COMPUTATION OF ACOUSTIC POWER, VIBRATION RESPONSE
AND ACOUSTIC PRESSURES OF FLUID-FILLED PIPES

BY

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Summary

The time-harmonic excitation is either an interior point source of sound or a mechanical point force located on the pipe's surface. The acoustic power radiated into the exterior fluid, the vibration response of the pipe's wall, and the pressures in the exterior and interior fluids are computed by a simple integration scheme. Numerical results are presented for the case of a water-filled steel pipe that is surrounded by air.
LIST OF SYMBOLS

(x,y,z) = cartesian coordinates
(r,θ,z) = cylindrical coordinates
(R,θ,Φ) = spherical coordinates
U(θ,z),V(θ,z),W(θ,z) = axial, tangential and radial displacements at shell mid-surface
a,n = axial and circumferential harmonic wavenumbers
U(ν,ν),V(ν,ν),W(ν,ν) = transforms of shell displacements
p_1(r,θ,z),p_2(r,θ,z) = interior and exterior pressures
ω = radian frequency of vibration
ρ_1, c_1 = density and sound velocity of interior fluid
ρ_2, c_2 = density and sound velocity of exterior fluid
k_i, i=1,2 = wavenumber in fluid, ω/c_i
a, h, ρ_s = radius, thickness and density of pipe
E_1 = Eh/(1-ν^2), E is Young's modulus and ν is Poisson's ratio
β^2 = h^2/12a^2
δ = Dirac delta function
ε_n = 1 when n=0, and 2 otherwise
F_0 = amplitude of point force located at cylindrical coordinates (a,0,0)
P_0 = amplitude of point source located at cylindrical coordinates (X_0,0,0)
R_0^2 = (X-X_0)^2+Y^2+Z^2 = r^2 + X_0^2 - 2rX_0 cos(θ) + Z^2
J_n, J'_n, Y_n, Y'_n, H_n, H'_n = Bessel functions and their derivatives. H_n = J_n + iY_n
ν_i, i=1,2 = (k_i^2 - a^2)^i with Im(ν_i) > 0 to satisfy the radiation condition
n = hysteretic loss factor of shell
INTRODUCTION

A previous report [1] has analysed the theoretical problem of far-field sound radiation from fluid-filled pipes that are excited by point forces and point acoustic sources. The salient points that emerge from the work are that the frequencies at which 'peaks' appear in the sound spectra depend on the location of the observation point, and that the far-field radiation characteristics are understood better when wavenumber versus frequency dispersion plots are available. In practice, however, measurements in the far-field are not usually possible because the environment is inevitably reverberant. It is thus necessary to average spatially the pressure measurements, in a suitable way, to give an estimate of the radiated acoustic power. Other responses of central interest to an experimentally based programme are the pressure fields near to the pipe's surface and the acceleration response of the pipe's wall.

In Section 2, the pressure fields and the pipe's displacements are represented by Fourier transforms, and the acoustic power in the surrounding fluid is obtained from the stationary phase approximation to the far-field pressure. The integral representations of the fluid pressures, pipe's displacements and radiated power are evaluated by a simple numerical quadrature scheme. In Section 3, numerical results are discussed for the case of a water-filled steel pipe that is radiating into the surrounding air. The numerical results are in the form of decibel level versus frequency plots for the responses in the individual circumferential harmonics, n=0, 1 and 2.

The numerical results are better understood by inspection of wavenumber versus frequency plots, but further numerical work is necessary. First, it is necessary to search the literature for measurements that have been obtained under controlled conditions, because a comparison with theoretical estimates would be particularly valuable. Secondly, plots of acoustic intensity vectors [6] would help to illustrate the essential physics of the fluid-pipe system. Thirdly, the eigenvectors should help to explain the curious maxima and minima that appear in the spectra of the responses at a distance from the source. Forthly, the range of applicability of the concept of matching wavenumbers of in-vacuo pipe and rigid-walled duct needs to be investigated in some detail because it is being used increasingly to explain certain aspects of measured data. Finally, the effect of fluid flow on the stability, vibration and sound radiation needs to be investigated.

