

NO-A143 911

EVALUATION OF THE EFFECTS OF STRESS STATE AND  
INTERFACIAL PROPERTIES ON T. (U) SOUTHWEST RESEARCH  
INST SAN ANTONIO TX G R LEVERANT ET AL. JUL 84

1/1

UNCLASSIFIED

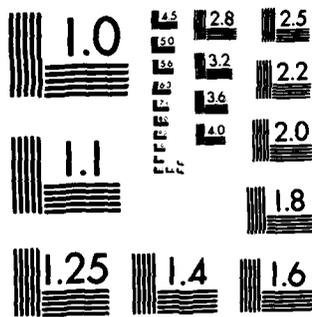
ARO-17207.4-NS DARG29-01-K-0049

F/B 11/4

NL







MICROCOPY RESOLUTION TEST CHART  
NATIONAL BUREAU OF STANDARDS-1963-A

ARO 17207.4-MS

②

Evaluation of the Effects of Stress State  
and Interfacial Properties on the Behavior  
of Advanced Metal Matrix Composites

FINAL REPORT

Gerald R. Leverant  
John E. Hack  
Richard A. Page

July 1984

Contract No. DAAG29-81-K-0049

Southwest Research Institute  
San Antonio, Texas

DTIC  
ELECTE  
AUG 02 1984  
S  
E  
D

Approved for Public Release: Distribution Unlimited

AD-A143 911

DTIC FILE COPY

UNCLASSIFIED

SECURITY CLASSIFICATION OF THIS PAGE (When Data Entered)

MASTER COPY - FOR REPRODUCTION PURPOSES

REPORT DOCUMENTATION PAGE		READ INSTRUCTIONS BEFORE COMPLETING FORM
1. REPORT NUMBER <b>ARO 17207-4-MS</b>	2. GOVT ACCESSION NO. N/A	3. RECIPIENT'S CATALOG NUMBER N/A
4. TITLE (and Subtitle) Evaluation of the Effects of Stress State and Interfacial Properties on the Behavior of Advanced Metal Matrix Composites		5. TYPE OF REPORT & PERIOD COVERED FINAL REPORT July 1981 - June 1984
		6. PERFORMING ORG. REPORT NUMBER 06-6621
7. AUTHOR(s) Gerald R. Leverant John E. Hack Richard A. Page		8. CONTRACT OR GRANT NUMBER(s) DAAG29-81-K-0049
9. PERFORMING ORGANIZATION NAME AND ADDRESS Southwest Research Institute 6220 Culebra Road San Antonio, TX 78284		10. PROGRAM ELEMENT, PROJECT, TASK AREA & WORK UNIT NUMBERS P-17207-MS
11. CONTROLLING OFFICE NAME AND ADDRESS U. S. Army Research Office Post Office Box 12211 Research Triangle Park, NC 27709		12. REPORT DATE July 1984
		13. NUMBER OF PAGES
14. MONITORING AGENCY NAME & ADDRESS (if different from Controlling Office)		15. SECURITY CLASS. (of this report) Unclassified
		15a. DECLASSIFICATION/DOWNGRADING SCHEDULE
16. DISTRIBUTION STATEMENT (of this Report)  Approved for public release; distribution unlimited.		
17. DISTRIBUTION STATEMENT (of the abstract entered in Block 20, if different from Report)  NA		
18. SUPPLEMENTARY NOTES  The view, opinions, and/or findings contained in this report are those of the author(s) and should not be construed as an official Department of the Army position, policy, or decision, unless so designated by other documentation.		
19. KEY WORDS (Continue on reverse side if necessary and identify by block number)  Metal Matrix Composite      Fatigue      Thermal Exposure Al <sub>2</sub> O <sub>3</sub> Fiber      Cyclic Cleavage      Processing Magnesium Alloys      Fracture Toughness      Defects Tensile Behavior		
20. ABSTRACT (Continue on reverse side if necessary and identify by block number)  The effects of fiber fraction, fiber orientation and matrix alloy additions on tensile and fatigue behavior were studied in commercially pure magnesium and ZE41A (Mg-4.25Zn-0.5Zr-1.25RE) that were both reinforced with FP alumina fibers. In general, axial properties were found to be dependent on fiber fraction while off-axis properties were not. Off-axis loading resulted in substantial reductions in tensile and fatigue strength in the commercially pure matrix material. Although failure in tensile overload occurred along the weak fiber/matrix interface in off-axis specimens, subcritical fatigue		

