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Technical Report No. 2

NONSPECULAR ULTRASONIC EFFECTS FOR LAYERED MEDIA

AND SECOND HARMONICS IN LAMB WAVES

by

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Key Words: Ultrasonics, solid plates, half-space, specular and nonspecular reflectivity and transmissivity, Lamb waves, harmonic generation, Rayleigh waves, acousto-optic interaction.

Abstract: Theoretical and experimental evidence is given to explain reflectivity and transmissivity of bounded ultrasonic beams impinging on flat solid thin and thick plates immersed in water. Effects are described which have to be considered when high amplitude ultrasonics is used or when the liquid-solid combination differs significantly from the usual metal-water combination of reflecting media.
This Technical Report describes work performed under Contract N00014-78-C-0584 "Linear and Nonlinear Ultrasonic Interactions on Liquid-Solid Boundaries." The Report consists of the following articles:


The first three of above listed papers describe both theoretical and experimental results which are used to evaluate and explain the reflectivity and transmissivity of bounded ultrasonic beams impinging on flat solid thin or thick plates which are surrounded by water. The angle of incidence of the incident beam determines what type of reflectivity or transmissivity may be expected. The last two papers describe effects which have to be taken into consideration when either high amplitude ultrasonics is used or the liquid-solid combination is changed significantly from the usual parameters describing a normal metal-water boundary.

W. G. Mayer
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Washington, DC, April 1980
Ultrasonic nonspecular reflectivity near longitudinal critical angle

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A numerical integration method is developed to determine the intensity profile of an ultrasonic beam reflected from a liquid-solid interface near the longitudinal critical angle. The profiles are calculated for different combinations of frequencies and beam widths with the angle of incidence being varied about the longitudinal critical angle for a water-Plexiglas interface. These calculations demonstrate the existence of nonspecular reflectivity near this particular critical angle and provide a quantitative description of its basic features. Theoretical results and experimental measurements are compared.

PACS numbers: 43.20.Fn, 43.35.Pt

I. INTRODUCTION

An analytical formulation to treat the reflection of a bounded ultrasonic beam at a liquid-solid interface was first established by Schoch.¹ That theory predicts a lateral displacement of the reflected beam for all total reflection angles. This simplified model fails to account for various experimentally observed phenomena at the Rayleigh critical angle. Making use of the existence of a pole-zero pair in the reflection coefficient at the Rayleigh critical angle, Bertoni and Tamir² describe an analytical approximation model that is able to explain all important features of the nonspecular reflectivity at the Rayleigh angle. Their approximation method for a liquid-solid interface has been subsequently extended to a study of reflectivity from a solid plate immersed in a liquid,³ the diffraction effect observed away from the interface,⁴ and the influence of attenuation¹ in the solid medium. Recently, the same Rayleigh-angle nonspecular reflectivity problem was investigated by means of a numerical integration method which produces results consistent with those of Bertoni and Tamir,¹ but also allows one to calculate changes in the profiles of the reflected beam for angles of incidence near the Rayleigh critical angle.⁴

While nonspecular reflectivity near the Rayleigh critical angle has been studied extensively, little is known about the possible nonspecularly reflected phenomena for bounded beams incident at or near the longitudinal and shear critical angles. Experimental evidence indicating the existence of nonspecular reflectivity similar to that at the Rayleigh angle was reported by Neubauer⁷ for Mallory 1000 at the shear critical angle and for Pyrex glass at the longitudinal critical angle.

This paper is intended to determine the reflected beam profile of a bounded sound beam incident at and near the longitudinal critical angle of a liquid-solid interface.

II. EFFECT OF ABSORPTION ON PLANE WAVE REFLECTION COEFFICIENT

A convenient form of the infinite plane wave amplitude reflection coefficient $R$ for a liquid-solid interface is given⁵ by

$$R(k_z) = \frac{(k_z^2 - 2k_z^2) + 4k_z^2\kappa_d\kappa - p k\kappa_d / \kappa}{(k_z^2 - 2k_z^2) + 4k_z^2\kappa_d\kappa + p k\kappa_d / \kappa},$$

where

$\rho$ = liquid density/solid density,
$k = \omega / v$,
$k_d = \omega / v_d$,
$k_0 = \omega / v_0$,
$k_z = k \sin \theta$,
$\kappa = (k_0^2 - k_z^2)^{1/2}$,
$\kappa_d = (k_d^2 - k_z^2)^{1/2}$,
$\kappa_0 = (k_0^2 - k_z^2)^{1/2}$.

FIG. 1. Magnitude and phase of the reflection coefficient. The absorptive cases are denoted by dashed curves and the lossless cases by solid curves. P denotes a water-Plexiglas interface for a frequency of 2 MHz and S denotes a water-steel interface for 15 MHz.

where $\omega$ is the angular frequency, $\theta$ is the incident angle, $v$ is the sound velocity in the liquid, and $v_s$ and $v_p$ are the velocities of the longitudinal and shear waves in the solid, respectively. The $x$ direction is along the interface, and the incidence is from the liquid half-space.

It is known that the behavior of $R(k_x)$ is critically dependent on the attenuation constants of the media, especially at the Rayleigh critical angle, where the so-called "least-reflection frequency" is normally observed for many solids. Near the longitudinal and shear critical angles, attenuation only slightly affects the magnitude and the phase of $R(k_x)$. However, it should be noted that inclusion of attenuation into the expression for $R(k_x)$ does smooth out the discontinuous behavior of the first derivatives of the magnitude and the phase of $R(k_x)$. In the complex plane of $k_x$, inclusion of attenuation results in the property that the branch points $\pm k_x, \pm k_d$, and $\pm k_s$ of $R(k_x)$ and their corresponding branch cuts are then located off the real axis. This shift of position of the branch points makes a numerical integration method possible.

The effects due to attenuation in materials can be incorporated into $R(k_x)$ by introducing the complex wave numbers $k, k_x, k_d,$ and $k_s$ as defined below. Denoting $\alpha$ as attenuation per wavelength for the sound wave in the liquid, and $\alpha_d$ and $\alpha_s$ for the longitudinal and shear waves in the solid, the complex wave numbers can be written as

$$k = \frac{\omega}{v} (1 + i\alpha/2\pi),$$

$$k_x = \frac{\omega}{v_s} (1 + i\alpha_s/2\pi),$$

$$k_d = \frac{\omega}{v_d} (1 + i\alpha_d/2\pi),$$

$$k_s = \frac{\omega}{v_p} (1 + i\alpha_s/2\pi).$$

To illustrate the influence of attenuation, the magnitude and the phase of $R(k_x)$ are plotted as functions of the angle of incidence for water-stainless steel and water-Plexiglas interfaces with and without attenuation (Fig. 1). The parameter values used to produce these curves and all subsequent figures are given in Tables I and II.

Figure 1 shows that the curves for stainless steel are typical of many other solid/liquid combinations as previous-
$E(A, X) = \exp(iCA, X)$, where

\[
A = 1 + ia/2\pi , \\
A_d = (v/v_o)(1 + ia_d/2\pi) , \\
A_r = (v/v_o)(1 + ia_r/2\pi) , \\
B = (A^2 - A_d^2)^{1/2} , \\
B_d = (A^2 - A_r^2)^{1/2} , \\
B_r = (A^2 - A_1^2)^{1/2} , \\
A_1 = \sin \theta_i , \\
X = x/w_o , \\
C = \omega w_p / v , \\
V_A = \pi / C ,
\]

with $x$ being the coordinate along the interface, $\theta_i$ the incident angle, and $2w_o$ the beam width projected onto the interface.

As expected from observing the behavior of $R(k_x)$ for incidence near $\theta_i$, the calculated reflected beam profiles for the case of water-stainless steel only show a small beam displacement of about 2% of $2w_o$ which is normally not observable. In contrast, for the case of water-Plexiglas, the displacement of the intensity peak in the reflected beam becomes much greater and can be observed experimentally. This characteristic of non specular reflection near $\theta_i$ is not only a strong function of the material properties, but also of the sound frequency, hence of $a_p$ and $a_s$, the beam width, and the incident angle.

In Fig. 2 a set of beam profiles are plotted for different frequencies while keeping the same incident beam width. The beam displacement is found to be larger for lower frequencies. Figure 3 shows another set of reflected beam profiles for a sound frequency of 2 MHz with the beam width 0.5, 0.75, and 1.0 in. The calculations point to the fact that as the beam width is decreased, the beam displacement becomes larger but the peak intensity is reduced.