2. PROBLEM FORMULATION

General

An infinite circular thin-walled pipe contains and is also surrounded by fluids of possibly differing densities and sound velocities. The displacements of the pipe's wall satisfy an eighth-order shell theory [2], and the interior and exterior sound pressures satisfy the Helmholtz wave-equation. The pipe is excited by time-harmonic surface stresses caused by mechanical point forces or interior acoustic sources. The time factor exp(-iωt) is omitted from all of the equations. The geometry is shown in Figure 1.
Pipe Displacements and Fluid Pressures

The axial, tangential and radial displacements of the pipe's wall and the interior and exterior acoustic pressures have the following Fourier transform representations:

\[
\begin{align*}
U(\theta,z) &= (1/2\pi) \sum_{n=0}^{\infty} \begin{bmatrix} \cos(n\theta) \\ \sin(n\theta) \\ \cos(n\theta) \end{bmatrix} \int_{-\infty}^{\infty} \exp(i\alpha z) \begin{bmatrix} \overline{U}(n,\alpha) \\ \overline{V}(n,\alpha) \\ \overline{W}(n,\alpha) \end{bmatrix} d\alpha \\
V(\theta,z) &= \left(1/2\pi\right) \sum_{n=0}^{\infty} \begin{bmatrix} \cos(n\theta) \\ \sin(n\theta) \\ \cos(n\theta) \end{bmatrix} \int_{-\infty}^{\infty} \exp(i\alpha z) \begin{bmatrix} \overline{V}(n,\alpha) \\ \overline{W}(n,\alpha) \end{bmatrix} d\alpha \\
W(\theta,z) &= \left(1/2\pi\right) \sum_{n=0}^{\infty} \begin{bmatrix} \cos(n\theta) \\ \sin(n\theta) \\ \cos(n\theta) \end{bmatrix} \int_{-\infty}^{\infty} \exp(i\alpha z) \begin{bmatrix} \overline{W}(n,\alpha) \end{bmatrix} d\alpha \\
p_1(r,\theta,z) &= (1/2\pi) \sum_{n=0}^{\infty} \cos(n\theta) \int_{-\infty}^{\infty} \exp(i\alpha z) \overline{p}_1(r,n,\alpha) d\alpha \\
p_2(r,\theta,z) &= (\omega^2 p_2/2\pi) \sum_{n=0}^{\infty} \cos(n\theta) \int_{-\infty}^{\infty} \exp(i\alpha z) \left[ H_n(u_2r)/u_2 H'_n(u_2) \right] \overline{W}(n,\alpha) d\alpha
\end{align*}
\]

Formulas for the transforms \(\overline{U}(n,\alpha), \overline{V}(n,\alpha), \overline{W}(n,\alpha)\) and \(\overline{p}_1(r,n,\alpha)\) are to be found in the Appendix.

Far-Field Acoustic Pressure

A stationary phase evaluation of the integral representation of the exterior pressure yields the far-field pressure as

\[
p_f(R,\theta,\phi) = -i\omega p_2 c_2 \exp(ik_2R/R) \sum_{n=0}^{\infty} \frac{\overline{W}(n,k_2 \cos \phi, \exp(-i\pi/2), \cos(n\theta))}{\sin(\theta) H'_n(\omega k_2 \sin \theta)}
\]

Radiated Power

The acoustic power radiated into the exterior fluid is defined as

\[
P = \int_0^\pi \int_0^{2\pi} |p_f(R,\theta,\phi)|^2 R^2 \sin(\theta) d\phi d\theta
\]

where the customary multiplication factor of one-half has been omitted because the excitation is to be specified in its rms form. Substituting equation (2.4) into equation (2.5) gives, after carrying out the integration in the \(\phi\)-coordinate,

\[
P = (2\omega^2 p_2 c_2/\pi) \sum_{n=0}^{\infty} e_n^{-1} \int_0^\pi \left| \overline{W}(n,k_2 \cos \phi) \right|^2 \sin(\theta) H'_n(\omega k_2 \sin \theta)^2 \right| d\theta
\]

When the excitation is located at \(z=0\), the symmetry of the sound field about \(\theta=\pi/2\) may be used to halve the computational time.
Numerical Evaluation of Integrals

The integrals in equations (2.1-2.3) and (2.6) must be evaluated numerically because closed-form expressions are not available. Their numerical approximation, on truncating the infinite limits in equations (2.1-2.3) to finite values, is based upon a simple adaptive quadrature scheme that has been used successfully to evaluate the response and acoustic field of a point-excited plate [3].