DD FORM 1 JAN 73 1473 EDITION OF 1 NOV 65 IS OBSOLETE

UNCLASSIFIED

SECURITY CLASSIFICATION OF THIS PAGE (When Data Entered)

04 07 31 238



## I. Background

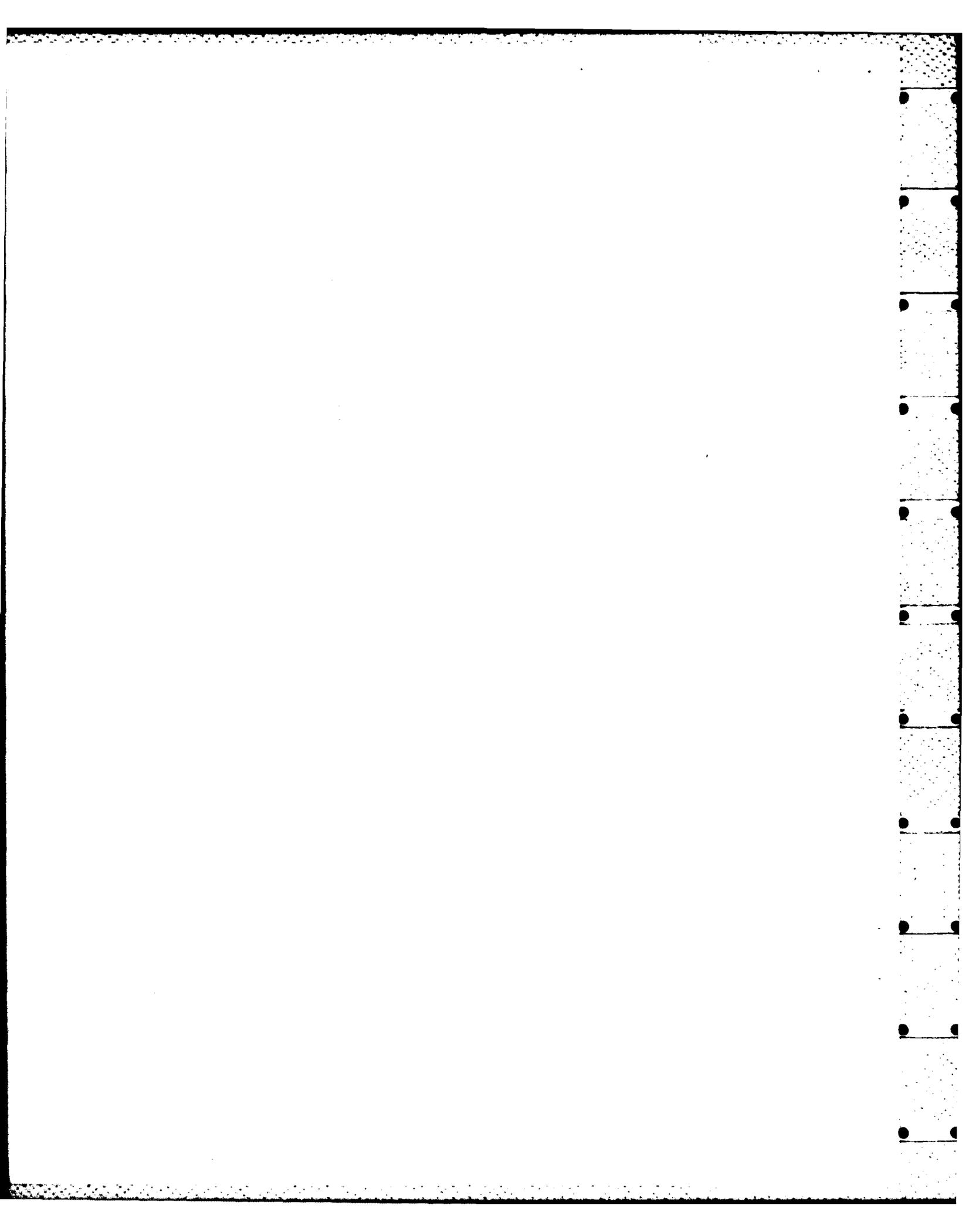
The use of metal matrix composites (MMC) in structures will require design information based on well-developed quantitative relationships between processing history, microstructure and mechanical properties. Based on available information in the literature, the establishment of such relationships in various MMC systems has proven very elusive. Although a great deal of work has been performed on fracture mechanisms in metal matrix composites, the results to date have been quite contradictory in nature. The inconsistencies arise from the complex interaction of the fiber/matrix interfacial region with various applied states of stress. Furthermore, the presence of the fiber/matrix interface in these composites introduces several mechanisms of crack propagation in addition to those observed in monolithic materials. Also, standard metallurgical techniques for modifying the properties of the matrix material (i.e., alloying, heat treatment, deformation processing, etc.) affect the metallurgical structure and properties of the interfacial region as well. Thus, the effects of processing variables on composite behavior through microstructural changes in the matrix are difficult to separate from the effects of those variables on the fiber/matrix interface.

