

AD A117585

(R)

ESL-TR-81-46

PREDICTIVE MODEL FOR JET ENGINE TEST CELL OPACITY

DR. GORDON A. LEWANDOWSKI
NEW JERSEY INSTITUTE OF TECHNOLOGY
323 HIGH STREET
NEWARK, NJ 07102

30 SEPTEMBER 1981

FINAL REPORT
1 JULY 1980 - 30 SEPTEMBER 1981

DTIC
SELECTED
S JUL 28 1982 D
A

APPROVED FOR PUBLIC RELEASE; DISTRIBUTION UNLIMITED

DTIC FILE COPY



ENGINEERING AND SERVICES LABORATORY
AIR FORCE ENGINEERING AND SERVICES CENTER
TYNDALL AIR FORCE BASE, FLORIDA 32403, 15

82-07-37

NOTICE

PLEASE DO NOT REQUEST COPIES OF THIS REPORT FROM
HQ AFESC/RD (ENGINEERING AND SERVICES LABORATORY).

ADDITIONAL COPIES MAY BE PURCHASED FROM:

NATIONAL TECHNICAL INFORMATION SERVICE
5285 PORT ROYAL ROAD
SPRINGFIELD, VIRGINIA 22161

FEDERAL GOVERNMENT AGENCIES AND THEIR CONTRACTORS
REGISTERED WITH DEFENSE TECHNICAL INFORMATION CENTER
SHOULD DIRECT REQUESTS FOR COPIES OF THIS REPORT TO:

DEFENSE TECHNICAL INFORMATION CENTER
CAMERON STATION
ALEXANDRIA, VIRGINIA 22314

Unclassified

SECURITY CLASSIFICATION OF THIS PAGE (When Data Entered)

REPORT DOCUMENTATION PAGE		READ INSTRUCTIONS BEFORE COMPLETING FORM
1. REPORT NUMBER ESL-TR-81-46	2. GOVT ACCESSION NO. AD-A117585	3. RECIPIENT'S CATALOG NUMBER
4. TITLE (and Subtitle) PREDICTIVE MODEL FOR JET ENGINE TEST CELL OPACITY		5. TYPE OF REPORT & PERIOD COVERED FINAL REPORT July 1 1980-Sept. 30, 1981
		6. PERFORMING ORG. REPORT NUMBER
7. AUTHOR(s) DR. GORDON A. LEWANDOWSKI		8. CONTRACT OR GRANT NUMBER(s) FO 8635-80-C0222
9. PERFORMING ORGANIZATION NAME AND ADDRESS New Jersey Institute of Technology 323 High Street Newark, NJ 07102		10. PROGRAM ELEMENT, PROJECT, TASK AREA & WORK UNIT NUMBERS P.E. 62601 F JON: 1900-90-11
11. CONTROLLING OFFICE NAME AND ADDRESS AFESC/RDVS Tyndall AFB, Florida 32403		12. REPORT DATE September 30, 1981
		13. NUMBER OF PAGES 72
14. MONITORING AGENCY NAME & ADDRESS (if different from Controlling Office)		15. SECURITY CLASS. (of this report) UNCLASSIFIED
		15a. DECLASSIFICATION/DOWNGRADING SCHEDULE
16. DISTRIBUTION STATEMENT (of this Report) Approved for public release; distribution unlimited		
17. DISTRIBUTION STATEMENT (of the abstract entered in Block 20, if different from Report)		
18. SUPPLEMENTARY NOTES Availability of this report is specified on verso of front cover.		
19. KEY WORDS (Continue on reverse side if necessary and identify by block number) Computer Program Test Cell Jet Engine Test Cell Visibility Light Scattering Opacity		
20. ABSTRACT (Continue on reverse side if necessary and identify by block number) A computer program (written in FORTRAN for a CDC 6600) was developed to predict the plume opacity of jet engine test cells. The data input required for the model includes: the particle density, concentration, and size distribution in the exhaust gas, and the effective stack diameter. Previous data obtained for J-57 engines were used to test the model, and the difference		

DD FORM 1 JAN 73 1473 EDITION OF 1 NOV 65 IS OBSOLETE

Unclassified

SECURITY CLASSIFICATION OF THIS PAGE (When Data Entered)

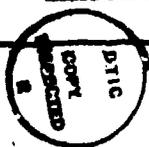
Unclassified

SECURITY CLASSIFICATION OF THIS PAGE(When Data Entered)

between the theoretical and measured transmittance was generally within one percent.

The program also predicts the theoretical effect of using electrostatic precipitators or venturi scrubbers to treat the exhaust emissions. These predictions indicate that control devices larger than the test cells would have to be installed to even achieve a minimal effect on the observed visibility.

Distribution For	
SECRET	<input checked="" type="checkbox"/>
CONFIDENTIAL	<input type="checkbox"/>
Unannounced	<input type="checkbox"/>
Justification	
By	
Distribution/	
Availability Codes	
Dist	Avail and/or Special
A	



Unclassified

SECURITY CLASSIFICATION OF THIS PAGE(When Data Entered)

PREFACE

This report was prepared at the Department of Chemical Engineering, New Jersey Institute of Technology, 323 High Street, Newark, New Jersey 07102, under contract No. FO 8635-80-CO222 with the Air Force Engineering and Services Center, Tyndall Air Force Base, Florida 32403. Capt. D. Berlinrut managed the program for the Air Force Engineering and Services Center. The work was begun July 1, 1980 and completed September 30, 1981.

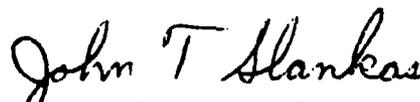
The author, Dr. Gordon A. Lewandowski, is indebted to the following individuals and organizations: Dr. W. Wong of New Jersey Institute of Technology for his valuable suggestions regarding stable generation of the complex Riccati-Bessel functions; Exxon Research & Engineering Co. (ERE) for release of their opacity computer program which was used to check the results of the program presented in this report; and S. Shaw of ERE for her efforts to obtain the release of their program and for guidance in its use.

This report has been reviewed by the Public Affairs Office and may be released to the National Technical Information Service (NTIS), where it will be available to the general public, including foreign nations.

This technical report has been reviewed and is approved for publication.



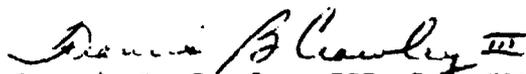
Daniel D. Berlinrut, Capt, USAF, BSC
Air Quality Research Engineer



John T. Slankas, Maj, USAF
Chief, Environmental Sciences
Branch



Michael J. Ryan, Lt Col, USAF, BSC
Chief, Environics Division



Francis B. Crowley, III, Col, USAF
Director, Engineering & Services
Laboratory

TABLE OF CONTENTS

Section		Page
I	Introduction	1
II	Smoke Number	4
III	Visibility Equations	7
IV	Electrostatic Precipitator Equations	14
V	Scrubber Equations	16
VI	Results & Discussion	19
	References	24
	Appendices	
	A Computer Input Data Format	27
	B Output Format	36
	C FORTRAN Listing	49
	D Computer Program Nomenclature	63

LIST OF FIGURES

Figure		Page
1	Generalized Test Cell Schematic	2
2	Relationship Between SAE Smoke Number & Sooty Density	5
3	SAE Smoke Number vs. Ringelmann Reading	6
4	Visibility	8
5	Black Plume Ringelmann Number Correlation with Transmittance	13
6	Log-Normal Plot of Size Distribution from Grems' Data	20

LIST OF TABLES

Table		Page
1	Comparison of Computer Predictions with Grems' Data for J57 Engine	21
2	Comparisons of Controlled and Uncontrolled Emissions	22

SECTION I
INTRODUCTION

A generalized schematic of a jet engine test cell is shown in Figure 1. Generally, the width of a plume issuing from a test cell is approximated by the stack dimension. The stacks are usually square; however, they contain acoustical baffles which considerably reduce the open area. For the calculations made in this report, the plume width was assumed to be the square root of the net open stack area.

Dimensions can vary considerably, depending upon the particular cell design, but the principle of operation is always the same. An engine that has been repaired, or otherwise maintained, is placed in the cell to test it under flight conditions before being remounted on the aircraft. The engine is considered a mobile emission source which is governed by Federal rather than state regulations. However, the test cell is immobile, and on that basis a U.S. District Court upheld the right of the State of California to regulate test cell emissions (Reference 1) which occasionally violate state visibility requirements of Ringelmann 1 (20% opacity). Since the U.S. Air Force has a large number of test cells in California, this court ruling can have a significant impact on Air Force operations and capital expenditures.

In order to satisfy state regulations, there are three possible alternatives: (1) design smokeless engines and install them on all existing aircraft; (2) introduce fuel additives to minimize soot formation; (3) use particulate control devices to treat the test cell exhaust.

The first of these alternatives is already being pursued as a result of the military incentive to reduce visibility of in-flight aircraft. However, replacement is very costly and time consuming, due to the variety and number of existing aircraft and aircraft engines.

The second alternative can be effective. However, fuel additives are organo-metallic compounds (e.g., Ferrocene), which deposit metallic oxides on engine surfaces. Considering the cost of the engine and its maintenance, and the cost of the aircraft, anything that may permanently alter engine parts is considered highly undesirable.

The third alternative does not affect the engine. Because of this, it is the only alternative which state regulatory authorities can impose. Nevertheless, particulate control

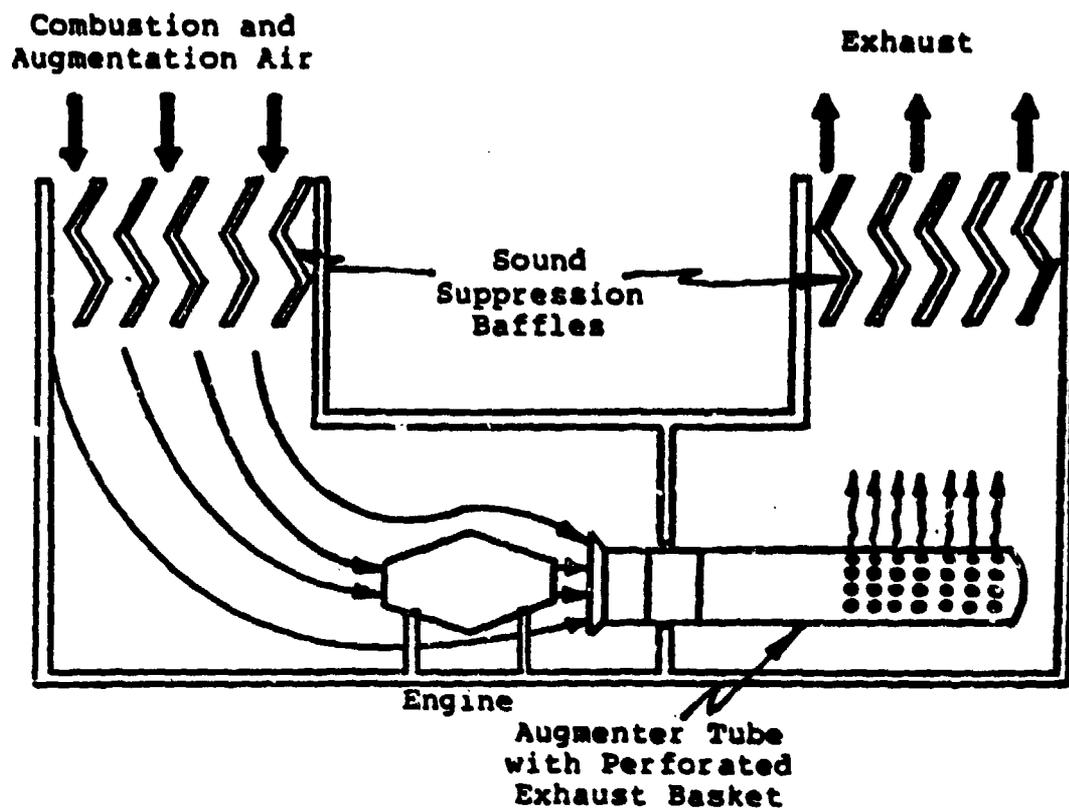


Figure 1. Generalized Test Cell Schematic

devices cannot be designed to meet an opacity requirement, only a specified degree of particulate removal.

The purpose of this study was to establish the connection between test cell particulate emissions and plume visibility as a basis for specifying control devices that could be mounted on the test cell exhaust stack. In addition, theoretical calculations were made to see under what conditions electrostatic precipitators or venturi scrubbers might satisfy opacity regulations.

SECTION II

SMOKE NUMBER

Much data on jet engine particulate emissions are in the form of SAE smoke numbers (SN), which measure the relative contrast of a standard filter paper exposed to the exhaust emissions for a standard period of time. A few investigators (References 2-6) have taken simultaneous measurements of particle loading ("soot density") and smoke number. Fewer studies (References 6-8) have determined plume opacity as a function of smoke number. These data, which are generally of poor quality, are plotted in Figures 2 and 3 for various engines.

