861	2.107958	2.080054	2.052151	2.024247

93 KSM	182.2735	181.8166	181.3526	180.8794	180.4062	179.9234	179.4360
94 KSM	238.1315	238.5017	238.8735	239.2478	239.6215	239.9960	240.3721
95 MM	65.18753	65.49113	65.79790	66.10735	66.41920	66.73405	67.05235

\*\*\*\*\*

BRWDL

1 TIME	2.812935	2.913380	3.013824	3.114269	3.214714	3.315159	3.415604
2 ADIFF	70.00000	70.00000	70.00000	70.00000	70.00000	70.00000	70.00000
3 ALINR	100.0000	100.0000	100.0000	100.0000	100.0000	100.0000	100.0000
4 DFAMB	-2.298051	-2.263397	-2.228763	-2.194085	-2.159644	-2.125183	-2.090957
5 OPT35	-93379.25	-93063.94	-92749.19	-92433.81	-92122.44	-91810.00	-91503.25
6 DPT4	114358.7	113772.9	113186.1	112596.6	112006.4	111412.9	110819.9
7 DTT35	-520858.0	-513310.1	-505831.9	-498421.4	-491054.4	-483747.4	-476489.1
8 DTT4	254148.8	251406.4	248697.2	246008.6	243324.2	240652.7	238002.4
9 FAMB	.1136815E-01	.1121194E-01	.1105582E-01	.1089980E-01	.1074396E-01	.1058820E-01	.1043263E-01
10 FA3	.0	.0	.0	.0	.0	.0	.0
11 FA35	.0	.0	.0	.0	.0	.0	.0
12 FPHPT	18.00000	18.00000	18.00000	18.00000	18.00000	18.00000	18.00000
13 GAMB	1.400000	1.400000	1.400000	1.400000	1.400000	1.400000	1.400000
14 GAMB35	1.400000	1.400000	1.400000	1.400000	1.400000	1.400000	1.400000
15 GAMB4	1.400000	1.400000	1.400000	1.400000	1.400000	1.400000	1.400000
16 IUPDAT	.0	.0	.0	.0	.0	.0	.0
17 PRODIFF	1.107004	1.106249	1.105493	1.104733	1.103958	1.103182	1.102387
18 PRLINR	1.293181	1.291553	1.289921	1.288285	1.286651	1.285007	1.283371
19 PT3	250.0000	250.0000	250.0000	250.0000	250.0000	250.0000	250.0000
20 PT35	225.8346	225.9888	226.1434	226.2987	226.4577	226.6171	226.7803
21 PT4	174.6348	174.9744	175.3157	175.6588	176.0054	176.3547	176.7067
22 R3	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000
23 R35	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000
24 R4	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000
25 TT3	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000
26 TT35	1822.737	1814.133	1805.539	1796.954	1788.352	1779.749	1771.137
27 TT4	2176.790	2164.382	2151.956	2139.528	2127.086	2114.637	2102.165
28 MFMB	1.996344	1.968440	1.940536	1.912633	1.884729	1.856826	1.828922
29 M3	178.9333	178.4287	177.9186	177.4030	176.8753	176.3419	175.7931
30 M35	240.7516	241.1315	241.5118	241.8901	242.2738	242.6566	243.0447
31 M4	67.37440	67.69867	68.02626	68.35704	68.69197	69.03059	69.37326

OUTLNR

32 CV35P	.1709999	.1709999	.1709999	.1709999	.1709999	.1709999	.1709999
33 OPT35	-93379.25	-93063.94	-92749.19	-92433.81	-92122.44	-91810.00	-91503.25
34 DTT35	-520858.0	-513310.1	-505831.9	-498421.4	-491054.4	-483747.4	-476489.1
35 FAMB	.1136815E-01	.1121194E-01	.1105582E-01	.1089980E-01	.1074396E-01	.1058820E-01	.1043263E-01
36 M3	240.0000	240.0000	240.0000	240.0000	240.0000	240.0000	240.0000
37 M35	437.4568	435.3918	433.3291	431.2688	429.2043	427.1396	425.0728
38 M4	522.4294	519.4514	516.4692	513.4866	510.5007	507.5127	504.5193
39 PT35	225.8346	225.9888	226.1434	226.2987	226.4577	226.6171	226.7803
40 QOBANC3	.0	.0	.0	.0	.0	.0	.0
41 RHO35	.1935917E-03	.1946426E-03	.1957029E-03	.1967729E-03	.1978582E-03	.1989546E-03	.2000661E-03
42 R35	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000
43 TT3	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000
44 TT35	1822.737	1814.133	1805.539	1796.954	1788.352	1779.749	1771.137

45 TT4	2176.790	2164.382	2151.956	2139.528	2127.086	2114.637	2102.165
46 U35	311.6890	310.2166	308.7471	307.2791	305.8081	304.3369	302.8643
47 M3	178.9333	178.4287	177.9186	177.4030	176.8753	176.3419	175.7931
48 M35	240.7516	241.1315	241.5118	241.8901	242.2738	242.6566	243.0447
BURNER							
49 CV4P	.1709999	.1709999	.1709999	.1709999	.1709999	.1709999	.1709999
50 DELAY	.0	.0	.0	.0	.0	.0	.0
51 DFAPB	-2.298051	-2.263397	-2.228763	-2.194085	-2.159644	-2.125183	-2.090957
52 OLYP	.0	.0	.0	.0	.0	.0	.0
53 OLYPRP	.0	.0	.0	.0	.0	.0	.0
54 DMAIR	174.1345	174.1834	174.2294	174.2701	174.3120	174.3492	174.3876
55 DMFUEL	1.239030	1.217822	1.196673	1.175588	1.154550	1.133574	1.112649
56 DPT4	114358.7	113772.9	113186.1	112596.6	112006.4	111412.9	110819.9
57 DTBIDL	519.9160	511.8416	503.7910	495.7703	487.7637	479.7644	471.8198
58 DTBLAG	522.5952	514.4819	506.3928	498.3286	490.2847	482.2634	474.2642
59 DTBURN	522.5447	514.4299	506.3396	498.2788	490.2327	482.2100	474.2100
60 DTT4	254148.8	251406.4	248697.2	246008.6	243324.2	240652.7	238002.4
61 ETAB	1.005056	1.005057	1.005059	1.005060	1.005062	1.005064	1.005066
62 FAO	.8292127E-02	.8163348E-02	.8034952E-02	.7907029E-02	.7779334E-02	.7652070E-02	.7525045E-02
63 FAP	.3316851E-01	.3265339E-01	.3213981E-01	.3162812E-01	.3111733E-01	.3060828E-01	.3010018E-01
64 FAPA	.3316851E-01	.3265339E-01	.3213981E-01	.3162812E-01	.3111733E-01	.3060828E-01	.3010018E-01
65 M35	437.4568	435.3918	433.3291	431.2688	429.2043	427.1396	425.0728
66 M4	522.4294	519.4514	516.4692	513.4866	510.5007	507.5127	504.5193
67 M4SS	562.8796	558.8674	554.8635	550.8677	546.8726	542.8828	538.8962
68 IUPDAT	.0	.0	.0	.0	.0	.0	.0
69 MBLITE	2.000000	2.000000	2.000000	2.000000	2.000000	2.000000	2.000000
70 PCBURN	1.000000	1.000000	1.000000	1.000000	1.000000	1.000000	1.000000
71 PT4	174.6348	174.9744	175.3157	175.6588	176.0054	176.3547	176.7067
72 QBURN	8450.289	8359.133	8267.531	8175.422	8082.859	7989.816	7896.305
73 QBHNP	.0	.0	.0	.0	.0	.0	.0
74 RMSS	.3259175	.3284234	.3309641	.3335381	.3361509	.3388011	.3414913
75 R4	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000
76 TEND	.0	.0	.0	.0	.0	.0	.0
77 TIME	2.812935	2.913380	3.013824	3.114269	3.214714	3.315159	3.415604
78 TLITE	2.812935	2.913380	3.013824	3.114269	3.214714	3.315159	3.415604
79 TT35	1822.737	1814.133	1805.539	1796.954	1788.352	1779.749	1771.137
80 TT4	2176.790	2164.382	2151.956	2139.528	2127.086	2114.637	2102.165
81 U4	372.2310	370.1091	367.9844	365.8591	363.7317	361.6028	359.4700
82 MFB	1.996344	1.968440	1.940536	1.912633	1.884729	1.856826	1.828922
83 M35	240.7516	241.1315	241.5118	241.8901	242.2738	242.6566	243.0447
84 M4	67.37440	67.69867	68.02626	68.35704	68.69197	69.03059	69.37326
SENSED PARAMETERS							
85 PT3H	250.0000	250.0000	250.0000	250.0000	250.0000	250.0000	250.0000
86 PT35H	225.8346	225.9888	226.1434	226.2987	226.4577	226.6171	226.7803
87 PT4H	174.6348	174.9744	175.3157	175.6588	176.0054	176.3547	176.7067
88 QBURNH	-5720.801	-5766.879	-5814.453	-5862.734	-5910.969	-5959.570	-6009.094
89 TT3H	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000
90 TT35H	1822.737	1814.133	1805.539	1796.954	1788.352	1779.749	1771.137
91 TT4H	2176.790	2164.382	2151.956	2139.528	2127.086	2114.637	2102.165
92 MFBH	1.996344	1.968440	1.940536	1.912633	1.884729	1.856826	1.828922
93 M3H	178.9333	178.4287	177.9186	177.4030	176.8753	176.3419	175.7931

94 N35H 240.7516 241.1315 241.5116 241.8901 242.2738 242.6566 243.0447  
 95 N4M 67.37440 67.69867 68.02626 68.35704 68.69197 69.03059 69.37326

\*\*\*\*\*

BRAND1

1 TIME	3.616493	3.716936	3.817383	3.917828	4.018272	4.118717
2 ADIFF	70.00000	70.00000	70.00000	70.00000	70.00000	70.00000
3 ALINR	100.0000	100.0000	100.0000	100.0000	100.0000	100.0000
4 DFAMB	-2.022459	-1.988378	-1.954441	-1.920522	-1.886525	-1.852839
5 DPT35	-90888.31	-90883.62	-90282.31	-89982.19	-89678.75	-89383.87
6 DPT4	10926.9	10925.5	108425.7	107821.7	107214.6	106606.6
7 DTT35	-462179.1	-455080.0	-448064.4	-441090.7	-434186.1	-427312.1
8 DTT4	232761.9	230142.4	227561.8	224983.9	222445.2	219894.6
9 FAMB	.1012167E-01	.9964418E-02	.9811245E-02	.9656172E-02	.9501159E-02	.9346325E-02
10 FA3	.0	.0	.0	.0	.0	.0
11 FA35	.0	.0	.0	.0	.0	.0
12 FPHPT	18.00000	18.00000	18.00000	18.00000	18.00000	18.00000
13 GAM3	1.400000	1.400000	1.400000	1.400000	1.400000	1.400000
14 GAM35	1.400000	1.400000	1.400000	1.400000	1.400000	1.400000
15 GAM4	1.400000	1.400000	1.400000	1.400000	1.400000	1.400000
16 IUPDAT	.0	.0	.0	.0	.0	.0
17 PRDIFF	1.101593	1.100792	1.099975	1.099150	1.097485	1.096624
18 PRLNR	1.281730	1.278432	1.274789	1.271333	1.273468	1.271809
19 PT3	250.0000	250.0000	250.0000	250.0000	250.0000	250.0000
20 PT35	226.9440	227.1092	227.2779	227.4484	227.7936	227.9722
21 PT4	177.0606	177.4168	177.7786	178.1409	178.5078	179.2503
22 R3	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000
23 R35	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000
24 R4	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000
25 TT3	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000
26 TT4	1762.537	1753.940	1745.305	1736.062	1719.437	1710.782
27 TT4	2089.697	2077.208	2064.694	2052.177	2039.639	2014.475
28 MFTB	1.601019	1.773116	1.745213	1.717309	1.689405	1.633598
29 M3	175.2413	174.6812	174.1048	173.5232	172.9271	171.7084
30 M35	243.4305	243.8180	244.2081	244.6015	244.9947	245.3835
31 M4	69.71924	70.06920	70.42456	70.78299	71.14639	71.88716

OUTLNR

32 CV35P	.1709999	.1709999	.1709999	.1709998	.1709999	.1709999
33 DPT35	-91194.75	-90888.31	-90282.31	-89982.19	-89678.75	-89383.87
34 DTT35	-469302.0	-462179.1	-455080.0	-448064.4	-441090.7	-427312.1
35 FAMB	.1027709E-01	.1012167E-01	.9964418E-02	.9811245E-02	.9656172E-02	.9501159E-02
36 M3	240.0000	240.0000	240.0000	240.0000	240.0000	240.0000
37 M35	423.0088	420.9453	418.8730	416.7346	412.6648	410.5876
38 M4	501.5273	498.5298	495.5266	489.5132	486.4944	483.4739
39 PT35	226.9440	227.1092	227.2779	227.4484	227.7936	227.9722
40 QBANC5	.0	.0	.0	.0	.0	.0
41 RHO35	.2011874E-03	.2023207E-03	.2034726E-03	.2058135E-03	.2070025E-03	.2082128E-03
42 R35	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000
43 TT3	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000
44 TT35	1762.537	1753.940	1745.305	1736.062	1719.437	1710.782
45 TT4	2089.697	2077.208	2064.694	2052.177	2039.639	2014.475

46 USS	301.3936	299.9236	296.4470	296.9741	295.6983	294.0237	292.5437
47 MS	175.2413	174.6812	174.1046	173.5232	172.9271	172.3298	171.7084
48 MS	243.4305	243.0180	244.2081	244.6015	244.9947	245.3851	245.7835
BURNER							
49 CV4P	.1709999	.1709999	.1709999	.1709999	.1709999	.1709999	.1709999
50 DELAY	.0	.0	.0	.0	.0	.0	.0
51 DFAPB	-2.056669	-2.022459	-1.988378	-1.954441	-1.920522	-1.886525	-1.852839
52 DLYP	.0	.0	.0	.0	.0	.0	.0
53 DLYPRP	.0	.0	.0	.0	.0	.0	.0
54 DHAIR	174.4205	174.4509	174.4785	174.5062	174.5287	174.5439	174.5620
55 DMFLKEL	1.091796	1.071005	1.050258	1.029586	1.008973	.9884285	.9679389
56 DPT4	110223.9	109626.9	109025.5	108425.7	107821.7	107214.6	106606.6
57 DTBIDL	463.8848	455.9722	448.0798	440.2063	432.3586	424.5413	416.7346
58 DTBLAG	466.2888	458.3369	450.4060	442.4978	434.6118	426.7517	418.9114
59 DTBURN	466.2361	458.2842	450.3530	442.4409	434.5547	426.6987	418.8535
60 DTT4	235367.7	232761.9	230142.4	227561.8	224983.9	222445.2	219894.6
61 ETAB	1.005069	1.005071	1.005074	1.005076	1.005078	1.005081	1.005085
62 FAO	.7398490E-02	.7272292E-02	.7146414E-02	.7020842E-02	.6895680E-02	.6770998E-02	.6646492E-02
63 FAP	.2959396E-01	.2908917E-01	.2858566E-01	.2808337E-01	.2758272E-01	.2708399E-01	.2658597E-01
64 FAPA	.2959396E-01	.2908917E-01	.2858566E-01	.2808337E-01	.2758272E-01	.2708399E-01	.2658597E-01
65 M35	423.0088	420.9453	418.8730	416.8059	414.7346	412.6648	410.5876
66 M4	501.5273	498.5298	495.5266	492.5222	489.5132	486.4944	483.4739
67 M4SS	534.9182	530.9463	526.9705	523.0054	519.0415	515.0852	511.1262
68 IUPBAT	.0	.0	.0	.0	.0	.0	.0
69 PBLITE	2.000000	2.000000	2.000000	2.000000	2.000000	2.000000	2.000000
70 PCBURN	1.000000	1.000000	1.000000	1.000000	1.000000	1.000000	1.000000
71 PT4	177.0606	177.4168	177.7786	178.1409	178.5078	178.8765	179.2503
72 GBURN	7802.238	7707.680	7612.711	7517.113	7421.059	7324.516	7227.430
73 GBURNLP	.0	.0	.0	.0	.0	.0	.0
74 M4SS	.3442167	.3469832	.3497979	.3526490	.3555471	.3584926	.3614858
75 R4	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000
76 TEND	.0	.0	.0	.0	.0	.0	.0
77 TIME	3.516048	3.616493	3.716938	3.817383	3.917828	4.018272	4.118717
78 TLITE	3.516048	3.616493	3.716938	3.817383	3.917828	4.018272	4.118717
79 TT35	1762.537	1753.940	1745.305	1736.692	1728.062	1719.437	1710.782
80 TT4	2089.697	2077.208	2064.694	2052.177	2039.639	2027.061	2014.475
81 U4	357.3361	355.2024	353.0625	350.9221	348.7781	346.6272	344.4751
82 MFB	1.601019	1.773116	1.745213	1.717309	1.689405	1.661502	1.633598
83 M35	243.4305	243.0180	244.2081	244.6015	244.9947	245.3851	245.7835
84 M4	69.71924	70.06920	70.42456	70.78299	71.14639	71.51422	71.88716
SENSED PARAMETERS							
85 PT3H	250.0000	250.0000	250.0000	250.0000	250.0000	250.0000	250.0000
86 PT35H	226.9440	227.1092	227.2779	227.4484	227.6213	227.7936	227.9722
87 PT4H	177.0606	177.4168	177.7786	178.1409	178.5078	178.8765	179.2503
88 GBURNH	-6059.055	-6110.414	-6160.797	-6213.051	-6265.219	-6319.539	-6372.891
89 TT3H	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000
90 TT35H	1762.537	1753.940	1745.305	1736.692	1728.062	1719.437	1710.782
91 TT4H	2089.697	2077.208	2064.694	2052.177	2039.639	2027.061	2014.475
92 MFBH	1.601019	1.773116	1.745213	1.717309	1.689405	1.661502	1.633598
93 M3H	175.2413	174.6812	174.1046	173.5232	172.9271	172.3298	171.7084
94 M35H	243.4305	243.0180	244.2081	244.6015	244.9947	245.3851	245.7835