FIG. 3. Reflected beam profiles for a bounded beam of frequency 2 MHz incident at $\theta_i$ onto a water-Plexiglas boundary. The beam width is (A) 0.5 in., (B) 0.75 in., (C) 1.0 in. The dashed line represents the incident beam.

IV. VARIATION OF BEAM PROFILE AS A FUNCTION OF ANGLE OF INCIDENCE NEAR $\theta_i$

Reflected beam profiles change as a function of the angle of incidence. Their variation can be characterized by changes in the beam displacement and the intensity peak, which are only substantial near the longitudinal critical angle for a water-Plexiglas interface. The numerical integration method is now used to determine the reflection profiles of a beam of 2-MHz frequency and 0.75-in. width when the angle of incidence is varied by steps of 0.5°. The phenomenon is also monitored by using the Schlieren visualization technique.

Figure 4 consists of a series of Schlieren photographs showing the gradual change in profiles of the beam reflected at angles first smaller than $\theta_i$, and becoming larger until exceeding $\theta_i$. A sharp increase in the beam displacement and the reflected intensity can be detected for an incident angle $\theta_i$.

FIG. 4. Schlieren photographs showing beam displacement and changes in sound intensity near $\theta_i$. The incident angle is (A) $\theta_i - 2^\circ$, (B) $\theta_i - 1^\circ$, (C) $\theta_i$, (D) $\theta_i + 1^\circ$.
Tables I and II and those which apply to a particular Plexiglas sample will cause the calculated displacement to differ from the observed one. The greatest uncertainty for this calculation lies in the values of the shear-wave velocity and absorption for Plexiglas. These values are not known to a high degree of accuracy.

V. CONCLUSION

The above calculated results obtained by a numerical integration method and experimental measurements establish the fact that there exists nonspecular reflectivity near the longitudinal critical angle for a liquid-solid interface. This phenomenon is found to be significant only for certain liquid-solid combinations, one of which is water-Plexiglas. For incidence near the longitudinal critical angle, nonspecular reflectivity is most prominently characterized by a substantial beam displacement which varies with the incident angle, the beam width, and the frequency-dependent attenuation in materials. In general, the numerical integration method is shown to be applicable at a range of incident angles, where no successful analytical formulation had been available.

ACKNOWLEDGMENT

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Numerical integration method for reflected beam profiles near Rayleigh angle

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A numerical integration method is developed to calculate the intensity profile of an ultrasonic beam reflected from a liquid—solid interface. This numerical treatment is used to calculate nonspecular reflectivity at a range of angles of incidence near and at the Rayleigh angle. Calculations for a water—stainless steel interface are compared to a known approximate analysis for various beamwidths and frequencies. The theoretical predictions of the reflected beam profile near the Rayleigh angle of incidence are compared to experimental results.

PACS numbers: 43.20.Fn, 43.20.Rz

INTRODUCTION

Nonspecular reflection effects for ultrasonic waves were originally predicted and experimentally verified by Schoch1 and later by Neubauer2 and Breazeale et al.3 for bounded beams incident on a liquid—solid interface. The simple model presented by Schoch does not account completely for different experimentally observed phenomena at Rayleigh critical angle, such as the point of intensity minimum and the trailing field. These discrepancies have been eliminated by Bertoni and Tamir4 who use a complex Laurent series expansion to simplify the mathematically complicated expression of reflection coefficient in terms of its complex pole and zero. Bertoni and Tamir's approximation gives excellent results near the Rayleigh critical angle which is associated with the complex pole and zero.

Attenuation in the media involved, when included in these analyses, further complicates the physical problem. Becker and Richardson5 pointed out the different behavior of the reflection coefficient with absorption being taken into account. In such cases, total reflection is not expected at any angle of incidence and the reflection amplitude and phase are shown to deviate significantly from the lossless case.

Recently, Bertoni and Hou6 have modified their formulation to incorporate attenuation into their theoretical investigation of nonspecular reflectivity. Pitts et al.7 have extended Bertoni and Tamir's results to a more complicated case of the liquid—solid—liquid plate structure. Breazeale et al.3 used Bertoni and Tamir's analysis to investigate the reflected beam profiles observed at large distances from the liquid—solid interface.

Bertoni and Tamir's approximation, although giving excellent results, has some limitations. It is not readily applicable to those critical angles other than the Rayleigh critical angle and presents considerable analytical problems when extended to non-Gaussian beam profiles.

In the following, the reflected profile is determined by numerically integrating the Fourier integral describing the reflected beam amplitude, taking into account the attenuation in the materials involved. The reflected profiles are calculated and compared to known results when one varies one of the physical parameters, namely, the beamwidth, the sound frequency, and the incident angle.

I. MATHEMATICAL REPRESENTATION OF BOUNDED BEAMS

In Fig. 1, a Gaussian sound beam with a beamwidth 2w is shown to be incident at an angle 01 from a liquid onto a solid medium. The amplitude field of a sound beam with finite lateral dimensions can be expressed as a sum of infinite plane waves via a Fourier transform pair.1 An incident bounded beam can then be represented by

\[ U_{in}(x, z) = (2\pi)^{-1} \int_{-\infty}^{\infty} V(k_x) \exp[i(k_x x + k_z z)] dk_x \]  
\[ (1) \]

and

\[ V(k_x) = \int_{-\infty}^{\infty} U_{in}(x, 0) \exp(-i k_x x) dx \]  
\[ (2) \]

where the time dependence \( \exp(-iwt) \) has been suppressed, \( k_x \) is the x component of the wave vector, \( V(k_x) \) is the Fourier transform of the incident beam amplitude at \( z = 0 \), and \( k_z \) is defined by

\[ k_z = (k^2 - k_x^2)^{1/2} \]

with \( k \) being the propagation wavenumber of the sound beam in the liquid.

The width of the area projected by such an incident beam onto the interface is \( 2w_0 \), which is given by

\[ w_0 = w \sec \theta_1 \]

Thus for well-defined beams, the range of integration in Eqs. (1) and (2) effectively is

\[ (k_1 - \pi/w_0) \leq k_x \leq (k_1 + \pi/w_0) \]  
\[ (3) \]

FIG. 1. A bounded sound beam incident onto the plane interface with a solid. The scale of the x axis is chosen arbitrarily for the center of the incident beam to be at 100.
and

\[ -w_0 \leq x \leq w_0, \]  

where

\[ k_i = k \sin \theta_i. \]

The Fourier integral for the incident beam \( U_{\text{inc}}(x, z) \) can be interpreted as being a superposition of an infinite number of incident plane waves with different amplitudes \( V(k_x) \), all of the same wavelength but incident at different angles within a narrow angular range about \( \theta_i \). Extending this interpretation, one can represent the reflected beam profile by integrating over individual reflected plane waves which are now affected by the plane wave reflection coefficient \( R(k_x) \) as

\[ U_2(x, z) = (2\pi)^{-1} \int_{-\infty}^{\infty} R(k_x) V(k_x) \exp[i(k_x x + k_z z)] \, dk_x. \]

At the interface \( z = 0 \) this expression reduces to

\[ U_2(x, 0) = (2\pi)^{-1} \int_{-\infty}^{\infty} R(k_x) V(k_x) \exp(ik_x x) \, dk_x, \]

with \( R(k_x) \) being given by

\[ R(k_x) = \frac{(k_x^2 - 2k_z^2)k + \rho k_x^2 k_x}{(k_x^2 - 2k_z^2)^2 + 4k_x^2 k_z^2 \rho}, \]

where

\[ \rho = \text{liquid density/solid density}, \]
\[ k = \omega/v_L, \quad k_z = \omega/v_S, \]
\[ \kappa = (k_x^2 - k_z^2)^{1/2}, \]
\[ k_0 = (k_x^2 - k_z^2)^{1/2}, \quad k_z = (k_x^2 - k_z^2)^{1/2}, \]

where \( v_L \) is the sound velocity in the liquid and \( v_S \) are the velocities of the longitudinal and shear waves in the solid, respectively.