Also presented in Figures 2 and 3 are empirical correlations based on Reference 9. The correlation in Figure 3 includes the results of Connor and Hodkinson's work (References 10, 11) relating observed visibility to plume transmittance.

As can be seen, there is about as much error in predicting the Ringelmann number directly from the smoke number, as there is in predicting mass loading. However, in order to determine opacity from loading, the particle size distribution must be known (involving an additional error), and a computer program used to make the calculation. In either case, the use of smoke numbers is a very unreliable tool in predicting test cell plume opacity.

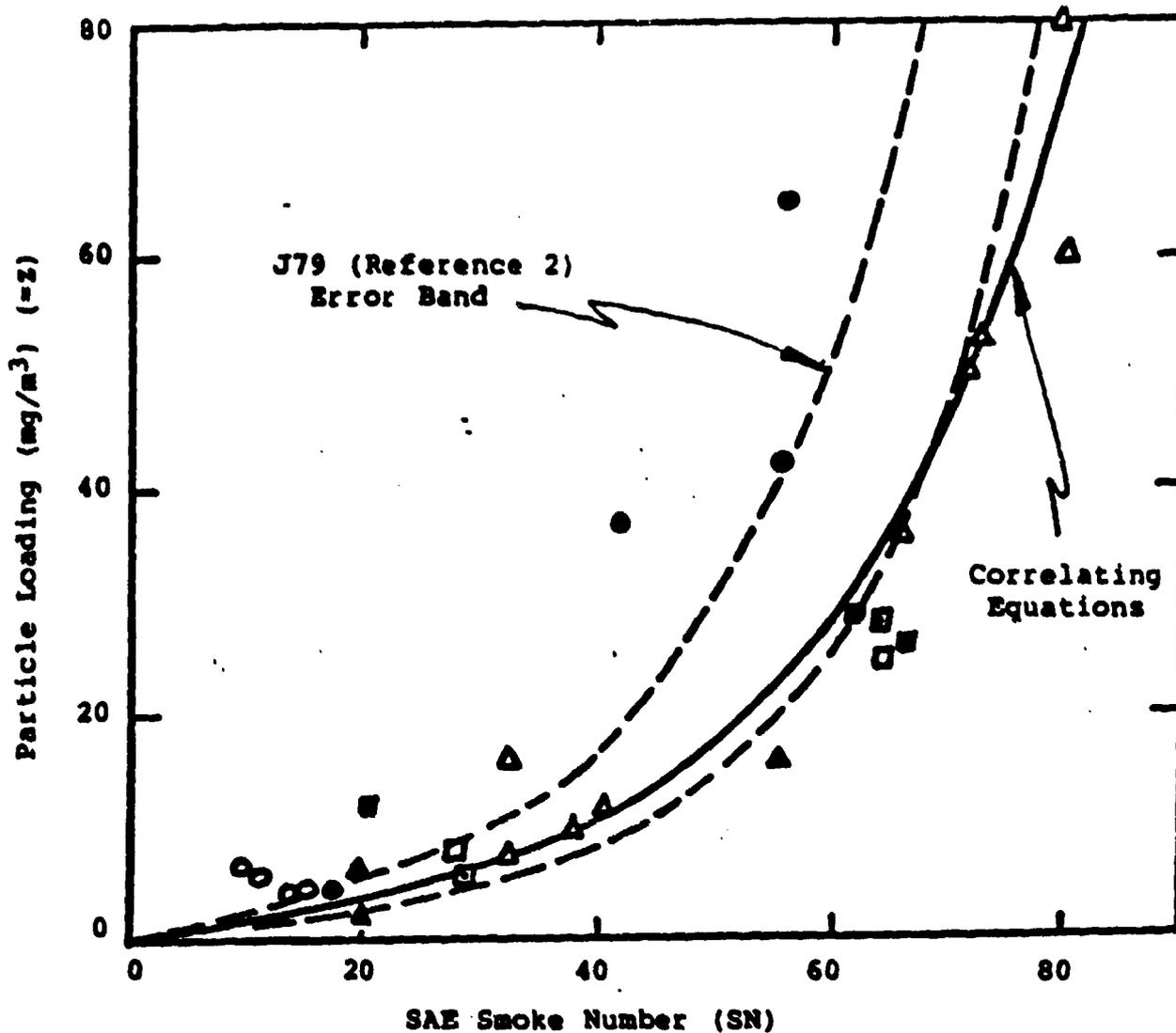


Figure 2. Relationship Between SAE Smoke Number and Root Density (Reference 6)

- JT8D & J52-P-6A
- J57-P-10 (Reference 4)
- J57-P-8
- T56 (501)
- ⊙ T58-GE-10
- ▲ T64-GE-413
- △ TF30 (Reference 5)
- ▲ T400 (at Max. Power)
- TPE331-5-X21 (Reference 3)

Correlating Equations:

$$\ln\left(1 - \frac{SN}{100}\right) = -(.04682)^{0.9} \} SN < 40$$

$$\ln\left(1 - \frac{SN}{100}\right) = -(.03162)^{0.6} \} SN > 40$$

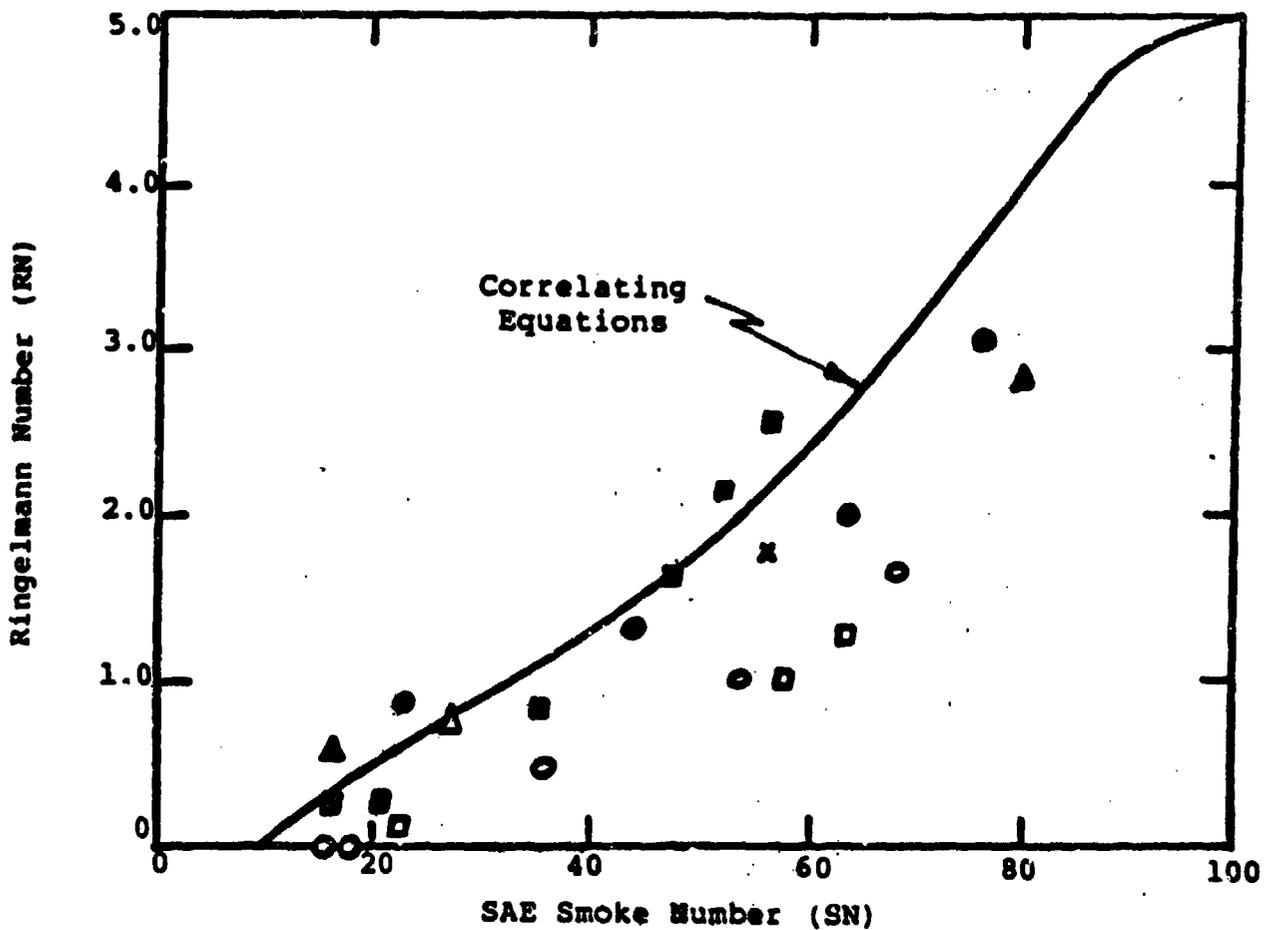


Figure 3. SAE Smoke Number vs. Ringelmann Reading (Reference 6)

Correlating Equations:

- | | |
|---------------------------|---|
| ○ J79 (Reference 2) | $\ln\left(\frac{T}{100}\right) = -(0.194L)\left[-\ln\left(1-\frac{SN}{100}\right)\right]^{2.08} \quad SN > 30$
(L = plume width = 3.5 m) |
| ● J79-GE-10 (Reference 8) | |
| □ J65 (clean) | $\ln\left(\frac{T}{100}\right) = -(0.0617L)\left[-\ln\left(1-\frac{SN}{100}\right)\right], \quad SN \leq 30$ |
| ■ J57-P-8 | |
| □ J57 (Reference 7) | $RN = -.0375T + 4.5, \quad 15 \leq T \leq 85$ |
| ○ J52-P-6A | |
| △ J52-P-6A (Smokeless) | $RN = -.0800T + 8.1, \quad 95 \geq T > 85$ |
| ▲ TF30-P-8 | $RN = -.0667T + 5.0, \quad T < 15$ |
| ▲ TF30-P-6 (Smokeless) | |
| X T56-A-10 | |

SECTION III

VISIBILITY EQUATIONS

Jet engine exhaust emissions largely consist of fine particles of unburned carbon. Because they are black, and therefore absorb much of the incident light intensity, carbon particles will exhibit very little back-scattering of ambient light. The visibility of black plumes is almost entirely a function of the relative contrast between the background skylight and the amount of such light transmitted through the plume (Figure 4). This relative contrast is independent of observer position and can be calculated by the following equation (Reference 11):

$$T = \frac{B_T}{B_0} = \exp \left[\left(\frac{-3WD}{2\rho_p} \right) \frac{1}{I} \sum_i \left(\frac{Q_{ext}}{d_p} \right)_i \right]$$

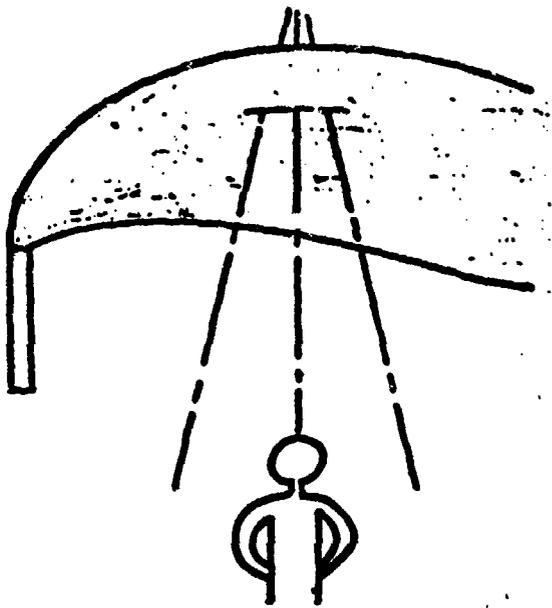
where: T = transmittance, or relative plume brightness
 B_T = brightness of light transmitted through the plume
 B_0 = background sky brightness
 W = particle loading
 D = plume diameter
 ρ_p = particle density
 I = total number of particle sizes
 d_p = particle size
 Q_{ext} = extinction coefficient, or ability of a given particle to reduce the intensity of the transmitted light

subscript i = i th particle size in the distribution

The extinction coefficient for any given particle is determined by the following equation (References 11 and 12):

Black Plumes

For transmitted light, the observer sees only relative contrast with background sky brightness.



White Plumes

Scattered light originates from both in front and behind the observer. Therefore, the relative position of the sun is significant.