95 MM 69.71924 70.06920 70.42456 70.78299 71.14639 71.51422 71.88716

\*\*\*\*\*

BANKDL

1 TIME	4.219162	4.319407	4.420052	4.520496	4.620941	4.721386	4.821831
2 ADIFF	70.00000	70.00000	70.00000	70.00000	70.00000	70.00000	70.00000
3 ALINR	100.00000	100.00000	100.00000	100.00000	100.00000	100.00000	100.00000
4 DFANB	-1.819263	-1.785377	-1.751880	-1.718612	-1.685014	-1.651942	-1.618705
5 DPT35	-89092.25	-86788.94	-84497.12	-82212.94	-80011.2	-77822.7	-75635.1
6 DPT4	105994.0	105377.4	104758.2	104139.1	103511.2	102882.7	102252.1
7 DTT4	-420463.4	-413719.5	-406976.2	-400281.3	-393655.9	-387023.3	-380483.3
8 DTT4	217336.3	214847.7	212323.7	209815.6	207341.6	204836.4	202399.7
9 FAMB	.9191632E-02	.9036873E-02	.8882359E-02	.8727949E-02	.8573510E-02	.8419357E-02	.8265138E-02
10 FA3	.0	.0	.0	.0	.0	.0	.0
11 FA35	18.00000	18.00000	18.00000	18.00000	18.00000	18.00000	18.00000
12 FPHPT	1.400000	1.400000	1.400000	1.400000	1.400000	1.400000	1.400000
13 GAM3	1.400000	1.400000	1.400000	1.400000	1.400000	1.400000	1.400000
14 GAM35	1.400000	1.400000	1.400000	1.400000	1.400000	1.400000	1.400000
15 GAM4	1.400000	1.400000	1.400000	1.400000	1.400000	1.400000	1.400000
16 IUPDAT	.0	.0	.0	.0	.0	.0	.0
17 PRODIFF	1.095746	1.094894	1.094001	1.093085	1.092190	1.091239	1.090302
18 PRLINR	1.270135	1.268445	1.266755	1.265069	1.263350	1.261637	1.259920
19 PT3	250.0000	250.0000	250.0000	250.0000	250.0000	250.0000	250.0000
20 PT35	228.1549	228.3324	228.5190	228.7103	228.8979	229.0973	229.2941
21 PT4	179.6303	180.0096	180.3971	180.7887	181.1832	181.5873	181.9909
22 R3	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000
23 R35	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000
24 R4	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000
25 TT3	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000
26 TT35	1702.098	1693.432	1684.713	1675.974	1667.239	1658.421	1649.634
27 TT4	2001.852	1989.184	1976.494	1963.771	1950.994	1938.169	1925.291
28 MFB	1.605695	1.577791	1.549888	1.521984	1.494081	1.466177	1.438273
29 M3	171.0669	170.4422	169.7818	169.1009	168.4284	167.7112	166.9960
30 M35	246.1852	246.5762	246.9782	247.3878	247.7850	248.2000	248.6086
31 M4	72.26634	72.64917	73.03889	73.43420	73.83501	74.24413	74.65758

OUTLNR

32 CV35P	.1709999	.1709999	.1709999	.1709999	.1709999	.1709999	.1709999
33 DPT35	-89092.25	-86788.94	-84497.12	-82212.94	-80011.2	-77822.7	-75635.1
34 DTT35	-420463.4	-413719.5	-406976.2	-400281.3	-393655.9	-387023.3	-380483.3
35 FAMB	.9191632E-02	.9036873E-02	.8882359E-02	.8727949E-02	.8573510E-02	.8419357E-02	.8265138E-02
36 M3	240.0000	240.0000	240.0000	240.0000	240.0000	240.0000	240.0000
37 M35	408.5010	406.4236	404.3308	402.2336	400.1372	398.0210	395.9121
38 M4	480.4443	477.4041	474.3584	471.3049	468.2385	465.1604	462.0698
39 PT35	228.1549	228.3324	228.5190	228.7103	228.8979	229.0973	229.2941
40 QBANC3	.0	.0	.0	.0	.0	.0	.0
41 RHO35	.2094440E-03	.2106783E-03	.2119417E-03	.2132252E-03	.2145183E-03	.2158467E-03	.2171829E-03
42 R35	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000
43 TT3	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000
44 TT35	1702.098	1693.432	1684.713	1675.974	1667.239	1658.421	1649.634
45 TT4	2001.852	1989.184	1976.494	1963.771	1950.994	1938.169	1925.291
46 U35	291.0569	289.5767	288.0857	286.5913	285.0977	283.5898	282.0874

47 M3	171.0469	170.4422	169.7818	169.1009	168.4284	167.7112	166.9960
48 M35	246.1852	246.5762	246.9782	247.3878	247.7850	248.2000	248.6086
BURNER							
49 CVAP	.1709999	.1709999	.1709999	.1709999	.1709999	.1709999	.1709999
50 DELAY	.0	.0	.0	.0	.0	.0	.0
51 DFAPB	-1.819263	-1.785377	-1.751880	-1.718612	-1.685014	-1.651942	-1.618705
52 DLYP	.0	.0	.0	.0	.0	.0	.0
53 DLYPRP	.0	.0	.0	.0	.0	.0	.0
54 DMAIR	174.5770	174.5776	174.5823	174.5889	174.5776	174.5758	174.5630
55 DMFUEL	.9474990	.9271492	.9068413	.8865992	.8664360	.8463079	.8262762
56 DPT4	105994.0	105377.4	104758.2	104139.1	103511.2	102882.7	102252.1
57 DTBLD1	408.9480	401.2041	393.4673	385.7434	378.0645	370.3833	362.7373
58 DTBLAG	411.0291	403.2991	395.5251	387.7708	380.0444	372.3337	364.6506
59 DTBURN	217336.3	214847.7	212323.7	209815.6	207361.6	204836.4	202399.7
61 ETAB	1.005089	1.005092	1.005095	1.005100	1.005104	1.005108	1.005113
62 FAO	.6522305E-02	.6398797E-02	.6275401E-02	.6152216E-02	.6029744E-02	.5907239E-02	.5785290E-02
63 FAP	.2608922E-01	.2559519E-01	.2510160E-01	.2460887E-01	.2411897E-01	.2362895E-01	.2314116E-01
64 FAPA	.2608922E-01	.2559519E-01	.2510160E-01	.2460887E-01	.2411897E-01	.2362895E-01	.2314116E-01
65 M35	408.5010	406.4236	404.3308	402.2336	400.1372	398.0210	395.9121
66 M4	480.4443	477.4041	474.3584	471.3049	468.2385	465.1604	462.0698
67 M4SS	507.1628	503.2153	499.2571	495.2986	491.3479	487.3811	483.4282
68 IUPDAT	.0	.0	.0	.0	.0	.0	.0
69 MBLITE	2.000000	2.000000	2.000000	2.000000	2.000000	2.000000	2.000000
70 PCBURN	1.000000	1.000000	1.000000	1.000000	1.000000	1.000000	1.000000
71 PT4	179.6303	180.0096	180.3971	180.7887	181.1832	181.5873	181.9909
72 QBURN	7129.250	7031.840	6933.309	6834.148	6734.539	6634.461	6533.738
73 QBURNP	.0	.0	.0	.0	.0	.0	.0
74 RM4SS	.3645365	.3676329	.3707898	.3740020	.3772726	.3806165	.3840137
75 R4	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000
76 TEND	.0	.0	.0	.0	.0	.0	.0
77 TIME	4.219162	4.319607	4.420052	4.520496	4.620941	4.721386	4.821831
78 TLITE	4.219162	4.319607	4.420052	4.520496	4.620941	4.721386	4.821831
79 TT35	1702.088	1693.432	1684.713	1675.974	1667.239	1658.421	1649.634
80 TT4	2001.852	1989.184	1976.494	1963.771	1950.994	1938.169	1925.291
81 U4	342.3167	340.1504	337.9802	335.8047	333.6199	331.4268	329.2246
82 MFB	1.605695	1.577791	1.549888	1.521984	1.494081	1.466177	1.438273
83 M35	246.1852	246.5762	246.9782	247.3878	247.7850	248.2000	248.6086
84 M4	72.26634	72.64917	73.03889	73.43420	73.83501	74.24413	74.65758
SENSED PARAMETERS							
85 PT3H	250.0000	250.0000	250.0000	250.0000	250.0000	250.0000	250.0000
86 PT35H	228.1549	228.3324	228.5190	228.7103	228.8979	229.0973	229.2941
87 PT4H	179.6303	180.0096	180.3971	180.7887	181.1832	181.5873	181.9909
88 QBURNH	-6425.719	-6482.516	-6537.082	-6592.559	-6650.012	-6705.703	-6765.516
89 TT3H	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000
90 TT35H	1702.088	1693.432	1684.713	1675.974	1667.239	1658.421	1649.634
91 TT4H	2001.852	1989.184	1976.494	1963.771	1950.994	1938.169	1925.291
92 MFBH	1.605695	1.577791	1.549888	1.521984	1.494081	1.466177	1.438273
93 M3H	171.0669	170.4422	169.7818	169.1009	168.4284	167.7112	166.9960
94 M35H	246.1852	246.5762	246.9782	247.3878	247.7850	248.2000	248.6086
95 M4H	72.26634	72.64917	73.03889	73.43420	73.83501	74.24413	74.65758

\*\*\*\*\*

BRAND

1	TINE	4.92276	5.02270	5.12315	5.223610	5.324055	5.424500	5.524944
2	ADIFF	70.00000	70.00000	70.00000	70.00000	70.00000	70.00000	70.00000
3	ALINR	100.0000	100.0000	100.0000	100.0000	100.0000	100.0000	100.0000
4	DFAMB	-1.586156	-1.550808	-1.606487	-1.401346	-1.365279	-1.334426	-1.303105
5	DP135	-87091.75	-86712.69	-92199.25	-87201.19	-87174.06	-87167.56	-87042.06
6	DP14	101621.0	100932.4	103465.6	102262.9	102096.3	102096.3	101751.6
7	DT135	-373910.5	-367483.7	-367802.0	-396737.5	-401642.7	-402835.9	-400810.8
8	DT14	199899.1	197443.0	200107.7	206742.1	206365.4	204999.1	203194.2
9	FAMB	.8111261E-02	.7956814E-02	.777949E-02	.7520188E-02	.7359661E-02	.7202473E-02	.7045541E-02
10	FA3	.0	.0	.0	.0	.0	.0	.0
11	FA35	.0	.0	.0	.0	.0	.0	.0
12	FPHPT	18.00000	18.00000	18.00000	18.00000	18.00000	18.00000	18.00000
13	GAM3	1.400000	1.400000	1.400000	1.400000	1.400000	1.400000	1.400000
14	GAM35	1.400000	1.400000	1.400000	1.400000	1.400000	1.400000	1.400000
15	GAM4	1.400000	1.400000	1.400000	1.400000	1.400000	1.400000	1.400000
16	IUPDAT	.0	.0	.0	.0	.0	.0	.0
17	PRODIFF	1.089293	1.089490	1.079644	1.093829	1.094801	1.095006	1.094771
18	PRLINR	1.258202	1.256267	1.266750	1.261795	1.262229	1.262239	1.261585
19	PT3	250.0000	250.0000	250.0000	250.0000	250.0000	250.0000	250.0000
20	PT35	229.5067	229.6758	231.5576	228.5548	228.3519	228.3089	228.3582
21	PT4	182.4084	182.8241	182.7964	181.1345	180.9116	180.8762	181.0088
22	R3	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000
23	R35	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000
24	R4	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000
25	TT3	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000
26	TT35	1640.727	1631.930	1632.614	1671.681	1678.313	1679.984	1677.391
27	TT4	1912.364	1899.375	1899.676	1937.168	1944.337	1945.771	1941.380
28	MFB	1.410370	1.382466	1.354563	1.326659	1.298756	1.270852	1.242949
29	M3	166.2181	165.5969	158.4385	169.6550	170.3741	170.5261	170.3533
30	M35	249.0434	249.3588	254.2231	246.6703	246.0790	245.9139	245.9827
31	M4	75.08134	75.50929	75.49193	74.07814	73.85043	73.80875	73.94635

OUTLNR

32	CV35P	.1709999	.1709999	.1709999	.1709999	.1709999	.1709999	.1709999
33	DP135	-87091.75	-86712.69	-92199.25	-87201.19	-87174.06	-87167.56	-87042.06
34	DT135	-373910.5	-367483.7	-367802.0	-396737.5	-401642.7	-402835.9	-400810.8
35	FAMB	.8111261E-02	.7956814E-02	.777949E-02	.7520188E-02	.7359661E-02	.7202473E-02	.7045541E-02
36	M3	240.0000	240.0000	240.0000	240.0000	240.0000	240.0000	240.0000
37	M35	393.7744	391.6631	391.8271	401.2034	402.7949	403.1960	402.5737
38	M4	458.9670	455.8499	455.9221	464.9202	466.6409	466.9851	465.9312
39	PT35	229.5067	229.6758	231.5576	228.5548	228.3519	228.3089	228.3582
40	QBNC5	.0	.0	.0	.0	.0	.0	.0
41	RH035	.2185643E-03	.2199043E-03	.2216131E-03	.2136276E-03	.2125944E-03	.2123430E-03	.2127171E-03
42	R35	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000
43	TT3	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000
44	TT35	1640.727	1631.930	1632.614	1671.681	1678.313	1679.984	1677.391
45	TT4	1912.364	1899.375	1899.676	1937.168	1944.337	1945.771	1941.380
46	U35	280.5642	279.0601	279.1770	285.8574	286.9915	287.2771	286.8337
47	M3	166.2181	165.5969	158.4385	169.6550	170.3741	170.5261	170.3533