II. AN APPROXIMATE ANALYTICAL SOLUTION TO THE REFLECTED BEAM INTEGRAL

Expanding \( R(k_x) \) in a Laurent series, Bertoni and Tamir were able to derive an approximate form for \( R(k_x) \) which yields the reflected beam profile expressed by Eq. (6) for a Gaussian beam incident at the Rayleigh critical angle. With the incident Gaussian beam being normalized to unity, the incident and reflected beam intensity profiles can be written as

\[ |U_{\text{inc}}(x, 0)|^2 = \exp[-2(x/w_0)^2] \]

and

\[ |U_2(x, 0)|^2 = |R_1 + R_2 R_2^*| |U_{\text{inc}}(x, 0)|^2, \]

where

\[ R_1 = (\text{Re} \, k_x - k_z)/(\text{Re} \, k_z - k_x), \]
\[ R_2 = (k_x - k_z)/(k_x - k_z), \]
\[ R_2 = 1 - [(\pi/2 \text{Im} \, k_z)/(\omega/w_0) \exp(\gamma^2) \text{erfc}(\gamma)], \]

where

\[ k_0 = \text{Pole of } R(k_x), \]
\[ k_z = \text{Zero of } R(k_x), \]
\[ \gamma = (w_0 \text{Im} \, k_z)/(x/w_0). \]

Briefly, the Bertoni and Tamir's formulation requires that the pole and zero locations be identified before the reflected profile can be determined by Eqs. (8) and (9).

III. INCORPORATION OF ATTENUATION IN THE MEDIA

Effects of attenuation in materials can be incorporated in the wave vectors \( k_x, k_z, \) and \( k_0 \). Denoting \( a_x, \) \( a_z, \) and \( a_0 \) as attenuation per wavelength for the sound waves in liquid, the longitudinal waves, and the shear waves in the solid, respectively, one can write the wave vectors for the absorptive media as

\[ k = (\omega/v_L)(1 - ia_x/2\pi), \]
\[ k_z = (\omega/v_S)(1 - ia_z/2\pi), \]
\[ k_0 = (\omega/v_0)(1 - ia_0/2\pi). \]

Inclusion of attenuation moves the branch points \( k_0, \) \( k_z, \) and \( k_z \) off the real axis. Analytically, introduction of attenuation into the terms contained in the reflection coefficient [Eq. (7)] makes \( R(k_x) \) differentiable even at the longitudinal and shear critical angles. For the lossless case, the reflection coefficient is frequency-independent. In practice, this is not the case since one usually observed in experiments different profiles for different frequencies.

Bertoni and Hou\(^*\) show that the values of \( k_0, \) \( k_z, \) and \( k_0 \) will change with the frequency-dependent values of \( a_x, \) \( a_z, \) and \( a_0, \) making the simplifying assumption that \( a_x = a_z = a_0 \) and that the absorption in the liquid medium can be neglected for a water-stainless steel interface. Under these simplifying conditions the value of \( k \) in Eq. (10) reduces to a real quantity, and the imaginary parts are the same for both \( k_0 \) and \( k_z. \) However, if one uses distinct values for \( a_x \) and \( a_z, \) and includes attenuation in the liquid, the loci of \( k_0, k_z, \) and \( k_0 \) are different from those obtained under these simplifying assumptions.

The results of such calculations for a water-stainless steel interface are shown in Fig. 2(a). The values of the longitudinal and shear wave absorption coefficients used are those given by Fitch\(^*\) (\( a_x = 0.00017, a_z = 0.00024 \)) and the value of the attenuation in water was

FIG. 2. Pole and zero locations at \( R(k_x) \) calculated for different attenuation parameters in the frequency range 1-40 MHz for water-stainless steel interface when \( a_x = a_z = a_0, \) and \( a = 0 \) (b) \( a_x = a_z = a_0 \) and \( a = 0. \)
assumed to be \( a = 0.00026 \). Figure 2(b), which is taken in part from Bertoni and Hou, is reproduced for comparison. The results shown in Fig. 2(b) are based on the assumption that \( a_l = a_r = 0 \) and that \( a \) is negligible. One notes that the loci of the poles and zeros in Fig. 2(a) no longer lie on a straight line. The values of \( \text{Re}(k_l)/\text{Re}(k) \) are of no particular importance for the present discussion since they depend only on the assumed value of the sound velocity in water.

IV. BEAM PROFILES CALCULATED BY NUMERICAL INTEGRATION

In the Bertoni and Tamir's derivation, it is important to note that one must identify the Rayleigh pole and zero and that the reflection coefficient is approximated in the following form:

\[
R(k_l) = (k_l - k_0)/(k_l - k_0).
\]

It is shown that Eq. (11) is only sufficiently accurate when the incident angle is at the Rayleigh critical angle. Away from the Rayleigh angle, the accuracy of Eq. (11) is reduced rapidly. It is realized that the reflected profile expressed by Eq. (6) can also be calculated by numerical integration. Since the exact form of the reflection coefficient is to be used, such a numerical approach can provide an accurate profile even away from the Rayleigh critical angle without a prior knowledge of the pole-zero location.

After normalization with respect to an incident beam with a Gaussian profile and making use of the effective integral limits given by Eqs. (3) and (4), Eq. (6) can be written as

\[
U_s(X) = (2\pi)^{-1/2} \int_{A_l-w_s}^{A_l+w_s} R(A_s) V(A_s) E(X, A_s) \, dA_s,
\]

where

\[
R(A_s) = \frac{(2A_s^2 - A_s^2 + 4B_s A_s^2 - A_s^2 B_s) / B}{(2A_s^2 - A_s^2 + 4B_s A_s^2 - A_s^2 B_s / B),}
\]

\[
V(A_s) = C \exp\left[-(C/2)^2(A_s - A_i)^2\right],
\]

\[
E(X, A_s) = \exp(iCA_s X),
\]

where the new dimensionless variables are defined as

\[
A_1 = \sin \theta_1, \quad X = x/w_0, \quad C = \omega w_0/v, \quad W_s = \pi/C.
\]

The reflected profiles which are shown in the next section are numerically determined according to Eq. (12).

V. CALCULATED AND EXPERIMENTAL RESULTS

In this section, the reflected profile is calculated by two approaches for different sets of physical parameters. First, the frequency is kept constant while the beamwidth is being changed. Then, the beamwidth is held constant as one varies the sound frequency. In these two cases, the beam is incident at the Rayleigh angle and the values of \( k_l \) and \( k_0 \) substituted in the results derived by Bertoni and Tamir are obtained from the attenuation data which are entered into the numerical integration approach. This was done to eliminate the possible discrepancy which may arise due to different data used for attenuation. Finally, the profiles of the reflected beam is calculated when the incident angle is varied near the Rayleigh critical angle.

In Fig. 3, the reflected beam profile is plotted for different beamwidths at 2 MHz for a water–stainless steel interface. The reflected profiles which are shown in the next section are numerically determined according to Eq. (12).
steel interface. Results obtained by both methods, Bertoni and Tamir's analysis and the numerical integration, are presented in this figure. Also for the water-stainless steel interface, the reflected profiles are plotted in Fig. 4 for various frequencies while the beamwidth remains unchanged. Again, both results are shown in the same figure. It is seen that Bertoni and Tamir's analysis, although using an approximate form for $R(k_\perp)$, agrees very well with an almost exact calculation.

Figure 5 shows the results for a 2-MHz beam incident on a water-stainless steel interface when the incident angle is varied from the Rayleigh critical angle. A gradual change in the beam profile is noticeable, in particular, the slow increase in intensity in the "null intensity" as the specularly and nonspecularly reflected sections of the beam merge into one with lessening beam displacement.

The same gradual change is observed experimentally. Figure 6 consists of a series of Schlieren photographs of the incident and reflected beams for a water-stainless steel interface. The beamwidth and the sound frequency used were 19 mm and 1.96 MHz. The incident angle was varied by steps of 0.25°.

It should be noted that the profile of the ultrasonic beam used was nearly Gaussian so that agreement between theory and experiment exists qualitatively. It should also be noted that the same type of agreement exists between experiment and the predictions of the present theory for reflection at angles greater or smaller than the Rayleigh angle by up to $2^\circ$. This range is sufficiently large so that differences between the approximate formulation and the numerical integration method become noticeable.

VI. CONCLUSION

It is shown that a numerical integration method can be used to determine accurately the reflected beam profile for a liquid-solid interface. The method is not restricted to calculations at the Rayleigh angle but can be extended to other angles of incidence, where the analytical analysis cannot be applied. The results obtained by the two methods do agree at and very near the Rayleigh critical angle. Schlieren photographs of the reflected beam at and near the Rayleigh critical angle show qualitative agreement between the results of the numerical integration method and experiments.