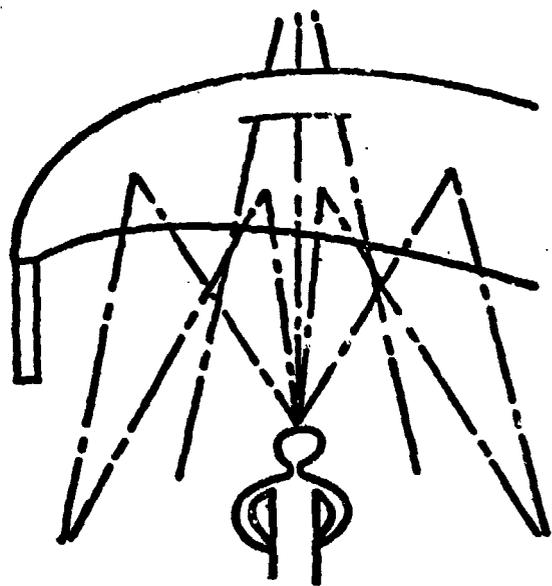


Figure 4. Visibility

$$Q_{\text{ext}} = \frac{2}{x^2} \sum_n (2n + 1) \text{Real}(a_n + b_n)$$

where: $x = \pi d_p / \lambda$

$\lambda =$ wavelength of light in which the plume is viewed (the computer program allows this value to be input, or if left blank assumes an average value for skylight of 0.550 microns)

a_n & b_n are complex Riccati-Bessel function of order " n ":

$$a_n = \frac{\psi_n'(y)\psi_n(x) - m\psi_n(y)\psi_n'(x)}{\psi_n'(y)\xi_n(x) - m\psi_n(y)\xi_n'(x)}$$

$$b_n = \frac{m\psi_n'(y)\psi_n(x) - \psi_n(y)\psi_n'(x)}{m\psi_n'(y)\xi_n(x) - \psi_n(y)\xi_n'(x)}$$

$y = mx$

$m =$ complex refractive index. This is a function of the wavelength of light (λ) at which it is measured, and also the method of generating the soot particles. The computer program allows this value to be input, or if left blank assumes a value for amorphous carbon at 0.550 microns of: $1.96 - 0.66i$ (References 13, 14). (Note: Because the transmitted light is altered both in phase and magnitude, a vector is needed to express the effect of the particles. As in electrical engineering, a vector can be expressed as a complex number with real and imaginary parts. This is the case for the particle refractive index.)

Since these equations involve series functions of complex numbers, their solution is not simple. Instabilities can easily arise (particularly with large values of x), which cause the extinction coefficient to oscillate wildly and even produce negative values. In order to generate stable functions, the

following method is used (References 12, 15):

$$P_n(z) = \frac{d[\ln \psi_n(z)]}{dz} = \frac{\psi_n'(z)}{\psi_n(z)}$$

$$Q_n(z) = \frac{d[\ln \xi_n(z)]}{dz} = \frac{\xi_n'(z)}{\xi_n(z)}$$

(where $z = y$ or x)

Therefore:
$$a_n = \frac{\psi_n(x)}{\xi_n(x)} \left[\frac{P_n(y) - mP_n(x)}{P_n(y) - mQ_n(x)} \right]$$

$$b_n = \frac{\psi_n(x)}{\xi_n(x)} \left[\frac{mP_n(y) - P_n(x)}{mP_n(y) - Q_n(x)} \right]$$

$$\psi_n'(z) = \psi_{n-1}(z) - \frac{n}{z} \psi_n(z)$$

$$\xi_n'(x) = \xi_{n-1}(x) - \frac{n}{x} \xi_n(x)$$

Therefore:
$$P_n(z) = \frac{\psi_{n-1}(z) - \frac{n}{z} \psi_n(z)}{\psi_n(z)} = \frac{\psi_{n-1}(z)}{\psi_n(z)} - \frac{n}{z}$$

$$= \frac{J_{\nu-1}(z)}{J_{\nu}(z)} - \frac{(\nu - 1/2)}{z}$$

where $J =$ Bessel function

$$\nu - 1/2 = n$$

Using Lentz's continued fraction method⁽¹⁵⁾:

$$\frac{J_{\nu-1}}{J_{\nu}} = \frac{|w_1|w_2, w_1| |w_3, w_2, w_1| \dots}{|w_2| |w_3, w_2| \dots}$$

$$w_p = (-1)^{p+1} \frac{2(v + p - 1)}{z}$$

$$|w_p, w_{p-1}, \dots, w_1| = w_p + \frac{1}{w_{p-1} + \frac{1}{w_{p-2} + \frac{1}{\dots}}}$$

Convergence is reached for J_{v-1}/J_v when $|w_p, \dots, w_1|$ in the numerator equals $|w_p, \dots, w_2|$ in the denominator.

$Q_n(x)$ is generated by the following recursion formula:

$$Q_n(x) = \frac{1}{\frac{n}{x} - Q_{n-1}(x)} - \frac{n}{x}$$

(where $Q_0(x) = -i$)

Since

$$\frac{\psi_{n-1}(x)}{\psi_n(x)} = \frac{J_{v-1}(x)}{J_v(x)}$$

$$\psi_n(x) = \psi_{n-1}(x) \left[\frac{J_v(x)}{J_{v-1}(x)} \right]$$

(where $\psi_0(x) = \sin x$)

Finally,

$$\xi_n(x) = \psi_n(x) + i\beta_n(x)$$

$$\beta_n(x) = \left(\frac{2n-1}{x} \right) \beta_{n-1}(x) - \beta_{n-2}(x)$$

(where $\beta_0(x) = \cos x$, and $\beta_1(x) = \frac{\cos x}{x} + \sin x$)

Following Wiscombe (Reference 12) the order (n) of these functions varies from 1 to N, where:

$$N = x + 4x^{1/3} + 2, \quad x \geq 4200$$

$$N = x + 4.05x^{1/3} + 2, \quad 8 < x < 4200$$

$$N = x + 4x^{1/3} + 1, \quad x \leq 8$$

These equations can be used to generate the requisite Riccati-Bessel functions, the extinction coefficient, and finally the transmittance.

Once the transmittance (T) is calculated, the Ringelmann number may be obtained from the empirical correlation of Connor and Hodkinson (References 10, 11), Figure 5.

A computer program was written to perform these calculations. The FORTRAN listing and sample runs are given in the Appendices.

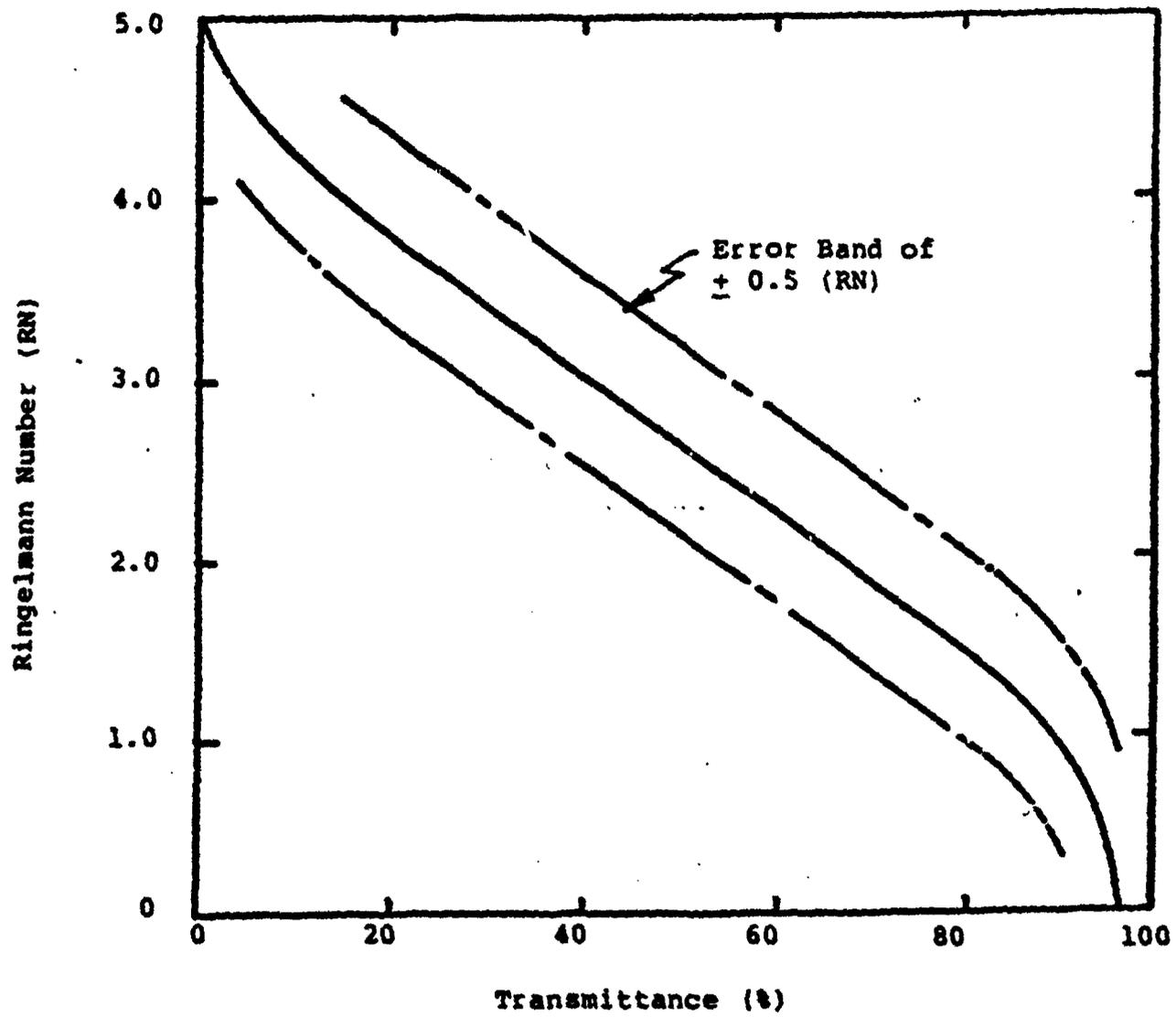


Figure 5. Black Plume Ringelmann Number Correlation with Transmittance (References 10, 11)

SECTION IV

ELECTROSTATIC PRECIPITATOR (ESP) EQUATIONS

A standard mathematical model (Reference 16) was used for predicting the ability of a wire-and-plate ESP to operate as a particle control device:

$$w_i = \frac{8.85 \times 10^{-5} E_c E_p d_{pi}}{\mu} \left(\frac{\xi}{\xi + 2} \right)$$

$$\eta_i = 1 - \exp\left(\frac{-A_p w_i K}{Q}\right)$$

$$\eta = \sum_i \eta_i m_i$$

where: η = overall fractional collection efficiency

η_i = fractional collection efficiency for particles of size d_{pi}

m_i = inlet mass fraction of particles of size d_{pi}

A_p/Q = specific collection area of the ESP (A_p = total collection surface in m^2 ; and Q = gas flow in m^3/sec)

w_i = theoretical migration velocity of particles of size d_{pi} , in m/sec

K = empirical constant

ξ = dielectric constant of the particles (dimensionless)

μ = gas viscosity, in cp

E_c = electric field strength near the discharge electrodes, in kV/cm

E_p = electric field strength near the collecting plates, in kV/cm

d_p = particles size, in microns

This model is based on a field-charging mechanism and is valid for particles larger than 0.5 microns.

If the collection efficiency for an ESP is plotted against the particle size, the resulting curve will exhibit a minimum in the range of 0.2 to 0.7 microns. Above that range, a field-charging mechanism predominates and the efficiency declines with particle size. Below that range, a diffusion-charging mechanism predominates and the efficiency increases with decreasing particle size. Since most of the particles emitted from a test cell are smaller than 0.2 μm , a large K value of 600 was used in order to compensate for the lack of a diffusion-charging mechanism in the model equations.

In order to reduce the amount of input data needed to run the computer program and avoid a prior design of the ESP, the following values were assumed:

$$\xi = 3, \text{ for carbon}$$

$$\mu = 0.024 \text{ cp, for air at } 350^{\circ}\text{F and } 1 \text{ atm}$$

$$E_c = E_p = \frac{40 \text{ kV}}{(4.5 \text{ inches})(2.54 \text{ cm/inch})} = 3.50 \text{ kV/cm}$$

(where 9 inches is generally used as the plate-to-plate spacing in utility-type ESP's, with a secondary voltage of 40 kV).

The computer program takes the uncontrolled test cell emission data, calculates the fractional efficiencies, and then determines the outlet particle size distribution and loading. This information then goes to the visibility portion of the program where the outlet Ringelmann number is calculated.