48 MSG	249.0434	249.3588	254.2231	246.6703	246.0790	245.9139	245.9827
BANNER							
49 CVAP	.1709999	.1709999	.1709999	.1709999	.1709999	.1709999	.1709999
50 DELAY	.0	.0	.0	.0	.0	.0	.0
51 DFAMB	-1.586156	-1.550808	-1.606487	-1.401346	-1.365279	-1.334426	-1.303105
52 DLYP	.0	.0	.0	.0	.0	.0	.0
53 DLYPRP	.0	.0	.0	.0	.0	.0	.0
54 DMHAI	174.4456	174.4456	179.3139	173.1450	172.7681	172.6329	172.5537
55 DMFUEL	.8062655	.7863956	.7718105	.7737356	.7592123	.7430480	.7256012
56 DPT4	101621.0	100932.4	103465.6	102262.9	102252.5	102096.3	101751.6
57 DTBIDL	355.0791	347.6135	334.0803	337.2170	330.9175	324.0251	316.8218
58 DTBLAG	356.9761	349.3689	339.5776	336.5862	330.2100	323.4019	316.2712
59 DTBURN	356.8960	349.3940	335.7959	338.9631	332.6367	325.7129	318.4751
60 DTT4	199899.1	197443.0	200107.7	206742.1	206365.4	204999.1	203194.2
61 ETAB	1.005117	1.005122	1.005136	1.005176	1.005196	1.005209	1.005219
62 FAO	.5663149E-02	.5544081E-02	.5328242E-02	.5378269E-02	.5277798E-02	.5167872E-02	.5052967E-02
63 FAP	.2265260E-01	.2217633E-01	.2131297E-01	.2151307E-01	.2111119E-01	.2067149E-01	.2021195E-01
64 FAPA	.2265260E-01	.2217633E-01	.2131297E-01	.2151307E-01	.2111119E-01	.2067149E-01	.2021195E-01
65 M35	393.7744	391.6631	391.8271	401.2034	402.7949	403.1960	402.5737
66 M4	458.9670	455.8499	455.9221	464.9202	466.6409	466.9851	465.9312
67 MASS	479.4487	475.5117	473.3259	481.9839	482.0454	480.8125	478.4790
68 IUPDAT	.0	.0	.0	.0	.0	.0	.0
69 MBLITE	2.000000	2.000000	2.000000	2.000000	2.000000	2.000000	2.000000
70 PCBURN	1.000000	1.000000	1.000000	1.000000	1.000000	1.000000	1.000000
71 PT4	182.4084	182.8241	182.7964	181.1345	180.9116	180.8762	181.0088
72 QBURN	6432.539	6331.348	6152.496	5984.070	5852.680	5728.770	5612.914
73 QBURNP	.0	.0	.0	.0	.0	.0	.0
74 RMSS	.3874967	.3910356	.3909145	.3798634	.3779970	.3776444	.3787760
75 R4	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000
76 TEND	.0	.0	.0	.0	.0	.0	.0
77 TIME	4.922276	5.022720	5.123165	5.223610	5.324055	5.424500	5.524944
78 TLITE	4.922276	5.022720	5.123165	5.223610	5.324055	5.424500	5.524944
79 TT35	1640.727	1631.930	1632.614	1671.681	1678.313	1679.984	1677.391
80 TT4	1912.364	1899.375	1899.676	1937.168	1944.337	1945.771	1941.380
81 U4	327.0142	324.7930	324.8445	331.2556	332.4814	332.7268	331.9758
82 MFVB	1.410370	1.382466	1.354563	1.326659	1.298756	1.270852	1.242949
83 M35	249.0434	249.3588	254.2231	246.6703	246.0790	245.9139	245.9827
84 M4	75.08134	75.50929	75.49193	74.07814	73.85043	73.80875	73.94635
SENSED PARAMETERS							
85 PT3H	250.0000	250.0000	250.0000	250.0000	250.0000	250.0000	250.0000
86 PT35H	229.5067	229.6758	231.5576	228.5548	228.3519	228.3089	228.3582
87 PT4H	182.4084	182.8241	182.7964	181.1345	180.9116	180.8762	181.0088
88 QBURNH	-6821.684	-6879.730	-7223.762	-7420.953	-7481.641	-7508.555	-7550.934
89 TT3H	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000
90 TT35H	1640.727	1631.930	1632.614	1671.681	1678.313	1679.984	1677.391
91 TT4H	1912.364	1899.375	1899.676	1937.168	1944.337	1945.771	1941.380
92 MFVBH	1.410370	1.382466	1.354563	1.326659	1.298756	1.270852	1.242949
93 M3H	166.2181	165.5969	158.4385	169.6550	170.3741	170.5261	170.3533
94 M35H	249.0434	249.3588	254.2231	246.6703	246.0790	245.9139	245.9827
95 M4H	75.08134	75.50929	75.49193	74.07814	73.85043	73.80875	73.94635

\*\*\*\*\*

BROWNDL

1	TIME	5.625309	5.826279	6.027168	6.127613	6.228058
2	ADIFF	70.00000	70.00000	70.00000	70.00000	70.00000
3	ALINR	100.00000	100.00000	100.00000	100.00000	100.00000
4	DFAMB	-1.272296	-1.214748	-1.168675	-1.249304	-1.056209
5	DPT35	-86835.25	-86455.69	-86485.12	-8636.31	-86581.56
6	DPT4	101233.9	99928.19	99362.44	98994.19	98440.87
7	DTT35	-395616.6	-387810.4	-373348.2	-364943.2	-374545.2
8	DTT4	200811.2	197971.9	194913.4	189283.9	186066.4
9	FAPB	.6889030E-02	.6733090E-02	.6424200E-02	.6128050E-02	.5905557E-02
10	FA3	.0	.0	.0	.0	.0
11	FA35	.0	.0	.0	.0	.0
12	FPHPT	18.00000	18.00000	18.00000	18.00000	18.00000
13	GAM3	1.400000	1.400000	1.400000	1.400000	1.400000
14	GAM35	1.400000	1.400000	1.400000	1.400000	1.400000
15	GAM4	1.400000	1.400000	1.400000	1.400000	1.400000
16	IURDAT	.0	.0	.0	.0	.0
17	PRODIFF	1.094038	1.091221	1.089428	1.087365	1.088809
18	PRLINR	1.260353	1.256679	1.255606	1.254912	1.252738
19	PT3	250.0000	250.0000	250.0000	250.0000	250.0000
20	PT35	228.5112	228.7690	229.4782	229.9134	229.6086
21	PT4	181.3072	181.7532	182.2626	183.2106	183.2854
22	R3	640.0000	640.0000	640.0000	640.0000	640.0000
23	R35	640.0000	640.0000	640.0000	640.0000	640.0000
24	R4	640.0000	640.0000	640.0000	640.0000	640.0000
25	TT3	1000.000	1000.000	1000.000	1000.000	1000.000
26	TT35	1670.557	1660.155	1637.927	1629.245	1645.032
27	TT4	1931.501	1917.010	1886.832	1875.308	1893.842
28	MFRB	1.215045	1.187141	1.159238	1.103431	1.047624
29	M3	169.8112	168.8905	167.6976	164.7190	165.8449
30	M35	246.3199	246.9156	248.5039	249.4419	247.3016
31	M4	74.25745	74.72095	75.23849	76.15295	75.81035

OUTLNR

32	CV35P	.1709999	.1709999	.1709999	.1709998	.1709998
33	DPT35	-86835.25	-86455.69	-86485.12	-8636.31	-86581.56
34	DTT35	-395616.6	-387810.4	-373348.2	-364943.2	-374545.2
35	FAPB	.6889030E-02	.6733090E-02	.6424200E-02	.6128050E-02	.5905557E-02
36	M3	240.0000	240.0000	240.0000	240.0000	240.0000
37	M35	400.9336	398.4370	393.1023	391.0188	394.8076
38	M4	463.5603	456.3215	452.8396	450.0737	454.5220
39	PT35	228.5112	228.7690	229.4782	229.9134	229.6086
40	QBNC3	.0	.0	.0	.0	.0
41	RHO35	.2137304E-03	.2153122E-03	.2171286E-03	.2204945E-03	.2180891E-03
42	R35	640.0000	640.0000	640.0000	640.0000	640.0000
43	TT3	1000.000	1000.000	1000.000	1000.000	1000.000
44	TT35	1670.557	1660.155	1637.927	1629.245	1645.032
45	TT4	1931.501	1917.010	1886.832	1875.308	1893.842
46	U35	285.6650	283.8862	280.0854	278.6008	281.3003
47	M3	169.8112	168.8905	167.6976	164.7190	165.8449
48	M35	246.3199	246.9156	248.5039	249.4419	247.3016



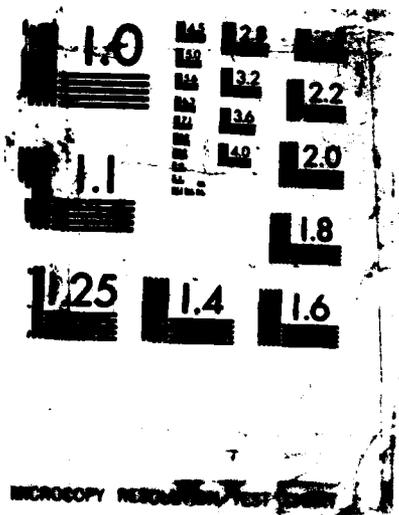
BASED

1 TIME	6.328503	6.428947	6.529392	6.629837	6.730282	6.830727	6.931171
2 ADIFF	70.00000	70.00000	70.00000	70.00000	70.00000	70.00000	70.00000
3 ALLNR	100.0000	100.0000	100.0000	100.0000	100.0000	100.0000	100.0000
4 DFABD	-1.104372	-1.9903963	-1.092928	-6652664	-9145156	-6517332	-7166873
5 DPT35	-96591.00	-86023.87	-95392.06	-82122.00	-87532.50	-71043.25	-81655.62
6 DPT4	99917.19	97502.94	98100.06	90286.87	87480.62	72554.81	76851.56
7 DTT35	-351933.9	-366694.2	-329628.4	-302804.2	-232102.4	-170485.0	-176609.7
8 DTT4	155100.2	183135.4	155668.9	161394.7	151101.1	121070.9	126757.3
9 FABD	.5807512E-02	.5585570E-02	.5467780E-02	.5258802E-02	.5113740E-02	.4840378E-02	.4695535E-02
10 FA3	.0	.0	.0	.0	.0	.0	.0
11 FA35	.0	.0	.0	.0	.0	.0	.0
12 FPHPT	18.00000	18.00000	18.00000	18.00000	18.00000	18.00000	18.00000
13 GAM3	1.400000	1.400000	1.400000	1.400000	1.400000	1.400000	1.400000
14 GAM35	1.400000	1.400000	1.400000	1.400000	1.400000	1.400000	1.400000
15 GAM4	1.400000	1.400000	1.400000	1.400000	1.400000	1.400000	1.400000
16 IUPDAT	.0	.0	.0	.0	.0	.0	.0
17 PRDFF	1.065784	1.068166	1.063315	1.080568	1.053871	1.061248	1.046179
18 PRLNR	1.257651	1.250793	1.253084	1.234638	1.228345	1.191024	1.205562
19 PT3	250.0000	250.0000	250.0000	250.0000	250.0000	250.0000	250.0000
20 PT35	234.5690	229.7442	235.1135	231.3597	237.2206	235.5716	238.9647
21 PT4	186.5135	183.6787	187.6277	187.3906	193.1221	197.7891	198.2185
22 R3	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000
23 R35	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000
24 R4	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000
25 TT3	1000.0000	1000.0000	1000.0000	1000.0000	1000.0000	1000.0000	1000.0000
26 TT35	1620.363	1633.863	1584.658	1538.175	1407.729	1293.635	1289.746
27 TT4	1911.145	1875.123	1851.989	1720.134	1544.273	1339.479	1339.222
28 MFB	1.019720	.9918165	.9639130	.9360094	.9081059	.8802023	.8522987
29 M3	145.9400	145.3459	143.5216	159.2133	133.6158	141.4453	124.6556
30 M35	255.9749	247.7367	258.1106	252.1011	267.9753	262.2268	272.8599
31 M4	76.79552	76.35130	78.47842	81.32782	88.45906	97.27625	97.49681

OUTLNR

32 CV35P	.1709998	.1709999	.1709998	.1709999	.1709999	.1709999	.1709999
33 DPT35	-96591.00	-86023.87	-95392.06	-82122.00	-87532.50	-71043.25	-81655.62
34 DTT35	-351933.9	-366694.2	-329628.4	-302804.2	-232102.4	-170485.0	-176609.7
35 FABD	.5807512E-02	.5585570E-02	.5467780E-02	.5258802E-02	.5113740E-02	.4840378E-02	.4695535E-02
36 M3	240.0000	240.0000	240.0000	240.0000	240.0000	240.0000	240.0000
37 H35	368.8870	392.1270	380.3179	369.1619	337.8550	310.4243	309.5391
38 M4	458.6746	450.0295	444.4773	412.8320	370.6255	321.4749	321.4133
39 PT35	234.5690	229.7442	235.1135	231.3597	237.2206	235.5716	238.9647
40 QDBNCS	.0	.0	.0	.0	.0	.0	.0
41 RH035	.2261927E-03	.2197096E-03	.2318259E-03	.2350185E-03	.2633014E-03	.2845759E-03	.2895005E-03
42 R35	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000
43 TT3	1000.0000	1000.0000	1000.0000	1000.0000	1000.0000	1000.0000	1000.0000
44 TT35	1620.363	1633.863	1584.658	1538.175	1407.729	1293.635	1289.746
45 TT4	1911.145	1875.123	1851.989	1720.134	1544.273	1339.479	1339.222
46 U35	277.0818	279.3904	270.9763	263.0278	240.7217	221.1774	220.5466
47 M3	145.9400	145.3459	143.5216	159.2133	133.6158	141.4453	124.6556
48 M35	255.9749	247.7367	258.1106	252.1011	267.9753	262.2268	272.8599





MICROCOPY RESOLUTION TEST CHART





18 PRLNR	1.190996	1.205500	1.190992	1.205441	1.190989	1.205386	1.190984
19 PT3	250.0000	250.0000	250.0000	250.0000	250.0000	250.0000	250.0000
20 PT35	235.5865	238.9688	235.6024	238.9725	235.6185	238.9762	235.6330
21 PT4	197.8062	196.2321	197.8202	198.2447	197.8342	198.2569	197.8472
22 R3	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000
23 R35	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000
24 R4	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000
25 TT3	1000.0000	1000.0000	1000.0000	1000.0000	1000.0000	1000.0000	1000.0000
26 TT35	1292.844	1289.249	1292.331	1288.756	1291.824	1288.268	1291.318
27 TT4	1338.718	1338.619	1338.111	1338.022	1337.517	1337.428	1336.915
28 MFTB	.8243952	.7964916	.7685881	.7406845	.7127810	.6848774	.6569738
29 M3	141.3742	124.6299	141.3034	124.6133	141.2290	124.5919	141.1616
30 M35	262.2891	272.8914	262.3572	272.9236	262.4272	272.9543	262.4910
31 M4	97.31236	97.52542	97.34128	97.55342	97.36975	97.58105	97.39813