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The nonspecular phenomena are investigated theoretically for an ultrasonic beam transmitted through a solid plate immersed in a liquid. The analytical method used by Bertoni and Tamir [3] to describe the nonspecular profile of a beam reflected from a liquid-solid interface is extended to the problem under consideration, taking into account all existing poles. To solve the integral representing the transmitted beam, the amplitude-plane wave transmission coefficient is replaced by a simpler approximate form. The transmitted beam profile is then calculated from both single-pole and multiple-pole formulations.
I. Introduction

Since the work of Goos and Hanchen in 1947, it has been known that light waves are not always reflected geometrically but may be shifted in the specular plane at some angles of incidence[1]. Similar displacements have been observed with sound beams as well as other non-specular phenomena, namely, a null strip in the reflected beam, and a trailing sound field which follows the primary reflection and decreases in intensity as it moves down the interface. In the early investigations of these effects, Schoch [2] proposed a simple theory which explained the displacement of the sound beam but failed to account for other non-specular features.

More recently Bertoni and Tamir [3] have proposed a theory which successfully accounts for all observed non-specular phenomena in reflection from a liquid-solid (L/S) interface. These authors emphasize the use of complex poles of the plane wave reflection coefficient in determining both an approximate mathematical form of the coefficient itself and in finding an accurate description of the reflected sound intensity profile. Pitts [4] has extended the approach of Bertoni and Tamir to the study of the reflection of a sound beam from a liquid-solid-liquid (L/S/L) system.

Experimental observation has shown that non-specular phenomena occur in the transmitted beam as well. In this investigation we will take the general theoretical approach used by Bertoni and Tamir and by Pitts and extend it to the problem of the transmission of a bounded acoustic beam through an L/S/L system [5].

II. Theory of the Reflection and Transmission of a Bounded Beam by a Solid Plate

Consider a bounded acoustic beam of width 2W incident at angle \( \theta \); on a
solid plate of width \( d \) immersed in a liquid. The coordinate system is as shown in Fig. 1 where the incident, reflected and transmitted beams are depicted. The plate's incident surface is taken to form the \( xy \) plane. The sound field is assumed to be uniform in the direction perpendicular to the plane of incidence and, therefore, has no spatial \( y \) dependence. Using Fourier integrals, the particle displacement amplitude \( U(x, z) \) associated with the incident sound field can be expressed as a sum of infinite plane waves,

\[
U(x, z) = \frac{1}{2\pi} \int_{-\infty}^{\infty} V(R_x) \exp \left[ i (xR_x + zR_z) \right] dR_x .
\]  

(1)

Here \( k_x \) is the projection of the incident wave vector on the \( x \)-axis, \( k_x = k \sin \theta \), with \( \theta \) the angle of incidence of the plane wave, and \( k_z^2 = k^2 - k_x^2 \). The amplitude \( V(k_x) \) of the plane wave components of the incident beam can be obtained by Fourier inversion from the initial beam on the \( z = 0 \) or incident surface of the plate,

\[
V(R_x) = \int_{-\infty}^{\infty} U(x, 0) \exp \left[ -i x R_z \right] dx .
\]  

(2)

The Fourier integral of the incident beam can be interpreted as being composed of an infinite number of plane waves, all of the same wavelength but at different incident angles \( \theta \) with respect to the normal to the surface. If this interpretation is extended to the sound fields reflected and transmitted by the plate, they too can be expressed as a sum over individual plane waves by using the reflection and transmission coefficients, \( R(k_x) \) and \( T(k_x) \) for the \( L/S/L \) system,

\[
U_R(x, z) = \frac{1}{2\pi} \int_{-\infty}^{\infty} R(R_x)V(R_x) \exp \left[ i (xR_x + zR_z) \right] dR_x ,
\]

\[
U_T(x, z) = \frac{1}{2\pi} \int_{-\infty}^{\infty} T(R_x)V(R_x) \exp \left[ i (xR_x + zR_z) \right] dR_x .
\]  

(3)
Any evaluation of the reflected and transmitted sound beams depends, among other things, on the amplitude distribution of the incident beam $U(x,0)$. Most transducers produce roughly a Gaussian beam for which a reasonable approximation is

$$U(x,z) \approx e^{i\lambda x} \exp \left[ -\left( \frac{x}{W} \right)^2 \right],$$

where $W$ is the half beam width and $z$ is a coordinate transverse to the beam propagation direction. A convenient form for the incident profile is

$$U(x,0) = e^{i\lambda x} \exp \left[ -\left( \frac{x}{W_0} \right)^2 + i \times \mathbf{R}_i \right], \quad (4)$$

where $W_0 = W \sin \theta_i$, $\mathbf{R}_i = \mathbf{R} \sin \theta_i$, and $\theta_i$ is the beam's incident angle as defined in Fig. 1. Combining (4) and (2) we get

$$V(\mathbf{R}_x) = \frac{W_0}{\sqrt{\pi}} \int \mathcal{R} \left[ -i \left( \mathbf{R}_i - \mathbf{R}_x \right) \left( \frac{W_0}{2} \right)^2 + i \times \mathbf{R}_x \right] d \mathbf{R}_x, \quad (5)$$

and for the reflected and transmitted sound field amplitudes at the appropriate surfaces

$$U_R(x,0) = \frac{W_0}{2\pi} \int \mathcal{R} \left[ i \left( \mathbf{R}_i - \mathbf{R}_x \right) \left( \frac{W_0}{2} \right)^2 + i \times \mathbf{R}_x \right] d \mathbf{R}_x, \quad (6)$$

$$U_T(x,0) = \frac{W_0}{2\pi} \int \mathcal{T} \left[ i \left( \mathbf{R}_i - \mathbf{R}_x \right) \left( \frac{W_0}{2} \right)^2 + i \times \mathbf{R}_x \right] d \mathbf{R}_x. \quad (7)$$

An evaluation of these integrals and their comparison with experiment is a central problem in the study of the L/S/L system. The reflected beam has been thoroughly investigated by Pitts and others, and in this study we shall extend their treatment to the transmitted sound field.

III. The Amplitude Transmission Coefficient

The reflection and transmission coefficients which appear in (6) and (7) are obtained from the boundary conditions at the surfaces of the plate. A
straightforward but laborious calculation yields

\[ R(R_x) = \frac{1}{f_s f_a} \left\{ \left( \frac{R_s^2 - 3R_x^2}{2} \right) + 4 \frac{R_x^2}{\sin \theta} K_s K_d \left( \frac{1 - \cos \theta \cos \theta}{\sin \theta} \right)^2 \right\} \]

\[ + 8 \left( \frac{R_s^2 - 3R_x^2}{2} \right) \frac{R_x^2}{\sin \theta} K_s K_d \left( \frac{1 - \cos \theta \cos \theta}{\sin \theta} \right)^2 \]

(8)

\[ T(R_x) = \frac{1}{f_s f_a} \left\{ \frac{\left( \frac{R_s^4}{K} \right)}{\sin \theta} \right\} \left\{ \frac{\left( \frac{R_x^2}{\sin \theta} \right)^2}{\sin \theta} + \frac{4 \frac{R_x^2}{\sin \theta} K_s K_d}{\sin \theta} \right\} \]

(9)

where

\[ f_s = \left( \frac{R_s^2}{R_x^2} \right)^2 \left\{ \frac{1 + \cos \theta}{\sin \theta} \right\} + 4 \frac{R_x^2}{\sin \theta} K_s K_d \left( \frac{1 + \cos \theta}{\sin \theta} \right) - \frac{\rho R_s^4}{K} \]

(10)

\[ f_a = \left( \frac{R_s^2}{R_x^2} \right)^2 \left\{ \frac{1 - \cos \theta}{\sin \theta} \right\} + 4 \frac{R_x^2}{\sin \theta} K_s K_d \left( \frac{1 - \cos \theta}{\sin \theta} \right) + \frac{\rho R_s^4}{K} \]

(11)

where the following variables have been defined:

\[ \rho = \rho_{\text{liquid}} / \rho_{\text{solid}}, \]

\[ \kappa = k \cos \theta = k(1 - \sin^2 \theta)^{\frac{1}{2}}, \]

\[ \kappa_s = k_s \cos \theta_s = k[(v/v_s)^2 - \sin^2 \theta_s]^\frac{1}{2}, \]

\[ \kappa_d = k_d \cos \theta_d = k[(v/v_d)^2 - \sin^2 \theta_d]^\frac{1}{2}, \]

\[ P = \Delta \kappa_d, \]

\[ Q = \Delta \kappa_s. \]

The coefficients (8) and (9) satisfy the relationship

\[ |R(R_x)|^2 + |T(R_x)|^2 = 1 \]

(13)
appropriate to a lossless medium.