SECTION V

SCRUBBER EQUATIONS

A standard mathematical model (Reference 17) was used to describe the particle collection efficiency of a high energy venturi scrubber. As with the ESP model, this also required an empirical factor (f) to make the model agree approximately with actual data:

$$\ln(1 - \eta_i) = -\left(\frac{18}{55}\right) \left(\frac{\rho_l}{\rho_p}\right) \left(\frac{Q_l}{Q_g}\right) \left(\frac{d_d}{d_{pi}}\right)^2 \frac{1}{C_i} \{ (k_i f + 0.7) - 1.4 \ln \left(\frac{k_i f + 0.7}{0.7} \right) - \left(\frac{0.49}{k_i f + 0.7} \right) \}$$

This equation is written in dimensionless form, and therefore any consistent set of units may be used:

η_i = fractional collection efficiency for particles of size d_{pi}

ρ_l = liquid density

ρ_p = particle density

Q_l = liquid flow rate

Q_g = gas flow rate

C_i = Cunningham correction factor for gas viscosity; for particles that are the same size or smaller than the mean free path of the gas molecules (λ)

$$= 1 + \frac{2\lambda}{d_{pi}} \left[1.23 + 0.41 \exp \left(\frac{-0.44 d_{pi}}{\lambda} \right) \right] \text{ (dimensionless)}$$

k_i = Stokes' parameter = $\frac{C_i \rho_p d_{pi}^2}{9\mu_g d} v_{gT}$ (dimensionless)

f = 0.5 (dimensionless empirical factor based on the author's experience)

V_{gT} = gas velocity in the venturi throat

$$= \sqrt{\frac{\Delta P_T}{\rho_L} \frac{Q_g}{Q_L} g_c}$$

ΔP_T = pressure drop across the venturi throat

g_c = Newton's Law conversion factor

The following equations require specific units:

d_d = mean drop size in the venturi throat (Reference 18), in microns

$$= \frac{1920}{V_{gT}} \sqrt{\frac{\sigma_L}{\rho_L}} + 3.69 \left(\frac{\mu_L}{\sqrt{\sigma_L \rho_L}} \right)^{0.45} \left(\frac{1000 Q_L}{Q_g} \right)^{1.5}$$

σ_L = surface tension of scrubbing liquid, in dynes/cm

ρ_L = density of scrubbing liquid, in g/cc

μ_L = viscosity of scrubbing liquid, in cp

Q_L = flow rate of scrubbing liquid, in gpm

Q_g = gas flow rate, in cfm

V_{gT} in ft/sec

λ = mean free path of gas molecules, in microns

$$= \frac{3.78 \mu_g}{\sqrt{P_g \rho_g}}$$

μ_g = gas viscosity, in cp

P_g = gas pressure, in psia

ρ_g = gas density, in lbm/ft³

Again, in order to minimize the input data requirements, the following operating conditions were assumed:

- (1) water is the scrubbing liquid at 70°F
- (2) gas properties are those of air at 350°F and 1 atm pressure.

For venturi scrubbers, Q_l/Q_g (the liquid-to-gas ratio) is generally 5 to 30 gpm/1000 cfm, and ΔP_T is 10 to 70 inches of water.

The computer program takes the uncontrolled test cell emission data, calculates the fractional efficiencies, and then determines the outlet particle size distribution and loading. This information then goes to the visibility portion of the program where the outlet Ringelmann number is calculated.

As with electrostatic precipitators, the primary collection mechanism for venturi scrubbers should theoretically change in the range 1.0 to 0.1 μm . Above 1 μm , the particles are collected by an inertial mechanism, while below 0.1 μm a diffusional mechanism should prevail. Again, this would imply a trough in the fractional efficiency curve for particles in the 1.0 to 0.1 μm range. However, in practice, the collection efficiency of venturi scrubbers continues to decline below 0.1 μm , indicating that the predominant mechanism remains inertial. This means that standard venturi scrubbers are inherently less efficient than ESP's in collecting particles smaller than 0.5 μm . One method of overcoming this deficiency has been to induce condensation in the gas stream, either before the scrubber (by quenching), or afterward (by utilizing a two-phase ejector). However, these methods cannot as yet be mathematically modelled with any confidence and have not been included in the computer program.

SECTION VI

RESULTS & DISCUSSION

Grems (Reference 19) measured the particle size distribution, loading, and transmittance from a test cell at McClellan Air Force Base. However, the particle density was unknown. In the present study, this density was used as an empirical parameter to fit the computer results to Grems' data. Excellent agreement was obtained for a particle density of 0.92 g/cc (Table 1). For comparison, computer results are also shown for a particle density of 1.0 g/cc.

Soot particles are porous spheres of carbon, having a high void fraction. Solid carbon has a density of 1.8 to 2.1 g/cc. Therefore, a particle density of 0.92 g/cc implies a void fraction of about 0.53.

Although Grems' data indicated a bimodal particle size distribution, a straight-line log-normal fit was made by the computer program (see Figure 6).

Table 2 shows computer predictions of plume visibility when an electrostatic precipitator or venturi scrubber is used. For the ESP, a specific collection surface of 3281/m² per 1000 m³/min of gas (1000 ft²/1000 cfm) corresponds to an upper limit in commercial applications. Since the gas flow from a test cell is on the order of 200,000 scfm, 200,000 ft² of collecting plate would be required. However, even such a large ESP has only a small effect on the Ringelmann number because of the small particle size. The size range which has the greatest effect on visibility matches the wavelength of visible light--i.e., 0.2 to 0.7 microns. This is precisely the range in which an ESP or any other control device, is least efficient. Therefore, purchase of an electrostatic precipitator larger than the test cell would only have a marginal effect on plume visibility during the few hours per week the test cell is in use.

Similarly, a venturi scrubber designed for the limit of its range of operability would also have a marginal effect on test cell plume visibility. At a liquid-to-gas ratio of 30 gpm/1000 cfm (4.01 m³/min water per 1000 m³/min gas) the scrubber would produce 6000 gpm of waste water for a typical installation; and at a pressure drop of 70 inches of water (131 mm mercury) a 2400 hp tail fan would be required. Under these conditions, the computer predicts an overall particle collection efficiency of 49% with a visibility improvement of 0.5

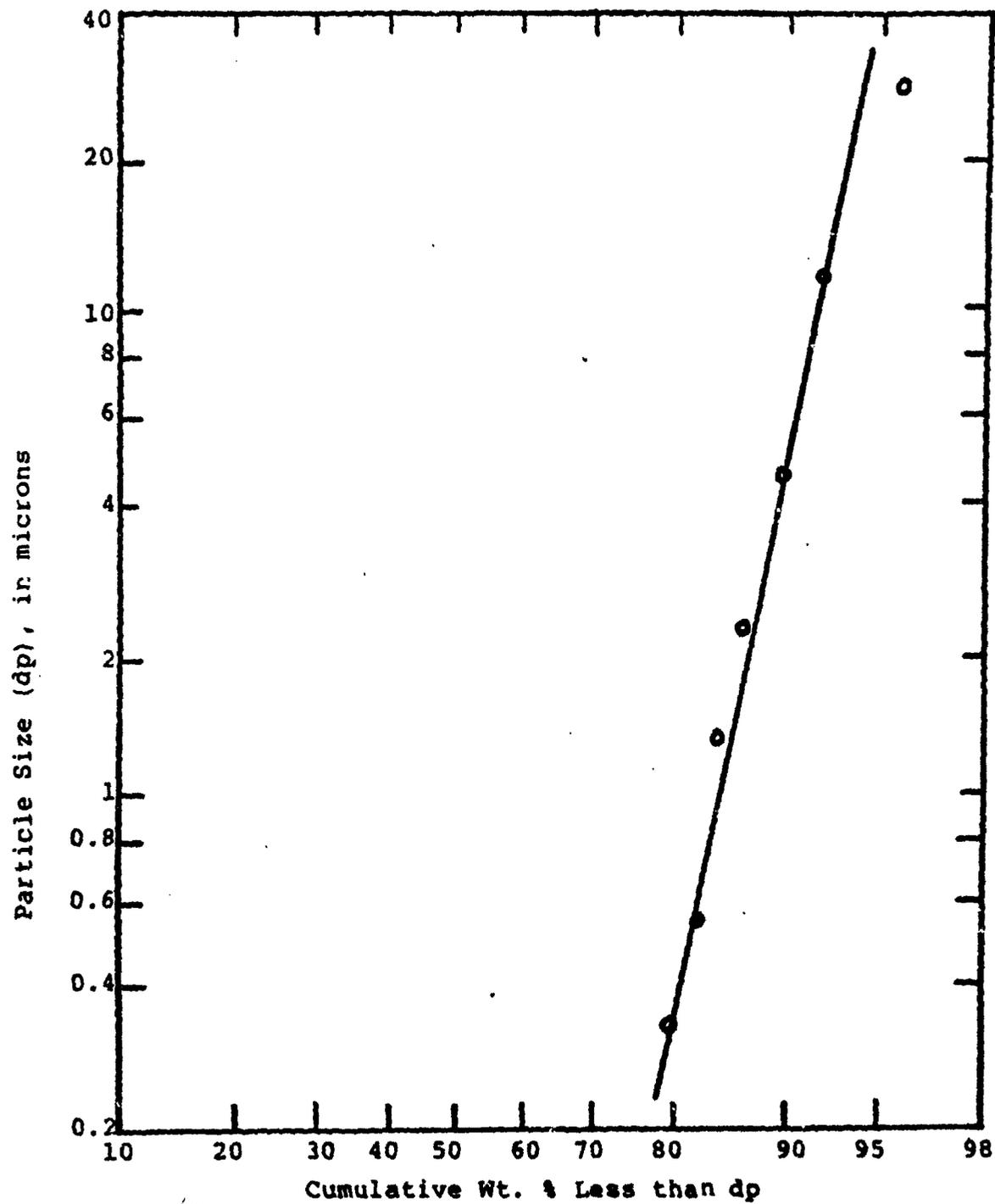


Figure 6. Log-Normal Plot of Size Distribution from Grems' Data (Reference 19)

TABLE 1. COMPARISON OF COMPUTER PREDICTIONS WITH GREN'S DATA FOR J57 ENGINE (Reference 19).

Fuel Firing Rate (lbm/hr)	Particle Loading (mg/m ³)	Measured Particle Size Distribution		Measured Transmittance	Predicted Transmittance (for $\rho_p = 1.0$ g/cc)	d_p^* (for $\rho_p = 0.92$ g/cc)	Predicted Transmittance (for $\rho_p = 0.92$ g/cc)
		Cum Mt % less than	Aerodynamic Diameter ^a ($=d_p$ for $\rho_p = 1.0$ g/cc)				
1000	2.16	93.5	23 μ	960	88.30	24.0 μ	87.40
		91.4	10.5			10.9	
		91.4	4			4.2	
		87.8	2			2.1	
		84.9	1.1			1.15	
		79.9	0.57			0.59	
		73.5	0.33			0.34	
2500	1.95	95.7	20	88	89.4	29.2	88.6
		92.1	11.5			12.0	
		89.7	4.5			4.69	
		86.6	2.2			2.29	
		84.5	1.3			1.36	
		82.7	0.54			0.56	
		79.3	0.33			0.34	
8620	6.00	98.6	22	66	67.9	22.9	65.6
		96.8	9.3			9.78	
		95.1	3.5			3.65	
		92.8	1.7			1.77	
		90.5	0.94			0.98	
		83.8	0.47			0.49	
		77.4	0.23			0.24	
8620	6.34	97.7	21	66	68.4	21.9	66.2
		97.7	9.4			9.80	
		97.4	3.5			3.65	
		95.8	1.7			1.77	
		94.8	0.92			0.96	
		94.2	0.46			0.48	
		89.2	0.22			0.23	

Net open exhaust area = 700 ft² (700 = 26.5 ft, or 8 meters); refractive index for amorphous carbon at 550 nm = 1.96-0.66i

^aFor a cascade injector, $d_{p1}\sqrt{\rho_{p1}} = d_{p2}\sqrt{\rho_{p2}}$

TABLE 2. COMPARISONS OF CONTROLLED AND UNCONTROLLED EMISSIONS.

UNCONTROLLED EMISSIONS: (Plume Width = 8 Meters)

Particle Loading (mg/m ³)	Particle Density (g/cc)	Particle Size (μ m)	Cum.Wt. % Less Than	Transmittance (%)	Ringelmann Number
6.34	0.92	21.9	97.7	66.2	1.5-2.5
		9.80	97.7		
		3.65	97.4		
		1.77	95.8		
		0.96	94.8		
		0.48	94.2		
		0.23	88.2		

CONTROLLED EMISSIONS:

(1) with ESP: SCA = 3281 m² /1000 m³/min

Outlet Particle Loading (mg/m ³)	Collection Efficiency (%)	Transmittance (%)	Ringelmann Number
2.93	53.8	82.8	0.9-1.9

(2) with Venturi Scrubber: L/G = 4.01 m³/min water per 1000 m³/min Gas
 Δ P = 131 mm Hg

Outlet Particle Loading (mg/m ³)	Collection Efficiency (%)	Transmittance (%)	Ringelmann Number
3.26	48.6	81.0	1.0-2.0

Ringelmann number. Operating data with a scrubber (not a venturi) at the Jacksonville Naval Air Station (Reference 20) indicate an average particle collection efficiency of about 75%. However, there was considerable uncertainty in the accuracy of the data. Stockham, et al. (Reference 21) also report an average collection efficiency of 48% with water injection into the augmentor tube. Again, this is not venturi scrubber data, but it does indicate the validity of the order of magnitude of the predicted results. It should be noted that opacity measurement with a scrubber operating is virtually impossible, since the scrubber will emit a large and obscuring steam plume of its own.