This report summarizes the results of a program which was initiated to overcome these difficulties by developing quantitative relationships between matrix microstructure, composition and properties, fiber orientation and volume percent, and fatigue crack growth behavior in these materials through direct observations of the effects of these variables on the strain field at the tip of a growing crack. The overall program is aimed at determining the local sequence of events at the tip of a fatigue crack (i.e., matrix failure vs. fiber failure vs. interfacial decohesion) and the critical strain accumulation required for these events to occur. However, very little data exist on the relationships between microstructure and fracture behavior in metal matrix composites. Thus, the efforts to date have concentrated on the baseline characterization of the tensile and fatigue behavior of Al<sub>2</sub>O<sub>3</sub> fiber reinforced commercially pure Mg (CPMg) and a magnesium alloy (ZE41A). In particular, the influences of fiber fraction, fiber orientation, alloy content and loading conditions were investigated.

## II. Summary of Program

The tensile and fatigue properties of CPMg and ZE41A as a function of orientation are shown in Figs. 1 and 2. Of particular note are the reduced monotonic and cyclic axial properties and enhanced off-axis properties for ZE41A compared to CPMg. This behavior is directly related to the size of the MgO reaction zone, Fig. 3, that forms at the fiber (Al<sub>2</sub>O<sub>3</sub>)/matrix interfaces, as well as the respective matrix strengths.

The manufacturer has indicated that the processing cycle (liquid metal infiltration) is identical for both composites. Since commercially-pure Mg will solidify congruently (i.e., no mushy zone) while ZE41A has a



mushy zone extending over a range of 120°C, this suggested that the reason for the thicker reaction zone and larger MgO particles in the latter were directly related to the time that the fibers were in contact with liquid magnesium. This has been confirmed by heat treating experiments conducted in this program. This result is also consistent with the fact that Auger spectroscopy has shown no preferential segregation of alloying elements in ZE41A to the fiber/matrix interface.

Detailed fractographic analysis of the off-axis CPMg fatigue specimens defined the extent of the subcritical and overload fracture zones. Combining these results with a fracture mechanics analysis, it was shown, Fig. 4, that the critical stress intensity ( $K_C$ ) is controlled by both the normal and shear stress components acting on the fiber/matrix interface as follows:

$$\left(\frac{K_I}{K_{Ic}}\right)^2 + \left(\frac{K_{III}}{K_{IIIc}}\right)^2 = 1$$

Similar  $K_{Ic}$  and  $K_{IIIc}$  values were determined for a variety of crack geometries, indicating that  $K_C$  for interfacial delamination is geometry independent. Therefore,  $K_C$  should be effective as a design parameter. The principal conclusions resulting from this study are as follows:

1. Process defects in the form of large clumps of Al<sub>2</sub>O<sub>3</sub> grains and fibers that were broken and lying on their sides were preferential sites for crack initiation for both monotonic and cyclic loading along the fiber axis. However, the presence of the defects was not detrimental to axial properties unless the size of the defect was a significant fraction of the specimen cross section. These defects did not play a role in crack initiation during off-axis loading.
2. For CPMg, a relatively weak fiber/matrix interface resulted in a large reduction in off-axis tensile properties. The reduction in strength is related to a change in the mode of fracture from flat fracture across fibers in the axial orientation to failure along the fiber/matrix interface in off-axis orientations. Interfacial failure occurred between the magnesium matrix and the MgO reaction zone.
3. For CPMg, fatigue crack initiation and propagation occurred primarily in the magnesium matrix for off-axis loading.
4. The critical stress intensity for unstable fracture of off-axis CPMg is controlled by both the normal and shear stress components acting on the fiber/matrix interface.