OUTLNR

32 CV35P	.1709999	.1709999	.1709999	.1709999	.1709999	.1709999	.1709999
33 DPT35	-71042.00	-81630.69	-71050.69	-81603.19	-71062.19	-81577.31	-71068.31
34 DTT35	-170247.7	-176404.0	-170051.2	-176199.6	-169860.1	-175997.9	-169665.1
35 FAMB	.4532907E-02	.4387487E-02	.4225552E-02	.4079594E-02	.3918316E-02	.3771816E-02	.3611222E-02
36 M3	240.0000	240.0000	240.0000	240.0000	240.0000	240.0000	240.0000
37 M35	310.2825	309.4197	310.1592	309.3013	310.0376	309.1843	309.9163
38 M4	321.2922	321.2686	321.1465	321.1252	321.0039	320.9827	320.8594
39 PT35	235.5865	238.9688	235.6024	238.9725	235.6185	238.9762	235.6330
40 GORNC	.0	.0	.0	.0	.0	.0	.0
41 RHO35	.2847242E-03	.2896171E-03	.2848544E-03	.2897524E-03	.2849877E-03	.2898467E-03	.2851170E-03
42 R35	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000
43 TT3	1000.0000	1000.0000	1000.0000	1000.0000	1000.0000	1000.0000	1000.0000
44 TT35	1292.844	1289.249	1292.331	1288.756	1291.824	1288.268	1291.318
45 TT4	1338.718	1338.619	1338.111	1338.022	1337.517	1337.428	1336.915
46 U35	221.0763	220.4616	220.9886	220.3773	220.9019	220.2939	220.8194
47 M3	141.3742	124.6299	141.3034	124.6133	141.2290	124.5919	141.1616
48 M35	262.2891	272.8914	262.3572	272.9236	262.4272	272.9543	262.4910

BURNER

49 CV4P	.1709999	.1709999	.1709999	.1709999	.1709999	.1709999	.1709999
50 DELAY	8562.953	11859.78	15145.45	10421.70	21687.00	24942.86	28187.85
51 DFAP	-.6100439	-.6691682	-.5685361	-.6217711	-.5270974	-.5744733	-.6856772
52 OLYP	.0	.0	.0	.0	.0	.0	.0
53 OLYPRP	.0	.0	.0	.0	.0	.0	.0
54 DMATR	165.4158	175.7919	165.4254	175.7665	165.4375	175.7400	165.4432
55 DMFUEL	.3852778	.3704692	.3443239	.3327444	.3327444	.3182027	.3065130
56 DPT4	72512.81	76800.75	72478.69	76750.69	72445.75	76700.81	72410.44
57 DTBIDL	.0	.0	.0	.0	.0	.0	.0
58 DTBLAG	.0	.0	.0	.0	.0	.0	.0
59 DTBURN	.0	.0	.0	.0	.0	.0	.0
60 DTT4	120984.1	126643.1	120894.2	126530.4	120805.4	126420.4	120716.1
61 ETAB	.0	.0	.0	.0	.0	.0	.0
62 FAO	.2704959E-02						
63 FAP	.1257231E-01	.1167485E-01	.1171819E-01	.1085556E-01	.1084443E-01	.1003651E-01	.1001137E-01
64 FAPA	.1257231E-01	.1167485E-01	.1171819E-01	.1085556E-01	.1084443E-01	.1003651E-01	.1001137E-01
65 M35	310.2825	309.4197	310.1592	309.3013	310.0376	309.1843	309.9163
66 M4	321.2922	321.2686	321.1465	321.1252	321.0039	320.9827	320.8594

67 MSS	310.2025	309.4197	310.1592	309.3013	310.0376	309.1843	309.9163
68 IUPDAT	.0	.0	.0	.0	.0	.0	.0
69 PBLITE	.0	.0	.0	.0	.0	.0	.0
70 PCBURN	.0	.0	.0	.0	.0	.0	.0
71 PT4	197.8062	196.2321	197.8202	198.2447	197.8342	198.2569	197.8472
72 GBURN	.0	.0	.0	.0	.0	.0	.0
73 GDBNLP	.0	.0	.0	.0	.0	.0	.0
74 RMAS5	.6002668	.6016033	.6005814	.6019102	.6008906	.6022145	.6012009
75 R4	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000
76 TEND	.0	.0	.0	.0	.0	.0	.0
77 TIME	7.051616	7.132061	7.232506	7.332951	7.433395	7.533840	7.634285
78 TLITE	6.778255	6.778255	6.778255	6.778255	6.778255	6.778255	6.778255
79 TT35	1292.844	1289.249	1292.331	1288.756	1291.824	1288.268	1291.318
80 TT4	1338.718	1338.619	1338.111	1338.022	1337.517	1337.428	1336.915
81 U4	228.9207	228.9039	228.8170	228.8018	228.7154	228.7001	228.6124
82 MFR8	.8243952	.7964916	.7685881	.7406845	.7127810	.6848774	.6569738
83 M35	262.2891	272.8914	262.3572	272.9236	262.4272	272.9543	262.4910
84 M4	97.31236	97.52542	97.34128	97.55342	97.36975	97.58105	97.39813

SENSED PARAMETERS

85 PT3H	250.0000	250.0000	250.0000	250.0000	250.0000	250.0000	250.0000
86 PT35H	235.5845	238.9688	235.6024	238.9725	235.6185	238.9762	235.6330
87 PT4H	197.8062	196.2321	197.8202	198.2447	197.8342	198.2569	197.8472
88 GBURNH	-12418.99	-13028.39	-12416.28	-13023.45	-12413.54	-13018.70	-12410.78
89 TT3H	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000
90 TT35H	1292.844	1289.249	1292.331	1288.756	1291.824	1288.268	1291.318
91 TT4H	1338.718	1338.619	1338.111	1338.022	1337.517	1337.428	1336.915
92 MFR8H	.8243952	.7964916	.7685881	.7406845	.7127810	.6848774	.6569738
93 M35H	141.3742	141.6299	141.3034	144.6133	141.2290	144.5919	141.1414
94 M35M	262.2891	272.8914	262.3572	272.9236	262.4272	272.9543	262.4910
95 M4H	97.31236	97.52542	97.34128	97.55342	97.36975	97.58105	97.39813

\*\*\*\*\*

BRNDL

1 TIME	7.734730	7.835175	7.935619	8.036064	8.136509	8.236954	8.337399
2 ADIFF	70.00000	70.00000	70.00000	70.00000	70.00000	70.00000	70.00000
3 ALINR	100.0000	100.0000	100.0000	100.0000	100.0000	100.0000	100.0000
4 OFAP8	-.5272769	-.4443207	-.4401826	-.4030405	-.4331827	-.3617926	-.3862958
5 DP135	-81551.87	-71077.31	-81526.12	-71089.81	-81498.12	-71098.94	-81472.69
6 DP14	76651.12	72376.50	74601.25	72344.56	76550.15	72310.75	76500.87
7 DT35	-175796.2	-169472.7	-175594.7	-169287.6	-175392.6	-169097.2	-175192.6
8 DT14	126309.7	126627.1	126196.7	120542.0	126083.2	120453.4	125971.2
9 FAV8	.3464161E-02	.3304226E-02	.3156638E-02	.2997362E-02	.2849246E-02	.2690621E-02	.2541962E-02
10 FA3	.0	.0	.0	.0	.0	.0	.0
11 FA35	.0	.0	.0	.0	.0	.0	.0
12 FPHT	18.00000	18.00000	18.00000	18.00000	18.00000	18.00000	18.00000
13 GAN3	1.400000	1.400000	1.400000	1.400000	1.400000	1.400000	1.400000
14 GAN35	1.400000	1.400000	1.400000	1.400000	1.400000	1.400000	1.400000
15 GAN4	1.400000	1.400000	1.400000	1.400000	1.400000	1.400000	1.400000
16 IUPDAT	.0	.0	.0	.0	.0	.0	.0
17 PROIFF	1.046111	1.060901	1.046095	1.060829	1.046080	1.060758	1.046062
18 PRLINR	1.205329	1.190980	1.205270	1.190983	1.205211	1.190981	1.205153

19 PT3	250.0000	250.0000	250.0000	250.0000	250.0000	250.0000	250.0000
20 PT35	238.9603	238.9639	235.6646	238.9875	235.6804	238.9914	238.9914
21 PT4	198.2697	198.2825	197.8741	198.2951	197.8875	198.3080	198.3080
22 R3	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000
23 R35	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000
24 R4	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000
25 TT3	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000
26 TT35	1287.779	1290.814	1287.291	1286.806	1289.818	1286.320	1286.320
27 TT4	1336.833	1336.318	1336.244	1335.735	1335.145	1335.073	1335.073
28 WPTB	.6290703	.6011667	.5732632	.5435596	.5174561	.4895525	.4616489
29 M3	124.5708	141.0905	124.5497	141.0157	140.9448	124.5117	124.5117
30 M35	272.9866	262.5579	273.0181	262.6277	262.6943	273.0789	273.0789
31 M4	97.60909	97.42667	97.63690	97.45432	97.66454	97.69225	97.69225

OUTLINE

32 CV35P	.1709999	.1709999	.1709999	.1709999	.1709999	.1709999	.1709999
33 DPT35	-81551.87	-71077.31	-81526.12	-71089.61	-81498.12	-71098.94	-81472.69
34 DTT35	-175796.2	-169472.7	-175594.7	-169287.6	-175392.6	-169097.2	-175192.6
35 FAMB	.3464161E-02	.3304226E-02	.3156638E-02	.2997362E-02	.2849248E-02	.2690621E-02	.2541982E-02
36 M3	240.0000	240.0000	240.0000	240.0000	240.0000	240.0000	240.0000
37 M35	309.0669	309.7952	308.9497	309.6763	308.8333	309.5562	308.7168
38 M4	320.8398	320.7163	320.6985	320.5762	320.5579	320.4346	320.4172
39 PT35	238.9603	238.6487	238.9839	235.6646	238.9875	238.6804	238.9914
40 GOBANC5	.0	.0	.0	.0	.0	.0	.0
41 RHO35	.2899617E-03	.2852471E-03	.2900760E-03	.2853761E-03	.2901898E-03	.2855058E-03	.2903040E-03
42 R35	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000
43 TT3	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000
44 TT35	1287.779	1290.814	1287.291	1286.806	1289.818	1286.320	1286.320
45 TT4	1336.833	1336.318	1336.244	1335.735	1335.145	1335.073	1335.073
46 U35	220.2102	220.7291	220.1268	220.6443	220.0438	220.5589	219.9608
47 M3	124.5708	141.0905	124.5497	141.0157	140.9448	124.5117	124.5117
48 M35	272.9866	262.5579	273.0181	262.6277	273.0474	262.6943	273.0789

BURNER

49 CV4P	.1709999	.1709999	.1709999	.1709999	.1709999	.1709999	.1709999
50 DELAY	31423.61	34648.51	37864.27	41069.38	44265.44	47450.96	50627.45
51 DFAMB	-.5272769	-.4443207	-.4801826	-.4030405	-.4331827	-.3617926	-.3062958
52 DLYP	.0	.0	.0	.0	.0	.0	.0
53 DLYPRP	.0	.0	.0	.0	.0	.0	.0
54 DMAIR	175.7144	165.4520	175.6883	165.4645	175.6602	165.4734	175.6343
55 DMFUEL	.2921039	.2803070	.2660284	.2541264	.2399763	.2279678	.2139466
56 DPT4	76653.12	72376.50	76601.25	72344.56	76550.56	72310.75	76500.87
57 DTB10L	.0	.0	.0	.0	.0	.0	.0
58 DTBLAG	.0	.0	.0	.0	.0	.0	.0
59 DTBURN	.0	.0	.0	.0	.0	.0	.0
60 DTT4	126309.7	120627.1	126196.7	120542.0	126083.2	120453.4	125971.2
61 ETAB	.0	.0	.0	.0	.0	.0	.0
62 FAO	.2704959E-02						
63 FAP	.9217598E-02	.9158615E-02	.839802E-02	.8306198E-02	.7580455E-02	.7454328E-02	.6762132E-02
64 FAPA	.9217598E-02	.9158615E-02	.839802E-02	.8306198E-02	.7454328E-02	.7454328E-02	.6762132E-02
65 M35	309.0669	309.7952	308.9497	309.6763	308.8333	309.5562	308.7168
66 M4	320.8398	320.7163	320.6985	320.5762	320.5579	320.4346	320.4172
67 M4SS	309.0669	309.7952	308.9497	309.6763	308.8333	309.5562	308.7168



20 PT35	235.6954	230.9946	235.7128	238.9986	235.7273	239.0025	235.7437
21 PT4	197.9007	196.3194	197.9146	196.3320	197.9267	196.3447	197.9398
22 R3	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000
23 R35	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000
24 R4	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000
25 TT3	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000
26 TT35	1289.322	1285.845	1288.826	1285.362	1288.336	1284.806	1287.846
27 TT4	1334.558	1334.491	1333.981	1333.908	1333.401	1333.337	1332.823
28 WPT8	.4337454	.4058418	.3779383	.3500347	.3221212	.2942276	.2663240
29 M3	140.8773	124.4949	140.7952	124.4737	140.7274	124.4524	140.6525
30 M35	262.7622	273.1099	262.8337	273.1411	262.8999	273.1716	262.9700
31 M4	97.51042	97.71919	97.53836	97.74677	97.56557	97.77390	97.59319

OUTLNR

32 CV35P	.1709999	.1709999	.1709999	.1709999	.1709999	.1709999	.1709999
33 DP735	-71107.81	-81446.44	-71123.62	-81421.06	-71132.44	-81396.06	-71145.31
34 DT735	-148909.9	-174996.9	-168726.6	-174797.9	-168542.3	-174601.6	-168359.7
35 FAMB	.2384005E-02	.2234844E-02	.2077514E-02	.1927837E-02	.1771147E-02	.1620955E-02	.1464903E-02
36 M3	240.0000	240.0000	240.0000	240.0000	240.0000	240.0000	240.0000
37 M35	309.4370	308.6028	309.3181	308.4868	309.2007	308.3726	309.0828
38 M4	320.2939	320.2778	320.1555	320.1379	320.0161	320.0007	319.8774
39 PT35	235.6954	230.9946	235.7128	238.9986	235.7273	239.0025	235.7437
40 GOBANC	.0	.0	.0	.0	.0	.0	.0
41 RHO35	.2856338E-03	.2904152E-03	.2857449E-03	.2905291E-03	.2858911E-03	.2906416E-03	.2860196E-03
42 R35	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000
43 TT3	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000
44 TT35	1289.322	1285.845	1288.826	1285.362	1288.336	1284.806	1287.846
45 TT4	1334.558	1334.491	1333.981	1333.908	1333.401	1333.337	1332.823
46 U35	220.4740	219.8795	220.3892	219.7969	220.3055	219.7156	220.2216
47 M3	140.8773	124.4949	140.7952	124.4737	140.7274	124.4524	140.6525
48 M35	262.7622	273.1099	262.8337	273.1411	262.8999	273.1716	262.9700

BURNER

49 CV4P	.1709999	.1709999	.1709999	.1709999	.1709999	.1709999	.1709999
50 DELAY	53793.50	56950.63	60097.43	63235.39	66362.37	69474.62	72577.75
51 DFAMB	-.3206045	-.3395109	-.2794820	-.2928271	-.2383928	-.2462426	-.1973640
52 DLYP	.0	.0	.0	.0	.0	.0	.0
53 DLYPRP	.0	.0	.0	.0	.0	.0	.0
54 DMAIR	165.4836	175.6085	165.4976	175.5824	165.5068	175.5559	165.5195
55 DMFUEL	.2018328	.1879416	.1757210	.1619574	.1496337	.1359970	.1235687
56 DPT4	72277.81	76452.19	72246.37	76402.56	72213.50	76353.19	72181.94
57 DTBIDL	.0	.0	.0	.0	.0	.0	.0
58 DTBLAG	.0	.0	.0	.0	.0	.0	.0
59 DTBURN	.0	.0	.0	.0	.0	.0	.0
60 DTT4	120366.9	125864.7	120279.8	125753.0	120195.6	125640.9	120111.2
61 ETAB	.0	.0	.0	.0	.0	.0	.0
62 FAO	.2704959E-02						
63 FAP	.6602857E-02	.5944006E-02	.5751744E-02	.5126063E-02	.4901197E-02	.4308317E-02	.4051015E-02
64 FAPA	.6602857E-02	.5944006E-02	.5751744E-02	.5126063E-02	.4901197E-02	.4308317E-02	.4051015E-02
65 M35	309.4370	308.6028	309.3181	308.4868	309.2007	308.3726	309.0828
66 M4	320.2939	320.2778	320.1555	320.1379	320.0161	320.0007	319.8774
67 M4SS	309.4370	308.6028	309.3181	308.4868	309.2007	308.3726	309.0828
68 IUPDAT	.0	.0	.0	.0	.0	.0	.0