Because of the complicated dependence of $R(k_x)$ and $T(k_x)$ upon $k_x$, a direct evaluation of the profile integrals (6) and (7) is very difficult. Complex contour integration is a possibility but the presence of branch cuts in $R$ and $T$ poses difficulties if this approach is taken. Bertoni and Tamir in the case of reflection from the L/S interface, and later Pitts treating the L/S/L case, have addressed this problem by using an approximation for $R(k_x)$. The approximate reflection coefficient of the general form

$$R'(k_x) = \frac{N}{j=1} \left( \frac{R_x - \frac{Z_j}{R_x - \frac{Z_j}{R_x}} \right)$$

is constructed to have the same poles $p_j$, $p_j = \beta_j + i\alpha_j$, and the same zeros $z_j$ as $R(k_x)$ but no other singularities. With $R(k_x)$ replaced by $R'(k_x)$, the integral (6) can be evaluated using the residue theorem.

We will follow that approach in this investigation and will construct approximations to the transmission coefficient. With these approximations the profile of the transmitted beam (7) will be evaluated and compared with experiment.

In a lossless medium the sum of reflected and incident energies must equal the incident energy, a condition expressed mathematically in Eq. (13). In the present investigation, where we shall try to use as much as possible the earlier results of Pitts, this equation will be the basis of our approximation procedure for $T$. Equation (13) states, among other things, that whenever $|R| = 0$, $|T| = 1$. Two very simple forms for $T$ which have this property are

$$T'(k_x) = 1 - \frac{N}{j=1} \left( \frac{R_x - \frac{Z_j}{R_x - \frac{Z_j}{R_x}} \right)$$

and

$$T''(k_x) = \frac{N}{j=1} \left( 1 - \frac{R_x - \frac{Z_j}{R_x - \frac{Z_j}{R_x}} \right)$$
in which \( z_j \) and \( p_j \) are respectively the zeroes and poles of \( R(k_x) \) already introduced in connection with (14).

Each of these forms has its shortcomings. Let us denote the relative phase of \( T, T', \) and \( T'' \) by \( \psi, \psi', \) and \( \psi'' \), respectively. Then numerical evaluation of (15), (16), (9) and comparison show good agreement between \(|T'|\) and \(|T|\) but poor agreement between the phases \( \psi' \) and \( \psi \). On the other hand, there is good agreement between \( \psi'' \) and \( \psi \), but poor agreement between \(|T''|\) and \(|T|\).

IV. The Transmitted Beam Profile

In addition to the complexities of \( T \), the integral (7) for the transmitted beam is complicated by the presence of the factor \( \exp (dk_z) \) in the integrand. This factor does not appear in (6) because of the choice of the incident surface of the plate as the \( z = 0 \) surface. This problem can be addressed, however, by expanding \( k_z \) in a Taylor series about \( \sin \theta \).

\[
R_z = R \cos \theta - (R_x R \sin \theta) \tan \theta + (R_x R \sin \theta)^2/2 R \cos^3 \theta + \ldots \quad (17)
\]

All the terms are now in the proper form for the evaluation of the transmitted beam profile integral. Because of the complexity of the expressions, it is worthwhile defining an integral which plays an important role in the evaluation. It describes the profile of the incident beam at the \( z = d \) plane as if there were no solid plate present. Substitution of Eqs. (2) and (17) into (1) yields

\[
U(x, d) = \frac{w}{2 \pi^2 \sin \theta} \int_{-\infty}^{\infty} \exp \left\{ -\left( R_x R \sin \theta \right)^2 \left( \frac{w}{2 \cos \theta} \right)^2 \right\} 
\]

\[
x \exp \left\{ i \pi x + i d \left[ R \cos \theta - (R_x R \sin \theta) \tan \theta + (R_x R \sin \theta)^2/2 R \cos^3 \theta \right] \right\} \exp \left\{ -\left( R_x R \sin \theta \right)^2/2 R \cos^3 \theta \right\} d R_x \quad (18)
\]
After some manipulation this can be shown to become

$$U(x, z) = \frac{W}{w_T} \exp \left\{ -\left( \frac{x_T}{w_T} \right)^2 + i R z_T \right\} $$

(19)

in which the variables

$$x_T = x \cos \theta - z \sin \theta,$$

$$z_T = x \sin \theta + z \cos \theta,$$

$$w_T = W\left[1 - (2iz/kw^2 \cos \theta) \right]^{1/2},$$

have been introduced. Equations (19) and (20) not only describe the Gaussian incident beam in terms of a new set of coordinates, $z_T$ parallel with and $x_T$ transverse to the direction of incidence, but also explicitly express the beam spreading caused by the geometrical diffraction of a bounded beam.

The transmitted sound field amplitude (7) may now be evaluated for either of the approximations (15) or (16). The procedure is essentially the same for both, and we will present it here only for $T''(k_x)$. First noting that

$$T'' = \frac{N}{\pi} \left(1 - \frac{R_x - \zeta_0}{R_x - \zeta_3} \right) = \frac{N}{\pi} \left( \frac{R_x - \zeta_3}{R_x - \zeta_3} \right)$$

then substituting into (7), we find

$$U_m^{(w)}(x, z) = \frac{W}{2\pi^{3/2} \cos \theta} \left\{ \frac{N}{\pi} \left( \frac{R_x - \zeta_0}{R_x - \zeta_3} \right) \right\} \sum_{m=1}^{\infty} \frac{d R_x}{\pi} \exp \left\{ -\left( \frac{R_x - \zeta_0}{\cos \theta} \right)^2 \left( \frac{W}{2 \cos \theta} \right)^2 + i x R_x + i z \left[ R_x \cos \theta \right] 

-(R_x - \zeta_0 \cos \theta) \zeta_0 \cos \theta + (R_x - \zeta_0 \cos \theta)^2 / 2 \cos \theta \right\} \right\}$$

$$= \frac{W_m}{2\pi^{3/2} \cos \theta} \left\{ \frac{N}{\pi} \left( \frac{R_x - \zeta_0}{R_x - \zeta_3} \right) U^{(o)}(x, z) \right\} \exp \left\{ \left( \frac{x}{w_T} \right)^2 \right\}$$

$$\times \sum_{m=1}^{\infty} \frac{d R_x}{\pi} \exp \left\{ -\left[ \frac{(R_x - \zeta_0 \cos \theta)}{2 \cos \theta} \right]^2 \right\} W_m^{1/2}$$

(21)

$$+ i (R_x - \zeta_0 \cos \theta) \zeta_0 / \cos \theta \right\}$$
Here $I^{(N)}$ is the integral

$$I^{(N)} = \exp \left\{ \left( \frac{\pi}{w_m} \right)^2 \sum_{m=1}^{N} \left( \Phi_m - \Phi_m' \right) \right\} \int_{-\infty}^{\infty} \frac{dk}{1 + \left( k - k_j \right)^2} \exp \left\{ -\left( \frac{1}{2} \frac{1}{w_m} \right) y^2 \right\}$$

where

$$k = \frac{\mathcal{R}_x - \mathcal{R}_y \sin \Theta}{2 \mathcal{R} \cos \Theta}, \quad k_j = \frac{\mathcal{R}_y - \mathcal{R}_x \sin \Theta}{2 \mathcal{R} \cos \Theta}.$$
$G^{(n)}(b-y)$ is integrable by complex contour integration. For $b > y$, the contour must enclose those poles with $a_1, \ldots, a_N > 0$ and for the $b < y$ case, it must enclose those with $a_1, \ldots, a_N < 0$. In forming both the approximations $T'(k_x)$ and $T''(k_x)$ only poles in the first quadrant were used. Therefore

$$G^{(n)}(b-y) = i \sum_{m=1}^{N} \left\{ \prod_{n=1 \atop m \neq m}^{N} \frac{1}{k_n - k_m} \right\} \exp \left\{ -i \left( b - y \right) k_m \right\}, \quad (26)$$

and the convolution integral can be carried through to yield

$$I^{(n)} = \exp \left\{ \left( \frac{\chi_m}{\omega r} \right)^2 \right\} \sum_{m=1}^{N} \left\{ \left( \frac{\omega r - 2\pi}{\omega r - 2\pi} \right)^{2} \exp \left\{ -i \left( \frac{\chi_m}{\omega r} \right)^2 \right\} \right\} \exp \left\{ \gamma_m^2 \right\} \exp \left\{ \gamma_m \right\}, \quad (27)$$

where

$$\gamma_m = \frac{\chi_m}{\omega r} - i \frac{\omega_r (\rho_m - R \cos \theta)}{2 \cos \theta}. \quad (28)$$