REFERENCES

- (1) Morse, H.N., "Legal Briefs," JAPCA, Vol. 31, p. 323. March 1981.
- (2) Shaffernocker, W.M. and Stanforth, C.M., "Smoke Abatement Techniques," SAE Transactions, Vol. 77, pp. 1059-1079. 1968.
- (3) Johansen, K.M. and Kumm, E.L., Determination of Aircraft Turbine Engine Particulates, EPA-650/2-75-055. May 1975.
- (4) Grems, B.C., Smoke Abatement for DOD Test Cells, CEEDO-TR-77-40. July 1977.
- (5) Fenton, D.L., Nordstrom, E.W., Luebcke, E.H., Turbine Engine Particulate Emission Characterization, FAA-RD-77-165. October 1977.
- (6) Lindenhofen, H.E., A Survey of the Air Pollution Potential of Jet Engine Test Facilities, NAPTC-PE-3. October 1972.
- (7) Stockham, J. and Bétz, H., "Study of Visible Exhaust Smoke from Aircraft Jet Engines," SAE paper No. 710428. May 1971.
- (8) Klarman, A.F., Rollo, A.J. and Scott, H.C., Evaluation of Water/Fuel Emulsion Concept for Test Cell Smoke Abatement, NAPC-PE-7. March 1978.
- (9) Wood, A.D., "Correlation Between Smoke Measurements and the Optical Properties of Jet Engine Smoke," paper No. 751119, presented at SAE National Aerospace Engineering & Manufacturing Meeting, Los Angeles, CA. 17-20 Nov. 1975.
- (10) Conner, W.A. and Hodgkinson, J.R., PHS Report No. 999-AP-30. 1967.
- (11) Halow, J.S. and Zeek, S.J., "Predicting Ringelmann number and Optical Characteristics of Plumes," JAPCA, Vol. 23, pp. 676-684. August 1973.
- (12) Wiscombe, W.J., "Improved Mie Scattering Algorithms," Applied Optics, Vol. 19, pp. 1505-1509. May 1980.

- (13) Ensor, D.S. and Pilat, M.J., "Calculation of Smoke Plume Opacity from Particulate Air Pollutant Properties," JAPCA, Vol. 21, pp. 496-501. 1971.
- (14) McDonald, J.E., "Visibility Reduction Due to Jet Exhaust Carbon Particles," J. Appl. Meteorology, Vol. 1, pp. 391-398. 1962.
- (15) Lentz, W.J., "Generating Bessel Functions in Mie Scattering Calculations Using Continued Fractions," Applied Optics, Vol. 15, pp. 668-671. March 1976.
- (16) Stern, A.C., ed. Air Pollution, 3rd ed., Vol. IV. New York: Academic Press. 1977.
- (17) Calvert, S., et al., Scrubber Handbook, NTIS Publication No. PB 213 016. August 1972.
- (18) Nukiyama, S. and Tanasawa, Y., Trans. Soc. Mech. Engrs. (Japan), Vol. 4, p. 86. 1938.
- (19) Grems, B.C., Plume Opacity and Particulate Emissions from a Jet Engine Test Facility, Masters Thesis, U. of California (Davis). 1976.
- (20) Kelly, J., and Chu, E., Jet Engine Test Cells-Emissions and Control Measures: Phase 2, EPA-340/1-78-001 b. April 1978.
- (21) Stockham, J.D., Lannis, M.D., MacNaughton, M.G., and Tarquinio, J.J., "Control of Particulate Emissions from Turbine Engine Test Cells by Cooling Water Injection," JAPCA, Vol. 31, pp. 675-678. June 1981.

APPENDIX A

COMPUTER INPUT DATA FORMAT

The following pages give the format for the input data needed to run the computer program. The particle loading should be determined by EPA Method 5, and the particle size distribution by cascade impactor (Reference 19). The plume width can be approximated by the square root of the net open stack area (the actual stack cross-section minus the area occupied by acoustical baffles). For soot particles, the refractive index is $1.96-0.66i$ at a wavelength of 550 nm, and the particle density was estimated empirically as 0.92 g/cc.

Examples are given for:

- (1) Grems' data (19)
- (2) an electrostatic precipitator with a specific collection area of 1000 ft²/1000 cfm (3281 m² of collecting plate per 1000 m³/min of gas)
- (3) a venturi scrubber with a liquid-to-gas ratio of 30 gpm/1000 cfm (4.01 m³/min of water per 1000 m³/min of gas) at a pressure drop of 70 inches of water (131 mm Hg).

CARD #1

This card contains the title of the case being run, inserted between columns 9 and 10.

CARD #2

This card contains the number of data pairs in the particle size distribution. The minimum number is 2, and the maximum is 100, inserted as an integer between columns 11 and 15 (right justified).

CARD #3 a, b, c, etc.

This card(s) contains the data pairs for the particle size distribution. Columns 1-10, 21-30, 41-50, and 61-70 contain values of the cumulative weight percent less than particle size d_p ; while columns 11-20, 31-40, 51-60, and 71-80 contain the corresponding values of d_p . Therefore, a maximum of four data pairs can fit on one card. If there are more data pairs (as per CARD #2), these are put on subsequent cards, until the total number of data pairs (cumulative weight percent less than d_p , and d_p) is equal to the number specified in CARD #2. All values are floating point numbers, with four digits (or blanks) to the right of the decimal point (right justified).

CARD #4

This card contains the following physical parameters:
columns 11-15--the effective stack diameter in meters
(based on the net open area) expressed as a floating
point number, with two digits (or blanks) to the right
of the decimal point (right justified)

columns 16-22--the particle loading in mg/m^3 , as a floating point number, with two digits (or blanks) to the right of the decimal point (right justified)

columns 23-28--the particle density in g/cm^3 , as a floating point number, with two digits (or blanks) to the right of the decimal point (right justified)

columns 29-46--the particle refractive index. Columns 29-37 contain the real part, and 38-46 the imaginary part; both as floating point numbers with two digits (or blanks) to the right of the decimal point (right justified). If these columns are left completely blank, a value for amorphous carbon of $1.96-0.66i$ is assumed by the program. Note that the refractive index and the wavelength that follows must be consistent

columns 47-53--the wavelength of light (in microns) at which the refractive index was measured, and at which the plume is presumed to be viewed. This must be expressed as a floating point number with three digits (or blanks) to the right of the decimal point (right justified). If columns 29-46 were left blank, these columns should also be left completely blank, in which case the program assumes a value of 0.550 microns.

CARD #5

This card contains (in column 2) an integer number which

indicates whether or not a particulate control device (electrostatic precipitator or venturi scrubber) has been installed on the test cell exhaust:

- zero (0) means no control device
- 1 means an electrostatic precipitator
- 2 means a venturi scrubber

These are the only permissible cases.

CARD #6

This card depends on the code given in CARD #5.

- (a) If there is no control device (0 in column 2 of CARD #5), CARD #6 does not exist.
- (b) If an electrostatic precipitator is indicated by CARD #5, CARD #6 must contain the specific collection area (in m^2 of plate area per 1000 m^3 /min of exhaust gas) in columns 11-17 as a floating point number with one digit (or blank) to the right of the decimal point (right justified).
- (c) If a venturi scrubber is indicated by CARD #5, CARD #6 must contain the liquid-to-gas ratio (in m^3 /min water per 1000 m^3 /min exhaust gas) in columns 11-15; and the scrubber pressure drop (in mm mercury) in columns 16-23; both expressed as floating point numbers, with two digits (or blanks) to the right of the decimal point (right justified).

CARD #7

This card contains (in column 2) a code which tells the computer if more cases are to follow:

zero (0) signifies no more cases

1 means an additional case follows

For each additional case, CARDS #1 to 7 must be repeated, even if some of the data remain the same.

PROGRAM Coding

PROGRAM NAME: **GREMS' DATA** DATE: _____ PAGE 1 OF 2

STATEMENT NUMBER	STATEMENT	OPERATION NUMBER
0	NJIT1	
1	7	
	88.2	0.22
	97.4	3.5
	8.0	6.34
	94.2	0.92
	0.46	
	94.8	
	97.7	21.0
	0.92	
	95.8	
	1.7	
0	NJIT2	
1	7	
	77.4	0.23
	95.1	3.5
	8.0	6.08
	83.8	0.92
	0.47	
	90.5	
	9.3	
	98.6	22.0
	0.94	
	92.8	
	1.7	
0	NJIT3	
1	7	
	73.5	0.33
	91.4	4.0
	8.0	2.16
	79.9	0.92
	0.57	
	84.9	
	10.5	
	93.5	23.0
	1.1	
	87.8	
	2.0	

32

Figure A-1. Computer Input Data Format

PARTIAL CODE

GRAMS' DATA (CONTINUED) 2-2

MEASUREMENT	PERCENT WATER						
79.3	0.33	82.7	0.54	84.5	1.3	86.6	2.2
89.7	1.5	92.1	11.5	95.7	20.9		
0	0.0	1.95	0.92				
0							

NJIT

33

Figure A-1. Continued. Computer Input Data Format

PROGRAM **SCRUBBER CASE**

POSTAGE CODE

1-1

POSTAGE STATEMENT

88.2	97.4	7	9.22	3.5	94.2	97.7	0.46	94.8	0.92	95.8	1.7
			9.0	6.34	0.92		9.4	97.7	21.0		
			4.01	131.0							

Figure A-1. Continued. Computer Input Data Format

APPENDIX B
OUTPUT FORMAT

Example outputs are given on the next few pages for the input data shown in the previous section (i.e. Grems' data, an ESP, and a venturi scrubber).

.....
N-111
.....

Figure B-1. Output Format

PLUME WIDTH (IN METERS) =	0.88	PARTICLE DENSITY (IN G/CC) =	.92
OUTLET LOADING (IN MG/M3) =	0.34	REFRACTIVE INDEX: REAL PART	1.96
LIGHT WAVELENGTH (IN MICRONS) =	1.50	REFRACTIVE INDEX: IMAGINARY PART	-.16

OUTLET PARTICLE SIZE DISTRIBUTION

CUMULATIVE WEIGHT PERCENT LESS THAN STATED PARTICLE SIZE	PARTICLE SIZE (MICRON)
1.0	1.000
2.0	.800
3.0	.600
4.0	.500
5.0	.400
6.0	.300
7.0	.250
8.0	.200
9.0	.150
10.0	.100
15.0	.080
20.0	.060
25.0	.050
30.0	.040
35.0	.030
40.0	.025
45.0	.020
50.0	.015
55.0	.010
60.0	.008
65.0	.006
70.0	.005
75.0	.004
80.0	.003
85.0	.002
90.0	.001
95.0	.001
99.0	.001
99.5	.001
99.9	.001
100.0	.001

RINGELMANN NUMBER RANGE

1.5 - 2.5

PERCENT LIGHT TRANSMITTED = 66.15

Figure B-1. Continued. Output Format

.....
NJ12
.....

Figure E-1. Continued. Output Format

PLUME WIDTH (IN METERS) = 8.00 PARTICLE DENSITY (IN G/CC) = .02
 OUTLET LOADING (IN MG/M3) = 2.16 REFRACTIVE INDEX: REAL PART = 1.96
 LIGHT WAVELENGTH (IN MICRONS) = 0.550 IMAGINARY PART = .100