The scheme splits the range of integration into a selected number of equal intervals each of which being repeatedly halved until an absolute or a relative convergence test is met by successive approximations to the integral in the interval - Gaussian quadrature of order two is suitable for those cases where low accuracy is sufficient. Vogel & Feit [4] have also used an adaptive scheme to evaluate the vibration response of a pipe with exterior fluid loading and point force excitation - their mesh size being controlled by the second derivative of the integrand.

3. NUMERICAL EXAMPLES

3.1 General

Plots of wavenumber, acoustic power, acceleration and acoustic pressure versus frequency are shown in Figures 2-9 for the particular case of a water-filled steel pipe that is surrounded by air. The material and geometric constants, in SI units, used in the computations are as follows:

- Pipe: \( E = 19.5 \times 10^7 \), \( \sigma = 0.29 \), \( \rho = 7700.0 \), \( h = 2.54 \times 10^{-2} \), \( a = 22.23 \times 10^{-2} \), \( n = 0.02 \)
- Water: \( \rho = 1000.0 \), \( c = 1500.0 \)
- Air: \( \rho = 1.21 \), \( c = 343.0 \)

Previous work [1] has presented plots of far-field acoustic radiation versus frequency, for the various combinations of the fluids, air and water, and the separate cases of point force and source excitation.

Wavenumber Plots

Figures 2 and 3 show the real branches of the wavenumber versus frequency plots for the circumferential harmonics \( n = 0-2 \). The loading caused by the surrounding air was neglected. The frequency range extends to 1.4 times the 'ring' frequency of the pipe. The physical interpretation of wavenumber-frequency plots is discussed in some detail elsewhere [5]; here, only a brief explanation of the 'cut-on' nature of the waves is necessary.

Figure 2 contains the plots for a water-filled pipe. First, for \( n = 0 \), the branches labelled 1-4 are close to a plane fluid wave, an axial wave in the pipe, a radial flexural wave in the pipe and a fluid wave, respectively: the branch '0' is a wave of pure torsion. Secondly, for \( n = 1 \), the branches are the familiar beam flexural wave, a fluid wave and a pipe torsional wave, respectively. Finally, for \( n = 2 \), the branches are the radial shell flexural wave, a fluid wave and a torsional wave, respectively.
Figure 3 contains the wavenumber versus frequency plots for the separate cases of the pipe in-vacuo, a rigid-walled waveguide of water and a pressure release waveguide of water. These plots, when compared with Figure 2, show the effect of the interior water on the real branches of the dispersion relation. They are also useful if the concept of equal wavenumber in the fluid and pipe is of value to the interpretation of the responses of the coupled fluid-pipe system.

**Airborne Acoustic Power**

Figures 4-5 show the airborne acoustic power radiated from the pipe's wall in the n=0-2 circumferential harmonics, for the separate cases of point-source and point-force excitation. The sound power decibel level is defined as $10\log(P) + 120$.

In Figure 4, the excitation is a point source, located at the cylindrical coordinates $(2a/3,0,0)$, whose free-field level in the interior fluid is 120 dB re 1 micropascal at 1m. The 'peaks' in the spectra occur at the frequencies at which wave-branches 'cut-on'. In particular, a large jump in the acoustic power occurs at the frequency of the n=2 harmonic where an essentially fluid-type wave 'cuts-on'. At frequencies below the 'cut-on' frequency of the n=2 flexural mode, the total power is dominated by the contribution of the n=0 harmonic.