Finally the profile of the transmitted beam for the approximation $T''$, using $N$ poles, is

$$U^{(n)}_{T''}(\chi, z) = U^{(0)}(\chi, z) \left[ \frac{i \pi \frac{z}{2 \cos \theta}}{\omega_r} \right] \omega_r \sum_{m=1}^{N} \left\{ \left( \frac{\omega_r - 2\pi}{\omega_r - 2\pi} \right)^{2} \exp \left\{ -i \left( \frac{\chi_m}{\omega r} \right)^2 \right\} \exp \left\{ \gamma_m^2 \right\} \exp \left\{ \gamma_m \right\} \right\}. \quad (29)$$

The other approximation $T'^{(n)}(R_x)$ can be written as

$$T'^{(n)}(R_x) = 1 - \sum_{m=1}^{N} \left( 1 + \frac{\rho_m - \xi_m}{\omega R_x - p_m} \right) \quad (30)$$
To express $T'(N)$ in terms of $T''(N)$, Eq. (16) is rearranged to give

$$T''(N) = \prod_{m=1}^{N} \left( \frac{\psi_m - z_m}{\tau_x - \psi_m} \right)$$

(31)

Denoting $\rho_m := \frac{\psi_m - z_m}{\tau_x - \psi_m}$, it is noted from Eqs. (30) and (31) that

$$T''(N)(\tau_x) = -\sum_{M=1}^{N} T''(M)(\tau_x)$$

(32)

Where $T''(M)(\tau_x)$ is defined as the sum of all possible products $a_{n_1} \cdots a_{n_M}$, in which $n_k$'s can assume any value from 1 to $N$ and must satisfy the requirement $a_{n_1} < a_{n_2} < \cdots < a_{n_M}$. For example, if $N = 3$, then $T''(1) = a_1 + a_2 + a_3$, $T''(2) = a_1a_2 + a_1a_3 + a_2a_3$, and $T''(3) = a_1a_2a_3$. Insertion of Eq. (32) into Eq. (7) would lead to the transmitted beam profile corresponding to $T'(N)(\tau_x)$,

$$\mathcal{U}'(N)(\chi, z) = -\sum_{M=1}^{N} \mathcal{U}''(M)(\chi, z)$$

(33)

where $\mathcal{U}''(M)(\chi, z)$ can be derived from Eq. (29).

V. Results and Conclusions

Theoretical predictions for these angles of incidence were obtained by numerically evaluating the transmitted beam profile $U_T(N)$ as given by Eq. (29). The approximation $T''$, Eq. (16) depends upon the complex poles $\rho_j = \beta_j + i\alpha_j$ of the transmission coefficient and, as shown by Pitts et al., $k_x = \beta_j$ corresponds to the condition for the existence of a vibrational mode of the plate. Since $k_x = k \sin \theta$, $\beta_j = k \sin \theta_j$ implies that at an angle of incidence $\theta_j$ the mode of vibration corresponding to the $j$th pole is excited. At an angle of incidence very near the $j$th pole, i.e.,
the contribution for the jth pole to the total intensity profile is five or more orders of magnitude greater than the contribution from other poles. At such angles the jth pole term dominates the entire expression for the transmission profile.

In our calculations, $U^{(N)}_{T^1}(x,z)$ is used since poor agreement between $|T|$ and $|T''|$ is not tolerable in the determination of the transmitted profile. The transmitted profile was first calculated for the case where $\theta_i = \theta_j$. Because of the dominance of the jth pole, only that pole was accounted for in $U^{(N)}_{T^1}(x,z)$ and the resulting profiles are shown in Fig. 2. The input values are chosen for a brass plate in water with $fd = 2 \text{ MHz.mm}$ and $\sin \theta_i = 0.36$ for two frequencies, 1 MHz and 7 MHz. The outstanding feature seen here is a lateral displacement of the transmitted beam relative to the incident beam. It is noted that in Fig. 2 the 7 MHz profile has a larger beam displacement but with a much lower intensity peak.

When there exist many poles in the neighborhood of $\sin \theta_i$, all the involved poles must be included in the evaluation of $U^{(N)}_{T^1}(x,z)$ to obtain the appropriate profiles. For example, the transmitted profile was again calculated for a water-brass-water system with $fd = 6.5 \text{ MHz.mm}$, $f = 2 \text{ MHz}$, and $\sin \theta_i$ varied from 0.31 to 0.36. In this angular range, there are two poles. The results of these calculations are shown graphically in Fig. 3. The profiles exhibit some interference effect and are seen to change with the incident angle.

The method employed in this work, while giving some reasonable results is limited in several ways. One is that the restraint (13) between $R$ and $T$ is bothersome and neither $T'$ or $T''$, the forms of which are suggested partly by it, are fully satisfactory. A more serious limitation is the dependence of this method on the location of the poles of $T$; the closer $\sin \theta_i$ is to a particular $p_j$ the better it works. A technique which does not involve the poles might be less restricted.
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Figure Captions

1. Coordinate system and schematic diagram of a bounded beam reflected from and transmitted through a solid plate immersed in a liquid.

2. Intensity profiles calculated for a beam transmitted through a solid plate in water with $f_d = 2 \text{ MHz.mm}$ and $\sin \theta_i = 0.36$ for two frequencies: (a) 1 MHz and (b) 7 MHz.

3. Intensity profiles calculated for a beam transmitted through a solid plate in water with $f_d = 6.5 \text{ MHz.mm}$, $f = 2 \text{ MHz}$, and $\sin \theta_i$ being varied from 0.31 to 0.36.
Restrictions on the Existence of Leaky Rayleigh Waves

NEAL G. BROWER, DOUGLAS E. HIMBERGER, AND WALTER G. MAYER

Abstract—The Rayleigh wave, an inhomogeneous surface wave, exists for all isotropic elastic solid infinite half-spaces. When the free surface of the solid is bounded by a liquid a leaky Rayleigh wave does not necessarily exist for all liquid/isotropic solid systems. The well-known condition for the existence of a leaky Rayleigh wave, the sound velocity in the liquid must be less than the shear wave velocity in the solid, is shown to be a necessary but not a sufficient condition. Additional conditions on density ratios and velocity ratios are given. Examples are listed showing liquid/solid combinations which satisfy the liquid-shear wave velocity condition but not the additional restrictions and thus do not support a leaky Rayleigh wave.

The purpose of this paper is an investigation of further conditions on the existence of a leaky Rayleigh wave. The conditions involve restrictions on the following ratios which describe the liquid/elastic solid system: $\rho_L/\rho_S$, $V_L/V_D$, $V_L/V_S$, where $\rho_L$ and $\rho_S$ are the densities of the liquid and the solid, respectively, and $V_L$, $V_D$, and $V_S$ are the sound velocity in the liquid, the longitudinal and shear bulk wave velocities in the solid.

Limits on these ratios are derived from numerical solutions of the secular equation for leaky Rayleigh waves and specific examples are given of liquid/elastic solid combinations which exceed the limits.

II. THEORETICAL BACKGROUND

Consider an unbounded isotropic bulk medium. In this case two and only two different waves can be propagated. If the medium is bounded, other waves may exist.