69 MBLITE	.0	.0	.0	.0	.0	.0	.0	.0	.0
70 PCBURN	.0	.0	.0	.0	.0	.0	.0	.0	.0
71 PT4	197.9007	190.3194	197.9146	198.3320	197.9267	198.3447	197.9398	197.9398	197.9398
72 GBURN	.0	.0	.0	.0	.0	.0	.0	.0	.0
73 QOBMLP	.0	.0	.0	.0	.0	.0	.0	.0	.0
74 RHASS	.6024250	.6037300	.6027281	.6040324	.6030272	.6043296	.6033268	.6033268	.6033268
75 R4	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000
76 TEND	.0	.0	.0	.0	.0	.0	.0	.0	.0
77 TIME	8.437843	8.538288	8.638733	8.739178	8.839622	8.940067	9.040512	9.040512	9.040512
78 TLITE	6.778255	6.778255	6.778255	6.778255	6.778255	6.778255	6.778255	6.778255	6.778255
79 TT35	1289.322	1285.845	1288.826	1285.362	1288.336	1284.886	1287.846	1287.846	1287.846
80 TT4	1334.558	1334.491	1333.981	1333.908	1333.401	1333.337	1332.823	1332.823	1332.823
81 U4	228.2095	228.1980	228.1108	228.0983	228.0116	228.0006	227.9127	227.9127	227.9127
82 MFMB	.4337454	.4058418	.3779383	.3500347	.3221312	.2942276	.2663240	.2663240	.2663240
83 MS5	262.7622	273.1099	262.8337	273.1411	262.8999	273.1716	262.9700	262.9700	262.9700
84 M4	97.51042	97.71919	97.53836	97.74677	97.56557	97.77390	97.59319	97.59319	97.59319

SENSED PARAMETERS

85 PT3H	250.0000	250.0000	250.0000	250.0000	250.0000	250.0000	250.0000	250.0000	250.0000
86 PT35H	235.6954	236.9946	235.7128	238.9986	235.7273	239.0025	235.7437	235.7437	235.7437
87 PT4H	197.9007	190.3194	197.9146	198.3320	197.9267	198.3447	197.9398	197.9398	197.9398
88 GBURNH	-12400.09	-12994.06	-12397.34	-12989.04	-12394.80	-12983.84	-12392.30	-12392.30	-12392.30
89 TT3H	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000
90 TT35H	1289.322	1285.845	1288.826	1285.362	1288.336	1284.886	1287.846	1287.846	1287.846
91 TT4H	1334.558	1334.491	1333.981	1333.908	1333.401	1333.337	1332.823	1332.823	1332.823
92 MFMBH	.4337454	.4058418	.3779383	.3500347	.3221312	.2942276	.2663240	.2663240	.2663240
93 MS5H	140.8773	124.4949	140.7952	124.4737	140.7274	124.4524	140.6525	140.6525	140.6525
94 MS5H	262.7622	273.1099	262.8337	273.1411	262.8999	273.1716	262.9700	262.9700	262.9700
95 M4H	97.51042	97.71919	97.53836	97.74677	97.56557	97.77390	97.59319	97.59319	97.59319

\*\*\*\*\*

BROWD

1 TIME	9.140957	9.241402	9.341846	9.442291	9.542736	9.643181	9.743626	9.743626	9.743626
2 ADIFF	70.00000	70.00000	70.00000	70.00000	70.00000	70.00000	70.00000	70.00000	70.00000
3 ALINR	100.0000	100.0000	100.0000	100.0000	100.0000	100.0000	100.0000	100.0000	100.0000
4 DFAMB	-1.997693	-1.1563779	-1.1533871	-1.154514	-1.1071100	-1.7457274E-01	-1.6093108E-01	-1.6093108E-01	-1.6093108E-01
5 DPT35	-81372.75	-71155.75	-81346.19	-71170.06	-81321.25	-71184.56	-81297.50	-81297.50	-81297.50
6 DPT4	76305.56	72149.25	76256.25	72118.87	76208.75	72088.75	76160.56	76160.56	76160.56
7 DTT35	-174410.2	-168177.6	-174215.2	-167999.8	-174025.2	-167824.1	-173834.8	-173834.8	-173834.8
8 DTT4	125535.4	120025.9	125423.8	119943.5	125319.4	119862.0	125212.1	125212.1	125212.1
9 FAMB	.1314207E-02	.1158783E-02	.1007585E-02	.8527846E-03	.7010929E-03	.5469141E-03	.3947297E-03	.3947297E-03	.3947297E-03
10 FAS	.0	.0	.0	.0	.0	.0	.0	.0	.0
11 FAS5	.0	.0	.0	.0	.0	.0	.0	.0	.0
12 FPMP	18.00000	18.00000	18.00000	18.00000	18.00000	18.00000	18.00000	18.00000	18.00000
13 GAMB	1.400000	1.400000	1.400000	1.400000	1.400000	1.400000	1.400000	1.400000	1.400000
14 GAMB5	1.400000	1.400000	1.400000	1.400000	1.400000	1.400000	1.400000	1.400000	1.400000
15 GAMB4	1.400000	1.400000	1.400000	1.400000	1.400000	1.400000	1.400000	1.400000	1.400000
16 IUPDAT	.0	.0	.0	.0	.0	.0	.0	.0	.0
17 PRDIFF	1.046000	1.060404	1.045983	1.060329	1.045970	1.060257	1.045954	1.045954	1.045954
18 PRLINR	1.204933	1.190989	1.204876	1.190994	1.204826	1.191001	1.204773	1.204773	1.204773
19 PTS	250.0000	250.0000	250.0000	250.0000	250.0000	250.0000	250.0000	250.0000	250.0000
20 PTS5	239.0058	235.7592	239.0093	235.7757	239.0125	235.7919	239.0162	239.0162	239.0162

21 PT4	198.3560	197.9525	198.3683	197.9654	198.3792	197.9778	198.3910
22 R3	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000
23 R35	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000
24 R4	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000
25 TT3	1000.0000	1000.0000	1000.0000	1000.0000	1000.0000	1000.0000	1000.0000
26 TT35	1284.415	1287.360	1283.943	1286.876	1283.479	1286.395	1283.012
27 TT4	1332.765	1332.250	1332.198	1331.682	1331.633	1331.119	1331.069
28 MFMB	.2384205	.2105169	.1826134	.1547098	.1268063	.9890270E-01	.7099915E-01
29 M3	124.4309	140.5811	124.4143	140.5063	124.3973	140.4310	124.3759
30 M35	273.2046	263.0361	273.2344	263.1079	273.2661	263.1790	273.2971
31 M4	97.80051	97.62038	97.82733	97.64760	97.85350	97.67435	97.88002

OUTLAR

32 CV35P	.1709999	.1709999	.1709999	.1709999	.1709999	.1709999	.1709999
33 DPT35	-81372.75	-71155.75	-81346.19	-71170.06	-81321.25	-71184.56	-81297.50
34 DTT35	-174410.2	-168177.6	-174215.2	-167999.8	-174025.2	-167824.1	-173834.8
35 FAMB	.1314207E-02	.1158783E-02	.1007585E-02	.8527846E-03	.7018929E-03	.5469141E-03	.3947297E-03
36 H3	240.0000	240.0000	240.0000	240.0000	240.0000	240.0000	240.0000
37 H35	308.2595	308.9663	308.1462	308.8501	308.8347	308.7349	307.9229
38 H4	319.8435	319.7400	319.7275	319.6038	319.5918	319.4685	319.4565
39 PT35	239.0058	235.7592	239.0093	235.7757	239.0125	235.7919	239.0162
40 QDBANC	.0	.0	.0	.0	.0	.0	.0
41 RHO35	.2907522E-03	.2861465E-03	.2908632E-03	.2862741E-03	.2909724E-03	.2864008E-03	.2910828E-03
42 R35	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000	640.0000
43 TT3	1000.0000	1000.0000	1000.0000	1000.0000	1000.0000	1000.0000	1000.0000
44 TT35	1284.415	1287.360	1283.943	1286.876	1283.479	1286.395	1283.012
45 TT4	1332.765	1332.250	1332.198	1331.682	1331.633	1331.119	1331.069
46 U35	219.6350	220.1385	219.5543	220.0558	219.4748	219.9756	219.3951
47 M3	124.4309	140.5811	124.4143	140.5063	124.3973	140.4310	124.3759
48 M35	273.2046	263.0361	273.2344	263.1079	273.2661	263.1790	273.2971

BURNER

49 CV4P	.1709999	.1709999	.1709999	.1709999	.1709999	.1709999	.1709999
50 DELAY	75673.37	78757.12	81832.12	84898.44	87957.12	91003.75	94042.81
51 DFAMB	-.1997693	-.1563779	-.1533871	-.1154514	-.1071100	-.7457274E-01	-.6093108E-01
52 DLYP	.0	.0	.0	.0	.0	.0	.0
53 DLYPRP	.0	.0	.0	.0	.0	.0	.0
54 DMAIR	175.5325	165.5287	175.5055	165.5435	175.4812	165.5580	175.4556
55 DMFUEL	.1100591	.9752715E-01	.8414328E-01	.7150847E-01	.5825001E-01	.4551241E-01	.3237822E-01
56 DPT4	76305.56	72149.25	76256.25	72118.87	76208.75	72088.75	76160.56
57 DTBIDL	.0	.0	.0	.0	.0	.0	.0
58 DTBLAG	.0	.0	.0	.0	.0	.0	.0
59 DTBURN	.0	.0	.0	.0	.0	.0	.0
60 DTT4	125535.4	120025.9	125423.8	119943.5	125319.4	119862.0	125212.1
61 ETAB	.0	.0	.0	.0	.0	.0	.0
62 FAO	.2704959E-02	.2704959E-02	.2704959E-02	.2704959E-02	.2704959E-02	.2704959E-02	.2704959E-02
63 FAP	.3490724E-02	.3201338E-02	.2673359E-02	.2352036E-02	.1856158E-02	.1503200E-02	.1039149E-02
64 FAPA	.3490724E-02	.3201338E-02	.2673359E-02	.2352036E-02	.1856158E-02	.1503200E-02	.1039149E-02
65 H35	308.2595	308.9663	308.1462	308.8501	308.8347	308.7349	307.9229
66 H4	319.8435	319.7400	319.7275	319.6038	319.5918	319.4685	319.4565
67 H4SS	308.2595	308.9663	308.1462	308.8501	308.8347	308.7349	307.9229
68 IUPDAT	.0	.0	.0	.0	.0	.0	.0
69 MBLITE	.0	.0	.0	.0	.0	.0	.0



FILE: FILE OUTPUT A PRATT AND WHITNEY AIRCRAFT

FUEL/AIR RATIO IN DELTAT LESS THAN ZERO  
 FUEL/AIR RATIO IN DELTAT LESS THAN ZERO

\*\*\*\*\*

BRNDL

1 TIME	9.844070	9.944515	10.04496
2 ADIFF	70.00000	70.00000	70.00000
3 ALINR	100.0000	100.0000	100.0000
4 DFAMB	-.3373873E-01	-.1485227E-01	.7044759E-02
5 DPT35	-71194.87	-81271.87	-71213.37
6 DPT4	72057.19	76112.31	72028.19
7 DTT35	-167647.8	-173644.3	-167475.2
8 DTT4	119781.0	125104.7	119700.1
9 FAMB	.2411665E-03	.8849695E-04	-.6445586E-04
10 FA3	.0	.0	.0
11 FA35	.0	.0	.0
12 FPHPT	18.00000	18.00000	18.00000
13 GAMB	1.400000	1.400000	1.400000
14 GAM35	1.400000	1.400000	1.400000
15 GAM4	1.400000	1.400000	1.400000
16 IUPDAT	.0	.0	.0
17 PRODIFF	1.060192	1.045938	1.060111
18 PRLINR	1.191004	1.204721	1.191015
19 PT3	250.0000	250.0000	250.0000
20 PT35	235.8062	239.0198	235.8242
21 PT4	197.9894	198.4026	198.0026
22 R3	640.0000	640.0000	640.0000
23 R35	640.0000	640.0000	640.0000
24 R4	640.0000	640.0000	640.0000
25 TT3	1000.000	1000.000	1000.000
26 TT35	1285.920	1282.549	1285.443
27 TT4	1330.556	1330.510	1329.999
28 MFMB	.4309559E-01	.1519203E-01	-.1271152E-01
29 M3	140.3630	124.3592	140.2807
30 M35	263.2456	273.3271	263.3196
31 M4	97.70074	97.90631	97.72772
- OUTLNR			
32 CV35P	.1709999	.1709999	.1709999
33 DPT35	-71194.87	-81271.87	-71213.37
34 DTT35	-167647.8	-173644.3	-167475.2
35 FAMB	.2411665E-03	.8849695E-04	-.6445586E-04
36 M3	240.0000	240.0000	240.0000
37 M35	308.6208	307.8118	308.5061
38 M4	319.3333	319.3223	319.1995

FILE: FILE OUTPUT A PRATT AND WHITNEY AIRCRAFT

39 PT35	235.8062	239.0198	235.8242
40 QDBNCS	.0	.0	.0
41 RH035	.2865240E-03	.2911922E-03	.2866525E-03
42 R35	640.0000	640.0000	640.0000
43 TT3	1000.000	1000.000	1000.000
44 TT35	1285.920	1282.549	1285.443
45 TT4	1330.556	1330.510	1329.999
46 U35	219.8923	219.3159	219.8107
47 M3	140.3630	124.3592	140.2807
48 M35	263.2456	273.3271	263.3196

BURNER

49 CV4P	.1709999	.1709999	.1709999
50 DELAY	97073.62	100094.7	101744.1
51 DFAYB	-.3373073E-01	-.1485227E-01	.7044759E-02
52 DLYP	.0	.0	.0
53 DLYPRP	.0	.0	.0
54 DHAIR	165.5683	175.4294	165.5856
55 DMFUEL	.1953910E-01	.6528389E-02	-.6411999E-02
56 DPT4	72057.19	76112.31	72028.19
57 DTBIDL	.0	.0	.0
58 DTBLAG	.0	.0	.0
59 DTBURN	.0	.0	.0
60 DTT4	119781.0	125104.7	119700.1
61 ETAB	.0	.0	1.007519
62 FAO	.2704959E-02	.2704959E-02	-.4827413E-04
63 FAP	.6548346E-03	.223274E-03	-.1930965E-03
64 FAPA	.6548346E-03	.223274E-03	-.1930965E-03
65 H35	308.6208	307.8118	308.5061
66 H4	319.3333	319.3223	319.1995
67 H4SS	308.6208	307.8118	308.5061
68 IUPDAT	.0	.0	.0
69 MBLITE	.0	.0	2.000000
70 PCBURN	.0	.0	1.000000
71 PT4	197.9894	198.4026	198.0026
72 QDBURN	.0	.0	.0
73 QDBNLP	.0	.0	.0
74 RFIASS	.6045081	.6057906	.6048019
75 R4	640.0000	640.0000	640.0000
76 TEND	.0	.0	.0
77 TIME	9.844070	9.944515	10.04496
78 TLITE	6.778255	6.778255	10.04496
79 TT35	1285.920	1282.549	1285.443
80 TT4	1330.556	1330.510	1329.999
81 U4	227.5251	227.5172	227.4297
82 MFMB	.4309559E-01	.1519203E-01	-.1271152E-01
83 M35	263.2456	273.3271	263.3196
84 M4	97.70074	97.90631	97.72772

SENSED PARAMETERS

85 PT3H	250.0000	250.0000	250.0000
86 PT35H	235.8062	239.0198	235.8242
87 PT4H	197.9894	198.4026	198.0026

FILE: FILE OUTPUT A PRATT AND WHITNEY AIRCRAFT

88 QBURNH	-12382.39	-12959.66	-12360.04
89 TT3H	1000.000	1000.000	1000.000
90 TT35H	1285.920	1282.549	1285.443
91 TT4H	1330.556	1330.510	1329.999
92 MFBHM	.4309559E-01	.1519203E-01	-.1271152E-01
93 M3H	140.3630	124.3592	140.2807
94 M35H	263.2456	273.3271	263.3196
95 M4H	97.70074	97.90631	97.72772

1

\*\*\* STATISTICS IN WRIT04 \*\*\*

TEMPORARY FILE FOR CREATING CAPAFILE CAPAFILE HAS BEEN WRITTEN WITH 71 VARIABLES AND 101 DATA POINTS.