5. The alloying elements in ZE41A resulted directly in an increased matrix strength and indirectly in an increased fiber/matrix interfacial strength and a decreased fiber strength compared to CPMg.
6. The increased interfacial strength was due to a thicker reaction zone of larger MgO particles and not to segregation of alloying elements to the reaction zone. The size of the reaction zone is controlled by the markedly different solidification characteristics of CPMg compared to ZE41A, resulting in considerably different exposure times of the Al<sub>2</sub>O<sub>3</sub> fibers to liquid magnesium.
7. The increased interfacial and matrix strengths combined with the decreased fiber strengths in ZE41A yielded reduced axial and increased off-axis tensile and fatigue properties compared to CPMg. In addition, this change in component strengths delayed the transition from fracture across the fibers to interfacial failure as the angle between the fiber direction and loading axis was increased.

### III. List of Publications

1. "Critical Stress Intensity for Off-Axis Fracture of Al<sub>2</sub>O<sub>3</sub> Fiber Reinforced Magnesium", by K. S. Chan, J. E. Hack and R. A. Page, Metallurgical Transactions A, Vol. 15A, pp. 756-759, 1984.
2. "Tensile and Fatigue Behavior of Aluminum Oxide Fiber Reinforced Magnesium Composites: Part I. Fiber Fraction and Orientation", by J. E. Hack, R. A. Page and G. R. Leverant, Metallurgical Transactions A, Vol. 15A, pp. 1389-1396, 1984.
3. "Tensile and Fatigue Behavior of Aluminum Oxide Fiber Reinforced Magnesium Composites: Part II. Alloying Effects", by R. A. Page, J. E. Hack, R. Sherman, and G. R. Leverant, Metallurgical Transactions A, Vol. 15A, pp. 1397-1405, 1984.
4. "The Influence of Thermal Exposure on Interfacial Reactions and Strength in Aluminum Oxide Fiber Reinforced Magnesium Alloy Composites", by J. E. Hack, R. A. Page and R. Sherman, Metallurgical Transactions A (submitted).

IV. Participating Scientific Personnel

1. Dr. Gerald R. Leverant (Principal Investigator)
2. Mr. John E. Hack
3. Dr. Richard A. Page
4. Dr. Kwai Chan
5. Dr. Robert Sherman

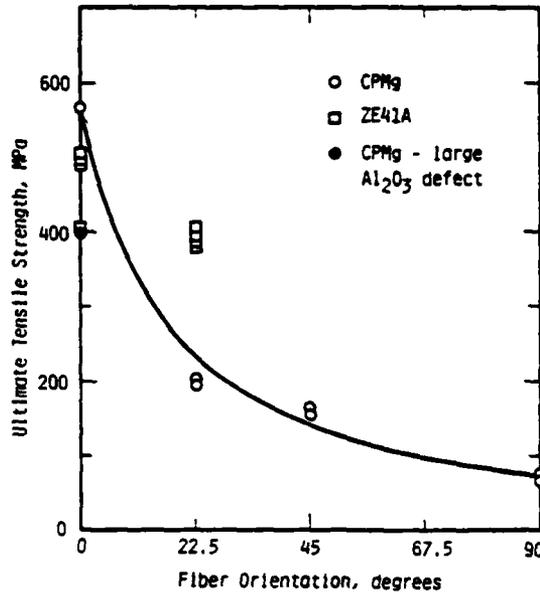


Fig. 1 — Effect of fiber orientation on ultimate tensile strength for both commercially pure magnesium and ZE41A matrix materials.