In Figure 5, the excitation is a radial 1 N point force that is located at the cylindrical coordinates $(a,0,0)$. Again, the peaks in the spectra occur at frequencies at which the wave-branches 'cut-on'. However, the jump in level at the 'cut-on' frequency of the n=2 fluid-type wave is less severe, and the radiated power of the n=0 harmonic goes to zero at low frequencies where the n=1 harmonic is dominant.

The effect, on the power radiation, of additional damping in the pipe's wall was investigated by increasing the hysteretic loss factor from 0.02 to 0.10. The power levels dropped by 7 dB which indicates a power reduction of $10\log(\eta_2/\eta_1)$, where $\eta_1$ is the power reduction would not necessarily have been obtained for the case of a water-filled pipe radiating into water, because the radiation loss factor could be of the same order of magnitude as the structural loss factor.

**Radial Acceleration of Pipe's Wall**

The radial acceleration of the pipe's wall in the n=0-2 harmonics is plotted in Figures 6-7 for the case of point force excitation located at $(a,0,0)$. The acceleration decibel levels are defined as $20\log(|\ddot{W}|) + 100$.

Figure 6 gives the drive point acceleration. It is evident that the maximum responses occur at the 'cut-on' frequencies of the modes. The plots, on adjustment to velocity response, are of similar appearance to the power radiation plots of Figure 5.

Figure 7 gives the acceleration response at an axial distance of 4m from the radial drive force. The responses are now quite complex at the higher frequencies, presumably due to modal interference effects and the changing nature of the wave-branches with frequency. It
is difficult to interpret these plots without the availability of the eigenvectors.

*Acoustic Pressures on Pipe's Surfaces*

Figures 8-9 show the acoustic pressures on the pipe's outer and inner surfaces, respectively. The excitation is a radial point force and the pressures are computed at an axial distance of 4m from the force, the decibel level being $20 \log(|p|) + 120$.

The plots of the pressure on the outside of the pipe, Figure 8, are of similar appearance to the transfer acceleration, Figure 7. At the higher frequencies, the maxima and minima in the plots of the inside pressure, Figure 9, do not coincide in frequency with those of the outer pressure plots. Indeed, it is curious that many maxima and minima are reversed in the two figures.

4. **ACKNOWLEDGEMENT**

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*J.H. James (PSO)*

*JHJ/jms*


3. MELOTTE, J.A., SPICER, W.J., unpublished work at AMTE (Teddington) on the vibration and sound radiation of point excited plates.

4. VOGEL, W., FEIT, D., Response of a Point Excited Infinitely-long Shell Immersed in an Acoustic Fluid, DTNSRDC(Bethesda), Report 80/061, November 1980.


The Spectral Displacements [1]

The excitation is either a radial point force, of magnitude $F_0$, that is located at the cylindrical coordinate $(a,0,0)$, or it is an interior point source of sound, of free-field pressure $p_0 \exp(ikR_0)/R_0$, that is located at $(x_0,0,0)$. The 'spectral' displacements are obtained, via Novozhilov shell theory [2], as the solution of the matrix equation

$$
\begin{bmatrix}
S_{11} & S_{12} & S_{13} \\
S_{21} & S_{22} & S_{23} \\
S_{31} & S_{32} & S_{33}
\end{bmatrix}
\begin{bmatrix}
U(n,a) \\
V(n,a) \\
W(n,a)
\end{bmatrix}
= 
\begin{bmatrix}
0 \\
0 \\
F_0 e_n/2\pi a \text{ or }
2p_0 e_n J_n(\mu_2 a)/\mu_1 a P_n(\mu_2 a)
\end{bmatrix}
$$

(A1)

where

$$
S_{11} = E_1(\alpha^2 + n^2(1-\sigma)/2a^2) - \rho_s \omega^2
$$

$$
S_{12} = -E_1 \alpha n(1+\sigma)/2a
$$

$$
S_{13} = -E_1 \alpha a/\alpha
$$

$$
S_{21} = -S_{12}
$$

$$
S_{22} = E_1(\alpha^2(1-\sigma)/2 + n^2/\alpha^2 + 2\alpha^2 \beta^2(1-\sigma) + \beta^2 n^2/\alpha^2) - \rho_s \omega^2
$$