THE MENU FOR THE CAPA DATA SET IS:

TIME	ADIFF	ALINR	CV35P	CV4P	DELAY	DFAMB	DLYPRP	DMAIR
DHFUEL	DPT35	DPT4	DTBIDL	DTBLAG	DTBURN	DTT35	ETAB	FAMB
FAO	FAP	FAPA	FA3	FA35	FPHPT	GAM3	GAM4	H3
H35	H4	M4SS	IUPDAT	MBLITE	PCBURN	PROIFF	PT3	PT3H
PT35	PT35H	PT4	PT4H	TLITE	QBLURN	QDBNCS	RHO35	RMASS
R3	R35	R4	TEND	TTITE	TT3	TT3H	TT35H	TT4
TT4H	U35	U4	MFBM	MFBHM	M3	M3H	M35H	M4
M4H								

1	SUMMARY OF ERRORS FOR THIS JOB	ERROR NUMBER	NUMBER OF ERRORS
0		208	4

APPENDIX B

SAMPLES OF STATE, STATIC, AND MEAS SUBROUTINES

```

SUBROUTINE STATIC(X,P,Y,U)
C*****
C SUBPROGRAM STATIC -- STEADY STATE BALANCE FOR SIMPLE BURNER MODEL
C
C INPUTS:  U = ARRAY OF INPUTS
C          P = ARRAY OF TWIDDLE FACTORS
C
C OUTPUT:  X = ARRAY OF STATES
C
C COMMENTS: NONE
C*****
      implicit double precision (a-h)
      implicit double precision (o-z)
      DIMENSION X(1),U(1),P(1)
      DIMENSION YSTATE(5),DYDX(5)
      DOUBLE PRECISION NAME(5)
      DIMENSION DELTAX(5), DXALOW(5), TOL(5)
      EQUIVALENCE (YSTATE(1), PT35),(YSTATE(2), TT35),(YSTATE(3), PT4),
      * (YSTATE(4), TT4),(YSTATE(5), FAMB)
      EQUIVALENCE ( DYDX(1),DPT35),( DYDX(2),DTT35),( DYDX(3),DPT4),
      * ( DYDX(4), DTT4),( DYDX(5),DFAMB)
      DATA NAME /
      1 'PT35', 'TT35', 'PT4', 'TT4', 'FAMB' /
C*****
C EXTRACT INPUTS FROM ARRAY *
C*****
      PT3U =U(1)
      TT3U =U(2)
      WFMBU=U(3)
C
      AP3  =P(6)
      BP3  =P(7)
      AT3  =P(12)
      BT3  =P(13)
      AWP  =P(21)
      BWP  =P(22)
C
C SENSOR MODELS FOR INPUTS ARE MOVED HERE SO THAT THE ERROR EFFECTS
C DRIVE THE MODEL - AS THEY WILL IN TESTING.
C
      PT3  = AP3 * PT3U + BP3
      TT3  = AT3 * TT3U + BT3
C*****
C MEASURED FUEL FLOW *
C*****
      WFMB = AWP*WFMBU+BWP
C
C*****
C INITIALIZATION FOR STATE VARIABLES *
C*****
      PT35=.97*PT3
      TT35=TT3
      PT4  =.94*PT3
      TT4  =TT3+1500.
      if( p(39) .eq. 0.0) tt4 = tt3
      FAMB=.02
C*****
C MISCELANIOUS FLAGS *
C*****
      TIME=0.

```

```

        IWAY=0
        NEQ=5
        TOLE=.10
        PCTDX=.001
        XBOUND=.1
        MXPASS=100
        ISSPNT=0
        dt = p(38) -
C*****
C  STEADY STATE BALANCE  *
C*****
10  CALL BRNMDL(0      ,-1      ,TIME, YSTATE,U,P,DYDX,Y,DT)
    CALL SSBL01(NEQ    , ISSPNT, YSTATE, DYDX  , IWAY  , TOLE  ,
    *           PCTDX  , XBOUND, MXPASS, NAME  , XBOUND, MXPASS,
    *           NAME   , NPASS )
    GO TO (10,10,15,11,13),IWAY
11  WRITE(6,12)
12  FORMAT(1X,' STEADY STATE BALANCE FAILED --- '/
    *      1X,' SINGULAR JACOBIAN' )
    STOP
13  WRITE(6,14)
14  FORMAT(1X,'STEADY STATE BALANCE FAILED WITH MAX. PASSES')
    STOP
15  CONTINUE
C*****
C  PUT STATES INTO X ARRAY  *
C*****
    DO 20 I=1,NEQ
20  X(I)=YSTATE(I)
    RETURN
    END

```

```

      SUBROUTINE STATE(X,U,P,XDOT,TIME)
C*****
C SUBPROGRAM STATE -- CALCULATED DERIVITIVES OF STATE VARIABLES
C
C INPUTS: IUPDAT = LAG/MODE UPDATE FLAG
C           -1 DURING STEADY STATE BALANCE OR ON FIRST CALL
C           0 DURING CONVERGENCE ATTEMPTS
C           1 AFTER COMPLETED DT
C
C           TIME = TIME
C           X = ARRAY OF STATES
C           U = ARRAY OF INPUTS
C           P = ARRAY OF DIDDLE FACTORS
C
C OUTPUTS: XDOT = ARRAY OF DERIVITIVES
C*****
      implicit double precision (a-h)
      implicit double precision (o-z)
      COMMON /dLAG/ iupdat
      COMMON /CLAG/ IPROP
      DIMENSION X(1),U(1),P(1),XDOT(1)
      IPRPL=0
      dt = p(38)
      CALL BRNMDL(IPRPL,IUPDAT,TIME,X,U,P,XDOT,Y,DT)
      RETURN
      END

```

```

SUBROUTINE MEAS(X,U,P,XDOT,Y)
C*****
C SUBPROGRAM MEAS -- LOADS OUTPUTS INTO Y ARRAY
C
C INPUTS:  TIME = TIME
C           X = ARRAY OF STATES
C           U = ARRAY OF INPUTS
C           P = ARRAY OF DIDDLE FACTORS
C
C OUTPUTS:  Y = OUTPUT ARRAY
C
C COMMENTS:
C 1. ANY AVAILIABLE PARAMETER CAN BE OUTPUT BE CALLING HEADLD
C    WITH THE PARAMETER.
C 2. ANY SYNTHESISED PARAMETER SHOULD BE CALCULATED IN THE
C    APPROPRIATE SUBROUTINE AND PASSED TO PRPL.
C 3. PARAMETERS ARE ARRANGED IN THE Y ARRAY IN THE SAME ORDER
C    AS THE CALLS TO HEADLD.
C*****
      implicit double precision (a-h)
      implicit double precision (o-z)
      DOUBLE PRECISION HEADER
      INCLUDE 'DUA2:[UTPWA.SCIDNT.SPL]NLMCOM.INC'
      COMMON/CLAG/ IPROP
      COMMON/PRPLBS/HEADER(100),VAROUT(100),NLOAD
      DIMENSION X(1),XDOT(1),U(1),P(1),Y(1)
      DATA IFIRST/0/
C
C
C LOAD NAMES OF OUTPUT VARIABLES
      IF(IFIRST.EQ.1)GO TO 1
      IFIRST=1
      CALL HEADLD('W3M      ')
      CALL HEADLD('PT35M    ')
      CALL HEADLD('PT4M     ')
      CALL HEADLD('TT4M     ')
      CALL HEADLD('W4M      ')
C
C LOAD Y ARRAY
1  CONTINUE
      CALL BRNMDL(1,0,TIME,X,U,P,XDOT,Y,DT)
C*****
C***** REMOVE WRITE LOOP BEFORE RUNNING TRANSIENTLY *****
C*****
      WRITE(6,*)IPROP,Y(3)
C %END SCT
      RETURN
      END

```

# APPENDIX C

## SAMPLE PARSEL OUTPUT

FOLLOWING PARAMETERS WILL BE USED AS ESTIMATE PARAMETERS

PARAMETER NO.	1	2	3	4	5	20	10	14	41
LABEL	PPHPT	ADIFF	ALIMR	VOLJ3	VOLJ	CSTABE	CSTABC	CEP7B	deloic
APRIORI UNCERT	1.000E+00	1.000E+06							
UNCERTAINTY	1.000E+00	7.000E+00	1.000E+01	2.500E+02	2.000E+02	1.000E-01	1.000E-01	4.000E-06	2.000E+00

FOLLOWING PARAMETERS WILL BE CONSIDERED GUIDANCE PARAMETERS

PARAMETER NO.	31	32
LABEL	CTAN	CTANM
UNCERTAINTY	6.000E-06	1.000E-01

FOLLOWING MEASUREMENT SET WILL BE USED IN THIS STUDY

MEAS NO.	1	2	3	4	5
LABEL	Y01	Y02	Y03	Y04	Y05
NOISE	3.00000E-01	1.50000E+00	1.00000E+00	1.00000E+01	3.00000E-01

PERFORMANCE INDEX :

$$J = v_1^2 + v_2^2 + v_3^2 + \dots + v_n^2$$

where  $v_i$  = weighting on  $i$ th component performance index,  $J_i$

$J_i$  =  $i$ th component of performance index given by:

$$J_i = d_i^T \cdot \text{delta}(\theta) + d_i^T \cdot \text{delta}(\phi)$$

where  $T(\theta, \phi)$  = vector of performance measure variables of the system

$\theta$  = set of estimate parameters

$\phi$  = set of nuisance parameters

$\text{delta}(\theta)$  = error in estimate parameters given by:

$$\text{delta}(\theta) = \text{covariance}(\theta) \cdot \text{nuisance error} \\ = \text{sqrt} \langle \text{diag}(\text{inv}(M11)) \rangle + \text{inv}(M11) \cdot M12 \cdot \text{delta}(\phi)$$

$\text{delta}(\phi)$  = error in nuisance parameters (nuisance level)

SUMMARY OF RUN  
OPTIMUM PERFORMANCE INDEX FOR 1 ESTIMATE PARAMETERS

performance index	number of estimate parameters	estimate parameters	number of nuisance parameters	nuisance parameters
1 349.22	1	1	11	2 3 4 5 29 30
2 350.48	1	41	11	33 34 41 31 32
3 376.22	1	4	11	1 2 3 4 5 29
				30 33 34 31 32
				1 2 3 5 29 30
				33 34 41 31 32

SUMMARY OF RUN  
OPTIMUM PERFORMANCE INDEX FOR 2 ESTIMATE PARAMETERS

performance index	number of estimate parameters	estimate parameters	number of nuisance parameters	nuisance parameters
1 47.664	2	1 41	10	2 3 4 5 29 30
2 295.46	2	5 41	10	33 34 31 32
3 339.53	2	1 2	10	1 2 3 4 29 30
				33 34 31 32
				3 4 5 29 30 33
				34 41 31 32

SUMMARY OF RUN  
OPTIMUM PERFORMANCE INDEX FOR 3 ESTIMATE PARAMETERS

	performance index	number of estimate parameters	estimate parameters	number of nuisance parameters	nuisance parameters
1	37.923	3	1 4 41	9	2 3 5 29 30 33 34 31 32
2	40.260	3	1 30 41	9	2 3 4 5 29 33 34 31 32
3	42.389	3	1 5 41	9	2 3 4 29 30 33 34 31 32

SUMMARY OF RUN  
OPTIMUM PERFORMANCE INDEX FOR 4 ESTIMATE PARAMETERS

Performance index	number of estimate parameters	estimate parameters	number of nuisance parameters	nuisance parameters
1 40.960	4	1 2 4 41	0	3 5 29 30 33 34 31 32
2 43.010	4	1 2 30 41	0	3 4 5 29 33 - 34 31 32
3 47.934	4	1 2 5 41	0	3 4 29 30 33 34 31 32

SUMMARY OF RUN  
OPTIMUM PERFORMANCE INDEX FOR 5 ESTIMATE PARAMETERS

Performance index	number of estimate parameters	estimate parameters	number of nuisance parameters	nuisance parameters
1 67.902	5	1 2 3 4 41	7	5 29 30 33 34 31
2 69.087	5	1 2 3 5 41	7	32 4 29 30 33 34 31
3 70.178	5	1 2 3 30 41	7	32 4 5 29 33 34 31

MEASUREMENT CONTRIBUTION TO SECOND DERIVATIVE OF THE COST FUNCTION

PARAMETER NO.	Y01	Y02	Y03	Y04	Y05
FRPT	7.3229E+01	3.7852E-02	1.0716E-01	1.4445E+02	1.0766E+02
ADIFF	5.2634E-04	7.9444E-04	8.2697E-04	2.0979E-03	1.5240E-03
ALIMR	6.1955E-04	1.0306E-06	2.4007E-04	2.4164E-03	6.1204E-04
VOL35	1.3690E-04	2.1006E-06	2.1997E-06	9.3003E-06	4.2605E-06
VOLA	9.5045E-05	2.2072E-06	9.3310E-06	4.3797E-05	1.1092E-06
CSTARS	3.1750E-02	1.7336E-05	8.0234E-05	5.6230E-01	1.0400E-02
CSTABC	2.0059E-02	1.7295E-07	2.1834E-10	2.7359E-04	1.0909E-05
CTAN	5.4074E+04	1.3064E+01	4.9269E+01	2.9699E+05	1.0179E+06
CTAMN	1.0025E+00	8.0000E-04	3.7097E-03	2.5727E+01	8.6243E-01
CEPPA	2.0509E-06	5.1032E-06	1.2200E-06	4.1700E-03	1.1363E-04
CEPPB	1.5051E-01	7.0117E-05	4.4653E-04	9.3366E+00	3.0362E-01
delocE	1.0035E+03	1.1407E+00	2.9907E+00	8.7057E+04	4.0703E+03



ESTIMATION PARAMETER SET

PARAMETER NO. 1 2 4 41  
 LABEL PPMPT ADIFF VOL15 deloff  
 APRIORI UNCERT 1.000E+06 1.000E+06 1.000E+06 1.000E+06

NUISANCE PARAMETER SET

3 : ALIBR 5 : VOL4 6 : AP3 7 : BP3 8 : AP35 9 : BP35 10 : AP4 11 : BP4  
 12 : AT3 13 : BT3 14 : AT4 15 : BT4 16 : TAUT4 17 : AMP 18 : BWP 19 : CSTABE  
 20 : CSTABC 21 : CTAD 22 : CTAM 23 : CEPA 24 : CEPA 25 : TAUP3 26 : TAUP35 27 : TAUP4  
 28 : betime

MEASUREMENT SET

MEAS NO. 1 2 3 4 5  
 LABEL Y01 Y02 Y03 Y04 Y05  
 NOISE 5.00000E-01 1.50000E+00 1.50000E+00 1.00000E+01 5.00000E-01

M11 MATRIX

	1	2	3	4
1	9.1631E+04	3.5264E+01	-3.2614E+01	-1.0055E+06
2	3.5264E+01	4.0579E+00	1.6827E-01	-2.0101E+03
3	-3.2614E+01	1.6827E-01	5.2095E-02	1.0007E+02
4	-1.0055E+06	-2.0101E+03	1.0007E+02	1.6310E+07