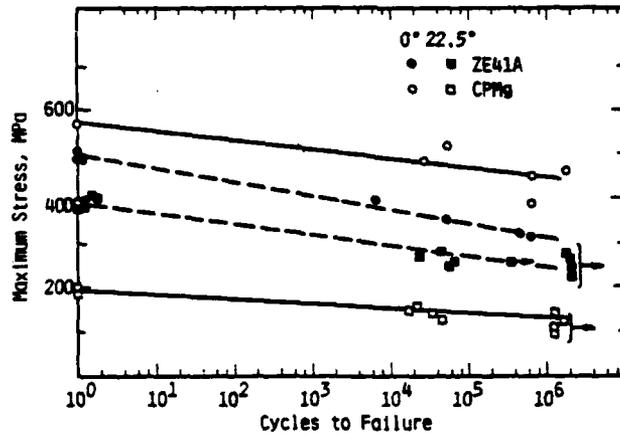
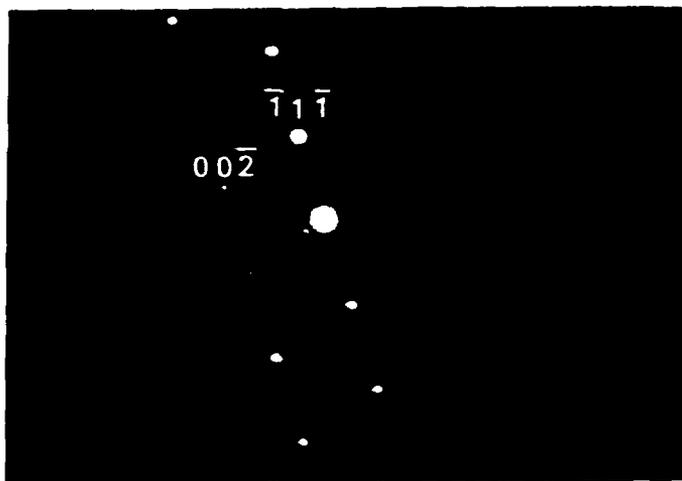


Fig. 2 — Effect of fiber orientation and matrix alloy addition on the fatigue behavior of Al<sub>2</sub>O<sub>3</sub> fiber-reinforced magnesium.



(a)



(b)

Fig. 3 — Transmission electron micrograph of the fiber/matrix reaction zone in ZE41A (a) and a diffraction pattern taken from a typical reaction zone particle (110 Zone) (b). The various constituents are indicated by: MG (ZE41A matrix); RZ (reaction zone product); and FP ( $\text{Al}_2\text{O}_3$  fiber).

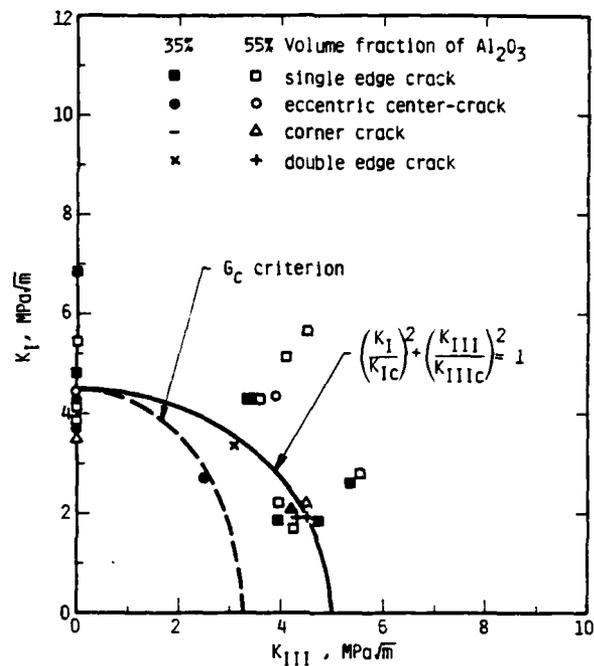


Fig. 4 — Calculated Mode I stress intensity factor,  $K_I$ , vs calculated Mode III stress intensity factor,  $K_{III}$ , for both 35 and 55 vol pct fiber. Solid line is for  $K_{Ic} = 4.5 \text{ MPa}\sqrt{\text{m}}$  and  $K_{IIIc} = 5 \text{ MPa}\sqrt{\text{m}}$ .

END

FILMED

9-84

DTIC