OUTLET PARTICLE SIZE DISTRIBUTION

CUMULATIVE WEIGHT PERCENT LESS THAN STATED PARTICLE SIZE	PARTICLE SIZE (MICRONS)
1.0	0.000
2.0	0.000
3.0	0.000
4.0	0.000
5.0	0.000
6.0	0.000
7.0	0.000
8.0	0.000
9.0	0.000
10.0	0.000
15.0	0.000
20.0	0.000
25.0	0.000
30.0	0.000
35.0	0.000
40.0	0.001
45.0	0.002
50.0	0.004
55.0	0.007
60.0	0.011
65.0	0.015
70.0	0.020
75.0	0.025
80.0	0.030
85.0	0.035
90.0	0.040
95.0	0.045
97.0	0.050
98.0	0.055
99.0	0.060
99.5	0.065
99.8	0.070
99.9	0.075
100.0	0.080
100.0	0.085
100.0	0.090
100.0	0.095
100.0	0.100
100.0	0.105
100.0	0.110
100.0	0.115
100.0	0.120
100.0	0.125
100.0	0.130
100.0	0.135
100.0	0.140
100.0	0.145
100.0	0.150
100.0	0.155
100.0	0.160
100.0	0.165
100.0	0.170
100.0	0.175
100.0	0.180
100.0	0.185
100.0	0.190
100.0	0.195
100.0	0.200
100.0	0.205
100.0	0.210
100.0	0.215
100.0	0.220
100.0	0.225
100.0	0.230
100.0	0.235
100.0	0.240
100.0	0.245
100.0	0.250
100.0	0.255
100.0	0.260
100.0	0.265
100.0	0.270
100.0	0.275
100.0	0.280
100.0	0.285
100.0	0.290
100.0	0.295
100.0	0.300
100.0	0.305
100.0	0.310
100.0	0.315
100.0	0.320
100.0	0.325
100.0	0.330
100.0	0.335
100.0	0.340
100.0	0.345
100.0	0.350
100.0	0.355
100.0	0.360
100.0	0.365
100.0	0.370
100.0	0.375
100.0	0.380
100.0	0.385
100.0	0.390
100.0	0.395
100.0	0.400
100.0	0.405
100.0	0.410
100.0	0.415
100.0	0.420
100.0	0.425
100.0	0.430
100.0	0.435
100.0	0.440
100.0	0.445
100.0	0.450
100.0	0.455
100.0	0.460
100.0	0.465
100.0	0.470
100.0	0.475
100.0	0.480
100.0	0.485
100.0	0.490
100.0	0.495
100.0	0.500
100.0	0.505
100.0	0.510
100.0	0.515
100.0	0.520
100.0	0.525
100.0	0.530
100.0	0.535
100.0	0.540
100.0	0.545
100.0	0.550
100.0	0.555
100.0	0.560
100.0	0.565
100.0	0.570
100.0	0.575
100.0	0.580
100.0	0.585
100.0	0.590
100.0	0.595
100.0	0.600
100.0	0.605
100.0	0.610
100.0	0.615
100.0	0.620
100.0	0.625
100.0	0.630
100.0	0.635
100.0	0.640
100.0	0.645
100.0	0.650
100.0	0.655
100.0	0.660
100.0	0.665
100.0	0.670
100.0	0.675
100.0	0.680
100.0	0.685
100.0	0.690
100.0	0.695
100.0	0.700
100.0	0.705
100.0	0.710
100.0	0.715
100.0	0.720
100.0	0.725
100.0	0.730
100.0	0.735
100.0	0.740
100.0	0.745
100.0	0.750
100.0	0.755
100.0	0.760
100.0	0.765
100.0	0.770
100.0	0.775
100.0	0.780
100.0	0.785
100.0	0.790
100.0	0.795
100.0	0.800
100.0	0.805
100.0	0.810
100.0	0.815
100.0	0.820
100.0	0.825
100.0	0.830
100.0	0.835
100.0	0.840
100.0	0.845
100.0	0.850
100.0	0.855
100.0	0.860
100.0	0.865
100.0	0.870
100.0	0.875
100.0	0.880
100.0	0.885
100.0	0.890
100.0	0.895
100.0	0.900
100.0	0.905
100.0	0.910
100.0	0.915
100.0	0.920
100.0	0.925
100.0	0.930
100.0	0.935
100.0	0.940
100.0	0.945
100.0	0.950
100.0	0.955
100.0	0.960
100.0	0.965
100.0	0.970
100.0	0.975
100.0	0.980
100.0	0.985
100.0	0.990
100.0	0.995
100.0	1.000

RINGELMANN NUMBER RANGE

10 - 100

PERCENT LIGHT TRANSMITTED = 87.37

Figure B-1. Continued. Output Format

.....
NJIT4
.....

Figure B-1. Continued. Output Format

NJTS

*** AN ELECTROSTATIC PRECIPITATOR HAS BEEN ASSUMED AS A CONTROL DEVICE ***
THE SPECIFIC COLLECTION SURFACE IS 3280.8 M² PER 1000 M³/MIN
WHICH PRODUCES AN OVERALL COLLECTION EFFICIENCY OF 53.77 %

ESP INLET LOADING = 6.34 MG/M³

INLET PARTICLE SIZE DISTRIBUTION

CUM.WT.%	MICRONS
1.0	0.030
2.0	0.330
3.0	0.000
4.7	0.000
5.0	0.000
10.0	0.000
15.0	0.000
20.0	0.000
30.0	0.000
40.0	0.000
50.0	0.000
60.0	0.001
70.0	0.004
80.0	0.022
85.0	0.060
90.0	0.213
95.0	1.382
96.7	2.381
97.0	4.650
98.0	11.317
99.0	65.977

Figure B-1. Continued. Output Format

PLUME WIDTH (IN METERS) =	0.00	PARTICLE DENSITY (IN G/CC) =	0.92
OUTLET LOADING (IN MG/M3) =	2.93	REFRACTIVE INDEX: REAL PART	1.94
LIGHT WAVELENGTH (IN MICRONS) =	0.550	IMAGINARY PART	-0.04

OUTLET PARTICLE SIZE DISTRIBUTION

CUMULATIVE WEIGHT PERCENT LESS THAN STATED PARTICLE SIZE	PARTICLE SIZE (MICRON)
---	---------------------------

1.0	0.000
2.0	0.000
3.0	0.000
4.0	0.000
5.0	0.000
10.0	0.000
15.0	0.000
20.0	0.000
30.0	0.000
40.0	0.000
50.0	0.000
60.0	0.000
70.0	0.000
80.0	0.000
85.0	0.000
90.0	0.000
95.0	0.000
96.0	0.000
97.0	0.000
98.0	0.000
99.0	0.000

RINSELMAN NUMBER RANGE

0.0 - 1.0

PERCENT LIGHT TRANSMITTED = 82.00

Figure B-1. Continued. Output Format

PLUME WIDTH (IN METERS) =	0.00	PARTICLE DENSITY (IN G/CC) =	0.02
OUTLET LOADING (IN MG/M3) =	3.24	REFRACTIVE INDEX: REAL PART	1.06
LIGHT WAVELENGTH (IN MICRONS) =	0.550	IMAGINARY PART	-0.66

OUTLET PARTICLE SIZE DISTRIBUTION

CUMULATIVE WEIGHT PERCENT LESS THAN STATED PARTICLE SIZE	PARTICLE SIZE (MICRON)
1.0	0.000
7.0	0.000
3.0	0.000
4.0	0.000
5.0	0.000
10.0	0.000
15.0	0.000
20.0	0.000
30.0	0.000
40.0	0.000
50.0	0.000
60.0	0.000
70.0	0.001
80.0	0.003
85.0	0.005
90.0	0.009
95.0	0.024
96.0	0.035
97.0	0.051
98.0	0.084
99.0	0.182

SINGELMANN NUMBER RANGE

1.0 - 2.0

PERCENT LIGHT TRANSMITTED = 86.00

Figure B-1. Continued. Output Format

PLUME WIDTH (IN METERS) =	0.00	PARTICLE DENSITY (IN G/CC) =	0.02
OUTLET LOADING (IN KG/H3) =	3.26	REFRACTIVE INDEX: REAL PART	1.06
LIGHT WAVELENGTH (IN MICRONS) =	0.550	IMAGINARY PART	-0.66

OUTLET PARTICLE SIZE DISTRIBUTION

CUMULATIVE WEIGHT PERCENT LESS THAN STATED PARTICLE SIZE	PARTICLE SIZE (MICRON)
---	---------------------------

1.0	0.000
2.0	0.000
3.0	0.000
4.0	0.000
5.0	0.000
10.0	0.000
15.0	0.000
20.0	0.000
30.0	0.000
40.0	0.000
50.0	0.000
60.0	0.000
70.0	0.001
80.0	0.003
85.0	0.005
90.0	0.009
95.0	0.024
96.0	0.035
97.0	0.051
98.0	0.084
99.0	0.182

SINGELMAN NUMBER RANGE

1.0 - 2.0

PERCENT LIGHT TRANSMITTED = 80.99

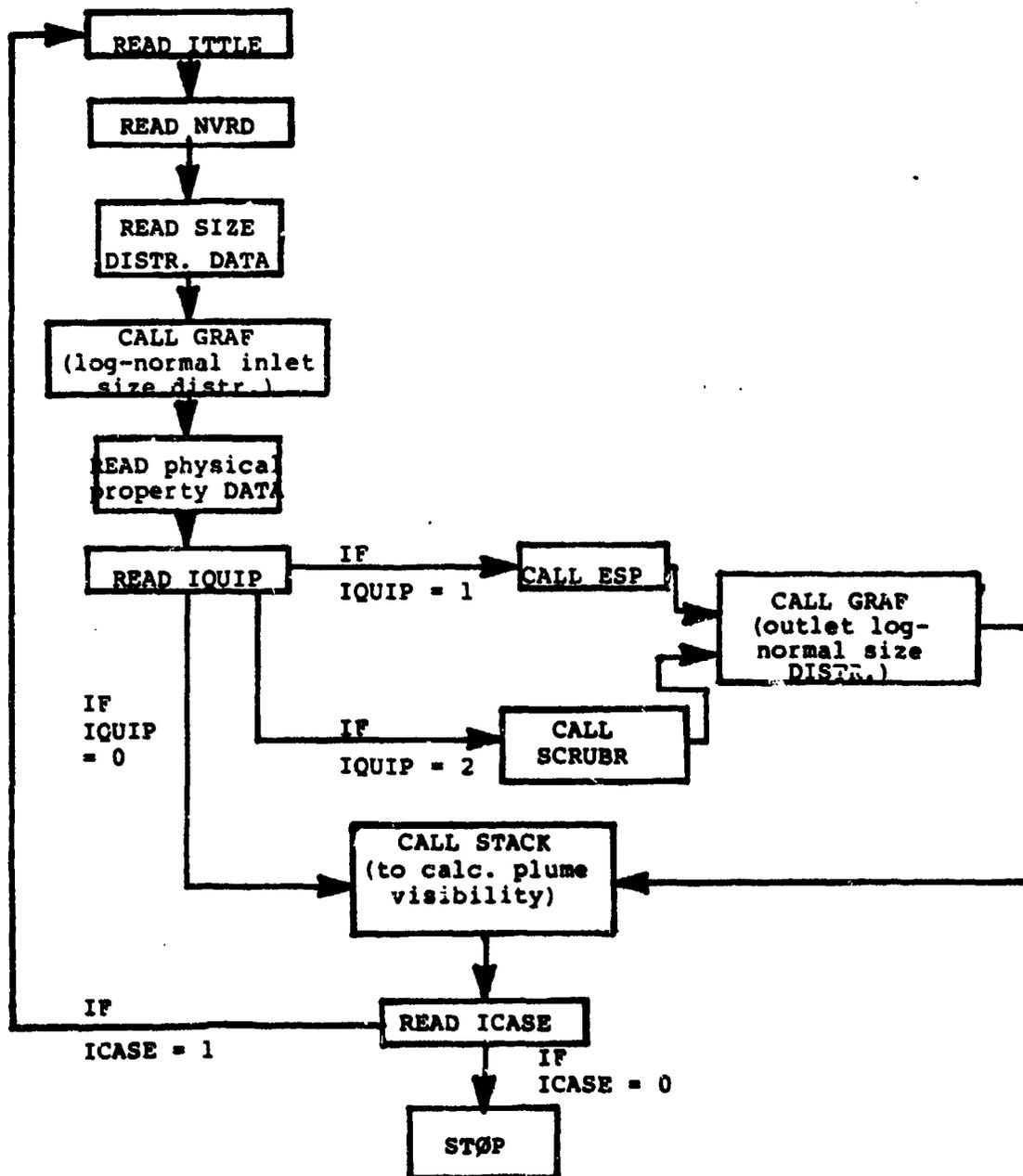
Figure B-1. Continued. Output Format

APPENDIX C
FORTRAN LISTING

The following FORTRAN program performs the various visibility and control calculations. It was originally written on a UNIVAC 90/80-3, and later converted for use on a CDC 6600. The version shown here is for the CDC 6600.

COMPUTER PROGRAM FLOW CHART

Figure C-1.




```

C VISIBLE LIGHT IS ASSUMED OF 0.550 MICRONS. *****
C
C     NIN = REAL (REFRAC)
C     NIM = REAL (REFRAC)
100 REFRAC (0.0) GO TO 0000
C     GO TO 0000
C     IF (IN (EU, 2.0)) GO TO 0007
C     GO TO 0000
101 REFRACT = 1.0000000000000000
C     NIN = REAL (REFRAC)
C     NIM = REAL (REFRAC)
102 REFRAC (0.0) REFRAC = 0.550
103 REFRAC (0.0) REFRAC = 0.550
104 FORMAT (1X, 11)
C     IF (IN (FR, 0.0)) GO TO 6
C     IF (IN (FR, 0.0)) GO TO 7
105 GO TO 8
C
C     READ ESP SPECIFIC COLLECTION AREA (SCA) AS WZ OF PLATE AREA
C     PER 1000 M3/MIN OF GAS *****
C     READ (1050) SCA
106 FINMAT (10X, 11)
C     CALL ESP (INPE, CVDZ)
C
C     SUBROUTINE *ESP* CALCULATES THE OUTLET PARTICLE SIZE
C     DISTRIBUTION AND LOADING FROM AN ELECTROSTATIC PRECIPITATOR.
C     THIS DISTRIBUTION IS THEN NORMALIZED BY CALLING *GRAFA*. ***
107 IF (IN (RO, 1)) GO TO 11
C     CALL GRAF (INPE, CVDZ, NE, DP)
108 GO TO 6
C
C     READ SCRUMBER D-TA: LIQUID/GAS RATIO (ALG) AS M3/MIN LIQUID
C     PER 1000 M3/MIN GAS AND PRESSURE DROP (PD) AS MM. MERCURY **
C     READ (1060) ALG, PD
109 FINMAT (10X, 11, 2, 2)
C
C     SUBROUTINE *SCRUM* CALCULATES THE OUTLET PARTICLE SIZE
C     DISTRIBUTION AND LOADING FROM A HIGH ENERGY VENTURI SCRUMBER.
C     THIS DISTRIBUTION IS THEN NORMALIZED BY CALLING *GRAFA*. ***
110 CALL SCRUM (INPE, UPS, CVDZ)
C     IF (IN (RO, 1)) GO TO 11
C     CALL GRAF (INPE, CVDZ, NS, DP)
111 GO TO 6
C
C     SUBROUTINE *STACK* PERFORMS THE VISIBILITY CALCULATIONS. ****
112 CALL STACK (INPE)
C     READ (1080) ICAP
113 FINMAT (1X, 11)
C     IF (IN (FR, 0.0)) GO TO 2
114 GO TO 1
C
C     STOP
115 END