M12 MATRIX

	1	2	3	4	5	6	7	8	9	10
1	-3.0711E+02	-0.7009E+01	-2.9508E+07	-7.2677E+04	-3.0796E+07	-7.7964E+04	3.0922E+03	1.3085E+03	-1.0670E+05	-3.3539E+02
2	1.0642E+00	1.0306E-01	1.8517E+05	4.3557E+02	1.8235E+02	4.2431E+02	2.4190E+03	1.0306E+01	-2.0090E+03	-1.8004E+00
3	2.5067E-01	5.3132E-02	3.8092E+04	9.2226E+01	3.9010E+04	1.0113E+02	1.6639E+02	7.8243E-02	-2.0006E+02	-1.8202E-01
4	2.7139E+03	9.0712E+02	1.1507E+06	2.6795E+05	1.3023E+06	3.2404E+05	-1.0137E+06	-0.1075E+03	5.4254E+06	7.4964E+03

M12 MATRIX

	11	12	13	14	15	16	17	18	19	20
1	-9.6952E+05	-3.2504E+02	-0.5477E+05	-7.8200E+05	-2.0320E+05	-4.4634E+03	-0.1359E+01	-3.0013E+06	-2.6473E+04	-6.7191E+02
2	-3.2063E+03	-1.0609E+00	2.9097E+03	-1.5363E+03	-5.5615E+02	3.7934E+00	6.5774E-02	2.6664E+03	2.2390E+01	-7.4770E-02
3	-2.8607E+01	-9.6522E-03	-0.2209E+01	7.8431E+01	3.1195E+01	3.8495E+00	7.6620E-02	2.6807E+03	2.4036E+01	5.5742E-03
4	1.7247E+07	5.7403E+03	2.4202E+06	1.2606E+07	4.7209E+06	3.5250E+04	5.8400E+02	2.3790E+07	1.9669E+05	9.8007E+02

M12 MATRIX

	21	22	23	24	25
1	0.5293E+03	-6.3902E+07	-7.4090E+07	2.3372E+06	-5.3720E+07
2	1.4695E+01	3.0300E+05	3.7452E+05	9.1000E+03	-3.0140E+04
3	-1.5750E+00	1.0764E+05	1.1151E+05	1.9267E+02	2.7600E+04
4	-1.3100E+05	1.6929E+00	2.7640E+00	-1.3065E+07	6.0760E+00

BIAS MATRIX

	1	2	3	4	5	6	7	8	9	10
1	-1.1103E-03	-4.4117E-04	-2.0037E-04	-4.2039E-01	-2.1621E+02	-4.0007E-01	1.2065E+01	5.8029E-02	1.7351E+00	6.7901E-04
2	1.0371E-01	1.6209E-02	2.1307E+04	4.1553E+01	2.0902E+04	3.8741E+01	6.4074E+02	2.7601E+00	-1.9373E+02	-2.6950E-02
3	3.0225E+00	6.3022E-01	5.5563E+05	1.5060E+03	5.4924E+05	1.5159E+03	4.7076E+03	2.2222E+01	-2.9240E+03	-3.9407E+00
4	9.9361E-05	2.6510E-05	-6.1079E-05	-1.4100E-02	-6.1340E+00	-1.4023E-02	7.3227E-01	3.2032E-03	4.3302E-01	5.2263E-04

BIAS MATRIX

	11	12	13	14	15	16	17	18	19	20
1	1.5920E+00	3.6073E-04	-4.5107E+01	-1.1740E-02	-4.7350E-02	-2.7734E-02	-5.1375E-04	-2.0112E+01	-1.7072E-01	-6.3259E-04
2	-1.9374E+02	-6.1002E-02	1.1066E+03	7.5055E+00	6.2910E+00	-1.3641E+00	-3.5500E-02	-9.8448E+02	-9.9018E+00	1.5320E-02
3	-1.1143E+03	-4.6001E-01	2.7745E+04	-1.3232E+01	-0.3479E+00	6.0195E+01	3.2607E+00	4.1070E+04	3.8490E+02	-2.0001E-01
4	1.3306E+00	3.6906E-04	-2.3290E+00	7.7012E-01	2.8705E-01	-0.0500E-05	-0.0437E-06	-1.6112E-01	-2.0639E-03	3.7046E-05

	21	22	23	24	25
1	0.4109E-04	2.1110E+01	-9.9421E+01	1.0209E+02	-2.5305E+02
2	3.3510E-01	1.1253E+04	9.1713E+03	1.9987E+03	-1.4231E+04
3	-1.5201E+01	2.0110E+06	2.0190E+06	4.9887E+04	3.7887E+05
4	-7.8903E-03	6.2920E-01	-5.3490E-01	5.4304E+00	1.7512E+01

MEASUREMENT CONTRIBUTION TO SECOND DERIVATIVE OF THE COST FUNCTION

PARAMETER NO.	Y01	Y02	Y03	Y04	Y05
FPHT	2.970E+04	1.6397E+01	4.7036E+01	3.7540E+04	2.4319E+04
ADIFF	2.3147E+00	3.244E-01	3.2852E-01	5.2537E-01	5.6406E-01
ALINE	2.1807E+00	6.750E-04	1.1046E-01	2.4928E+00	1.2288E-01
VOLIS	5.2701E-02	1.675E-05	1.6059E-05	4.1162E-05	3.0294E-05
VOL4	5.7899E-02	1.9302E-05	7.2045E-05	0.0268E-02	3.5176E-03
AP3	3.4560E+10	4.4292E+06	4.3190E+06	9.0123E+06	7.0214E+06
BP3	2.0026E+05	0.5707E+01	0.3519E+01	1.1393E+02	1.3565E+02
AP35	3.4960E+10	4.5016E+06	0.0002E+00	0.0002E+00	0.0002E+00
BP35	2.1651E+05	0.0002E+00	0.0002E+00	0.0002E+00	0.0002E+00
AP4	0.0000E+00	0.0000E+00	4.4393E+06	0.0000E+00	4.3120E+06
BP4	0.0000E+00	0.0000E+00	0.8892E+01	0.0000E+00	0.6075E+01
AT3	0.1400E+03	3.7409E+02	0.7973E+02	2.2103E+06	1.5706E+03
BT3	3.2504E+00	2.1902E-04	5.2577E-04	3.7536E+00	2.3724E-01
AT4	0.0000E+00	0.0000E+00	0.0000E+00	1.7932E+07	1.0620E+06
BT4	0.0000E+00	0.0000E+00	0.0000E+00	2.0000E+00	1.2037E-01
TAUT4	0.0000E+00	0.0000E+00	0.0000E+00	3.9615E+08	2.0392E+07
AWF	4.3421E+03	3.2413E+02	0.2490E+02	0.9460E+06	4.9621E+03
BWF	6.0015E+04	5.1916E+01	1.2917E+02	1.2365E+06	6.0794E+04
CSTABE	4.3563E+02	1.2007E-01	5.4077E-01	3.8099E+02	1.6081E+01
CSTABC	3.0444E-01	6.0410E-05	2.0302E-04	1.7460E-01	7.3234E-03
CTAU	2.1790E+00	4.0070E+04	2.1430E+05	1.9965E+08	7.9230E+06
CTAUN	1.7617E+04	5.2002E+00	2.1775E+01	1.4539E+04	6.1204E+02
CEPPA	2.5104E-01	5.5479E-05	9.0475E-05	5.7692E-02	2.0032E-03
CEPPB	6.9230E+01	2.5095E-02	0.2657E-02	9.9136E+02	5.4070E+01
TAUP3	2.2002E+11	4.0079E+08	3.6492E+08	7.7806E+08	7.5913E+08
TAUP35	2.3777E+11	4.6717E+08	0.0000E+00	0.0000E+00	0.0000E+00
TAUP4	0.0000E+00	0.0000E+00	4.5097E+08	0.0000E+00	4.7319E+08
deloff	6.3041E+05	4.4072E+02	1.1541E+03	1.4056E+07	0.2171E+05
botino	2.4515E+10	0.1313E+06	2.9275E+07	4.6931E+10	2.1617E+09

RUN SUMMARY

ESTIMATION				NOISANCE		
PARAMETER NO.	LABEL	STANDARD ERRORS	ESTIMATION BIAS	PARAMETER NO.	LABEL	LEVEL
1	FPHT	7.031E-03	4.073E+00	3	ALNR	1.000E+01
2	ADIV	5.603E-01	2.807E+02	5	VOLA	2.600E+02
4	VOL35	6.222E+00	8.030E+03	6	AP3	2.000E-03
41	401off	5.271E-04	2.729E-01	7	BP3	5.000E-01
				8	AP35	2.000E-03
				9	BP35	5.000E-01
				10	AP4	2.000E-03
				11	BP4	5.000E-01
				12	AT3	3.000E-03
				13	BT3	3.000E+00
				14	AT4	3.000E-03
				19	BT4	3.000E+00
				20	TAUT4	5.000E-05
				21	AMP	2.500E-02
				22	BWP	7.000E-02
				29	CSTABE	1.000E-01
				30	CSTABC	1.000E-01
				31	CTAU	6.000E-04
				32	CTAUM	1.000E-01
				33	CSPPA	4.000E-06
				34	CSPPB	4.000E-04
				35	TAUP3	5.000E-05
				36	TAUP35	5.000E-05
				37	TAUP4	5.000E-05
				42	bottom	1.000E-02



ESTIMATION PARAMETER SET

PARAMETER NO. 1 2 4 41  
 LABEL PPUFT ADIFF VOLJS GLOFF  
 PRIORI UNCERT 1.000E+06 1.000E+06 1.000E+06 1.000E+06

MISSANCE PARAMETER SET

3 : ALJMB 5 : VOLG 6 : AP3 7 : BP3 8 : AP35 9 : BP35 10 : AP4 11 : BP4  
 12 : AP3 13 : BP3 18 : AP4 19 : BT4 20 : TAUTA 21 : AMF 22 : DMF 29 : CSTABE  
 30 : CSTABC 31 : CTAM 32 : CTAMH 33 : CEFFA 34 : CEFFB 35 : TAMP3 36 : TAMP35 37 : TAMP4  
 42 : DOLIMO

MEASUREMENT SET

MEAS NO. 1 2 3 4 5  
 LABEL Y01 Y02 Y03 Y04 Y05  
 NOISE 5.00000E-01 1.50000E+00 1.50000E+00 1.00000E+01 5.00000E-01

M11 MATRIX

	1	2	3	4
1	9.1631E+04	3.5264E+01	-3.2614E+01	-1.0055E+06
2	3.5264E+01	4.0579E+00	1.6027E-01	-2.0101E+03
3	-3.2614E+01	1.6027E-01	3.2005E-02	1.0007E+02
4	-1.0055E+06	-2.0101E+03	1.0007E+02	1.6310E+07

M12 MATRIX

	1	2	3	4	5	6	7	8	9	10
1	-3.0711E+02	-0.7009E+01	-2.9508E+07	-7.2677E+04	-3.0796E+07	-7.7064E+04	3.0922E+05	1.3005E+03	-1.0070E+05	-3.3539E+02
2	1.0042E+00	1.0048E-01	1.0517E+05	4.3557E+02	1.0235E+05	4.2403E+02	2.4198E+03	1.0106E+01	-2.0000E+03	-1.0000E+00
3	2.5067E-01	3.3132E-02	3.0009E+04	9.7226E+01	3.9010E+04	1.0113E+02	1.6639E+01	7.0243E-02	-2.0006E+02	-1.0202E-01
4	2.7139E+03	9.0712E+02	1.1507E+00	2.6795E+05	1.1023E+06	3.2004E+05	-1.0137E+06	-0.1075E+03	5.4254E+06	7.4054E+03

M13 MATRIX

	11	12	13	14	15	16	17	18	19	20
1	-9.6952E+05	-3.2504E+02	-0.5478E+05	-7.0200E+05	-2.0120E+05	-4.0536E+03	-0.1359E+01	-3.0013E+06	-2.6473E+04	-6.7191E+01
2	-3.2063E+03	-1.0609E+00	2.9097E+01	-1.5363E+03	-5.5615E+02	3.7034E+00	6.5770E-02	3.6648E+03	2.2390E+01	-7.4770E-02
3	-2.0607E+01	-9.6522E-03	-4.2205E+01	7.0431E+01	3.1195E+01	3.0098E+00	7.6620E-02	2.6007E+03	2.0036E+01	5.5742E-03
4	1.7267E+07	5.7403E+03	2.0200E+06	1.2606E+07	4.7209E+06	3.5236E+04	5.0409E+02	3.3790E+07	1.0609E+05	9.0007E+02

M14 MATRIX

	21	22	23	24	25
1	0.5293E+03	-6.3902E+07	-7.0090E+07	2.3372E+06	-5.1720E+07
2	1.4695E+01	1.0360E+05	1.7452E+05	9.1000E+03	-3.0140E+04
3	-1.0759E+00	1.0764E+05	1.1151E+05	1.9207E+02	2.7000E+04
4	-1.3100E+05	1.6929E+00	2.7640E+00	-1.3065E+07	6.0760E+00

M15 MATRIX

	1	2	3	4	5	6	7	8	9	10
1	-1.1103E-03	-4.0117E-04	-2.0037E-04	-0.2039E-01	-2.1621E+02	-4.0078E-01	1.2065E+01	5.0029E-02	1.7351E+00	6.7901E-04
2	1.0371E-01	1.0209E-02	2.1307E+04	4.1553E+01	2.0002E+04	3.0741E+01	6.4070E+02	2.7601E+00	-1.9373E+02	-2.6500E-02
3	3.0329E+00	6.3002E-01	5.5562E+05	1.5060E+03	5.0924E+05	1.5199E+03	4.7070E+03	2.2222E+01	-2.9240E+03	-3.9407E+00
4	9.0361E-05	2.6516E-05	-6.1079E+00	-1.0100E-02	-6.1509E+00	-1.0023E-02	7.3227E-01	3.2032E-03	4.3302E-01	5.2263E-04

M16 MATRIX

	11	12	13	14	15	16	17	18	19	20
1	1.5920E+00	3.6073E-04	-6.5107E+01	-1.1740E-02	-4.7350E-02	-2.7740E-02	-5.1375E-04	-2.0112E+01	-1.7072E-01	-4.3259E-04
2	-1.0374E+02	-6.1002E-02	1.1066E+03	7.5055E+00	6.2910E+00	-1.3041E+00	-3.5500E-02	-0.0440E+02	-9.0010E+00	1.5329E-02
3	-1.1143E+03	-4.6001E-01	-2.7745E+04	-3.2322E+01	-0.1479E+00	6.0195E+01	1.2607E+00	4.1070E+04	3.0490E+02	-2.0001E-01
4	1.1306E+00	3.6906E-04	-2.3290E+00	7.7012E-01	2.0705E-01	-0.0500E-05	-0.0437E-06	-1.6112E-01	-2.0639E-03	3.7046E-05

BIAS MATRIX

	21	22	23	24	25
1	0.4100E-04	2.1110E+01	-0.0421E+01	1.0200E+02	-2.5305E+02
2	3.3510E-01	1.1253E+04	9.1713E+03	1.9907E+03	-1.4231E+04
3	-1.5201E+01	2.0110E+06	2.0190E+06	4.0087E+04	3.7007E+05
4	-7.0003E-03	6.2920E-01	-5.3490E-01	5.4304E+00	1.7512E+01