A. The Rayleigh Wave

For an isotropic elastic solid with a plane boundary, a surface wave always exists. The wave is known as a Rayleigh wave. Briefly, one seeks a solution to the equations of motion for an elastic solid where the displacement amplitude decays exponentially perpendicular to the surface. At the boundary, all stress components must vanish. This leads to a secular equation for the velocity of the surface wave $V$. The secular equation is given by

$$4(V_S/V)^2[1 - (V_D/V)^2]^{1/2}((V_S/V_D)^2 - (V_S/V)^2)^{1/2} + [1 - 2(V_S/V)^2]^2 = 0$$

where $V_D$ and $V_S$ are the longitudinal and shear bulk wave velocities, respectively. There are three roots to the secular equation (1). The range of the Poisson ratio $\sigma$ determines the nature of the roots. The Poisson ratio may be written in terms of the bulk velocities as

$$\sigma = (V_D^2 - 2V_S^2)/(2V_D^2 - V_S^2).$$

If the Poisson ratio is in the range where

$$\sigma < 0.263 \ldots$$

then there exist three real roots to (1). However, if the Poisson ratio falls in the range where

$$\sigma > 0.263 \ldots$$
there exist one real, and two complex conjugate roots. For all real media the Poisson ratio is bounded by

$$0 < \sigma < 0.5.$$  

(5)

Complex roots are not acceptable for they imply attenuation or damping which is not the case for the free surface here. Further the root \( V \) must have a magnitude less than the velocity of the bulk shear wave in the medium.

$$V < V_S.$$  

(6)

The constraint is a manifestation of the radiation condition. Thus there is only one root \([2]\) satisfying (6). The root, given by

$$V = V_R$$  

(7)

corresponds to the Rayleigh surface wave propagating with velocity \( V_R \). The root (7) always exists. The Rayleigh velocity varies approximately from

$$0.87 V_S \rightarrow 0.96 V_S$$  

(8)
as the Poisson ratio varies from 0 to 0.5.

B. The Leaky Rayleigh Wave

Consider the elastic isotropic solid above, now loaded by a liquid half-space. Again one seeks a solution to the equations of motion in which the displacement amplitude attenuates exponentially normal to the interface. At the interface, normal stresses and displacements must be continuous. Further tangential stresses in the solid must vanish. Upon application of the boundary conditions, a new secular equation is derived. The secular equation is given by

$$4(V_S/V)^2[1 - (V_S/V)^2]^{3/2}[(V_S/V_D)^2 - (V_S/V)^2]^{1/2} + [1 - 2(V_S/V)^2] + (\rho_L/\rho_S)(V_S/V_D)^2$$  

= \[31\]

$$- (V_S/V)^2 [(V_S/V_L)^2 - (V_S/V)^2]^{1/2} = 0$$  

(9)

where \( V_L \) is the phase velocity in the liquid. The densities \( \rho_L \) and \( \rho_S \) correspond to those of the liquid and solid, respectively.

Two permissible roots exist for (9); one of them is real. This root always exists as in the Rayleigh wave case. The surface wave associated with this solution is known as the Stoneley wave. It is a pure surface wave propagating parallel to the interface without attenuation and exponentially damped in both directions normal to the boundary. The Stoneley phase velocity \( V_{ST} \) is constrained to be

$$V_{ST} < V_L.$$  

(10)

There is a second root which is complex \([3]\). The value of this complex root approaches the Rayleigh wave velocity as the density of the liquid approaches zero. The inhomogeneous wave associated with the root is the leaky Rayleigh wave.

The complex root is an allowed solution for a liquid/elastic longitudinal wave velocity. Since the Poisson ratio has an upper limit of \( 1/2 \), the elastic bulk shear velocity, \( V_S \), is given by

$$V_S < V_D/\sqrt{2}.$$  

(18)

Due to this constraint the range of \( V_L/V_S \) is limited by the parameter \( V_L/V_D \). This is expressed in Table 1 by the cutoff in calculated maxima in the density ratio.

As can be seen from Table 1, as the bulk longitudinal wave velocity increases for a given shear wave and liquid velocity, the maximum density ratio increases. Likewise, as the elastic...
TABLE I
MAXIMUM ALLOWED \( \rho_L/\rho_S \) AS A FUNCTION OF \( V_L/V_D \) (HORIZONTAL HEADING) AND \( V_L/V_S \) (VERTICAL HEADING).

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<td>0.43</td>
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<td>0.49</td>
<td>0.52</td>
<td>0.55</td>
</tr>
<tr>
<td>Hg/Ice</td>
<td>0.25</td>
<td>0.28</td>
<td>0.31</td>
<td>0.34</td>
<td>0.37</td>
<td>0.40</td>
<td>0.43</td>
<td>0.46</td>
<td>0.49</td>
<td>0.52</td>
<td>0.55</td>
</tr>
</tbody>
</table>

IV. CONCLUSION

The leaky Rayleigh wave, which is the complex solution to (9), approaches the Rayleigh wave as the density of the liquid goes to zero. While the free Rayleigh wave always exists, the leaky Rayleigh wave does not. It is well known that if the liquid velocity is greater than the bulk elastic shear wave velocity, then there is no admissible leaky Rayleigh solution. However, as a result of parametric investigation of the secular equation via numerical analysis a second set of conditions is evident. Existence of a solution does not only depend on \( V_L/V_S \) but on the values of the three ratios: \( V_L/V_D \), \( V_L/V_S \), and \( \rho_L/\rho_S \).

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Shear velocity increases for given \( V_L \) and \( V_D \), the maximum density ratio increases. Further, for constant \( V_D \) and \( V_S \), i.e., for a given solid, an increase in the liquid velocity decreases the maximum allowable density ratio.

Fig. 1 shows the range of admissible parameter sets. All sets of the parameters under the curved surface yield a valid solution to (17). However, if for a given set of \( V_L/V_D \) and \( V_L/V_S \) the density ratio lies above the curve, then no leaky Rayleigh wave is possible. Thus one obtains the condition for existence of the leaky Rayleigh wave.

To illustrate the condition, several examples are given in Table II. For real media, there are cases where the velocity and density ratio parameters are such that no leaky Rayleigh propagation is possible. It is evident from Table II that there are cases where a leaky Rayleigh wave cannot be generated even if the usual condition \( (V_L < V_D) \) is satisfied.
ACOUSTO-OPTIC INTERACTION OF SECOND HARMONICS IN LAMB WAVES

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Abstract. - The possibility of harmonic generation in Lamb waves propagating on isotropic plates is investigated. The conditions for harmonic generation are discussed. A detection scheme using the acousto-optic interaction is designed by which asymmetries in the diffraction pattern are measured to determine the existence of second harmonics. The results of the experiments are presented.

1. Introduction. - Finite amplitude mechanical vibrations in an elastic solid are nonlinear in nature which may give rise to a number of interesting interactions. One such interaction is the generation of harmonics as a finite amplitude acoustic wave is propagating in an elastic solid.

Acoustic bulk second harmonic generation was observed by Gedroits and Krasil'nikov\cite{1} when an initially monochromatic longitudinal ultrasonic bulk wave was launched in an elastic medium, a secondary longitudinal wave was observed. The second harmonic amplitude increased linearly with the interaction length, i.e., with the distance of travel.

Harmonic generation of acoustic surface waves in crystals was observed by Lüpen\cite{2}. His experimental results agreed with the theoretical analysis. As in the case of bulk waves, the amplitude of the second harmonic surface wave varied linearly with the interaction length and quadratically with the fundamental surface wave amplitude.

The possibility of harmonic generation in elastic isotropic plates is investigated in the present experiment. As an introduction, general features of second harmonic generation are presented as a theoretical background. Next, conditions for generation are discussed. The detection scheme utilizing a laser optical probe is outlined. A brief description of the experimental design is given. Finally results of the experiment are presented.

2. Theoretical Background. - Using a perturbative approximation\cite{3}, the nonlinear equations of motion for an isotropic elastic solid bulk medium reduce to

\[ \frac{\partial^2 \dddot{U}}{\partial t^2} = C_L^2 \left( \dddot{U} + C_T^2 \frac{\partial^2 \dddot{U}}{\partial x^2} \right) + \frac{C_L^2}{C_T^2} \frac{\partial^2 \dddot{U}}{\partial y^2} = Q, \]  

where \( \dddot{U} \) is the particle displacement amplitude and \( C_L \) and \( C_T \) are the longitudinal and transverse bulk wave velocities. The term on the right-hand side of eq. (1), \( Q \), is the source or forcing term. In the case of harmonic generation, \( Q \) consists of terms quadratic in the fundamental plane wave amplitude. An iterative solution\cite{4} to the nonlinear equation of motion yields

\[ U_2 = C U_1^2 x. \]  

The amplitude of the second harmonic, \( U_2 \), is proportional to the square of the fundamental wave amplitude, \( U_1 \), and also to the interaction length, \( x \), with \( C \) a proportionality constant. This result is also obtained by Lüpen\cite{2} for surface waves in which case \( U_2 \) is the fundamental surface wave amplitude. Similarly, for plate mode interactions, eq. (2) is valid, where \( U_1 \) is the Lamb mode fundamental amplitude. The characteristics implied by eq. (2) will be used to identify a generated harmonic.