```

Figure C-2. Continued. Computer Source Code Listing


```

1      SUBROUTINE FSP (IMP, DPE, CWD)
2
3      *** FSP CALCULATES THE INLET PARTICLE SIZE DISTRIBUTION AND LOADING
4      FROM AN ELECTROSTATIC PRECIPITATOR WITH SPECIFIC COLLECTION SURFACE
5
6      COMMON/RESP/FF, SCA, NE
7      COMMON/ANON/ININ, CLC(101)
8      COMMON/ACON/??, MOP
9      DIMENSION DP(101), DPE(101), CWD(101), SUME(101)
10
11     *** EQUATIONS USED TO CALCULATE ESP EFFICIENCY
12
13     FRACTIONAL EFFICIENCY = 1 - EXP(-SCA*DK)
14     BY FIELD CHARGING MECHANISM
15     V = MIGRATION VELOCITY (IN/SEC) = 8.05E-05*E*EP*DP*(CD/(CD+2))/V
16     F = 40 KV DISCHARGE VOLTAGE AND 6 INCH PLATE-TO-PLATE SPACING
17     EC = EP = FIELD STRENGTH = 3.50 KV/CM
18     DP = INLET PARTICLE SIZE IN MICRONS
19     CD = DIELECTRIC CONSTANT OF PARTICULATE, WHICH FOR CARBON IS ABOUT
20     FOR AIR AT 750 F
21     V = GAS VISCOSITY = 1.024 CP
22     K = FUDGE FACTOR, SINCE THE MODEL EQUATION FOR THE
23     MIGRATION VELOCITY APPLIES TO PARTICLES LARGER THAN
24     0.5 MICRONS (WHICH IS LARGER THAN MOST OF THE PARTICLES
25     ENTERED BY THE TEST CELL), A "K" VALUE OF 600 WAS USED
26     IN THE PROGRAM
27     CONTINUE
28     FRACTIONAL EFFICIENCY = 1 - EXP(-.075*SCA*DP)
29     WHERE SCA IS IN M2 OF COLLECTING PLATE PER 1000 M3/MIN OF GAS FLOW
30     DP IS IN MICRONS
31     95
32     96
33     97
34     98
35     99
36     100
37     101
38     102
39     103
40     104
41     105
42     106
43     107
44     108
45     109
46     110
47     111
48     112
49     113
50     114
51     115
52     116
53     117
54     118
55     119
56     120
57     121
58     122
59     123
60     124
61     125
62     126
63     127
64     128
65     129
66     130
67     131
68     132
69     133
70     134
71     135
72     136
73     137
74     138
75     139
76     140
77     141
78     142
79     143
80     144
81     145
82     146
83     147
84     148
85     149
86     150
87     151
88     152
89     153
90     154
91     155
92     156
93     157
94     158
95     159
96     160
97     161
98     162
99     163
100    164
101    165
102    166
103    167
104    168
105    169
106    170
107    171
108    172
109    173
110    174
111    175
112    176
113    177
114    178
115    179
116    180
117    181
118    182
119    183
120    184
121    185
122    186
123    187
124    188
125    189
126    190
127    191
128    192
129    193
130    194
131    195
132    196
133    197
134    198
135    199
136    200
137    201
138    202
139    203
140    204
141    205
142    206
143    207
144    208
145    209
146    210
147    211
148    212
149    213
150    214
151    215
152    216
153    217
154    218
155    219
156    220
157    221
158    222
159    223
160    224
161    225
162    226
163    227
164    228
165    229
166    230
167    231
168    232
169    233
170    234
171    235
172    236
173    237
174    238
175    239
176    240
177    241
178    242
179    243
180    244
181    245
182    246
183    247
184    248
185    249
186    250
187    251
188    252
189    253
190    254
191    255
192    256
193    257
194    258
195    259
196    260
197    261
198    262
199    263
200    264
201    265
202    266
203    267
204    268
205    269
206    270
207    271
208    272
209    273
210    274
211    275
212    276
213    277
214    278
215    279
216    280
217    281
218    282
219    283
220    284
221    285
222    286
223    287
224    288
225    289
226    290
227    291
228    292
229    293
230    294
231    295
232    296
233    297
234    298
235    299
236    300
237    301
238    302
239    303
240    304
241    305
242    306
243    307
244    308
245    309
246    310
247    311
248    312
249    313
250    314
251    315
252    316
253    317
254    318
255    319
256    320
257    321
258    322
259    323
260    324
261    325
262    326
263    327
264    328
265    329
266    330
267    331
268    332
269    333
270    334
271    335
272    336
273    337
274    338
275    339
276    340
277    341
278    342
279    343
280    344
281    345
282    346
283    347
284    348
285    349
286    350
287    351
288    352
289    353
290    354
291    355
292    356
293    357
294    358
295    359
296    360
297    361
298    362
299    363
300    364
301    365
302    366
303    367
304    368
305    369
306    370
307    371
308    372
309    373
310    374
311    375
312    376
313    377
314    378
315    379
316    380
317    381
318    382
319    383
320    384
321    385
322    386
323    387
324    388
325    389
326    390
327    391
328    392
329    393
330    394
331    395
332    396
333    397
334    398
335    399
336    400
337    401
338    402
339    403
340    404
341    405
342    406
343    407
344    408
345    409
346    410
347    411
348    412
349    413
350    414
351    415
352    416
353    417
354    418
355    419
356    420
357    421
358    422
359    423
360    424
361    425
362    426
363    427
364    428
365    429
366    430
367    431
368    432
369    433
370    434
371    435
372    436
373    437
374    438
375    439
376    440
377    441
378    442
379    443
380    444
381    445
382    446
383    447
384    448
385    449
386    450
387    451
388    452
389    453
390    454
391    455
392    456
393    457
394    458
395    459
396    460
397    461
398    462
399    463
400    464
401    465
402    466
403    467
404    468
405    469
406    470
407    471
408    472
409    473
410    474
411    475
412    476
413    477
414    478
415    479
416    480
417    481
418    482
419    483
420    484
421    485
422    486
423    487
424    488
425    489
426    490
427    491
428    492
429    493
430    494
431    495
432    496
433    497
434    498
435    499
436    500
437    501
438    502
439    503
440    504
441    505
442    506
443    507
444    508
445    509
446    510
447    511
448    512
449    513
450    514
451    515
452    516
453    517
454    518
455    519
456    520
457    521
458    522
459    523
460    524
461    525
462    526
463    527
464    528
465    529
466    530
467    531
468    532
469    533
470    534
471    535
472    536
473    537
474    538
475    539
476    540
477    541
478    542
479    543
480    544
481    545
482    546
483    547
484    548
485    549
486    550
487    551
488    552
489    553
490    554
491    555
492    556
493    557
494    558
495    559
496    560
497    561
498    562
499    563
500    564
501    565
502    566
503    567
504    568
505    569
506    570
507    571
508    572
509    573
510    574
511    575
512    576
513    577
514    578
515    579
516    580
517    581
518    582
519    583
520    584
521    585
522    586
523    587
524    588
525    589
526    590
527    591
528    592
529    593
530    594
531    595
532    596
533    597
534    598
535    599
536    600
537    601
538    602
539    603
540    604
541    605
542    606
543    607
544    608
545    609
546    610
547    611
548    612
549    613
550    614
551    615
552    616
553    617
554    618
555    619
556    620
557    621
558    622
559    623
560    624
561    625
562    626
563    627
564    628
565    629
566    630
567    631
568    632
569    633
570    634
571    635
572    636
573    637
574    638
575    639
576    640
577    641
578    642
579    643
580    644
581    645
582    646
583    647
584    648
585    649
586    650
587    651
588    652
589    653
590    654
591    655
592    656
593    657
594    658
595    659
596    660
597    661
598    662
599    663
600    664
601    665
602    666
603    667
604    668
605    669
606    670
607    671
608    672
609    673
610    674
611    675
612    676
613    677
614    678
615    679
616    680
617    681
618    682
619    683
620    684
621    685
622    686
623    687
624    688
625    689
626    690
627    691
628    692
629    693
630    694
631    695
632    696
633    697
634    698
635    699
636    700
637    701
638    702
639    703
640    704
641    705
642    706
643    707
644    708
645    709
646    710
647    711
648    712
649    713
650    714
651    715
652    716
653    717
654    718
655    719
656    720
657    721
658    722
659    723
660    724
661    725
662    726
663    727
664    728
665    729
666    730
667    731
668    732
669    733
670    734
671    735
672    736
673    737
674    738
675    739
676    740
677    741
678    742
679    743
680    744
681    745
682    746
683    747
684    748
685    749
686    750
687    751
688    752
689    753
690    754
691    755
692    756
693    757
694    758
695    759
696    760
697    761
698    762
699    763
700    764
701    765
702    766
703    767
704    768
705    769
706    770
707    771
708    772
709    773
710    774
711    775
712    776
713    777
714    778
715    779
716    780
717    781
718    782
719    783
720    784
721    785
722    786
723    787
724    788
725    789
726    790
727    791
728    792
729    793
730    794
731    795
732    796
733    797
734    798
735    799
736    800
737    801
738    802
739    803
740    804
741    805
742    806
743    807
744    808
745    809
746    810
747    811
748    812
749    813
750    814
751    815
752    816
753    817
754    818
755    819
756    820
757    821
758    822
759    823
760    824
761    825
762    826
763    827
764    828
765    829
766    830
767    831
768    832
769    833
770    834
771    835
772    836
773    837
774    838
775    839
776    840
777    841
778    842
779    843
780    844
781    845
782    846
783    847
784    848
785    849
786    850
787    851
788    852
789    853
790    854
791    855
792    856
793    857
794    858
795    859
796    860
797    861
798    862
799    863
800    864
801    865
802    866
803    867
804    868
805    869
806    870
807    871
808    872
809    873
810    874
811    875
812    876
813    877
814    878
815    879
816    880
817    881
818    882
819    883
820    884
821    885
822    886
823    887
824    888
825    889
826    890
827    891
828    892
829    893
830    894
831    895
832    896
833    897
834    898
835    899
836    900
837    901
838    902
839    903
840    904
841    905
842    906
843    907
844    908
845    909
846    910
847    911
848    912
849    913
850    914
851    915
852    916
853    917
854    918
855    919
856    920
857    921
858    922
859    923
860    924
861    925
862    926
863    927
864    928
865    929
866    930
867    931
868    932
869    933
870    934
871    935
872    936
873    937
874    938
875    939
876    940
877    941
878    942
879    943
880    944
881    945
882    946
883    947
884    948
885    949
886    950
887    951
888    952
889    953
890    954
891    955
892    956
893    957
894    958
895    959
896    960
897    961
898    962
899    963
900    964
901    965
902    966
903    967
904    968
905    969
906    970
907    971
908    972
909    973
910    974
911    975
912    976
913    977
914    978
915    979
916    980
917    981
918    982
919    983
920    984
921    985
922    986
923    987
924    988
925    989
926    990
927    991
928    992
929    993
930    994
931    995
932    996
933    997
934    998
935    999
936    1000
937    1001
938    1002
939    1003
940    1004
941    1005
942    1006
943    1007
944    1008
945    1009
946    1010
947    1011
948    1012
949    1013
950    1014
951    1015
952    1016
953    1017
954    1018
955    1019
956    1020
957    1021
958    1022
959    1023
960    1024
961    1025
962    1026
963    1027
964    1028
965    1029
966    1030
967    1031
968    1032
969    1033
970    1034
971    1035
972    1036
973    1037
974    1038
975    1039
976    1040
977    1041
978    1042
979    1043
980    1044
981    1045
982    1046
983    1047
984    1048
985    1049
986    1050
987    1051
988    1052
989    1053
990    1054
991    1055
992    1056
993    1057
994    1058
995    1059
996    1060
997    1061
998    1062
999    1063
1000   1064
1001   1065
1002   1066
1003   1067
1004   1068
1005   1069
1006   1070
1007   1071
1008   1072
1009   1073
1010   1074
1011   1075
1012   1076
1013   1077
1014   1078
1015   1079
1016   1080
1017   1081
1018   1082
1019   1083
1020   1084
1021   1085
1022   1086
1023   1087
1024   1088
1025   1089
1026   1090
1027   1091
1028   1092
1029   1093
1030   1094
1031   1095
1032   1096
1033   1097
1034   1098
1035   1099
1036   1100
1037   1101
1038   1102
1039   1103
1040   1104
1041   1105
1042   1106
1043   1107
1044   1108
1045   1109
1046   1110
1047   1111
1048   1112
1049   1113
1050   1114
1051   1115
1052   1116
1053   1117
1054   1118
1055   1119
1056   1120
1057   1121
1058   1122
1059   1123
1060   1124
1061   1125
1062   1126
1063   1127
1064   1128
1065   1129
1066   1130
1067   1131
1068   1132
1069   1133
1070   1134
1071   1135
1072   1136
1073   1137
1074   1138
1075   1139
1076   1140
1077   1141
1078   1142
1079   1143
1080   1144
1081   1145
1082   1146
1083   1147
1084   1148
1085   1149
1086   1150
1087   1151
1088   1152
1089   1153
1090   1154
1091   1155
1092   1156
1093   1157
1094   1158
1095   1159
1096   1160
1097   1161
1098   1162
1099   1163
1100   1164
1101   1165
1102   1166
1103   1167
1104   1168
1105   1169
1106   1170
1107   1171
1108   1172
1109   1173
1110   1174
1111   1175
1112   1176
1113   1177
1114   1178
1115   1179
1116   1180
1117   1181
1118   1182
1119   1183
1120   1184
1121   1185
1122   1186
1123   1187
1124   1188
1125   1189
1126   1190
1127   1191
1128   1192
1129   1193
1130   1194
1131   1195
1132   1196
1133   1197
1134   1198
1135   1199
1136   1200
1137   1201
1138   1202
1139   1203
1140   1204
1141   1205
1142   1206
1143   1207
1144   1208
1145   1209
1146   1210
1147   1211
1148   1212
1149   1213
1150   1214
1151   1215
1152   1216
1153   1217
1154   1218
1155   1219
1156   1220
1157   1221
1158   1222
1159   1223
1160   1224
1161   1225
1162   1226
1163   1227
1164   1228
1165   1229
1166   1230
1167   1231
1168   1232
1169   1233
1170   1234
1171   1235
1172   1236
1173   1237
1174   1238
1175   1239
1176   1240
1177   1241
1178   1242
1179   1243
1180   1244
1181   1245
1182   1246
1183   1247
1184   1248
1185   1249
1186   1250
1187   1251
1188   1252
1189   1253
1190   1254
1191   1255
1192   1256
1193   1257
1194   1258
1195   1259
1196   1260
1197   1261
1198   1262
1199   1263
1200   1264
1201   1265
1202   1266
1203   1267
1204   1268
1205   1269
1206   1270
1207   1271
1208   1272
1209   1273
1210   1274
1211   1275
1212   1276
1213   1277
1214   1278
1215   1279
1216   1280
1217   1281
1218   1282
1219   1283
1220   1284
1221   1285
1222   1286
1223   1287
1224   1288
1225   1289
1226   1290
1227   1291
1228   1292
1229   1293
1230   1294
1231   1295
1232   1296
1233   1297
1234   1298
1235   1299
1236   1300
1237   1301
1238   1302
1239   1303
1240   1304
1241   1305
1242   1306
1243   1307
1244   1308
1245   1309
1246   1310
1247   1311
1248   1312
1249   1313
1250   1314
1251   1315
1252   1316
1253   1317
1254   1318
1255   1319
1256   1320
1257   1321
1258   1322
1259   1323
1260   1324
1261   1325
```



```

1      SIMMITHINE STACK (DDP)
2      C
3      *** SYSTEM PROGRAM THE VISIBILITY CALCULATIONS BASICALLY USING
4      THE METHOD OF HALOW & ZELIN (JAPCS, VOL. 23, PP. 974-984,
5      AUGUST 1973). HOWEVER, THERE HAVE BEEN A NUMBER OF IMPORTANT
6      MODIFICATIONS AS DESCRIBED BELOW. *****
7
8      COMMON/ACHAR/EL,AM,PATH,REFRAC
9      COMMON/ACOM/71,NMOP
10     COMMON/ANODD/MIND,CLC(10)
11     DIMENSION NO(11),NO1(100),NO2(100),NO3(100),NO4(100)
12     COMPLEX YV,REFRAC,NO1,NO2,NO3,NO4,NO5,NO6,NO7,NO8,NO9,NO10,NO11
13     COMPLEX Q1(100),QMC(100),QMC1(100),QMC2(100)
14     COMPLEX NIMAR
15     NIMAR = (0.0,1.0)
16     RIFPR = 0.0
17     DO 100 K=1,100
18     QP1 = (DP(K),OP(K))/Z*Q
19     IF (INT(PT,50.0) .GT. 24)
20     C
21     *** PARTICLES LARGER THAN 50 MICRONS HAVE A NEGOTIABLE EFFECT
22     ON VISIBILITY, AND THEREFORE, HAVE BEEN EXCLUDED FROM THE
23     CALCULATIONS IN ORDER TO SAVE COMPUTATION TIME. *****
24
25     XV = 3.1415927*NO1/RLAM
26     YV = 0.5*NO2*YV
27     USQ = 0.0
28     NRI2 = SIN(XV)
29     NRI3 = -SIN(XV)
30     NRI4 = COS(XV)
31     Q11 = -NIMAR
32     Q1 = 1.07*Q11
33
34     C
35     *** MAXIMUM VALUES (NMAX) OF THE ORDER (N) OF THE RICCATI-BESSEL
36     FUNCTIONS ARE GENERATED BY THE FOLLOWING THREE EQUATIONS
37     ACCORDING TO W. COMBE, W.J., APPLIED OPTICS, VOL. 19, PP. 1505-
38     1508, MAY 1980. *****
39
40     IF (XV.GE.4200.0) NMAX=XV*4.0*(XV**E3)+2.0
41     IF (XV.LT.4200.0) NMAX=XV*0.00012*(XV**E3)+2.0
42     IF (XV.LE.0.0) NMAX=XV*4.0*(XV**E3)+1.0
43     DO 99 N=1,NMAX
44
45     C
46     *** THE NEXT 14 LINES GENERATE BESSEL FUNCTIONS OF THE ARGUMENT
47     XV ACCORDING TO THE METHOD OF LENTZ, W.J., APPLIED OPTICS,
48     VOL. 19, PP. 648-671 (MARCH 1976). *****
49
50     UP(1) = 2.0*(PLUAT(N)+0.5)*XV
51     UN(1) = UP(1)
52     UP(2) = -2.0*(FLOAT(N-1)+0.5)/XV
53     UN(2) = UP(2)
54     UN(3) = UP(2)*UP(1)
55     UN(4) = UN(3)*UN(2)*UP(2)
56
57     10  UP(5) = 2.0*(FLOAT(N+1)+0.5)/XV
58     UN(5) = UN(4)*UN(3)*UN(2)*UN(1)
59     UN(6) = UN(5)*UN(4)*UN(3)*UN(2)*UN(1)
60     IF (ABS(UN(5)-UN(4)) .LE. 1.0E-10) GO TO 13
61     N = N + 1
62     GO TO 10
63
64     C
65     *** PNR AND Q ARE LOGARITHMIC DERIVATIVE FUNCTIONS USED
66     TO CALCULATE AN & NM.