MEASUREMENT CONTRIBUTION TO SECOND DERIVATIVE OF THE COST FUNCTION

PARAMETER NO.	Y01	Y02	Y03	Y04	Y05
FBPRT	1	1.6307E+01	4.7036E+01	3.7540E+04	2.4319E+04
ADIPP	2	3.244E-01	3.2052E-01	5.2537E-01	5.640E-01
ALIER	3	2.1067E+00	1.106E-01	2.4920E+00	1.220E-01
VOIJS	4	5.2701E-02	1.605E-05	4.1162E-05	3.0294E-05
VOI4	5	5.709E-02	7.2042E-05	0.026E-02	3.517E-03
AP3	6	3.456E+10	4.423E+06	9.0123E+06	7.8214E+06
BP3	7	2.0036E+05	0.5707E+01	1.1393E+02	1.3565E+02
BP35	8	3.496E+10	4.501E+06	0.000E+00	0.000E+00
AP4	9	2.1651E+05	0.000E+00	0.000E+00	0.000E+00
AP4	10	0.000E+00	0.000E+00	0.000E+00	4.3120E+06
BP4	11	0.000E+00	0.000E+00	0.000E+00	0.000E+00
AT3	12	0.140E+05	3.740E+02	0.7973E+02	2.2103E+06
BT3	13	3.250E+00	2.100E-04	5.2577E-04	3.7536E+00
AT4	14	0.000E+00	0.000E+00	0.000E+00	1.7932E+07
BT4	15	0.000E+00	0.000E+00	0.000E+00	1.0620E+06
TAPT4	16	0.000E+00	0.000E+00	2.000E+00	1.2037E-01
AWF	17	0.000E+00	0.000E+00	3.9615E+00	2.0392E+07
AWF	18	4.3421E+05	0.245E+02	0.946E+06	4.9621E+05
SWF	19	6.0015E+04	3.1910E+01	1.2017E+02	1.2365E+06
CSTABE	20	4.3563E+02	1.2007E-01	3.009E+02	6.079E+04
CSTARC	21	3.044E-01	0.041E-05	2.030E-04	1.6001E+01
CTAN	22	2.170E+00	6.0070E+04	2.140E+05	7.3234E-03
CTAN	23	1.7617E+04	3.202E+00	2.177E+01	7.9230E+06
CEPFA	24	2.510E-01	5.547E-05	9.047E-05	6.124E+02
CEPFB	25	6.9230E+01	2.509E-02	0.265E-02	2.0032E-03
TAPF3	26	2.200E+11	4.0070E+00	3.644E+00	5.407E+01
TAPF3	27	2.377E+11	4.6717E+00	0.000E+00	7.5013E+00
TAPF4	28	0.000E+00	0.000E+00	4.509E+00	0.000E+00
do1off	29	6.3041E+05	4.007E+02	1.154E+03	4.7319E+00
betime	30	2.4515E+10	0.1315E+06	2.927E+07	0.2171E+05
	31			4.6931E+10	2.1617E+00

RUN SUMMARY

ESTIMATION				BIASANCE		
PARAMETER NO.	LABEL	STANDARD ERRORS	ESTIMATION BIAS	PARAMETER NO.	LABEL	LEVEL
1	FRPT	7.631E-03	2.724E+00	3	ALIEE	1.000E+01
2	ADIFF	5.603E-01	1.527E+02	5	VOLA	2.600E+02
4	VOLJ5	6.222E+00	4.237E+03	6	AP3	2.000E-05
41	deloff	5.271E-04	2.282E-01	7	BP3	5.000E-04
				8	AP35	2.000E-06
				9	BP35	5.000E-04
				10	AP4	2.000E-04
				11	BP4	5.000E-01
				12	AP3	3.000E-04
				13	BT3	3.000E+00
				16	AT4	3.000E-04
				19	BT4	3.000E+00
				20	TAUT4	5.000E-05
				21	AMP	2.500E-02
				22	BWF	7.000E-02
				29	CSTABE	1.000E-02
				30	CSTABC	1.000E-01
				31	CTAU	6.000E-05
				32	CTAUN	1.000E-03
				33	CEFFA	4.000E-06
				34	CEFFB	4.000E-04
				35	TAUP3	5.000E-05
				36	TAUP35	5.000E-05
				37	TAUP4	5.000E-05
				42	botime	1.000E-02

SENSITIVITY DATA FOR THE FOLLOWING PARAMETERS  
WAS GENERATED BY THE SCIDYT PROGRAM

PARAMETER NO.	1	2	3	4	5	6	7	8	9	10
LABEL	FPFPT	ADIV7	ALINE	VOLJ5	VOL4	APJ	BPJ	APJ5	BPJ5	AP4
NORMALIZATION	1.000E+01	7.000E+01	1.000E+02	2.500E+03	2.500E+03	1.000E+03	1.000E+00	1.000E+00	1.000E+00	1.000E+00
PARAMETER NO.	11	12	13	16	19	20	21	22	29	30
LABEL	BP4	ATJ	BTJ	AT4	BT4	TAUTA	AV7	BM7	CSTABE	CSTABC
NORMALIZATION	1.000E+00	5.000E+00	5.000E+00							
PARAMETER NO.	31	32	33	34	35	36	37	41	42	
LABEL	CTAU	CTAUM	CEP7A	CEP7B	TAUPJ	TAUPJ5	TAUP4	deleoff	botime	
NORMALIZATION	1.000E+00	1.000E+00	1.963E+00	2.501E+00	1.000E+00	1.000E+00	1.000E+00	1.000E+00	5.000E-03	

FOLLOWING MEASUREMENTS WERE USED BY THE SCIDYT PROGRAM

1 : Y01                    2 : Y02                    3 : Y03                    4 : Y04                    5 : Y05

ESTIMATION PARAMETER SET

PARAMETER NO. 1 2 4 41  
 LABEL FREQ ADIFF VOL35 delocf  
 APRIORI UNCERT 1.000E+06 1.000E+06 1.000E+06 1.000E+06

NUISANCE PARAMETER SET

3 : ALIEN 5 : VOL4 6 : AP3 7 : BP3 8 : AP35 9 : BP35 10 : AP4 11 : BP4  
 12 : AT3 13 : BT3 14 : AT4 15 : BT4 16 : TAUT4 17 : AMT 18 : DMF 19 : CSTABE  
 20 : CSTABC 21 : CTAU 22 : CTAUM 23 : CBPA 24 : CBPA 25 : TAUP3 26 : TAUP3 27 : TAUP4  
 28 : bctimo

MEASUREMENT SET

MEAS NO. 1 2 3 4 5  
 LABEL Y01 Y02 Y03 Y04 Y05  
 NOISE 5.00000E-01 1.50000E+00 1.50000E+00 1.00000E+01 5.00000E-01

M11 MATRIX

	1	2	3	4
1	9.1631E+04	3.526E+01	-3.2614E+01	-1.0055E+06
2	3.5264E+01	4.0579E+00	1.6027E-01	-2.0101E+03
3	-3.2614E+01	1.6027E-01	5.2005E-02	1.0007E+02
4	-1.0055E+06	-2.0101E+03	1.0007E+02	1.6310E+07

M12 MATRIX

	1	2	3	4	5	6	7	8	9	10
1	-3.0711E+02	-0.7089E+01	-2.9508E+01	-7.2677E+04	-3.0786E+07	-7.7964E+04	3.0922E+05	1.3085E+03	-1.0070E+05	-3.3539E+02
2	1.0042E+00	1.0306E-01	1.0517E+05	4.3557E+02	1.0235E+05	4.2403E+02	2.4190E+03	1.0366E+01	-2.0090E+03	-1.0004E+00
3	2.507E-01	5.3152E-02	3.0009E+04	9.9226E+01	3.9010E+04	1.0113E+02	1.6639E+01	7.0203E-02	-2.0006E+02	-1.0202E-01
4	2.7139E+03	9.0712E+02	1.1507E+08	2.6795E+05	1.3023E+08	3.2404E+05	-1.0137E+06	-0.1075E+03	5.4254E+06	7.4964E+03

M13 MATRIX

	11	12	13	14	15	16	17	18	19	20
1	-9.6952E+05	-3.2504E+02	-0.5477E+05	-7.0200E+05	-2.9328E+05	-4.4636E+03	-0.1359E+01	-3.0013E+06	-2.6473E+04	-6.7191E+01
2	-3.2063E+03	-1.0609E+00	2.9097E+03	-1.5303E+03	-5.5615E+02	3.7934E+00	6.5774E-02	2.6644E+03	2.2390E+01	-7.4770E-02
3	-2.0607E+01	-9.6552E-03	-4.2205E+01	7.931E+01	3.1195E+01	3.0495E+00	7.6620E-02	2.6007E+03	2.4036E+01	5.5742E-03
4	1.7247E+07	5.7403E+03	2.4200E+06	1.2666E+06	4.7209E+07	3.5256E+04	5.0409E+02	2.3790E+07	1.9609E+05	9.0007E+02

M14 MATRIX

	21	22	23	24	25
1	0.5293E+03	-6.3902E+07	-7.4090E+07	2.3720E+06	-5.3720E+07
2	1.4695E+01	3.0360E+05	3.7452E+05	9.1000E+03	-1.0140E+04
3	-1.5799E+00	1.0764E+05	1.1151E+05	1.9207E+02	2.7600E+04
4	-1.3100E+05	1.6922E+08	2.7646E+08	-1.3065E+07	6.0769E+04

BIAS MATRIX

	1	2	3	4	5	6	7	8	9	10
1	-1.1192E-03	-4.4117E-04	-2.0037E+02	-4.2039E-01	-2.1621E+02	-6.0007E-01	1.2065E+01	5.0029E-02	1.7351E+00	6.7901E-04
2	1.0371E-01	1.6209E-02	2.1307E+04	4.1553E+01	2.0992E+04	3.0741E+01	6.0874E+02	2.7601E+00	-1.9373E+02	-2.6950E-02
3	3.4325E+00	6.3002E-01	5.5563E+05	1.5069E+03	5.4924E+05	1.5159E+03	4.7076E+03	2.222E+01	-2.9240E+03	-3.9407E+00
4	9.9361E-05	2.6516E-05	-6.1079E+00	-1.0100E-02	-6.1540E+00	-1.4023E-02	7.3227E-01	3.2032E-03	4.3302E-01	5.2263E-04

BIAS MATRIX

	11	12	13	14	15	16	17	18	19	20
1	1.5920E+00	3.6073E-04	-4.5107E+01	-1.1740E-02	-4.7350E-02	-2.7734E-02	-5.1375E-04	-2.0112E+01	-1.7073E-01	-4.3259E-04
2	-1.9370E+02	-6.1002E-02	1.1066E+03	7.955E+00	6.2910E+00	-1.3641E+00	-3.5500E-02	-9.0400E+02	-9.9010E+00	1.5329E-02
3	-1.1143E+03	-4.6001E-01	-2.7745E+04	-1.3322E+01	-0.3479E+00	6.0195E+01	1.2607E+00	4.1070E+04	3.0490E+02	-2.0001E-01
4	1.1300E+00	3.6996E-04	-2.3290E+00	7.7012E-01	2.0705E-01	-0.0500E-05	-0.0437E-06	-1.6112E-01	-2.0639E-03	3.7046E-05

DIAS MATRIX

	21	22	23	24	25
1	0.4109E-04	2.1114E+01	-9.9421E+01	1.0209E+02	-2.5365E+02
2	3.3510E-01	1.1253E+04	9.1713E+03	1.9987E+03	-1.4231E+04
3	-1.5281E+01	2.0114E+06	2.0190E+06	4.9887E+04	3.7887E+05
4	-7.8903E-03	6.2920E-01	-5.3490E-01	5.4304E+00	1.7512E+01

MEASUREMENT CONTRIBUTION TO SECOND DERIVATIVE OF THE COST FUNCTION

PARAMETER NO.	Y01	Y02	Y03	Y04	Y05
FPEFT	2.5700E+04	1.6397E+01	4.7036E+01	3.7540E+04	2.4319E+04
ADIFF	2.3147E+00	3.2444E-01	3.2852E-01	5.2537E-01	5.6485E-01
ALIER	2.1887E+00	6.7588E-04	1.1046E-01	2.4928E+00	1.2288E-01
VOL35	5.2701E-02	1.6758E-03	1.6059E-03	4.1162E-03	3.0294E-03
VOLA	5.7899E-02	1.9302E-03	7.2045E-03	0.0268E-02	3.5176E-03
AP3	3.4560E+10	4.4329E+06	4.3198E+06	9.0123E+06	7.0214E+06
BP3	2.0826E+03	0.5707E+01	0.3519E+01	1.1393E+02	1.3565E+02
AP33	3.4960E+10	4.5016E+06	0.0000E+00	0.0000E+00	0.0000E+00
BP33	2.1651E+03	0.8089E+01	0.0000E+00	0.0000E+00	0.0000E+00
AP4	0.0000E+00	0.0000E+00	4.4393E+06	0.0000E+00	0.3120E+06
BP4	0.0000E+00	0.0000E+00	0.8089E+01	0.0000E+00	0.6076E+01
AS3	0.1400E+03	3.7409E+02	0.7973E+02	2.2103E+06	1.5706E+05
BT3	3.2584E+00	2.1900E-04	3.2577E-04	3.7536E+00	2.3724E-01
AS4	0.0000E+00	0.0000E+00	0.0000E+00	1.7932E+07	1.0620E+06
BT4	0.0000E+00	0.0000E+00	0.0000E+00	2.8000E+00	1.3037E-01
TAMT4	0.0000E+00	0.0000E+00	0.0000E+00	3.9615E+00	2.0392E+07
AMP	4.3421E+03	3.2413E+02	0.2458E+02	0.9460E+06	4.9621E+05
BNP	6.8015E+04	3.1916E+01	1.2917E+02	1.2365E+06	6.0794E+04
CSTAB	4.3563E+02	1.2887E-01	5.4077E-01	3.8099E+02	1.6081E+01
CSTAB	3.0444E-01	6.0410E-03	2.0302E-04	1.7460E-01	7.3234E-03
CTAB	2.1798E+00	4.8078E+04	2.1430E+03	1.9965E+00	7.9230E+06
CTAMB	1.7617E+04	5.2082E+00	2.1775E+01	1.4539E+04	6.1244E+02
CEPPA	2.5184E-01	5.5479E-03	9.0475E-03	5.7692E-02	2.8032E-03
CEPPB	6.9230E+01	2.5095E-02	0.2657E-02	9.9136E+02	5.4078E+01
TAMP3	2.2082E+11	4.0879E+00	3.6449E+00	7.7806E+00	7.5913E+00
TAMP33	2.3777E+11	4.6717E+00	0.0000E+00	0.0000E+00	0.0000E+00
TAMP4	0.0000E+00	0.0000E+00	4.5097E+00	0.0000E+00	4.7319E+00
deloff	6.3041E+03	4.4872E+02	1.1541E+03	1.4856E+07	0.2171E+05
bottom	2.4515E+10	0.1313E+06	2.9275E+07	4.6931E+10	2.1617E+09

RUN SUMMARY

ESTIMATION				BIASANCE		
PARAMETER NO.	LABEL	STANDARD ERRORS	ESTIMATION BIAS	PARAMETER NO.	LABEL	LEVEL
1	PPHPT	7.031E-03	2.300E-02	3	ALIEE	5.000E-01
2	ADIFF	5.603E-01	1.500E+00	5	VOLA	2.600E+00
4	VOL35	6.222E+00	2.463E+01	6	AP3	2.000E-06
41	6010EE	5.271E-04	4.125E-02	7	BP3	5.000E-04
				8	AP35	2.000E-06
				9	BP35	5.000E-04
				10	AP4	2.000E-04
				11	BP4	5.000E-02
				12	AT3	3.000E-04
				13	BT3	3.000E-01
				16	AT4	3.000E-04
				19	BT4	3.000E-01
				20	TAUT4	5.000E-05
				21	AM7	2.500E-02
				22	BM7	7.000E-02
				29	GTABE	1.000E-02
				30	GTABC	1.000E-01
				31	CTAU	6.000E-05
				32	CTAUM	1.000E-03
				33	CEP7A	4.000E-01
				34	CEP7B	4.000E-04
				35	TAUP3	1.000E-06
				36	TAUP35	1.000E-06
				37	TAUP4	5.000E-05
				42	bot100	1.000E-06

END

FEB.

1988

DTIC