In order that a harmonic be generated, the velocities of the fundamental and the harmonic must be equal, that is
v(ω₁) = v(2ω₁).

This is a resonance condition. Two means of satisfying eq. (3) are possible for Lamb mode interaction. The curves in Fig. 1 show the Lamb mode velocity dispersion for an unloaded brass plate. One means of satisfying eq. (3) is by exciting a locally non-dispersive Lamb mode. For example, if the fundamental were excited at 7 MHz in the A₁ mode (point 1) then a second harmonic at 14 MHz (point 2) will satisfy the resonance condition. The resonance condition may also be satisfied for dispersive regions provided there is mode coupling. For example, consider a fundamental in the A₁ mode at 5 MHz (point 3) the second harmonic is in the A₂ mode (point 4). Whenever Lamb modes couple, eq. (3) is satisfied. This coupling has been observed by Brower and Mayer /5/ for the “isotropic three-phonon interaction”.

The two cases of possible resonance are experimentally investigated here. Table I lists the fundamentals used and the possible harmonics generated.

<table>
<thead>
<tr>
<th>Fundamental Mode</th>
<th>Second Harmonic</th>
</tr>
</thead>
<tbody>
<tr>
<td>A₁ at 7 MHz</td>
<td>A₁ at 14 MHz</td>
</tr>
<tr>
<td>A₂ at 5 MHz</td>
<td>A₂ at 10 MHz</td>
</tr>
</tbody>
</table>

Fig. 1. - Velocity dispersion curves for a free brass plate as a function of product frequency times plate thickness (fd in units of MHz.mm). Selected points for harmonic generation are indicated by numbers 1 to 4 - see text.


A. Basic Theory.

The optical detection system is based on the fact that a light beam (laser) will be diffracted upon reflection from a periodic surface corrugation. This corrugation may be an acoustic surface wave (a Rayleigh wave) or a plate vibrational mode (a Lamb wave). A theoretical treatment of this acousto-optic interaction for a Rayleigh wave has been given by Mayer, Lamers, and Auth /6/. This treatment can be extended to include Lamb waves /7/. Fig. 2 is a schematic diagram of the laser-surface wave interaction. The light is incident into the surface at an angle Φ and is diffracted into discrete diffraction orders with angular spacing θ, centered around the reflection angle Φ. The light intensity in the mth diffraction order due to a single-frequency Lamb mode of amplitude U₁ is given by

\[ I_m = |J_m^2(γ_1)|, \]

where

\[ γ_1 = 2\pi K_\lambda \cos Φ, \]

with K the wavenumber of the light and Jₘ the mth order Bessel function. It is evident from eq. (4) that the diffraction pattern is symmetric with respect to the central order reflected at angle Φ.

Now consider a surface corrugation caused by a Lamb wave, which, in addition to its fundamental...
frequency, contains a second harmonic component of amplitude \(U_2\). In this case the particle displacement will no longer be pure sinusoidal and a distorted Lamb wave will interact with the incident light beam. The resulting diffraction pattern will be asymmetric with respect to the central order. The amount of asymmetry will depend on the magnitude of \(U_2\) relative to \(U_1\).

This process is similar to that described by Neighbors and Mayer /8/ for nonsinusoidal Rayleigh waves. Using the same approach for nonsinusoidal Lamb waves and assuming that only the fundamental and a second harmonic component are present in the wave, the light intensities in the positive and negative first orders are given by

\[
I_{+1} = \left[ \cdots - J_3(y_1)J_2(y_2) - J_1(y_1)J_1(y_2) + J_0(y_1)J_0(y_2) \right]^2,
\]

\[
I_{-1} = \left[ \cdots - J_3(y_1)J_2(y_2) - J_1(y_1)J_1(y_2) - J_0(y_1)J_0(y_2) \right]^2,
\]

where \(y_2 = 2KU_2 \cos \varphi\).

The series of products of Bessel functions in above equations can safely be truncated to the terms shown since both \(y_1\) and \(y_2\) are usually small so that insignificant errors in the values of \(I_{+1}\) and \(I_{-1}\) are introduced by neglecting terms in \(J_rJ_s\) with \(r,s > 3\).

Equations (5-6) are the bases in the optical detection of the presence of second harmonics. If \(U_2 \neq 0\), one finds that \(I_{-1} > I_{+1}\), giving rise to an asymmetric diffraction pattern. It should be noted that eqs. (5-6) reduce to eq. (4) if \(U_2 = 0\). In this case all terms vanish except \(J_1(y_1)J_0(y_2)\) which equals \(J_1(y_1)\).

This analysis provides a method of determining the presence or absence of second harmonics in a vibrating solid plate. Probing the surface with a light beam and measuring the intensity distribution in the diffraction pattern yields information about the magnitude of \(U_2\) compared to \(U_1\). Moreover, probing the surface at different distances from the source of the Lamb wave generation, one can measure the change in harmonic content as a function of distance.

Due to the method of generating the fundamental Lamb mode, i.e., by means of the "liquid wedge" method /9/, the possibility exists that second harmonic components are generated in the liquid prior to incidence on the plate surface. However, if such components should be present and should continue to propagate along the plate they would at best decrease in amplitude as a function of distance traveled on the plate. Their amplitude cannot increase as a function of distance on the plate unless harmonic generation on the plate is possible. Thus eq. (2) will be satisfied only if harmonic generation on the plate itself occurs.

Using eq. (2), it may be shown that \(I_p\) is a linear function of distance from the source of the Lamb wave /7/ if the plate is considered to be a dissipationless medium. Therefore, \(I_p\) can be measured as a function of distance.
B. Experimental Design and Results.

The design for the optical probing system is shown in Fig. 3. Light from a He-Ne laser is expanded and collimated (details of the necessary optical components not shown in Fig. 3.) The collimated light beam is reflected from a highly polished brass plate which supports the Lamb wave. The diffracted-reflected light is focussed approximately 8 m from the plate. This distance is sufficient to properly separate the diffraction orders so that intensity measurements of I_+1 and I_-1 can be made with a photodiode.

Measurements of I_p were made at various distances x on the plate for the different fundamental frequencies and modes listed in Table I. Excitation of the modes was accomplished by adjusting the quartz transducer to the required angular position for the generation of the desired Lamb mode.

The polished brass plate, excited in the A_1 mode at 7 MHz, was probed with the He-Ne laser. A diffraction pattern was observed and the asymmetry between the first diffraction orders was measured. At a relative interaction length of 0.8 cm, the percent asymmetry was approximately 15%. The laser probe was then moved along the plate in the positive x-direction and the asymmetry increased. At a distance of 1.2 cm, a 29% asymmetry was observed. Likewise, for the plate in the fundamental A_1 mode, excited at 5 MHz, asymmetry was observed. In this case the asymmetry also increased with interaction distance. The results are listed in Table II.

<table>
<thead>
<tr>
<th>x</th>
<th>I_p</th>
<th>A_1 mode at 7 MHz</th>
<th>A_1 mode at 5 MHz</th>
</tr>
</thead>
<tbody>
<tr>
<td>0.8 cm</td>
<td>15%</td>
<td></td>
<td></td>
</tr>
<tr>
<td>1.2 cm</td>
<td>29%</td>
<td></td>
<td></td>
</tr>
<tr>
<td>0.9 cm</td>
<td>14%</td>
<td></td>
<td></td>
</tr>
<tr>
<td>1.3 cm</td>
<td>39%</td>
<td></td>
<td></td>
</tr>
</tbody>
</table>

As a control, the S_0 mode at 3 MHz was excited (point 5 in Fig. 1). As can be seen in Fig. 1, there is no second harmonic which satisfies Eq. (3) because the required location (point 6) does not fall on a curve. Therefore, there should be no second harmonic content in the detected mode. The measured diffraction pattern showed this to be true: no detectable asymmetry in the first orders could be found.

In conclusion, second harmonic generation in Lamb modes was detected. Harmonics were observed for both a locally dispersionless mode and also for the mode coupling case, provided the appropriate conditions for harmonic generation in plates were satisfied.

Acknowledgment. - This work was supported by the Office of Naval Research, U.S. Navy.

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Asymmetry will be influenced or be absent under certain conditions of the phase relationships between the fundamental and the second harmonic, as pointed out by ALIPPI A., et al. J. Phys. 46 (1977) 2182.
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