(A2)

$$
S_{23} = E_1(n/\alpha^2 + \alpha^2 \beta^2(2-\sigma) + n^3 \beta^2/\alpha^2)
$$

$$
S_{31} = -S_{13}
$$

$$
S_{32} = S_{23}
$$

$$
S_{33} = E_1[1/\alpha^2 + \alpha^4 \beta^2 a^2 + 2n^2 \alpha^2 \beta^2 + n^4 \beta^2/\alpha^2] - \rho_s \omega^2 h
$$

An axial point force excitation is obtained by setting the right-hand side of the matrix equation to $[F_0 e_n/2\pi a,0,0]^T$. 

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The Interior Spectral Pressure [1]

The interior 'spectral' pressure is given by the formula

\[ p_1(r,n,a) = \rho_1 \omega^2 \left[ J_n(\mu_1 r)/\mu_1 J'_n(\mu_1 a) \right] \bar{W}(n,a) + A(n,a) \]  

(A3)

where \( A(n,a) \) is identically zero for point force excitation, and

\[ A(n,a) = \pi \rho_0 d_n \left\{ J_n(\mu_1 x_0)/J_n(\mu_1 a) \right\} \left\{ [J_n(\mu_1 r)Y_n'(\mu_1 a) - Y_n(\mu_1 r)J'_n(\mu_1 a)] \right\}, \]  

\[ r > x_0 \]  

(A4)

\[ A(n,a) = \pi \rho_0 d_n \left\{ J_n(\mu_1 r)/J_n(\mu_1 a) \right\} \left\{ [J_n(\mu_1 x_0)Y_n'(\mu_1 a) - Y_n(\mu_1 x_0)J'_n(\mu_1 a)] \right\}, \]  

\[ r < x_0 \]  

(A5)

for a point source excitation.

An alternative to equation (2.2) for the particular case of point source excitation is

\[ p_1(r,0,z) = \rho_0 \exp(\frac{ikx_0}{x_0}) + \left( \frac{1}{2\pi} \right) \sum_{n=0}^{\infty} \cos(n\theta) \int_{-\infty}^{\infty} \left\{ b(r,n,a) \exp(iaz) \right\} \right] \]  

(A6)

where

\[ b(r,n,a) = \left\{ J_n(\mu_1 r)/J_n(\mu_1 a) \right\} \left( \rho_1 \omega^2 \bar{W}(n,a)/\mu_1 - \pi \rho_0 d_n \left\{ J_n(\mu_1 x_0)H'_n(\mu_1 a) \right\} \right] \]  

(A7)
FIG. 1 GEOMETRY AND COORDINATE SYSTEMS

\[ x = r \cdot \cos(\phi) \quad x = R \cdot \sin(\theta) \cdot \cos(\phi) \]
\[ y = r \cdot \sin(\phi) \quad y = R \cdot \sin(\theta) \cdot \sin(\phi) \]
\[ z = z \quad z = R \cdot \cos(\theta) \]
FIG. 2 AXIAL WAVENUMBER VERSUS FREQUENCY.
STEEL PIPE. INTERIOR WATER. EXTERIOR VACUUM.
FIG. 3 AXIAL WAVENUMBER VERSUS FREQUENCY.
PIPE IN VACUO ★★★★
RIGID-WALLED DUCT OF WATER ★★★★
PRESSURE-RELEASE DUCT OF WATER ————
FIG. 4 AIRBORNE ACOUSTIC POWER. SOURCE EXCITATION.
FIG. 5 AIRBORNE ACOUSTIC POWER. FORCE EXCITATION.
FIG. 6 DRIVE POINT RADIAL ACCELERATION, FORCE EXCITATION.
FIG. 7 TRANSFER RADIAL ACCELERATION. FORCE EXCITATION.
FIG. 8 EXTERIOR PRESSURE ON PIPE SURFACE, FORCE EXCITATION.
FIG. 9 INTERIOR PRESSURE ON PIPE SURFACE. FORCE EXCITATION.