```

Figure C-2. Continued. Computer Source Code Listing

APPENDIX D
COMPUTER PROGRAM NOMENCLATURE

A1	=	real part of a_n denominator
A2	=	a_n numerator/A1
A3	=	a_n denominator/A1
AN	=	a_n
AK	=	Stokes' parameter (k) in Calvert's scrubber equations (dimensionless)
AKF	=	AK(f) + 0.7, where f = 0.5
ALG	=	liquid/gas ratio
B	=	intercept of straight-line log-normal equation (in GRAF)
BIMAG	=	i
B1	=	real part of b_n denominator
B2	=	b_n numerator/B1
B3	=	b_n denominator/B1
BJ	=	$J_{\nu-1}/J_{\nu}$ (with real argument)
BJC	=	$J_{\nu-1}/J_{\nu}$ (with complex argument)
BN	=	b_n
CC	=	Cunningham correction factor in scrubber equations (dimensionless)
CLCW	=	cum. wt. % in 1% increments
CN	=	n/x
CWDE	=	outlet cumulative wt. % from ESP
CWDS	=	outlet cumulative wt. % from scrubber
CWRD	=	raw cum. wt. % input data

DD = scrubber drop size (by Nukiyama-Tanasawa equation), in microns

DHUN is defined by line 7 of function EINV

DP = normalized particle size from subroutine GRAF (in microns)

DPE = outlet particle size from ESP (in microns)

DPRD = raw particle size data (in microns)

DPS = outlet particle size from scrubber (in microns)

DPI = average value of DP in interval (microns)

EFF = calculated ESP efficiency (%)

EFFS = calculated scrubber efficiency (%)

EINV(RCW) = inverse normal distribution function (RCW = variable)

ETA is defined by line 17 of function EINV

ETASQ is defined by line 16 of EINV

E3 = 1/3

EF2 is defined by line 15 of function EINV

FND = FLOAT(NVRD)

G is defined by line 37 of subroutine ESP

GRAF is a subroutine that uses the method of least squares to fit the raw particle size distribution data to a straight-line equation

I, IJ, IK, IL are all counters

ICASE = 0, when there are no further cases, and = 1 when another set of data cards (i.e. another case) follows

INOGO = 1 if the electrostatic precipitator, or scrubber, has such a high efficiency that virtually all of the original particle size distribution is collected. Otherwise, it is 0.

IP = p (in generating values of w)
 IQUIP = 0, if there is no particle collection device; = 1, if there is an electrostatic precipitator; = 2, if there is a venturi scrubber
 ITTLE = alphanumeric variable (maximum of 10 letters) corresponding to the case title.
 J,JI are counters
 K = counter
 L = $(-1)^{p+1}$ (in generating values of w)
 N = n (order of Riccati-Bessel functions)
 NE = number of data pairs for outlet distribution from ESP
 NMAX = max. value of n
 NS = number of data pairs for outlet distribution from scrubber
 NVRD = number of raw data pairs for inlet size distribution
 PATH = plume width = effective stack diameter (in meters)
 PD = pressure drop across scrubber (in mm. mercury)
 PNx = $P_n(x) = \psi_n'(x)/\psi_n(x)$
 PNY = $P_n(y) = \psi_n'(y)/\psi_n(y)$
 PTLN = natural log of scrubber penetration
 Q = $Q_n(x) = \xi_n'(x)/\xi_n(x)$
 QEXT = extinction coefficient = $\frac{2}{x^2} \zeta(2n+1) \text{real}(a_n+b_n)$
 QSUM = $\zeta(2n+1) \text{real}(a_n+b_n)$
 RB1x = $\psi_n(x)$
 RB2 = $\beta_n(x)$
 RB3 = $\xi_n(x)$

REFRAC = complex refractive index of particle (dimensionless)
 RHOP = particle density (in g/cc)
 RIIM = imaginary part of refractive index
 RIRL = real part of refractive index
 RNM = calculated Ringelmann number
 RNMAX = upper estimate of Ringelmann number (=RNM+0.5)
 RNMIN = lower estimate of RNM
 S = fractional penetration
 SCA = specific collection area of ESP (in m² per 1000 m³/min of gas)
 SGN is defined by lines 11 and 14 of EINV
 SLOPE = slope of straight-line log-normal equation (in GRAF)
 STACK is the subroutine that calculates plume visibility
 STEPB = $\sum_i \left(\frac{Q_{ext}}{d_p}\right)_i$
 SUM = % penetration
 SUME = cumulative wt % penetration for particles of size DPE
 SUMS = cumulative wt % penetration for particles of size DPS
 SUMX = EX
 SUMXY = EXY
 SUMX2 = $\sum(X^2)$
 SUMY = EY
 TRANS = calculated fractional transmittance =

$$\exp\left(\frac{-3WD}{2\rho p}\right) \frac{1}{100} \sum_{i=1}^{100} \left(\frac{Q_{ext}}{d_p}\right)_i$$

TRANSW = % transmittance
 VGT = gas velocity in venturi scrubber throat (in meters/sec)
 WD = $|w_p, \dots, w_2|$, with real arguments
 WDC = $|w_p, \dots, w_2|$, with complex arguments
 WI = IMAG(WNC-WDC)
 WN = $|w_p, \dots, w_1|$, with real arguments
 WNC = $|w_p, \dots, w_1|$, with complex arguments
 WP = $w_p = (-1)^{p+1} \frac{2(v+p-1)}{x}$
 WPC = $(-1)^{p+1} \frac{2(v+p-1)}{y}$
 WR = real (WNC-WDC)
 X = EINV(CWRD)
 X1 = EINV(CLCW)
 X2 = x^2
 XLAM = wavelength of light used to view the plume (λ) (in microns)
 XV = $\pi d_p / \lambda$ (dimensionless)
 Y = $\log_{10} \text{DPRD}$
 YV = $m^2 d_p / \lambda$ (dimensionless)
 ZI = particle loading (in mg/m^3)